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EXPERIMENTAL DETERMINATION OF THE WAVE ELEVATION NEXT
TO A MODEL OF THE S..(U) MICHIGAN UNIV ANN ARBOR DEPT
OF NAVAL ARCHITECTURE AND MARINE.. A W TROESCH ET AL.

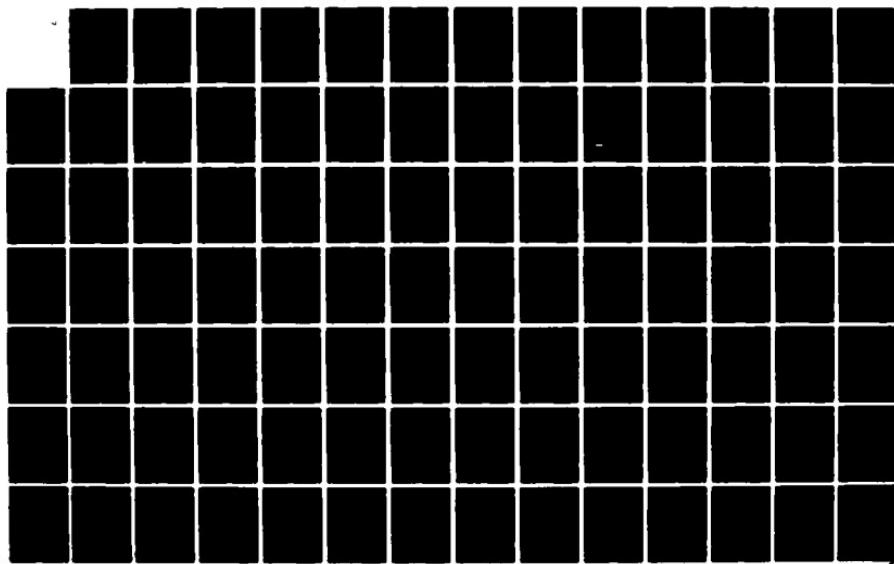
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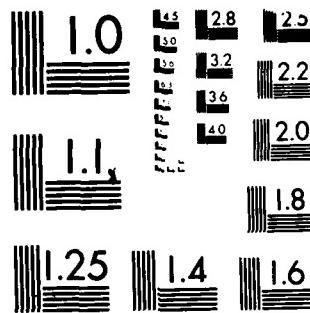
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Final Contract Report N00014-83-K-0305

April 1984
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EXPERIMENTAL DETERMINATION OF THE WAVE ELEVATION NEXT TO A MODEL OF THE SL-7

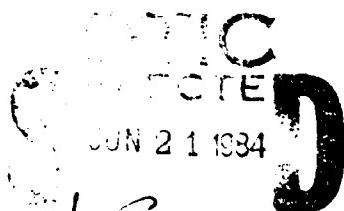
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| 20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The experimental results presented in this report describe the relative water elevation along the side of a model of the SL-7. Three relative motion components were measured and the results discussed. The components are the following: one due to steady forward speed, one due to forced oscillation, and one due to incident waves. Capacitance wave probes were attached to the model at ten locations starting at the forward perpendicular and extending to mid-ships. The forced oscillation or radiation relative motion was measured for three different forward speeds. The reactive motion associated with the incident wave | | |

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was measured in both regular and irregular, transient waves. In all cases, the wave records were Fourier analyzed on line, between runs. The results are graphed as functions of wavelength to ship-length ratios and again as functions of longitudinal hull location. A number of repeatability tests and linearity checks were run.

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Final Contract Report N00014-83-K-0305
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EXPERIMENTAL DETERMINATION OF THE WAVE ELEVATION
NEXT TO A MODEL OF THE SL-7

Department of Naval Architecture and Marine Engineering
The University of Michigan
Ann Arbor, Michigan

Prepared under contract for:

Office of Naval Research
800 N. Quincy St.
Arlington, VA 22217



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Experimental Determination of the Wave Elevation Next to a Model
of the SL-7
by
A.W. Troesch, T.B. Kelly, B.K. King

INTRODUCTION

The deficiencies in the methods used in predicting deck-wetness have been known for some time. Hoffman and MacLean (1970) noted that the usual relative motion analysis yielded poor correlation between experiment and theory. Takaishi, et.al. (1972) also conducted relative wave elevation experiments and stated that the disagreement between measured and calculated values was relatively small for bow and stern seas but large in quartering seas. More recently, Andrew and Lloyd (1980) compare the calculated average deck-wetness interval with sea trial results and report that the theory is inadequate. Oliver (1983), in a report to the Ship Structure Committee, has formulated a non-linear theory that includes above-water line effects. However, some of the assumptions in Oliver's model are suspect and the computer analysis can take hours of CPU time. More rational approaches to theoretical estimates are given by Beck (1982), Lee (1982), and Lee, O'Dea and Meyers (1982). While all of these theoretical models address fundamental aspects of the deck wetness problem, they will have difficulty including effects such as increase wave-steepness, large amplitude motions, and non-wall-sided vessels. Consequently the use of model experiments with actual ship hull forms is expected to remain an important part in the prediction of lost freeboard and deck-wetness. See, for example, O'Dea (1983), O'Dea and Jones (1983), and Webster and Zhu (1983).

The experimental results presented in this report describe the relative water elevation along the side of a model of the SL-7. Any analysis of lost freeboard and relative motion will divide the total response into various

components. Generally these will be a component due to steady forward speed, one due to body motion, and one due to incident waves. As shown in this report, all three of these components were measured and the results are discussed. Capacitance wave probes were attached to the model at ten locations, starting at the forward perpendicular and ending at mid-ships. The forced oscillation or radiation relative motion was measured for three different forward speeds. The relative motion associated with the incident waves was measured in both regular and irregular, transient waves. In all cases, the wave records were Fourier analyzed on line, between runs. The results are graphed as functions of wavelength to ship length ratios and again as functions of longitudinal hull location. A number of repeatability tests and linearity checks were run. These are also described in the text. The large amount of data collected and presented represents a useful tool in determining the relative motion of SL-7 type hull forms.

TECHNICAL DISCUSSION

Relative motion is defined as the distance from the mean waterline of the vessel to the instantaneous elevation of the water surface. Generally, the total displacement of the water at any point along the hull may be attributed to a number of components. Under a linear hypothesis, the following components are assumed to be additive:

- i) Steady forward motion wave profile.
- ii) Incident wave.
- iii) Diffracted wave.
- iv) Radiated or forced oscillation wave.

All of the above components were measured in the experiments described in this report. The relative motion at ten stations were found through the use of capacitance wave probes. The steady wave profile was determined at three Froude numbers, 0.1, 0.2, and 0.3 respectively.

The relative motion resulting from the diffracted wave and incident wave were measured together. In this type of experiment, the model is restrained from moving in heave, pitch, sway, roll, yaw, and surge. With the model rigidly attached to the carriage, it is then towed through waves generated by a wave maker located at the far end of the tank. Typically the incident wave is measured at some distance in front of the carriage. The wave probes on the model record the sum of the incident and diffracted waves. In the experiments described in this report, the incident plus diffracted wave relative motion was measured for a Froude number of 0.2 in regular and transient waves. The transient wave tests were conducted in a manner similar to that described by Takezawa and Takekawa (1976).

The radiated wave is the wave train generated by rigid body motion. In this series of tests, the model is oscillated in heave or pitch by some type of shaker mechanism. The relative motion is then defined in terms of a ratio of water displacement to hull displacement. Results for both heave and pitch at Froude numbers 0.1, 0.2, and 0.3 are included.

EXPERIMENTAL APPARATUS AND TECHNIQUES

Towing Tank and Wavemaker

The University of Michigan, Department of Naval Architecture and Marine Engineering maintains a towing tank of 190.7m (360 ft) in length, approximately 6.7m (22 ft) wide at the normal waterline, and 4.3m (14 ft) deep. The towing carriage is capable of speeds up to 4.88m (16 ft/sec), and models up to 9.1m (30 ft) in length have been tested.

A vertically oscillating wedge-shaped steel tank is installed as a wavemaker. It is electrically actuated, and gives acceptably linear response over a wavelength band from 0.9 to 5.5m (3 to 18 ft), with wave heights up to approximately 0.15m (0.5 ft), depending somewhat on the wavelength involved. For nominally sinusoidal regular waves, input to the electric control system is from an oscillator. Alternatively, random waves can be generated using prerecorded input derived from a given sea spectrum.

Forced-Oscillation Mechanism

The forced-oscillation device was developed in connection with recent ABS-supported experimental work. The device is capable of forcing oscillation in either pitch or heave. The mechanism is electrically actuated, with its frequency controlled by varying the speed of the electric drive. The amplitude of the impressed motion could be varied by discrete increments, using a scotch yoke mechanism.

Model

The model used in this study was that of the SL-7 class containership. Dimensions, weights, hull-form coefficients, and pitch gyradii for the ship and model are shown in Table I. A body plan of the ship is shown in Figure 1.

TABLE I: SL-7 Hull Particulars

| | Ship | Model |
|--------------------------|--------------|---------------|
| LOA m(ft) | 288.5(946.6) | 3.607(11.83) |
| LBP m(ft) | 268.4(880.5) | 3.355(11.01) |
| Beam m(ft) | 32.16(105.5) | 0.402(1.319) |
| Draft @ LCF m (ft) | 9.94(32.6) | 0.124(.408) |
| Trim by stern mm(ft) | 43.0(0.14) | 0.53(0.0018) |
| LCG Aft of O m(ft) | 11.7(38.4) | 0.146(.480) |
| Displacement MT(LT) | 48364(47500) | 0.0945(.0928) |
| Pitch Radius of Gyration | 0.21 LBP | 0.21 LBP |

Heave and Pitch Measurements

For the forced-oscillation tests, the vertical displacement at a known location of the model was measured by a linear voltage differential transformatal (LVDT). This displacement, then, could be easily related to heave and pitch motions.

Wave Probes

Two types of wave probes were used in the experiments. For the diffraction tests, the incident wave was measured by a Wesmar Ultrasonic Level Monitor, Model LM4000 sonic wave probe. It was fitted under the forward end of the towing carriage, approximately 2.1m (7 ft) ahead of the model. For the relative motion measurements, a capacitance-type wave probe was used. It is described below.

Capacitance wave probes have certain advantages over other commonly used water level measurement devices, i.e. resistance and sonic probes. As de-

scribed by Luft (1968), they require no aging, show only moderate sensitivity to dirt, and several probes can be operated close to each other simultaneously without interference. In the past, disadvantages included the choice of material used for the dielectric and a relatively complex excitation circuit.

While carrying out the terms of an ONR/GHR contract (N00014-78-C-0109) contract, a different method of constructing and using capacitance probes was found. Following an IC design suggested by Pipil and Curzon (1979) for use in measuring standing or capillary waves in a basin, a modified circuit was produced for the ship motion experiments. The circuit is simple, inexpensive, and reliable. It is built around an IC chip used in automobile tachometers. Ten such circuits were manufactured for the experiments described in this report.

The probes were made of 32 gauge magnet wire and supported away from the model by small poles. The distance that the wires were from the hull surface averaged 3-4mm (0.12-0.16 in). The non-linear gain of the probes due to the changing hull shape was factored into the computer program that did the data reduction.

Data Recording

The entire measurement system was calibrated twice a day during the experimental program. A schematic showing the instrumentation set up is shown in Figure 2. On each run, twelve channels of data were sampled. Ten were capacitance wave probes, one was the incident wave probe or shaker LVDT, and one was the carriage speed tachometer. The records were Fourier analyzed and the results printed out between the runs. Interim plots were kept in order to identify general trends and bad data.

EXPERIMENTAL RESULTS

The test matrix for the SL-7 model is shown in Table II. The regular waves covered a range of wavelength to ship length ratios of 0.25 to 1.9. Some of the data points were lost due to infrequent hardware difficulties. These are represented by gaps in the curves. Only one forward speed was used for the diffraction experiments.

TABLE II: Model Test Conditions

| | Froude number | Regular waves | Transient waves |
|-------------|---------------|---------------|-----------------|
| | 0.1 0.2 0.3 | | |
| Diffraction | X | X | X |
| Radiation | X X X | X | |

The longitudinal location of the wave probes are shown in Figure 3. Also shown are the directions of positive heave, pitch, and relative motion. Heave is defined as the vertical displacement of the model measured at midship, and pitch as the angular rotation about an axis located at the intersection of the water plane and the midship section. Heave is positive up, pitch is positive bow down, and relative motion is positive for the water surface above the mean water line. (Positive heave produces a negative relative motion.)

The actual tests results are presented in tabular and graphical form. In the following figures, the crosses represent the unfaired data. (The points are connected by straight line segments for visual effect.) The following non-dimensional scheme was used:

Forced Heave -

$$\text{Relative motion} = \frac{\text{relative motion amplitude}}{\text{heave amplitude}}$$

$$\text{Phase Angle} = \text{Arg(rel. motion)} - \text{Arg(heave)}$$

Forced Pitch -

$$\text{Relative motion} = \frac{\text{relative motion amplitude}}{\text{amplitude of vertical displacement at Station 20}}$$

$$\text{Phase Angle} = \text{Arg(rel. motion)} - \text{Arg(pitch)}$$

Diffraction -

$$\text{Relative motion} = \frac{\text{relative motion amplitude}}{\text{incident wave amplitude}}$$

$$\text{Phase Angle} = \text{Arg(rel. motion)} - \text{Arg(incident wave)}$$

The argument of the incident wave is zero when a wave trough is at midships.

The actual data points for the various runs are listed in Table III. The magnitudes for the time dependent relative motions are normalized as described above. The approximate excitation amplitude listed in the table refers to the heave amplitude for forced heave, the amplitude of the vertical displacement at station 20 for forced pitch, and the incident wave amplitude for the diffraction tests. The units of the excitation amplitudes are inches. The steady state wave profiles in inches are also listed.

The figures that follow are the plotted results listed in Table III. The relative motion is given as a function of wavelength to ship length for various stations, and again as a function of longitudinal location along the hull for various wavelength to ship lengths. This method of cross plotting helps illustrate a number of interesting points. The order that the plots

appear is shown below:

- i) Forced Heave . Fn = 0.1
- ii) Forced Pitch . Fn = 0.1
- iii) Forced Heave . Fn = 0.2
- iv) Forced Pitch . Fn = 0.2
- v) Forced Heave . Fn = 0.3
- vi) Forced Pitch . Fn = 0.3
- vii) Forced Heave . Fn = 0.2 (linearity test)
- viii) Forced Pitch . Fn = 0.2 (linearity and repeatability tests)
- ix) Diffraction . Fn = 0.2 (regular waves)
- x) Diffraction . Fn = 0.2 (transient test, three different runs)
- xi) Steady Wave Profiles.

CONCLUSIONS

There are large increases in the relative motion magnitudes for $F_n = 0.1$ when wavelength to ship length equals 1.445. The parameter that compares the wave period to hull speed, i.e. $(\text{frequency of encounter}) * (\text{vessel speed}) / (\text{gravitational constant})$, has a value of 1/4 for this case. This implies that the radiated wave speed and hull speed are nearly equal and tank-wall reflection becomes a major problem. Consequently, the results for $F_n = 0.1$ with wavelengths to ship lengths ratios equal to and greater than 1.445 are contaminated with tank-wall reflection effects. For the higher Froude numbers, the parameter was greater than 1/4 in all wavelengths tested.

The forced pitch records show that the point of rotation, as far as relative motion is concerned, is at station 12, rather than station 10. The plots show that the relative motion magnitude reaches a minimum there. Also, the phase angles change 180° from stations 14 to 10 indicating when the relative motion at station 14 is positive, it is negative at station 10.

Both repeatability and linearity checks were run for the forced oscillation tests. The repeatability appeared to be adequate. For the linearity tests, the heave amplitude was increased by 50% and the pitch amplitude was more than doubled. No significant difference was observed for heave, but the larger pitch motions seemed to reduce the pure pitch relative motion response amplitude operator. This was particularly true in the bow region where interactions between the hull shape, steady state wave, and time dependent wave are more pronounced.

The transient test results were inconclusive. Three different runs were made, each meeting the wave group in a different location in the tank. The overall comparison with the regular wave results is not good. More work is needed in this area to identify the cause of disagreement.

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Table III: Tabulated Results of Relative Motion Experiments

DATA FOR FORCED HEAVE FN=0.1
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.00

| MAGNITUDES | LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.5 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 12.0 | STA. 10.0 |
|------------|----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| 1.883 | 0.737 | 0.911 | 0.895 | 0.759 | 0.731 | 0.788 | 0.749 | 0.784 | 0.878 | 0.990 | | | | |
| 1.445 | 1.059 | 1.593 | 1.839 | 2.065 | 1.877 | 1.751 | 1.715 | 1.538 | 1.434 | 1.409 | | | | |
| 1.152 | 0.907 | 1.109 | 1.242 | 1.209 | 1.173 | 1.099 | 1.055 | 1.120 | 1.249 | 1.163 | | | | |
| 0.865 | 0.865 | 1.038 | 1.202 | 1.177 | 1.180 | 1.165 | 1.085 | 0.973 | 1.027 | 1.027 | | | | |
| 0.734 | 0.956 | 1.128 | 1.290 | 1.256 | 1.235 | 1.230 | 1.141 | 1.043 | 1.059 | 1.119 | | | | |
| 0.634 | 0.882 | 1.035 | 1.221 | 1.205 | 1.211 | 1.194 | 1.145 | 1.088 | 1.139 | 1.217 | | | | |
| 0.376 | 0.882 | 1.005 | 1.221 | 1.229 | 1.322 | 1.340 | 1.392 | 1.451 | 1.532 | 1.575 | | | | |
| 0.257 | 0.887 | 0.969 | 1.141 | 1.147 | 1.248 | 1.393 | 1.532 | 1.754 | 1.754 | 1.754 | | | | |

DATA FOR FORCED PITCH FN=0.1
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.44

| MAGNITUDES | LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.5 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 12.0 | STA. 10.0 |
|------------|----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| 1.883 | 1.323 | 1.638 | 1.427 | 1.275 | 1.077 | 0.977 | 0.846 | 0.383 | 0.151 | 0.260 | | | | |
| 1.445 | 1.585 | 2.105 | 1.922 | 1.740 | 1.370 | 1.104 | 0.912 | 0.371 | 0.408 | 0.638 | | | | |
| 1.152 | 1.010 | 1.285 | 1.158 | 1.102 | 0.964 | 0.763 | 0.701 | 0.384 | 0.231 | 0.414 | | | | |
| 0.865 | 1.008 | 1.270 | 1.148 | 1.129 | 1.004 | 0.814 | 0.713 | 0.273 | 0.076 | 0.146 | | | | |
| 0.734 | 0.942 | 1.201 | 1.075 | 1.079 | 0.956 | 0.783 | 0.671 | 0.273 | 0.091 | 0.166 | | | | |
| 0.634 | 1.003 | 1.262 | 1.143 | 1.116 | 1.023 | 0.824 | 0.746 | 0.339 | 0.134 | 0.132 | | | | |
| 0.376 | 1.002 | 1.255 | 1.147 | 1.144 | 1.083 | 0.939 | 0.909 | 0.485 | 0.177 | 0.175 | | | | |
| 0.248 | 0.983 | 1.147 | 0.947 | 0.965 | 0.884 | 0.857 | 1.109 | 0.502 | 0.127 | 0.244 | | | | |

Table III: Tabulated Results of Relative Motion Experiments (con't.)

DATA FOR FORCED HEAVE FN=0.2
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.00

| LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 12.0 | STA. 10.0 |
|--------------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| 1.503 | 0.894 | 1.260 | 1.086 | 1.192 | 1.331 | 1.101 | 1.012 | 1.130 | 1.063 | 1.057 | | |
| 1.503 | 0.901 | 1.259 | 1.110 | 1.198 | 1.333 | 1.192 | 1.012 | 1.110 | 1.040 | 1.040 | | |
| 1.154 | 0.890 | 1.257 | 1.068 | 1.176 | 1.349 | 1.103 | 0.986 | 1.176 | 1.127 | 1.157 | | |
| 0.867 | 0.861 | 1.147 | 1.027 | 1.150 | 1.357 | 1.132 | 1.068 | 1.270 | 1.252 | 1.304 | | |
| 0.766 | 0.866 | 0.926 | 1.001 | 1.132 | 1.341 | 1.204 | 1.240 | 1.353 | 1.383 | 1.424 | | |
| 0.618 | 0.904 | 0.967 | 1.008 | 1.166 | 1.395 | 1.418 | 1.493 | 1.480 | 1.561 | 1.634 | | |
| 0.379 | 0.907 | 0.913 | 0.911 | 1.066 | 1.284 | 1.510 | 1.666 | 1.963 | 1.991 | 1.899 | | |
| 0.248 | 0.865 | 0.872 | 0.668 | 0.683 | 0.810 | 1.192 | 1.625 | 2.126 | 2.052 | 1.757 | | |
| PHASE ANGLES | LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 10.0 |
| 1.503 | 174.412 | 168.971 | 166.400 | 167.589 | 175.628 | 179.497 | 184.206 | 197.605 | 202.264 | 201.403 | | |
| 1.503 | 175.282 | 169.961 | 166.620 | 168.379 | 176.488 | 180.707 | 179.860 | 197.885 | 201.964 | 201.223 | | |
| 1.154 | 173.336 | 166.822 | 164.098 | 165.735 | 174.071 | 176.757 | 183.734 | 200.340 | 204.906 | 203.583 | | |
| 0.867 | 175.734 | 169.240 | 165.096 | 167.012 | 174.887 | 178.773 | 188.349 | 204.964 | 210.500 | 205.846 | | |
| 0.766 | 172.902 | 168.204 | 161.385 | 163.556 | 170.637 | 171.118 | 186.459 | 204.040 | 209.982 | 204.283 | | |
| 0.618 | 171.401 | 165.762 | 157.592 | 159.733 | 166.253 | 171.784 | 183.684 | 203.935 | 208.985 | 202.216 | | |
| 0.379 | 169.635 | 165.635 | 157.085 | 156.085 | 161.855 | 168.415 | 185.995 | 199.775 | 196.985 | 184.076 | | |
| 0.248 | 167.461 | 167.209 | 153.156 | 148.513 | 157.660 | 167.857 | 185.984 | 189.691 | 188.538 | 181.025 | | |

DATA FOR FORCED PITCH FN=0.2
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.44

| LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 12.0 | STA. 10.0 |
|--------------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| 1.852 | 1.138 | 1.109 | 1.121 | 1.109 | 1.133 | 0.964 | 0.681 | 0.317 | 0.044 | 0.224 | | |
| 1.503 | 1.070 | 1.116 | 1.099 | 1.098 | 1.146 | 0.975 | 0.766 | 0.323 | 0.048 | 0.210 | | |
| 1.154 | 1.082 | 1.181 | 1.083 | 1.081 | 1.180 | 1.001 | 0.0 | 0.353 | 0.094 | 0.217 | | |
| 0.867 | 1.062 | 1.139 | 1.065 | 1.070 | 1.204 | 0.990 | 0.692 | 0.397 | 0.119 | 0.200 | | |
| 0.766 | 1.054 | 1.126 | 1.041 | 1.064 | 1.209 | 1.031 | 0.792 | 0.417 | 0.133 | 0.171 | | |
| 0.618 | 1.070 | 1.147 | 1.035 | 1.138 | 1.191 | 1.080 | 0.870 | 0.485 | 0.179 | 0.154 | | |
| 0.379 | 0.998 | 1.069 | 0.898 | 1.022 | 1.077 | 1.014 | 1.025 | 0.646 | 0.211 | 0.261 | | |
| PHASE ANGLES | LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 10.0 |
| 1.851 | 173.464 | 174.938 | 169.021 | 172.075 | 179.948 | 183.172 | 194.555 | 200.269 | 206.932 | 1.026 | | |
| 1.503 | 174.662 | 173.371 | 166.730 | 170.269 | 178.188 | 180.907 | 188.166 | 201.765 | 246.134 | 2.313 | | |
| 1.154 | 175.206 | 174.092 | 166.258 | 171.145 | 176.471 | 181.347 | 119.154 | 212.100 | 268.646 | 353.573 | | |
| 0.867 | 172.994 | 172.550 | 162.846 | 169.312 | 170.287 | 177.013 | 196.709 | 212.154 | 254.440 | 354.156 | | |
| 0.766 | 169.042 | 169.154 | 159.335 | 165.936 | 165.677 | 175.108 | 193.409 | 210.240 | 241.222 | 355.393 | | |
| 0.618 | 169.551 | 168.102 | 158.872 | 161.343 | 165.783 | 174.814 | 195.374 | 210.945 | 234.355 | 357.476 | | |
| 0.379 | 171.695 | 165.015 | 155.905 | 154.095 | 159.025 | 164.885 | 186.475 | 204.795 | 196.115 | 13.295 | | |

Table III: Tabulated Results of Relative Motion Experiments (con't.)

DATA FOR FORCED HEAVE FN=0.3
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.00

| MAGNITUDES | | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 12.0 | STA. 10.0 |
|--------------|---------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| LAMBDA/L | | | | | | | | | | | | | |
| 1.831 | 0.963 | 0.971 | 1.159 | 1.309 | 1.340 | 1.504 | 1.446 | 1.358 | 1.248 | 1.171 | | | |
| 1.541 | 0.873 | 0.889 | 1.005 | 1.171 | 1.200 | 1.374 | 1.337 | 1.256 | 1.152 | 1.100 | | | |
| 1.156 | 0.950 | 0.865 | 1.119 | 1.279 | 1.327 | 1.557 | 1.568 | 1.546 | 1.431 | 1.426 | | | |
| 0.867 | 0.973 | 0.917 | 1.106 | 1.233 | 1.278 | 1.562 | 1.598 | 1.661 | 1.602 | 1.623 | | | |
| 0.751 | 0.974 | 0.839 | 1.070 | 1.204 | 1.263 | 1.555 | 1.639 | 1.738 | 1.784 | 1.800 | | | |
| 0.633 | 1.018 | 0.809 | 1.094 | 1.198 | 1.253 | 1.586 | 1.706 | 1.973 | 2.033 | 2.023 | | | |
| 0.381 | 0.949 | 0.808 | 0.950 | 0.990 | 1.140 | 1.426 | 1.747 | 2.405 | 2.337 | 2.125 | | | |
| PHASE ANGLES | | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 12.0 | STA. 10.0 |
| LAMBDA/L | | | | | | | | | | | | | |
| 1.831 | 173.072 | 167.281 | 167.840 | 165.909 | 171.238 | 179.727 | 188.876 | 188.005 | 202.864 | 201.053 | | | |
| 1.541 | 166.080 | 158.805 | 161.969 | 159.463 | 165.878 | 173.992 | 181.557 | 194.181 | 200.485 | 200.250 | | | |
| 1.156 | 174.859 | 167.813 | 167.036 | 166.470 | 171.484 | 180.178 | 191.882 | 204.455 | 210.519 | 209.443 | | | |
| 0.867 | 168.340 | 162.417 | 162.523 | 160.340 | 167.537 | 173.703 | 186.040 | 200.896 | 208.673 | 204.230 | | | |
| 0.751 | 168.444 | 163.558 | 162.432 | 160.926 | 167.990 | 171.684 | 186.148 | 202.112 | 210.236 | 203.340 | | | |
| 0.633 | 167.462 | 167.386 | 163.039 | 161.612 | 169.656 | 173.439 | 186.313 | 202.316 | 208.069 | 200.173 | | | |
| 0.381 | 166.125 | 165.987 | 155.319 | 155.442 | 163.304 | 166.776 | 180.779 | 186.901 | 185.013 | 171.975 | | | |

DATA FOR FORCED PITCH FN=0.3
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.44

| MAGNITUDES | | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 12.0 | STA. 10.0 |
|--------------|---------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| LAMBDA/L | | | | | | | | | | | | | |
| 1.831 | 1.065 | 1.230 | 1.084 | 1.120 | 1.059 | 0.996 | 0.778 | 0.351 | 0.074 | 0.202 | | | |
| 1.541 | 1.017 | 1.142 | 1.053 | 1.083 | 1.026 | 0.996 | 0.787 | 0.330 | 0.059 | 0.230 | | | |
| 1.156 | 0.984 | 1.165 | 1.036 | 1.037 | 0.991 | 1.015 | 0.793 | 0.389 | 0.169 | 0.311 | | | |
| 0.867 | 1.027 | 1.190 | 1.056 | 1.040 | 1.010 | 1.089 | 0.881 | 0.487 | 0.172 | 0.231 | | | |
| 0.751 | 1.013 | 1.167 | 1.064 | 1.027 | 1.012 | 1.107 | 0.881 | 0.531 | 0.194 | 0.188 | | | |
| 0.633 | 1.010 | 1.125 | 1.034 | 0.987 | 0.999 | 1.094 | 0.968 | 0.595 | 0.240 | 0.171 | | | |
| 0.391 | 0.990 | 0.985 | 0.795 | 0.753 | 0.773 | 0.752 | 1.206 | 0.732 | 0.211 | 0.346 | | | |
| PHASE ANGLES | | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 13.0 | STA. 12.0 | STA. 10.0 |
| LAMBDA/L | | | | | | | | | | | | | |
| 1.831 | 174.722 | 175.031 | 172.680 | 169.879 | 173.078 | 185.497 | 199.796 | 205.755 | 256.984 | 2.173 | | | |
| 1.541 | 173.070 | 172.715 | 170.089 | 166.963 | 171.008 | 182.762 | 197.097 | 205.711 | 264.725 | 6.620 | | | |
| 1.156 | 177.369 | 176.833 | 172.996 | 169.710 | 175.384 | 187.508 | 192.212 | 229.305 | 292.049 | 347.263 | | | |
| 0.867 | 172.850 | 171.527 | 167.443 | 163.580 | 169.787 | 178.543 | 190.780 | 216.486 | 259.983 | 337.910 | | | |
| 0.751 | 171.234 | 167.398 | 163.032 | 159.696 | 167.760 | 175.004 | 185.498 | 211.072 | 248.176 | 339.240 | | | |
| 0.633 | 172.012 | 167.166 | 162.489 | 161.252 | 168.936 | 172.509 | 189.483 | 210.176 | 238.589 | 347.573 | | | |
| 0.391 | 168.579 | 167.740 | 162.100 | 163.340 | 168.151 | 172.841 | 174.571 | 202.512 | 208.552 | 359.872 | | | |

Table III: Tabulated Results of Relative Motion Experiments (con't..)

DATA FOR DIFFRACTION FN=0.2
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.00

| MAGNITUDES | LAMBDA/L | STA. 20.0 | STA. 19.0 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 12.0 | STA. 10.0 |
|------------|----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| 1.845 | 0.920 | 1.068 | 1.206 | 1.153 | 1.195 | 1.311 | 1.113 | 1.107 | 1.043 | 0.955 |
| 1.503 | 0.916 | 1.045 | 1.227 | 1.121 | 1.183 | 1.217 | 1.199 | 1.082 | 1.029 | 0.929 |
| 1.154 | 0.948 | 0.978 | 1.271 | 1.131 | 1.242 | 1.366 | 1.278 | 1.133 | 1.076 | 0.960 |
| 0.926 | 1.013 | 1.082 | 1.300 | 1.150 | 1.302 | 1.352 | 1.276 | 1.150 | 1.088 | 0.968 |
| 0.766 | 0.973 | 1.158 | 1.265 | 1.098 | 1.273 | 1.356 | 1.259 | 1.139 | 1.066 | 0.960 |
| 0.650 | 0.964 | 1.121 | 1.254 | 1.105 | 1.283 | 1.347 | 1.255 | 1.160 | 1.086 | 0.970 |
| 0.406 | 1.003 | 1.274 | 1.232 | 1.149 | 1.339 | 1.426 | 1.410 | 1.333 | 1.244 | 1.103 |
| 0.261 | 0.983 | 1.372 | 1.097 | 1.065 | 1.352 | 1.433 | 1.583 | 1.336 | 1.244 | 1.122 |

DATA FOR FORCED HEAVE FN=0.2
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.50

| MAGNITUDES | LAMBDA/L | STA. 20.0 | STA. 19.0 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 15.0 | STA. 14.0 | STA. 12.0 | STA. 10.0 |
|------------|----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| 1.852 | 1.020 | 0.858 | 1.142 | 1.197 | 1.289 | 1.255 | 1.155 | 1.102 | 1.024 | 0.975 |
| 0.514 | 0.947 | 0.981 | 1.017 | 1.115 | 1.302 | 1.419 | 1.418 | 1.590 | 1.626 | 1.685 |
| 0.306 | 0.913 | 0.941 | 0.928 | 1.056 | 1.206 | 1.489 | 1.600 | 1.968 | 1.836 | 1.706 |

Table III: Tabulated Results of Relative Motion Experiments (con't.)

DATA FOR FORCED PITCH FN=0.2
THE APPROXIMATE EXCITATION AMPLITUDE IS 3.09

| MAGNITUDES | LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 14.0 | STA. 12.0 | STA. 10.0 |
|------------|----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| | 1.852 | 0.911 | 1.012 | 0.942 | 0.978 | 0.941 | 0.797 | 0.495 | 0.349 | 0.110 | 0.090 |
| | 1.506 | 0.904 | 1.007 | 0.935 | 0.966 | 0.961 | 0.812 | 0.570 | 0.371 | 0.138 | 0.073 |
| | 1.154 | 0.894 | 1.001 | 0.897 | 0.948 | 0.947 | 0.825 | 0.661 | 0.379 | 0.162 | 0.110 |
| | 0.867 | 0.868 | 0.983 | 0.873 | 0.927 | 0.941 | 0.836 | 0.712 | 0.418 | 0.198 | 0.085 |
| | 0.766 | 0.886 | 0.995 | 0.862 | 0.945 | 0.966 | 0.859 | 0.755 | 0.490 | 0.257 | 0.076 |
| | 0.618 | 0.868 | 0.976 | 0.834 | 0.916 | 0.948 | 0.833 | 0.814 | 0.539 | 0.297 | 0.074 |
| | 0.379 | 0.864 | 0.880 | 0.759 | 0.815 | 0.862 | 0.872 | 0.949 | 0.734 | 0.359 | 0.032 |

| MAGNITUDES | LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 14.0 | STA. 12.0 | STA. 10.0 |
|------------|----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| | 1.852 | 169.640 | 172.080 | 168.190 | 171.790 | 180.360 | 185.690 | 192.280 | 199.760 | 207.550 | 1.510 |
| | 1.503 | 170.200 | 170.550 | 166.840 | 170.310 | 179.190 | 184.250 | 191.280 | 202.120 | 213.130 | 345.020 |
| | 1.154 | 170.120 | 171.410 | 166.710 | 170.500 | 179.310 | 186.530 | 198.890 | 214.250 | 237.090 | 326.840 |
| | 0.867 | 165.170 | 168.920 | 163.380 | 167.950 | 175.960 | 183.790 | 198.910 | 216.070 | 234.060 | 310.360 |
| | 0.766 | 167.750 | 167.130 | 162.230 | 166.390 | 173.890 | 180.980 | 199.540 | 215.490 | 227.860 | 283.180 |
| | 0.618 | 167.870 | 166.700 | 161.610 | 165.510 | 172.700 | 181.900 | 198.660 | 218.110 | 227.310 | 266.220 |
| | 0.379 | 166.320 | 162.450 | 158.800 | 159.200 | 162.560 | 172.320 | 189.690 | 210.070 | 203.930 | 69.100 |

DATA FOR FORCED PITCH FN=0.2
THE APPROXIMATE EXCITATION AMPLITUDE IS 1.44

| MAGNITUDES | LAMBDA/L | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 14.0 | STA. 12.0 | STA. 10.0 |
|------------|----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| | 1.852 | 0.941 | 1.211 | 1.099 | 1.131 | 1.115 | 0.784 | 0.561 | 0.277 | 0.029 | 0.173 |
| | 1.154 | 0.995 | 0.954 | 1.062 | 1.097 | 1.117 | 0.739 | 0.514 | 0.278 | 0.060 | 0.195 |
| | 0.867 | 170.120 | 174.830 | 169.420 | 171.210 | 179.850 | 183.280 | 194.430 | 201.960 | 241.000 | 9.250 |
| | 1.155 | 175.270 | 171.910 | 163.830 | 166.190 | 174.040 | 183.280 | 197.420 | 215.080 | 281.510 | 3.330 |

| STEADY WAVE PROFILES | FN | STA. 20.0 | STA. 19.5 | STA. 19.0 | STA. 18.5 | STA. 18.0 | STA. 17.0 | STA. 16.0 | STA. 14.0 | STA. 12.0 | STA. 10.0 |
|----------------------|-----|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|
| | 0.1 | 0.087 | -0.127 | 0.128 | 0.004 | 0.090 | 0.049 | 0.349 | 0.022 | -0.017 | -0.033 |
| | 0.2 | 0.509 | 0.516 | 0.084 | -0.113 | -0.063 | 0.318 | 0.365 | 0.021 | 0.013 | -0.148 |
| | 0.3 | 0.646 | 1.197 | 0.835 | 0.623 | 0.477 | 0.497 | 0.502 | 0.071 | -0.087 | -0.396 |

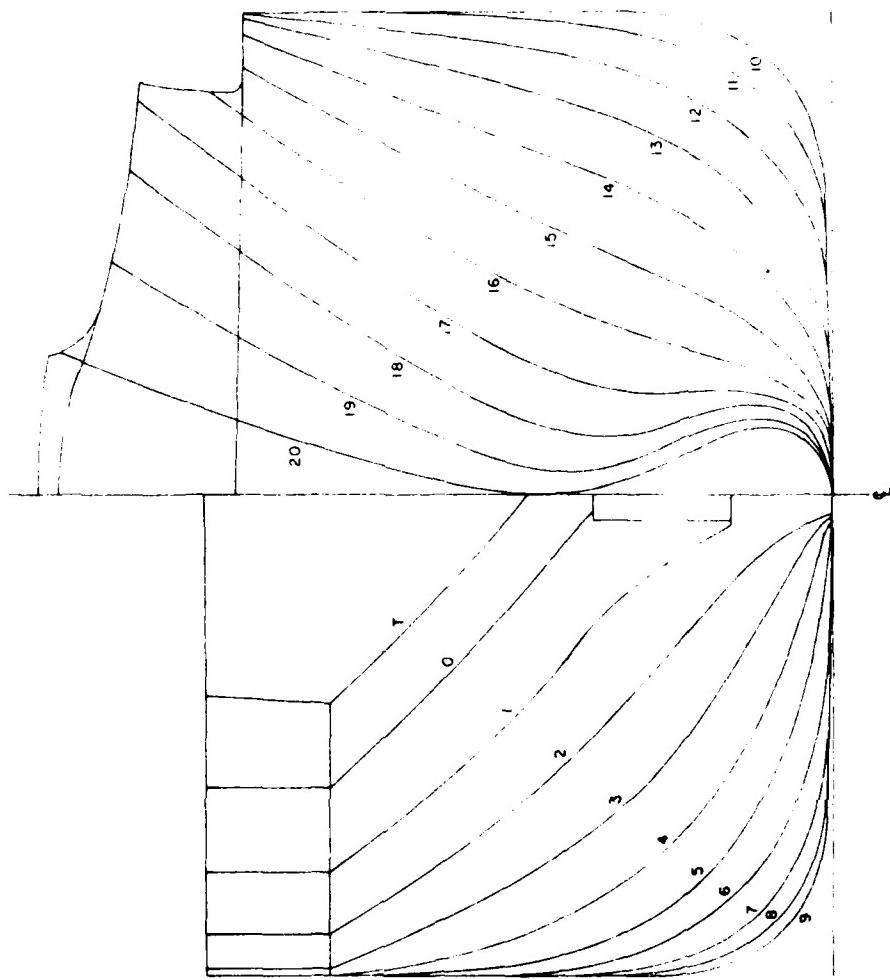


Figure 1: Body Plan of the SL-7 (with station numbers)

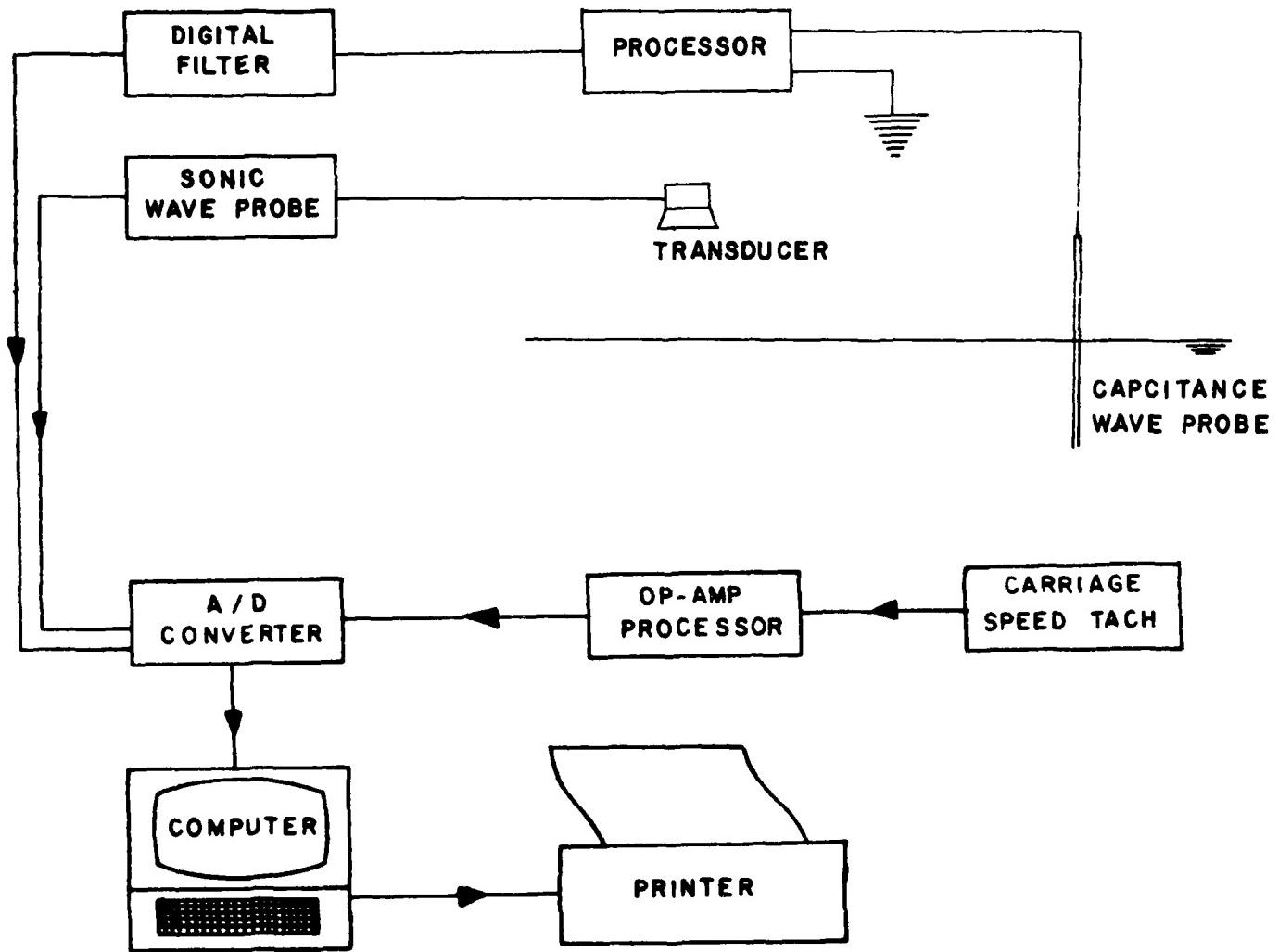


Figure 2: Signal Processing/Data Analysis Configuration

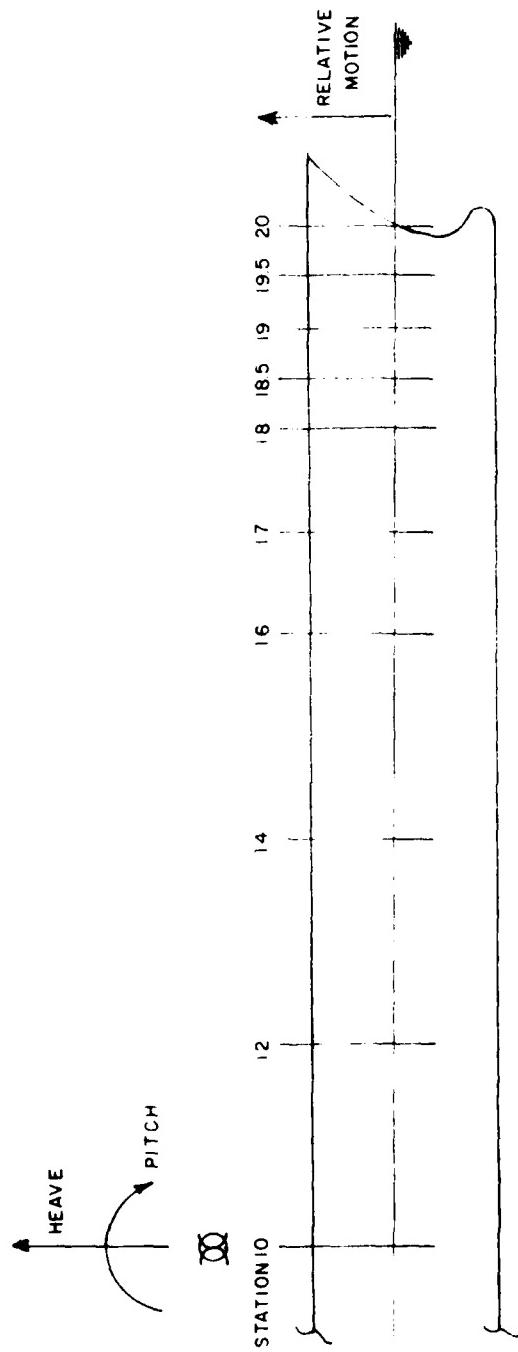
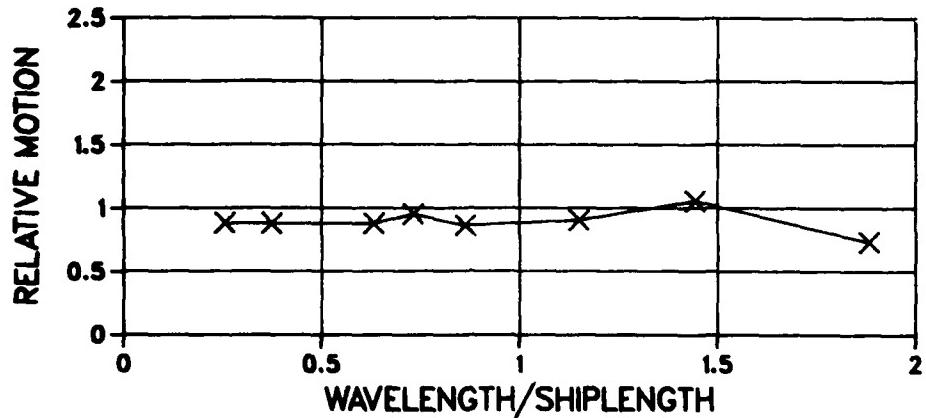


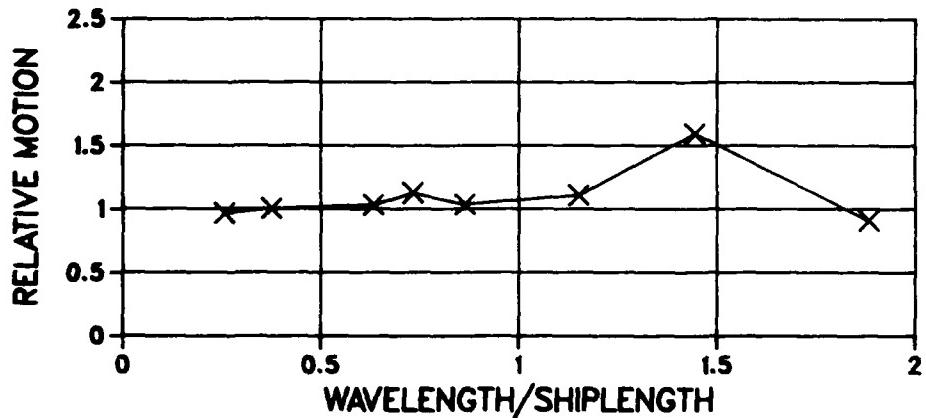
Figure 3: Location of Wave Probes and Definition of Response

FORCED HEAVE. FN=0.1

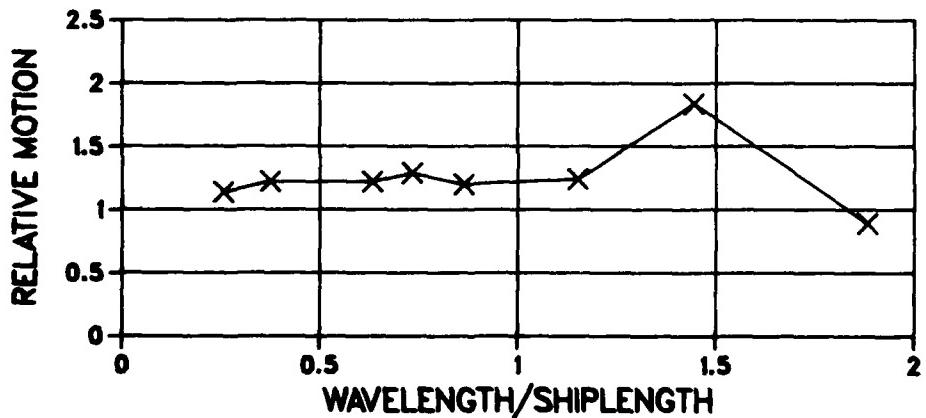
STA.20



STA.19.5

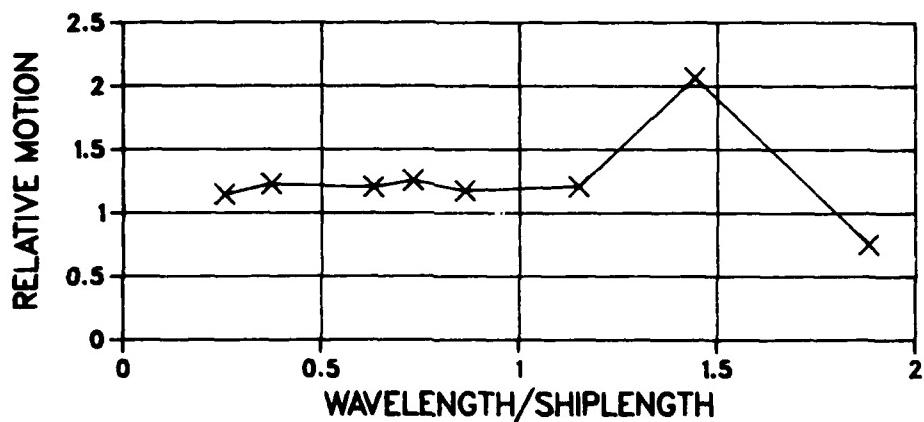


STA.19

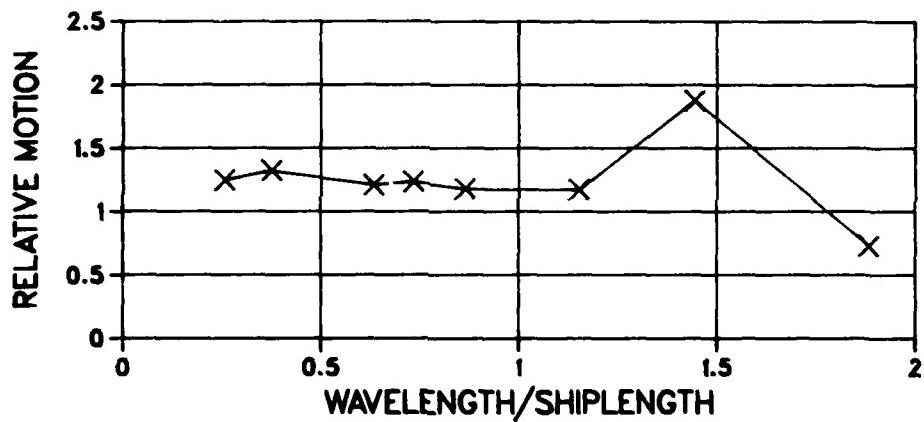


FORCED HEAVE. FN=0.1

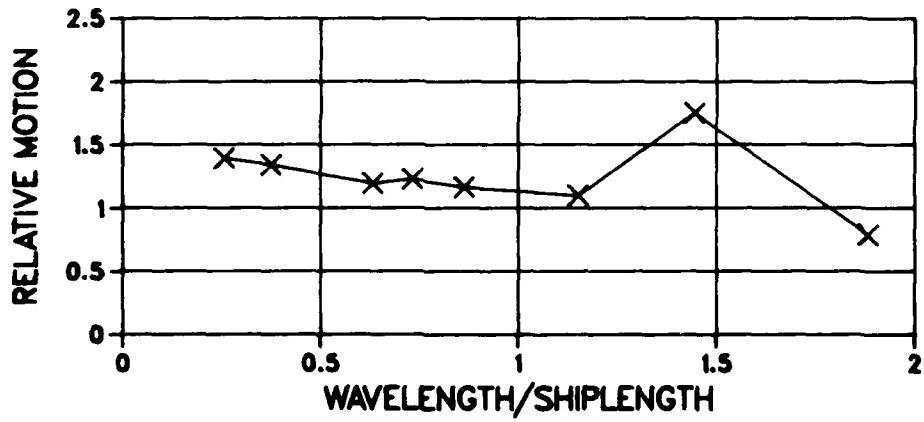
STA.18.5



STA.18

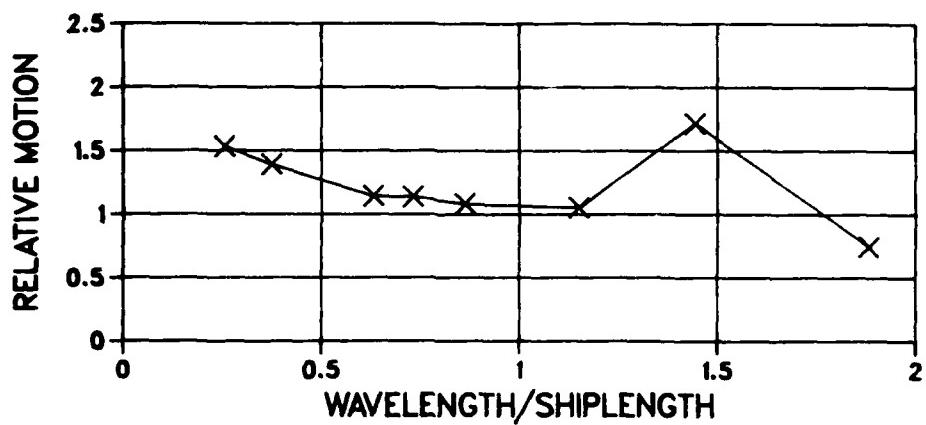


STA.17

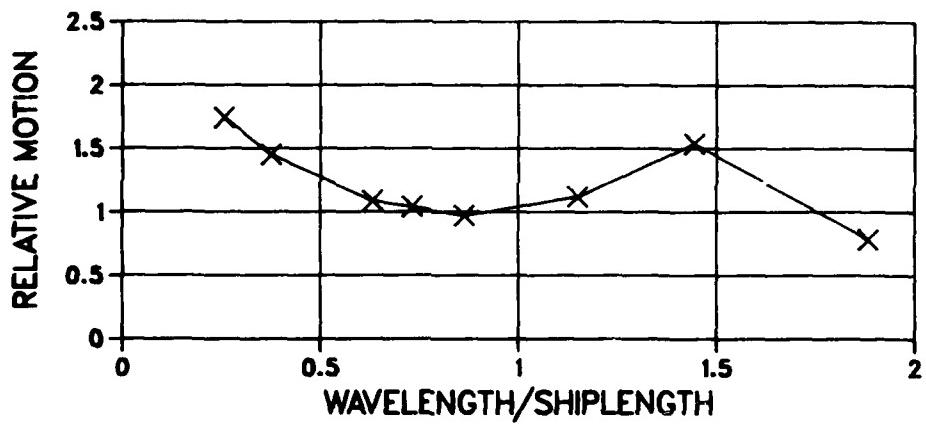


FORCED HEAVE. FN=0.1

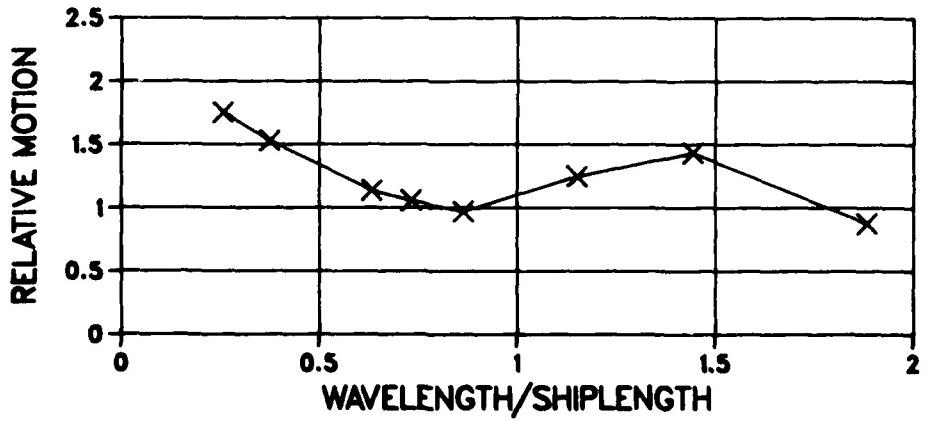
STA.16



STA.14

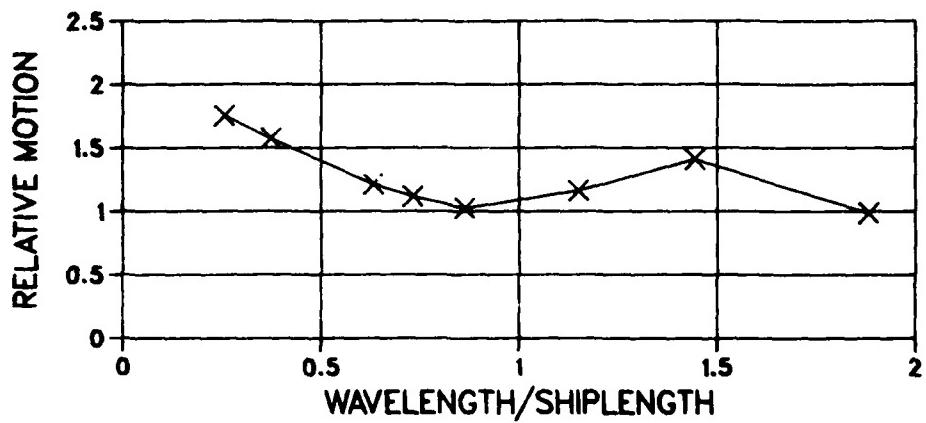


STA.12

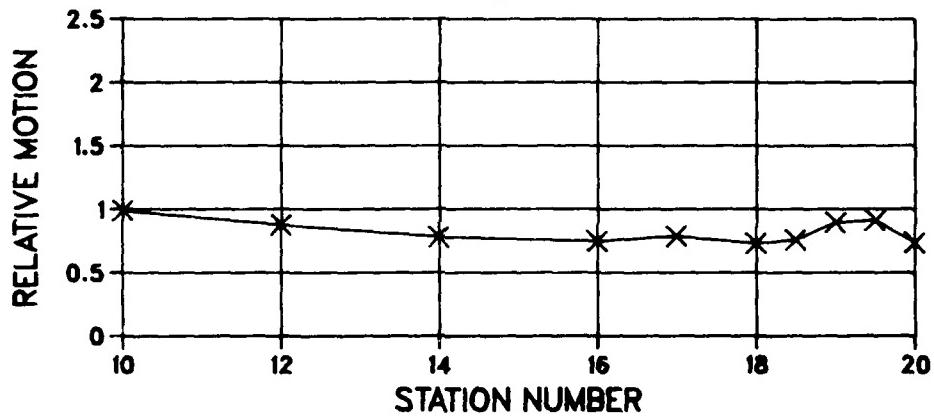


FORCED HEAVE. FN=0.1

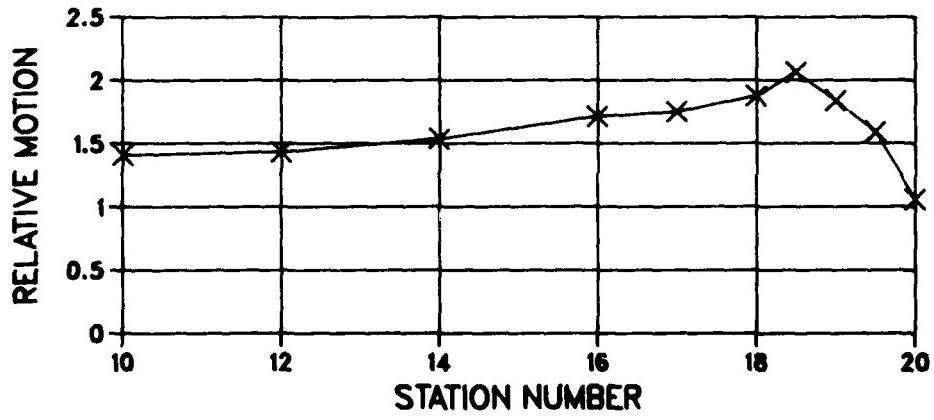
STA.10



LAMBDA/L = 1.883

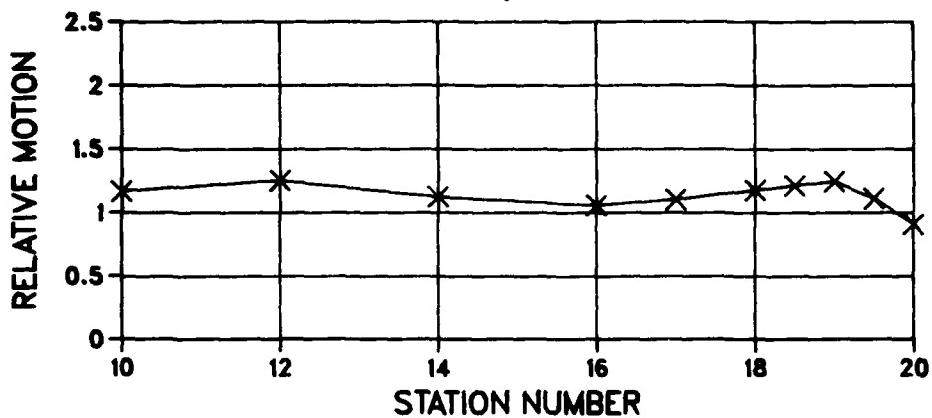


LAMBDA/L = 1.445

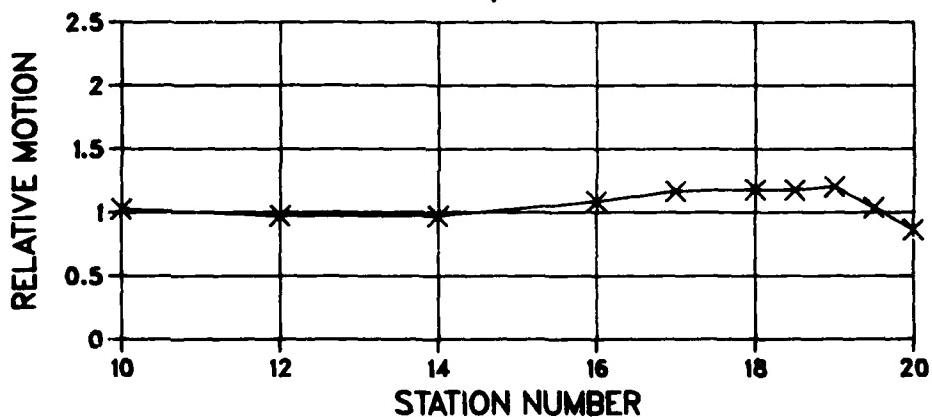


FORCED HEAVE. FN=0.1

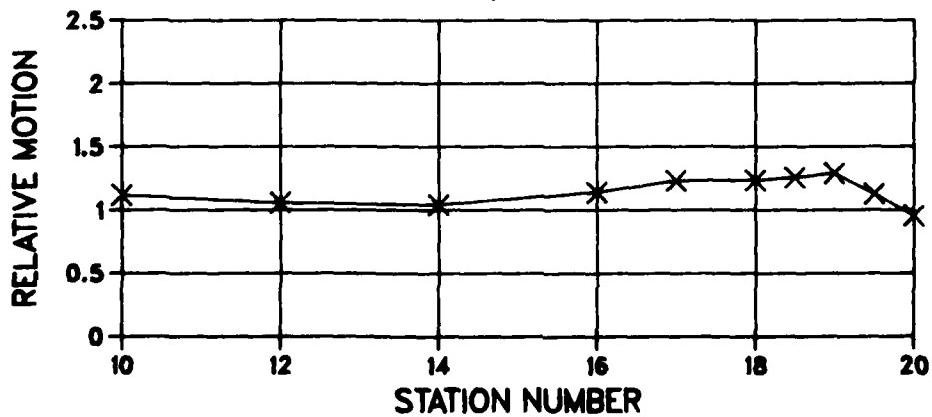
LAMBDA/L = 1.152



LAMBDA/L = 0.865

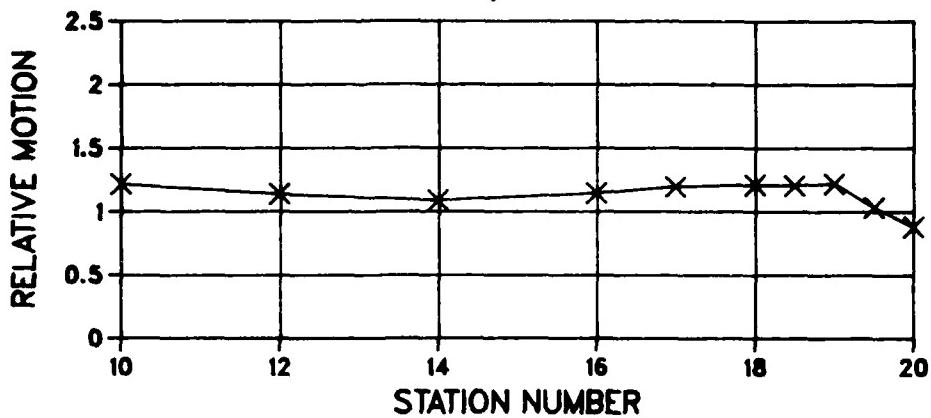


LAMBDA/L = 0.734

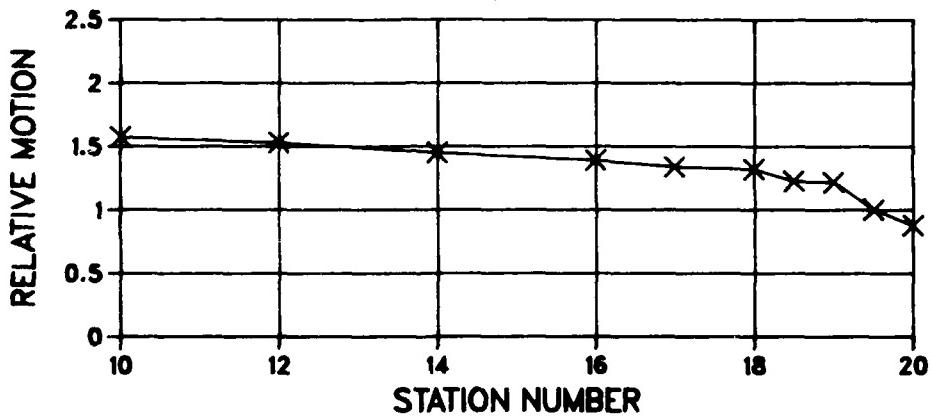


FORCED HEAVE. FN=0.1

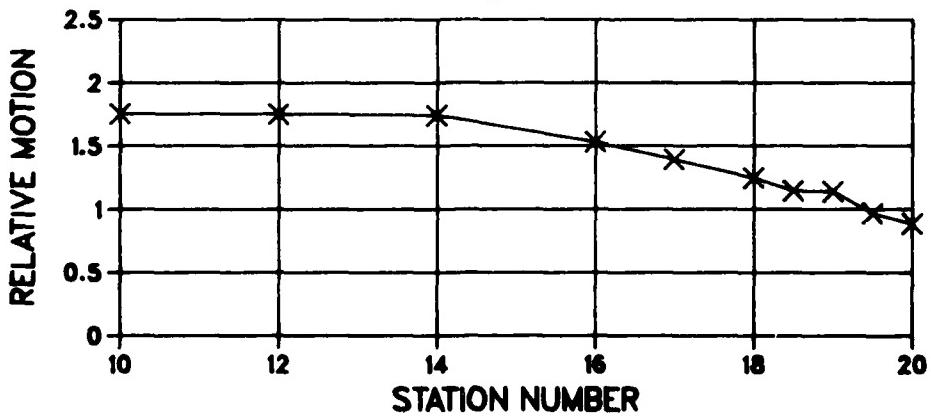
LAMBDA/L = 0.634



LAMBDA/L = 0.376

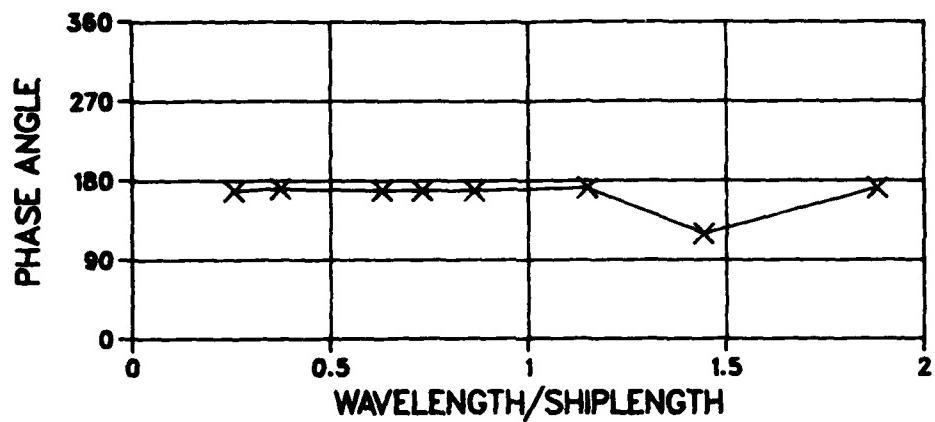


LAMBDA/L = 0.257

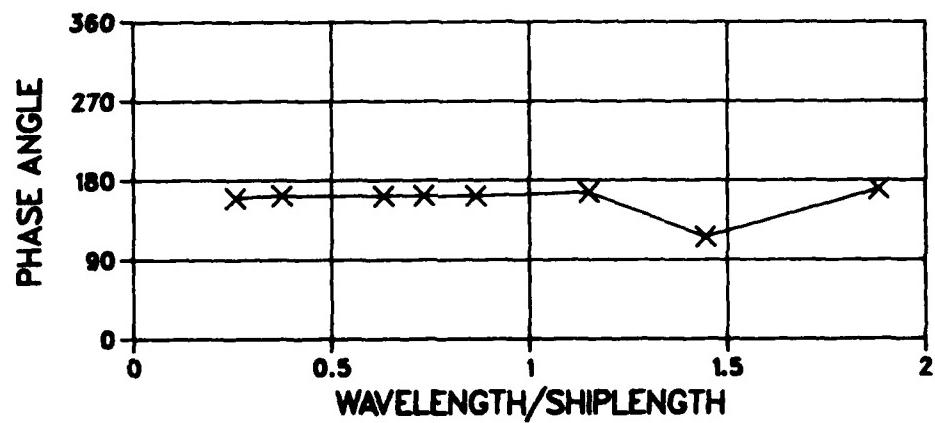


FORCED HEAVE. FN=0.1

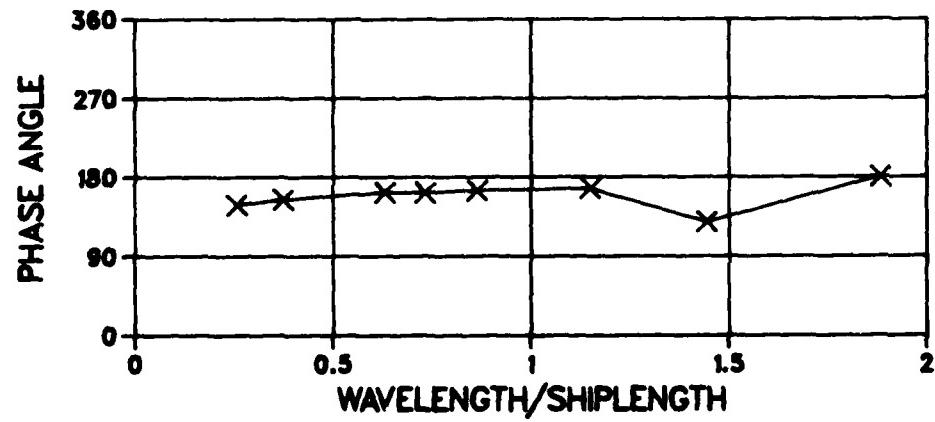
STA.20



STA.19.5

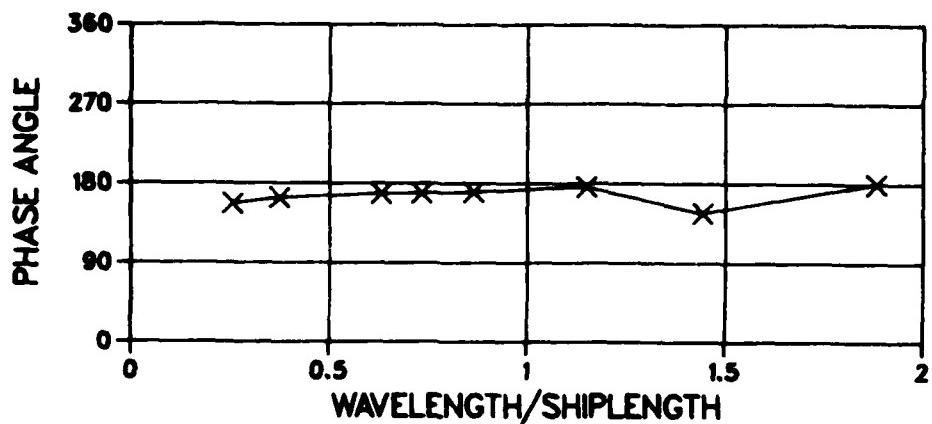


STA.19

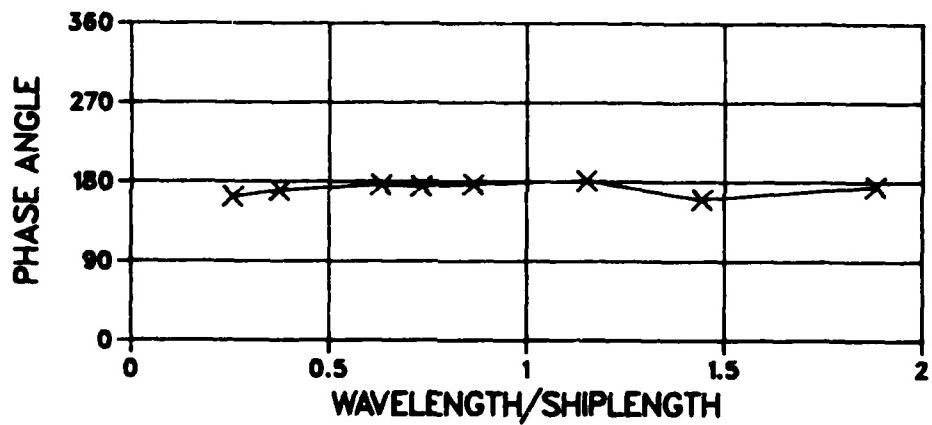


FORCED HEAVE. FN=0.1

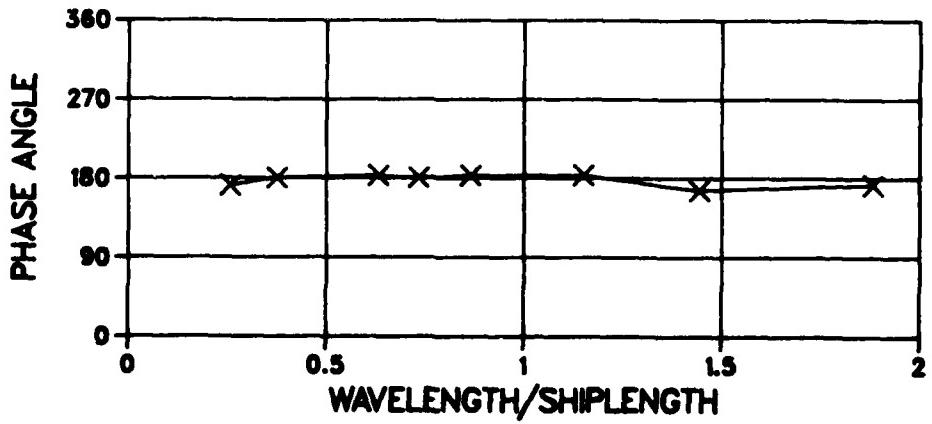
STA.18.5



STA.18

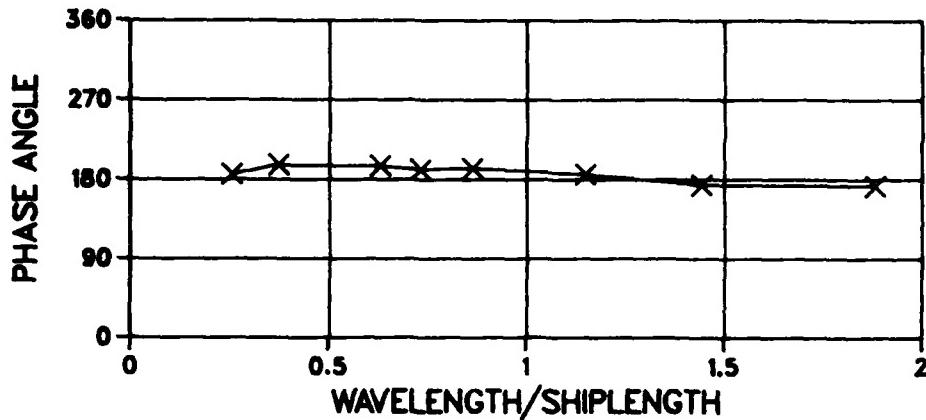


STA.17

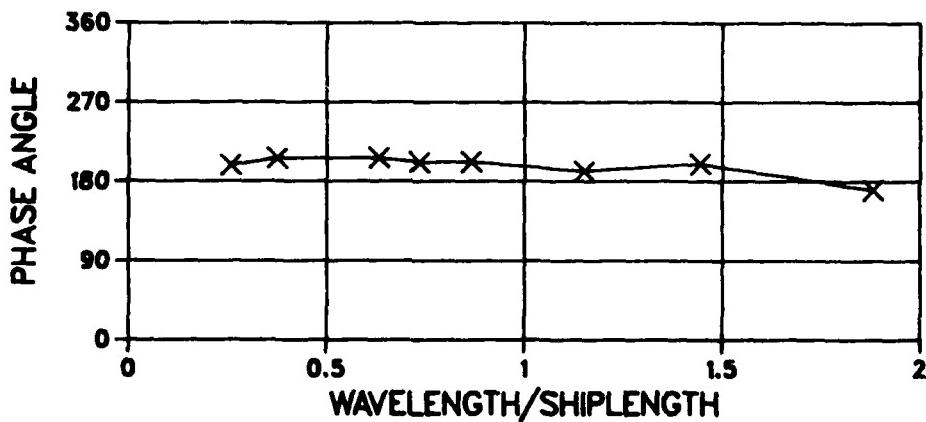


FORCED HEAVE. FN=0.1

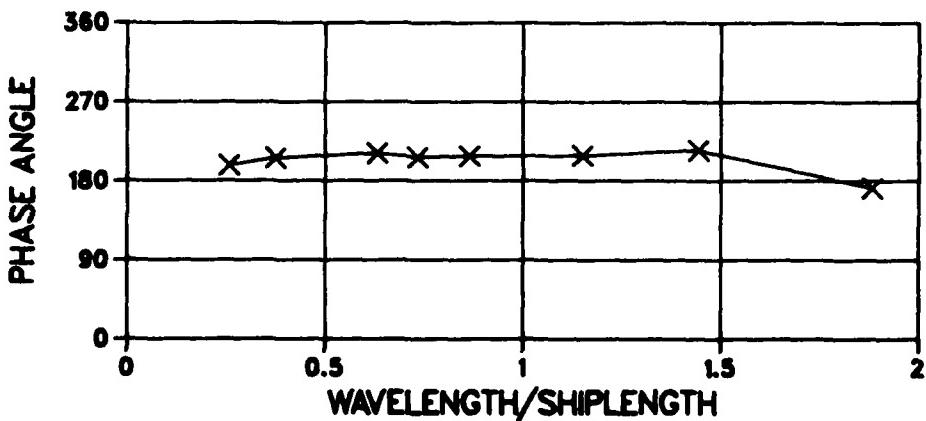
STA.16



STA.14

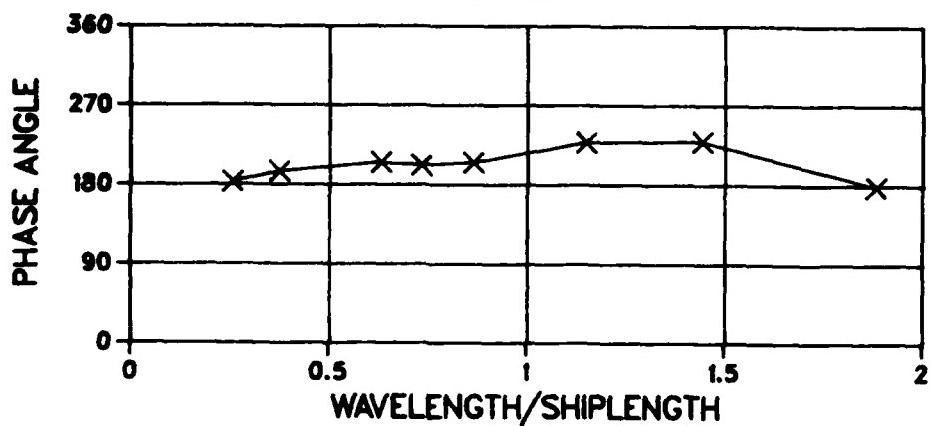


STA.12

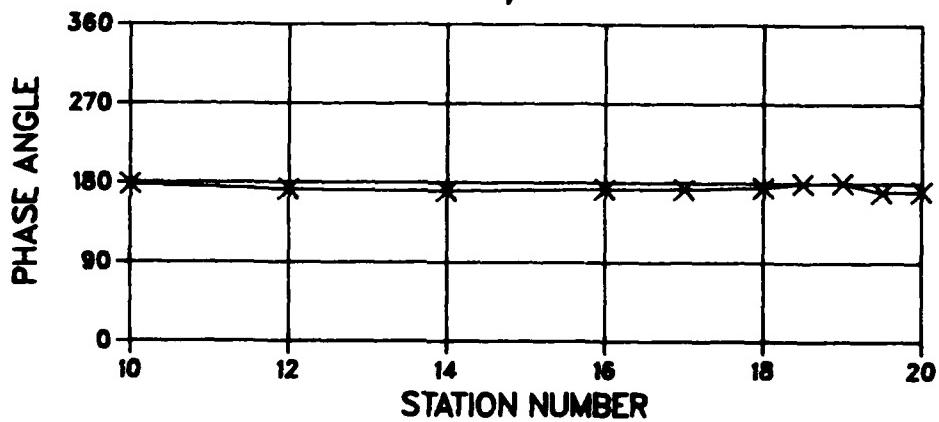


FORCED HEAVE. FN=0.1

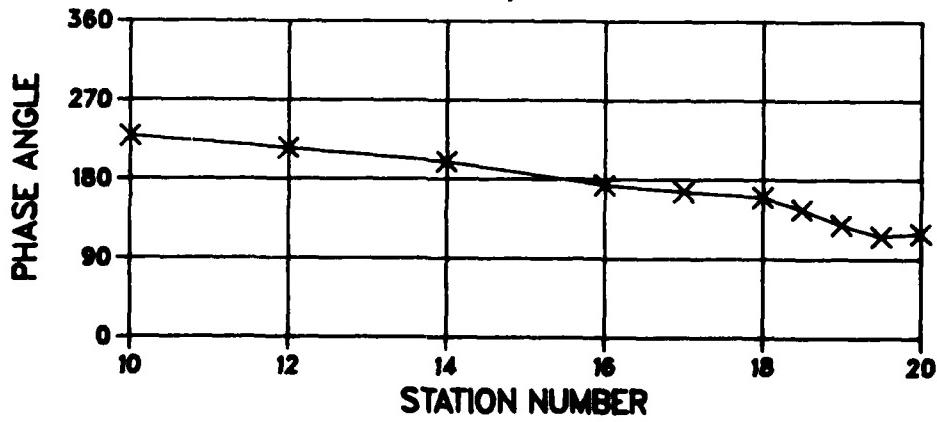
STA.10



LAMBDA/L = 1.883

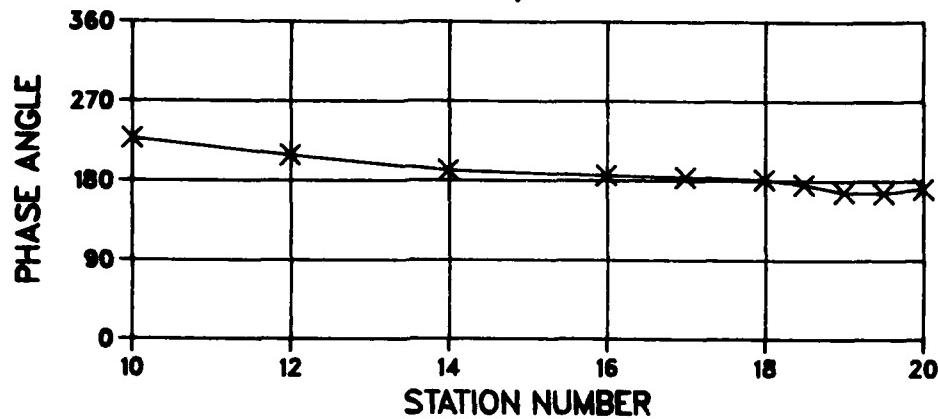


LAMBDA/L = 1.445

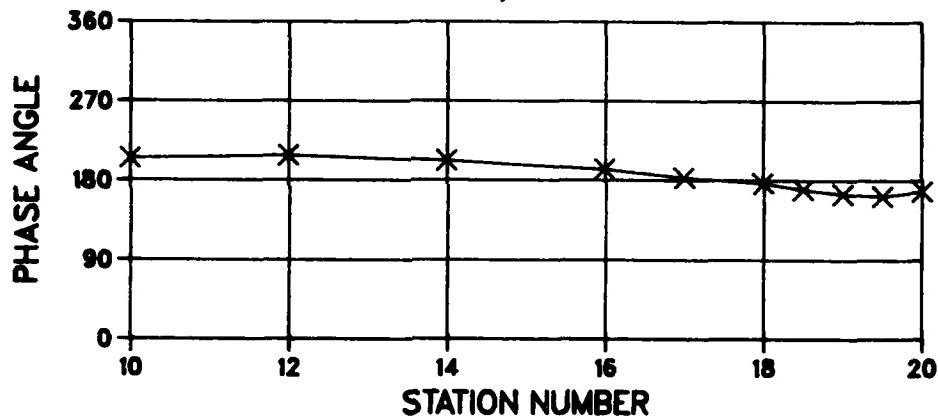


FORCED HEAVE. FN=0.1

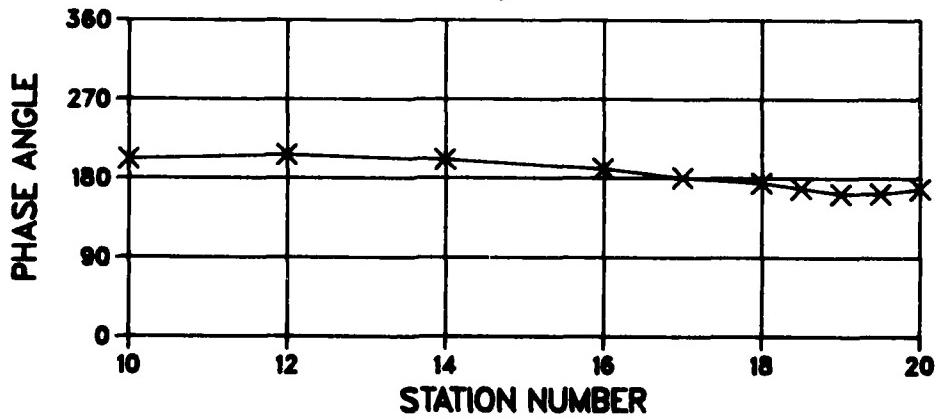
LAMBDA/L = 1.152



LAMBDA/L = 0.865

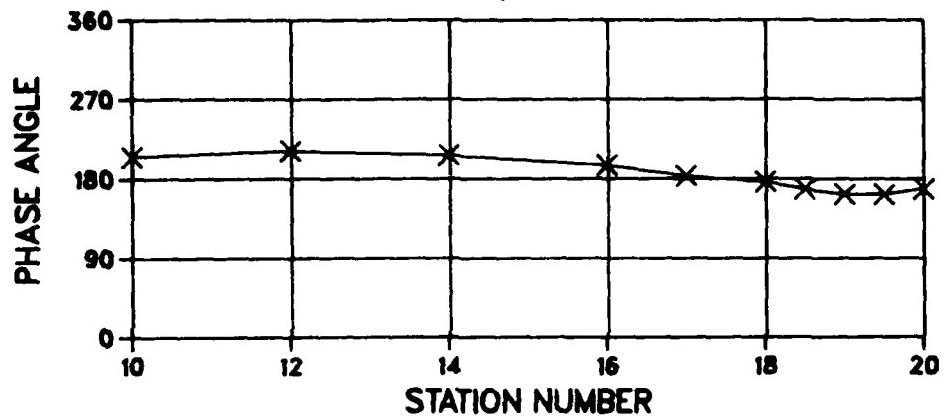


LAMBDA/L = 0.734

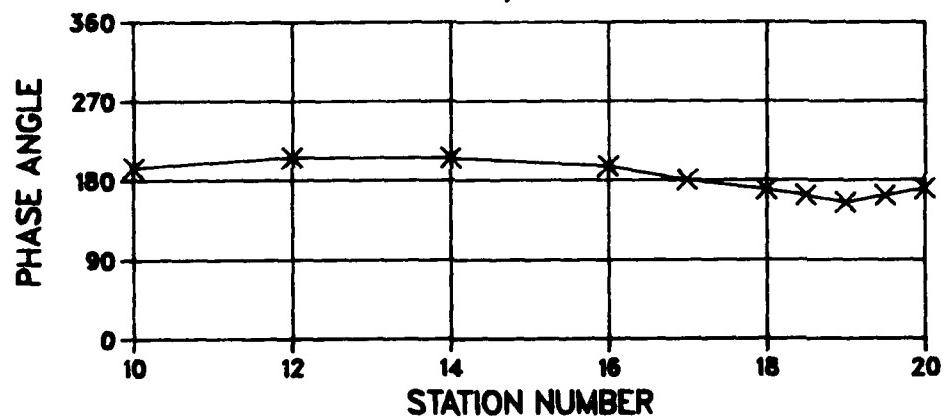


FORCED HEAVE. FN=0.1

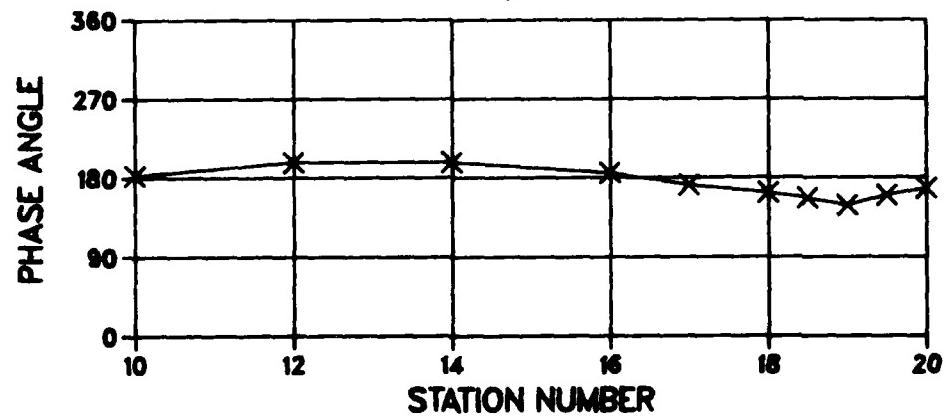
LAMBDA/L = 0.634



LAMBDA/L = 0.376

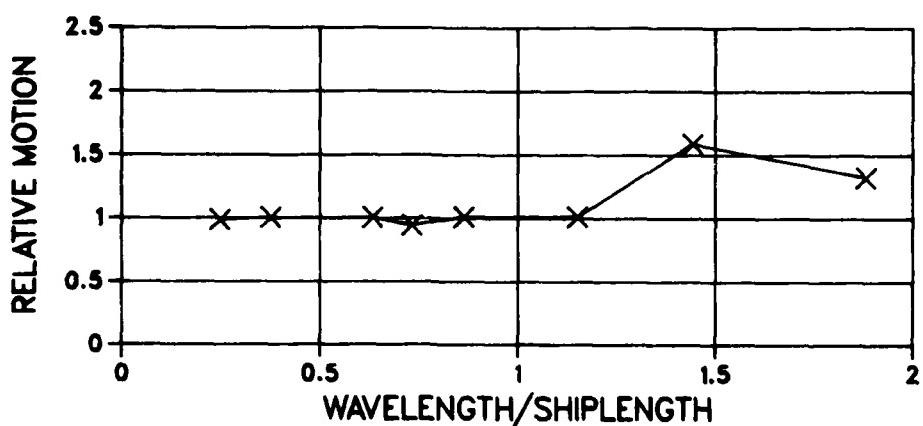


LAMBDA/L = 0.257

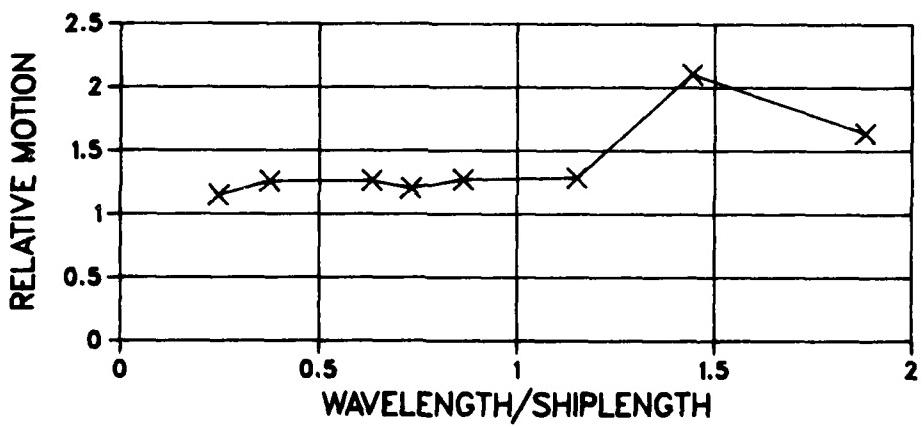


FORCED PITCH. FN=0.1

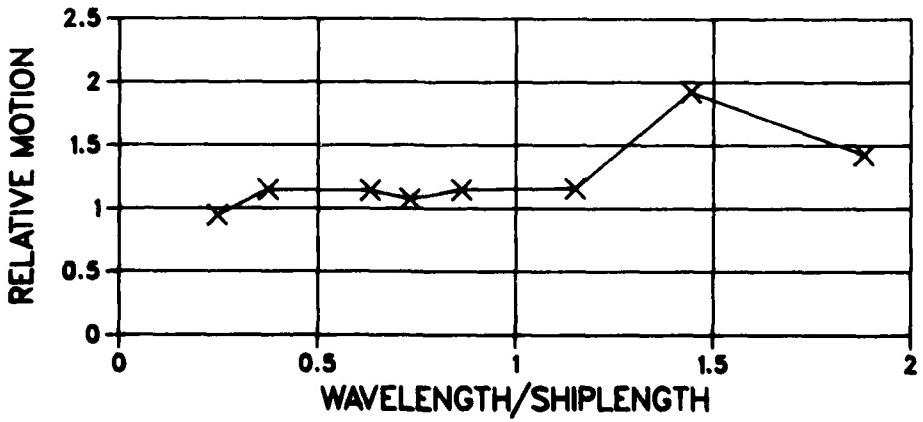
STA.20



STA.19.5

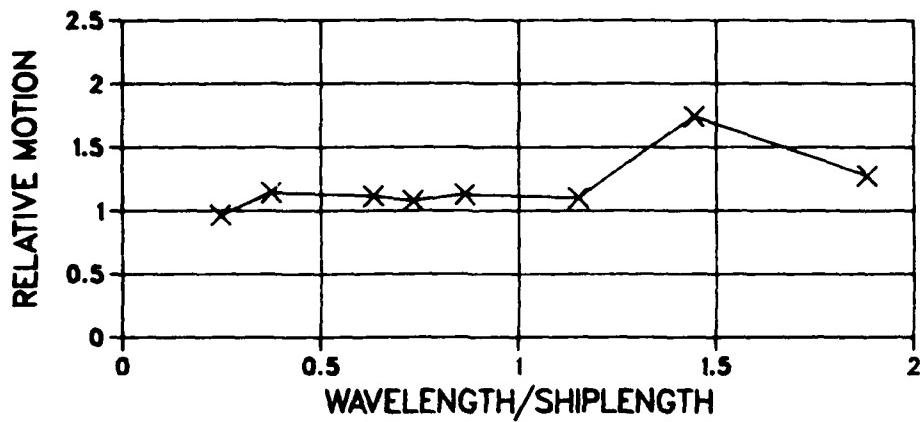


STA.19

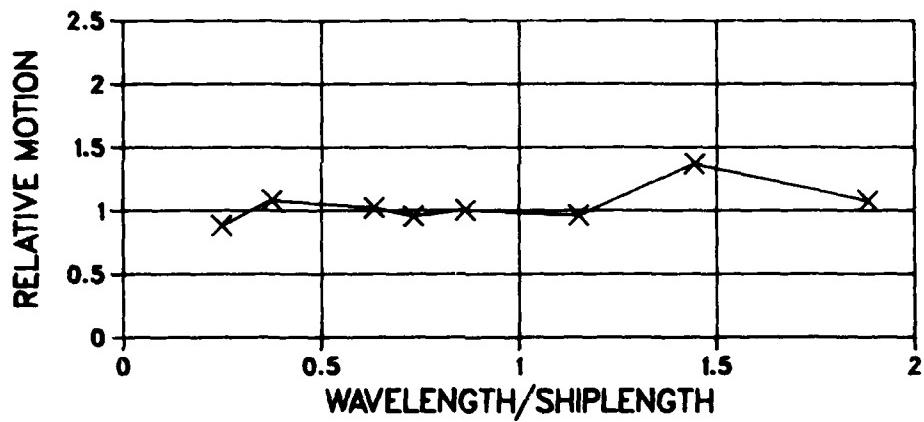


FORCED PITCH. FN=0.1

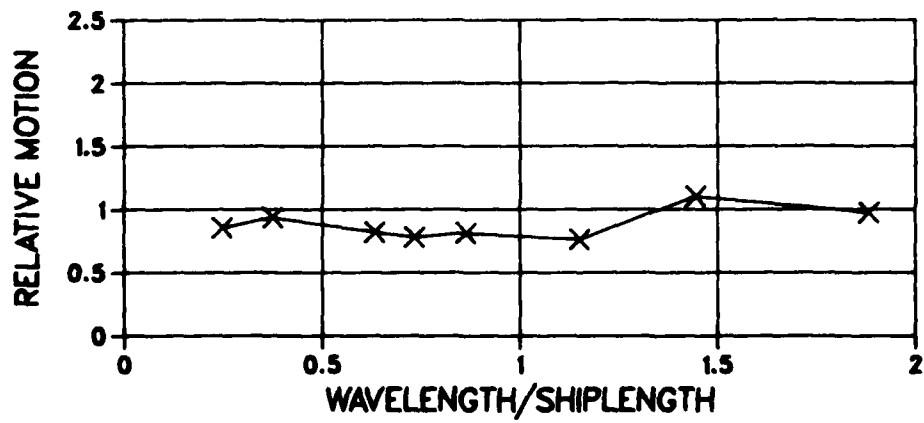
STA.18.5



STA.18

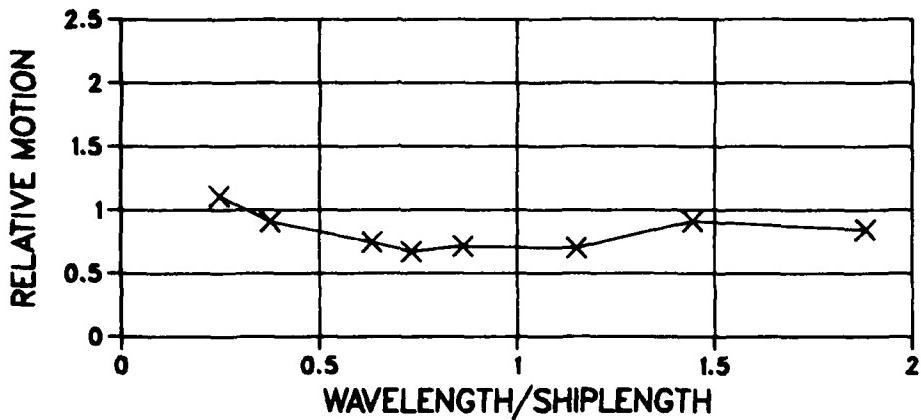


STA.17

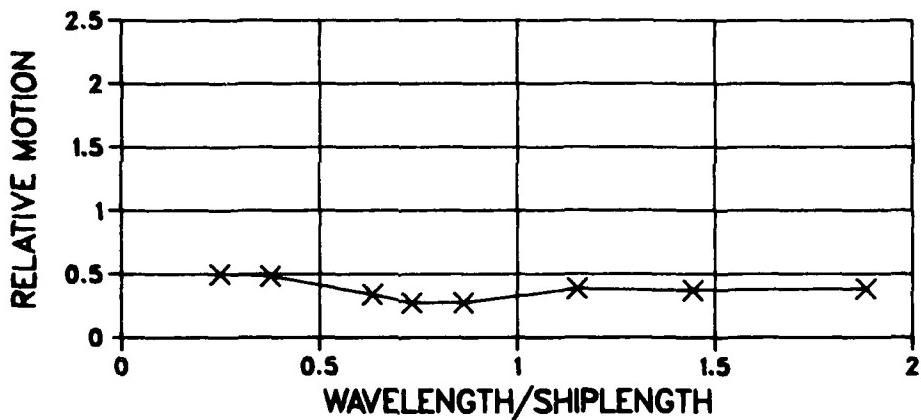


FORCED PITCH. FN=0.1

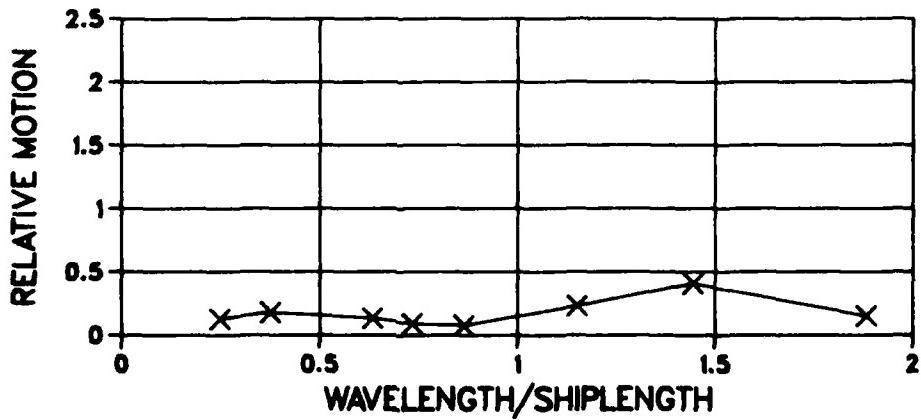
STA.16



STA.14

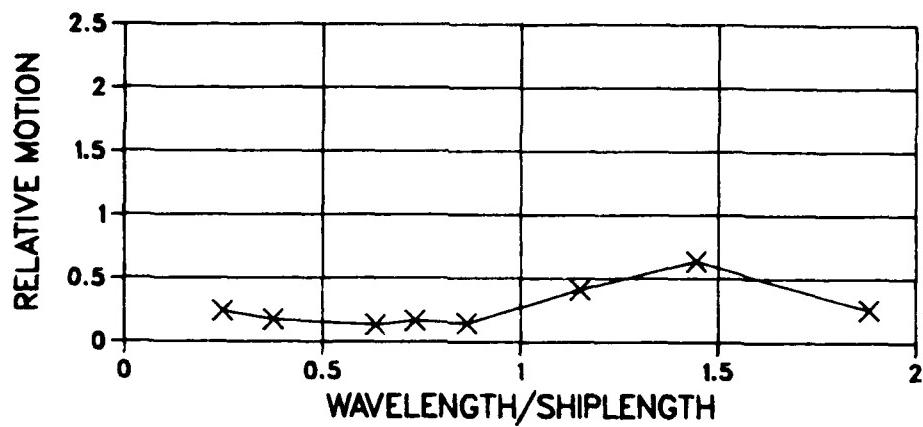


STA.12

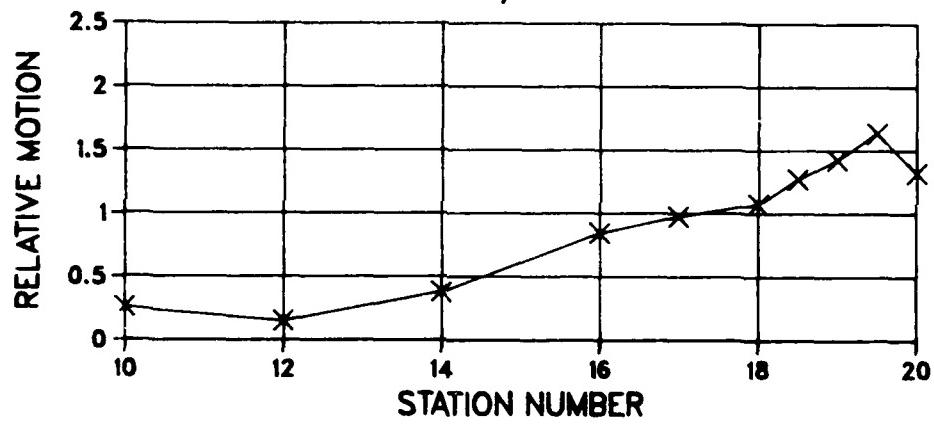


FORCED PITCH. FN=0.1

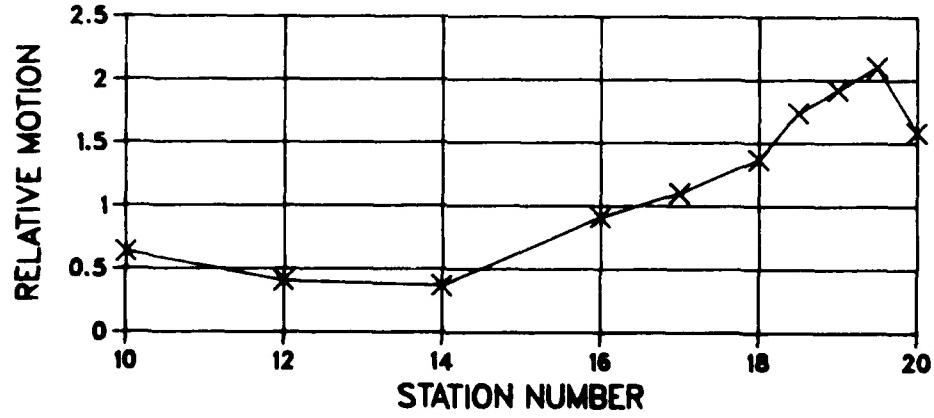
STA.10



$\lambda/L = 1.883$

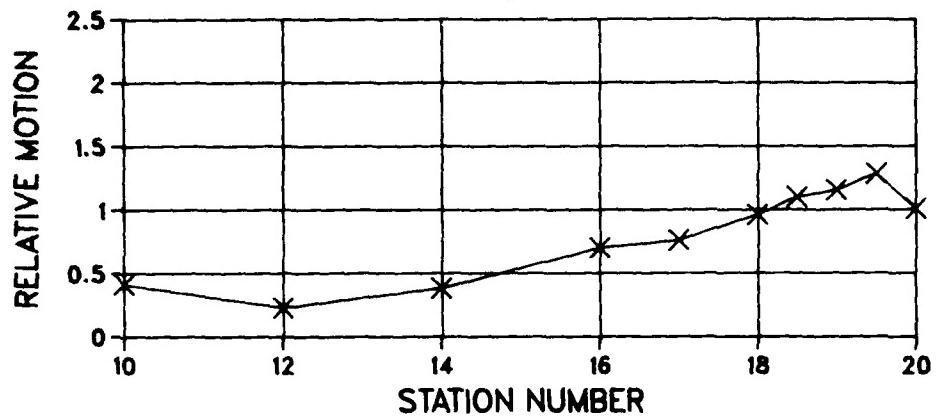


$\lambda/L = 1.445$

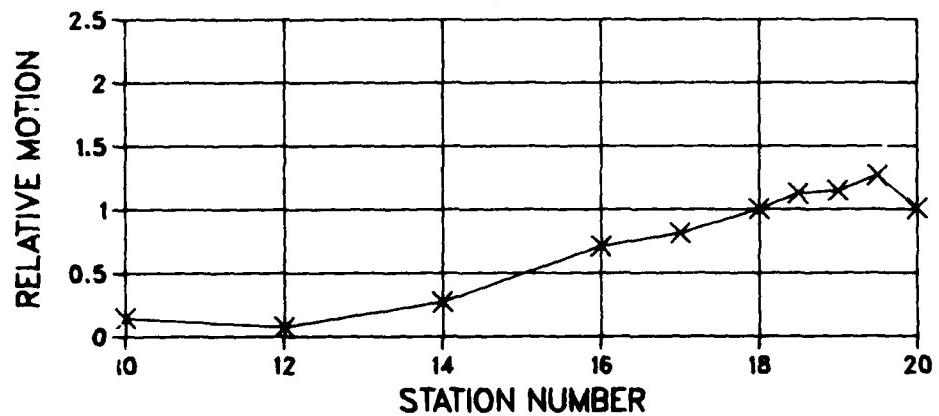


FORCED PITCH. FN=0.1

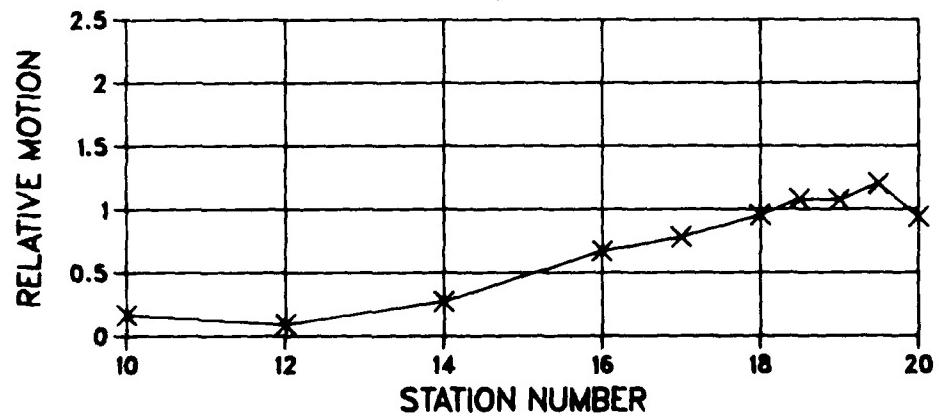
LAMBDA/L = 1.152



LAMBDA/L = 0.865

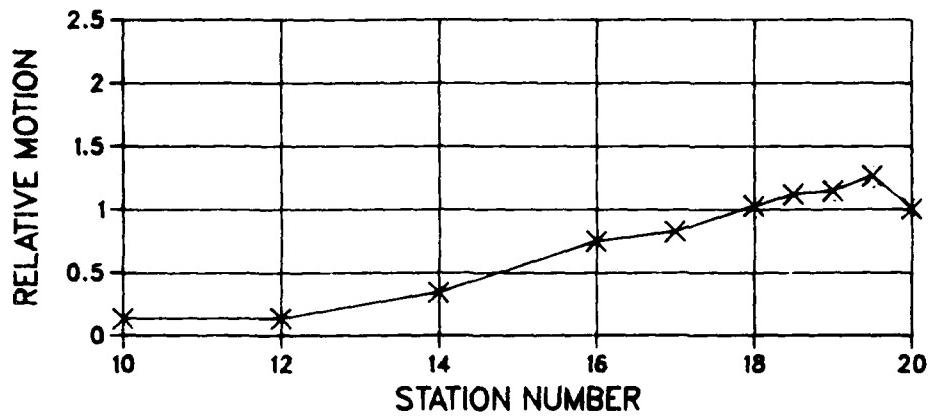


LAMBDA/L = 0.734

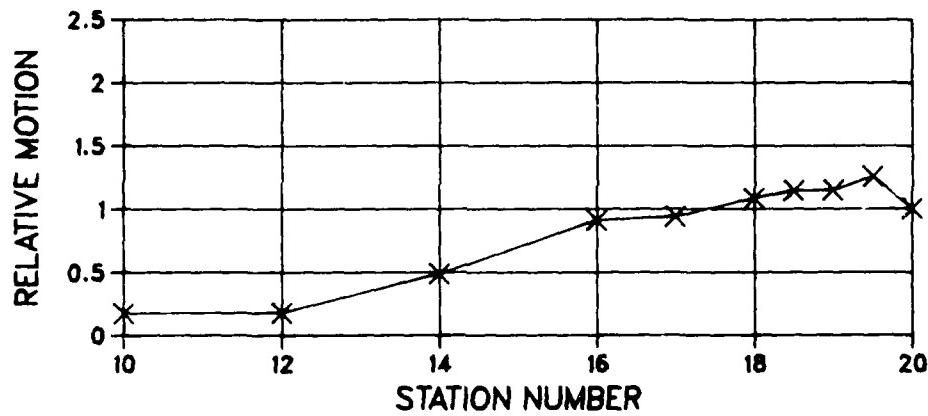


FORCED PITCH. FN=0.1

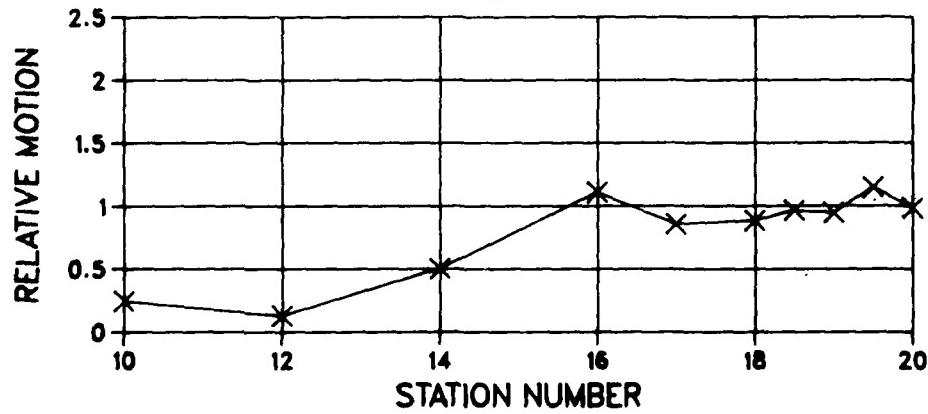
LAMBDA/L = 0.634



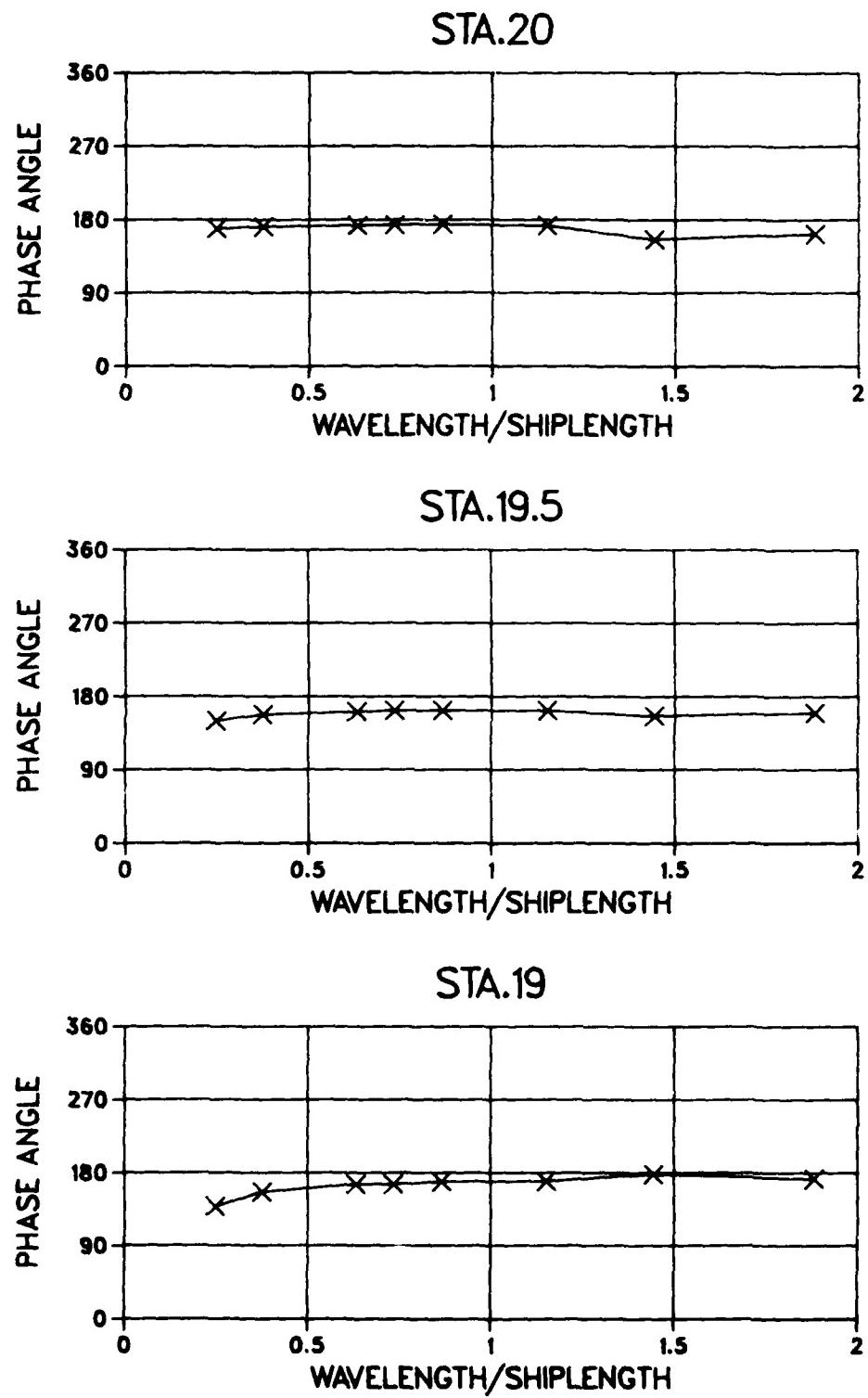
LAMBDA/L = 0.376



LAMBDA/L = 0.248

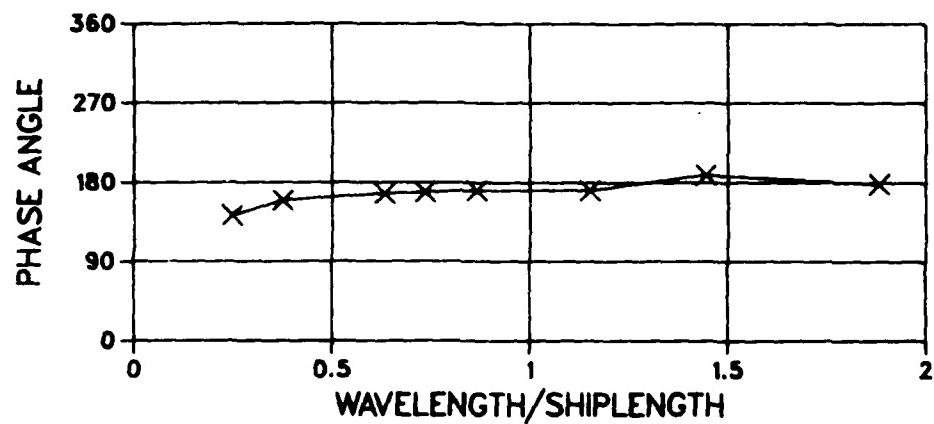


FORCED PITCH. FN=0.1

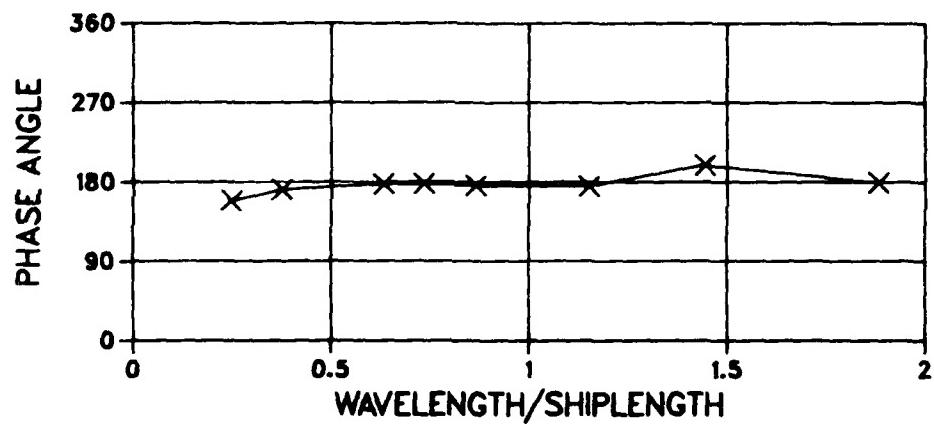


FORCED PITCH. FN=0.1

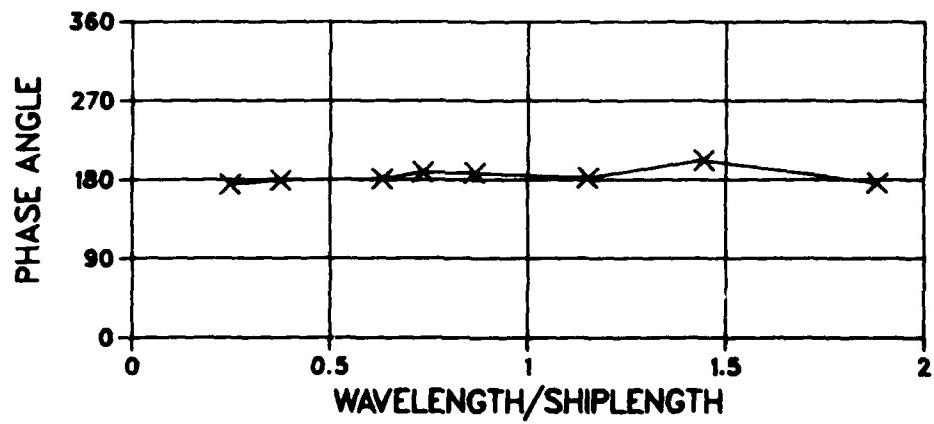
STA.18.5



STA.18

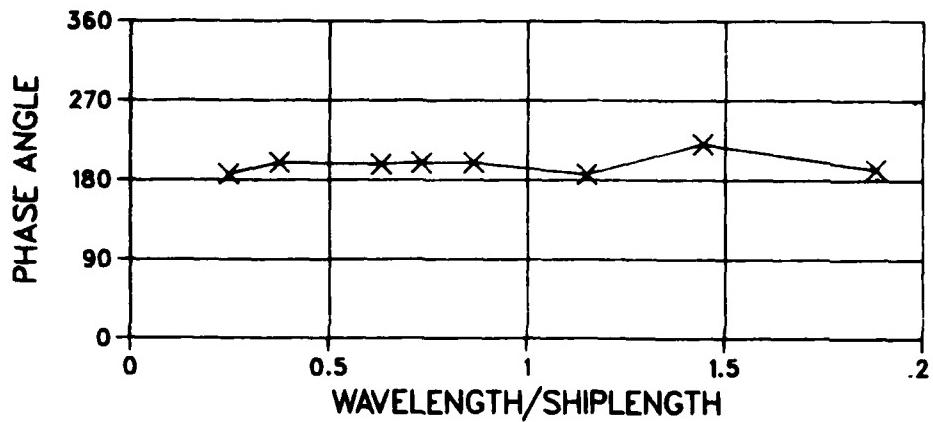


STA.17

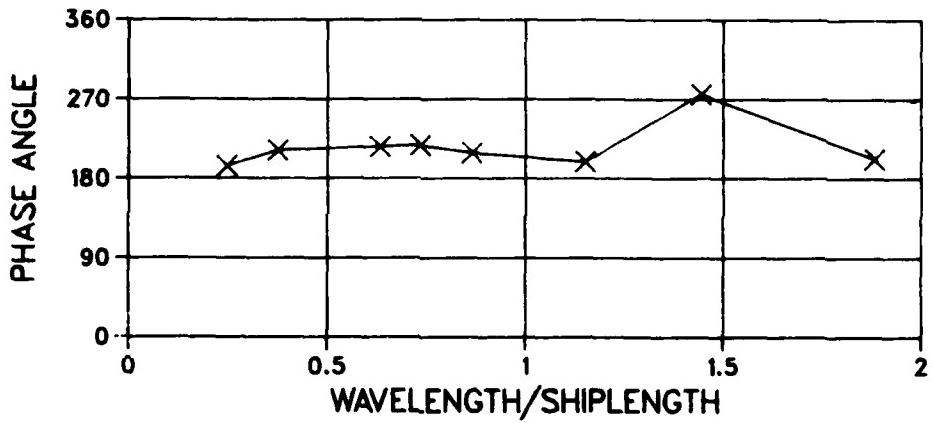


FORCED PITCH. FN=0.1

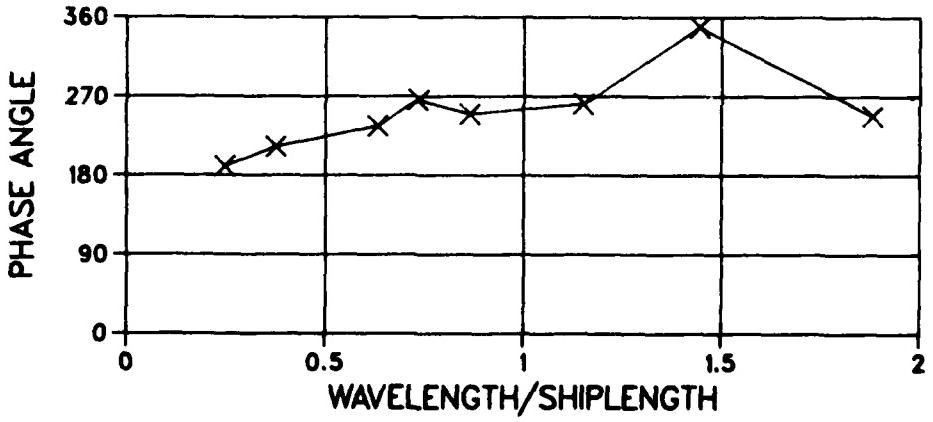
STA.16



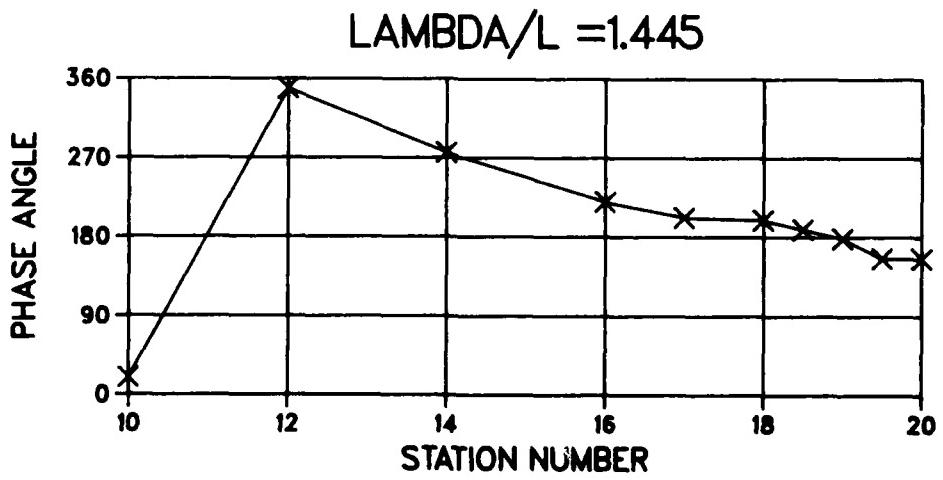
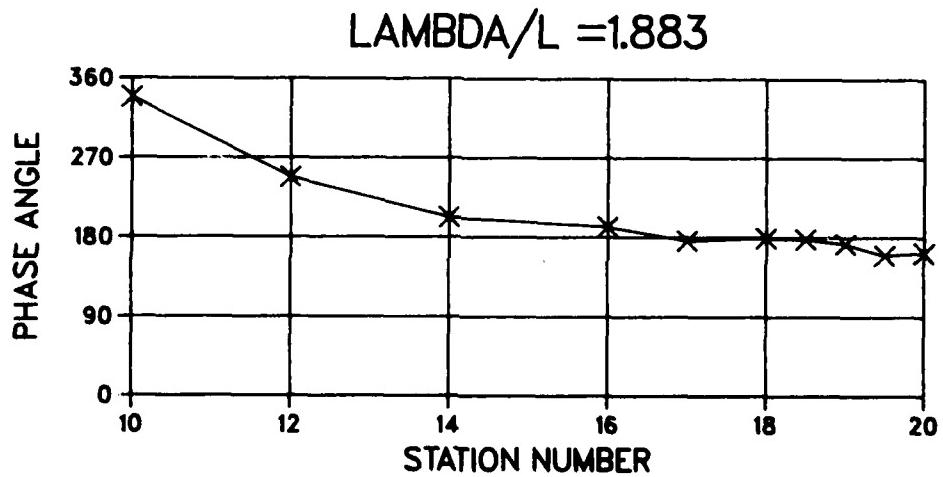
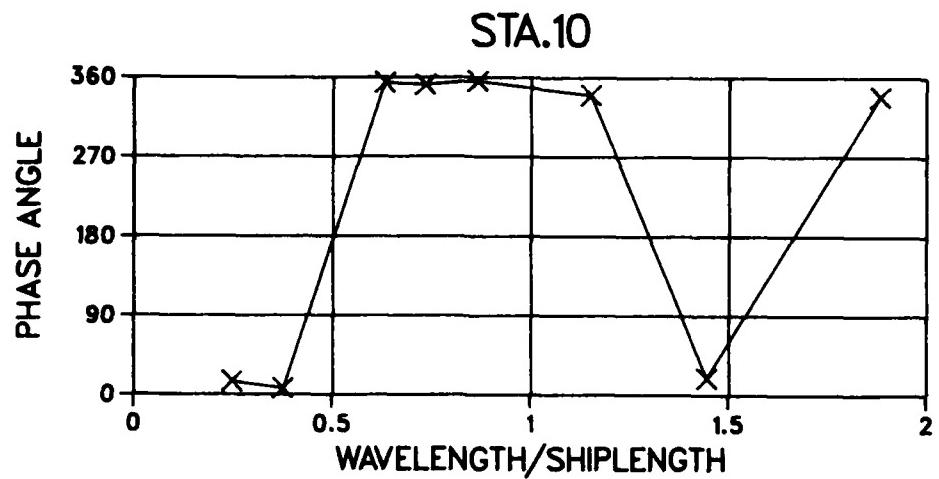
STA.14



STA.12

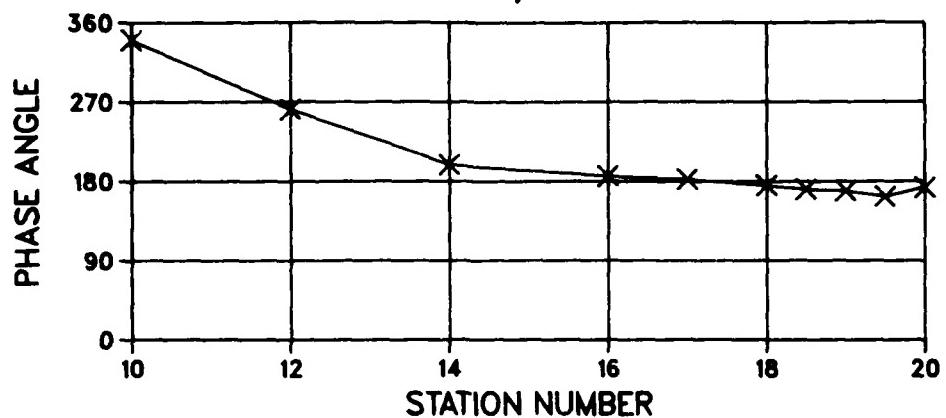


FORCED PITCH. FN=0.1

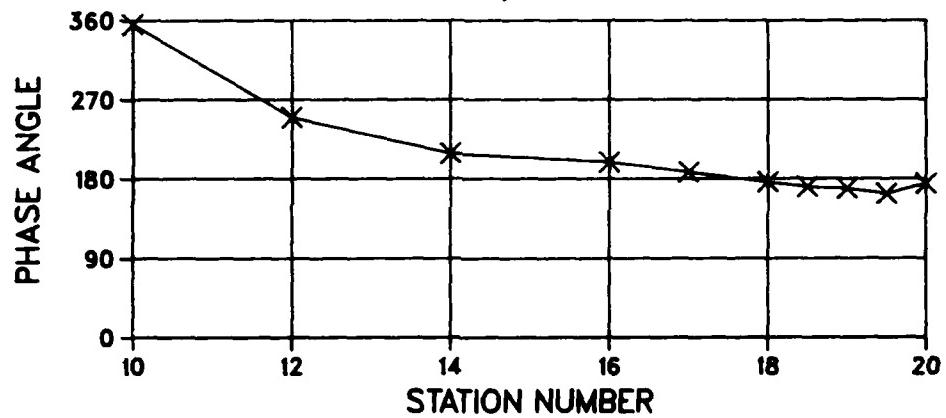


FORCED PITCH. FN=0.1

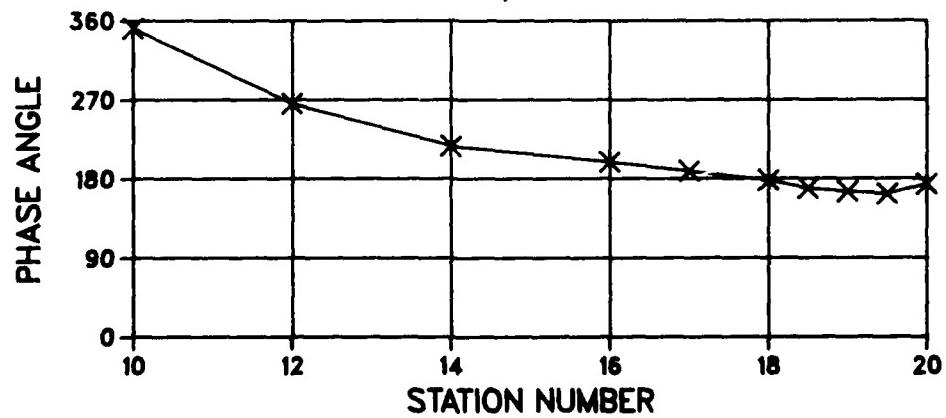
LAMBDA/L = 1.152



LAMBDA/L = 0.865

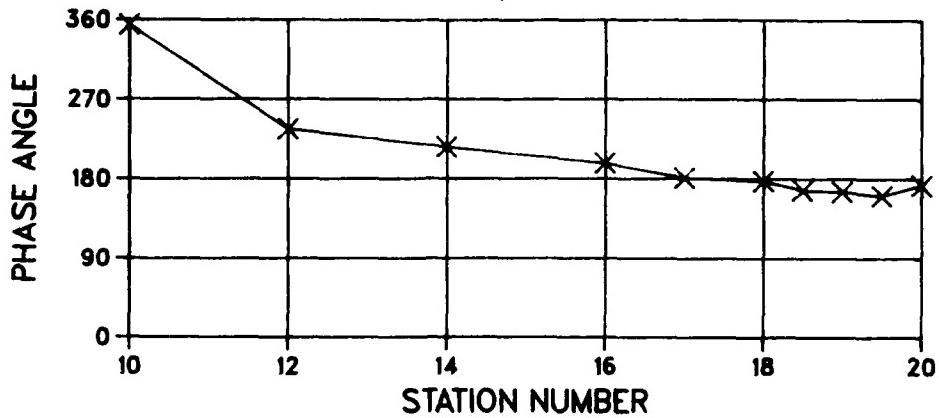


LAMBDA/L = 0.734

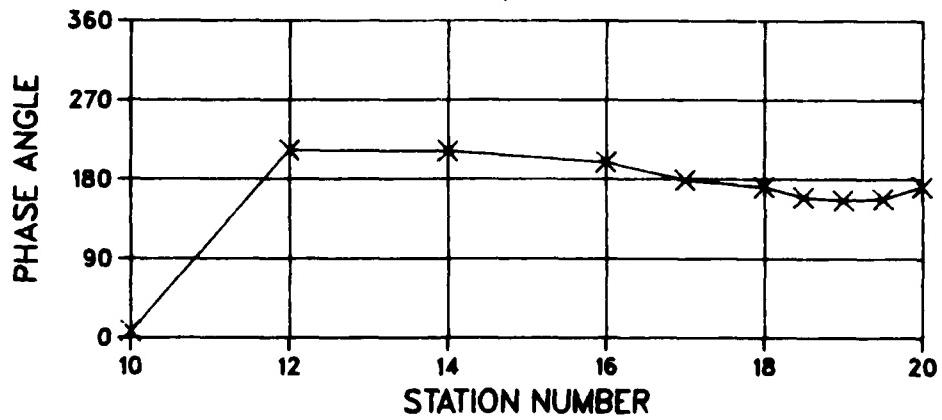


FORCED PITCH. FN=0.1

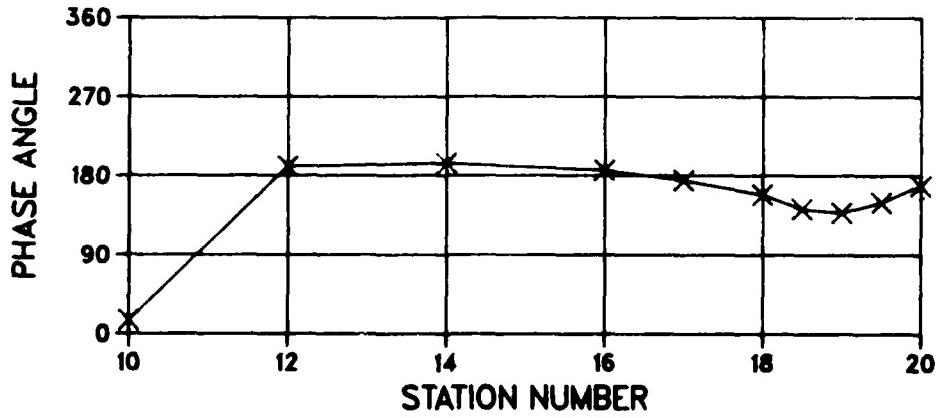
LAMBDA/L = 0.634



LAMBDA/L = 0.376

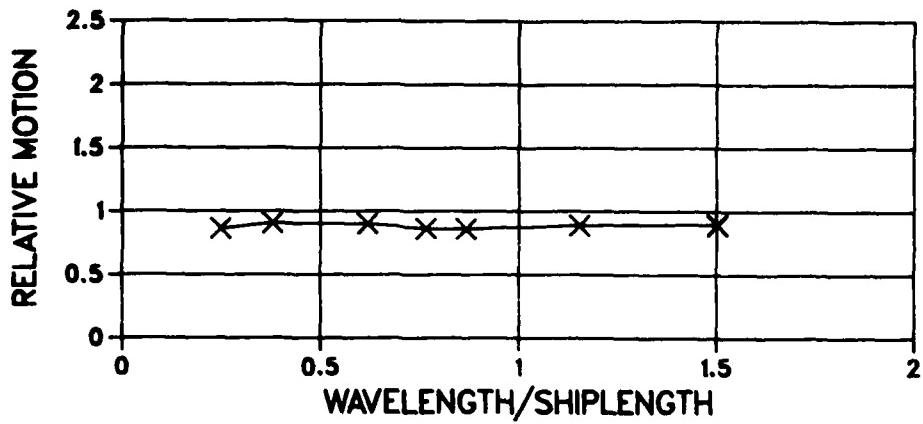


LAMBDA/L = 0.248

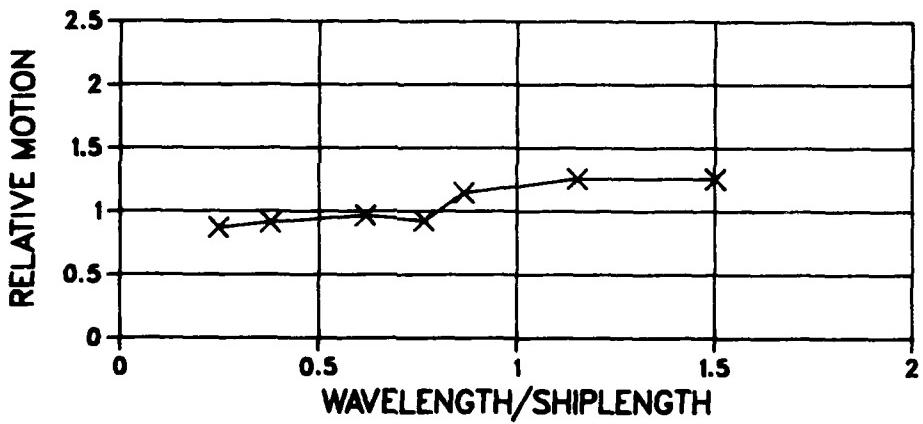


FORCED HEAVE. FN=0.2

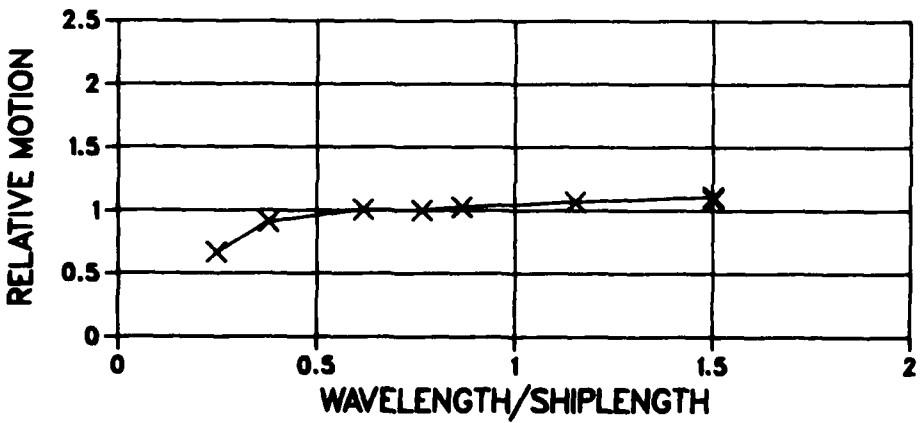
STA.20



STA.19.5

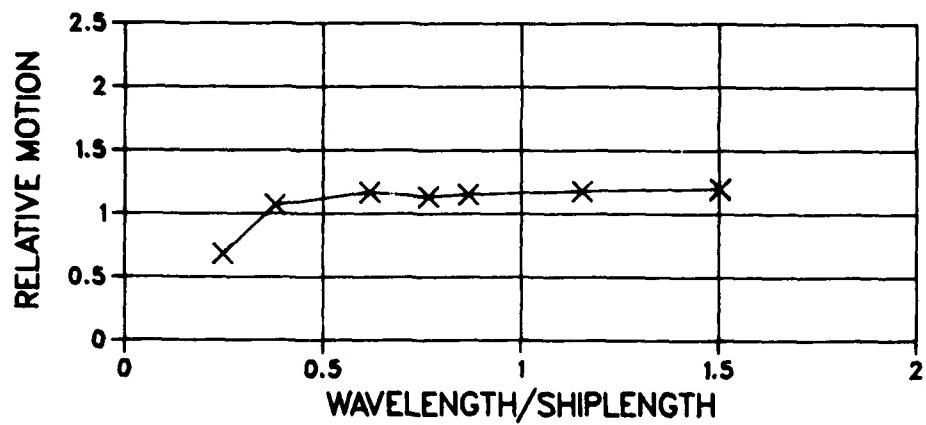


STA.19

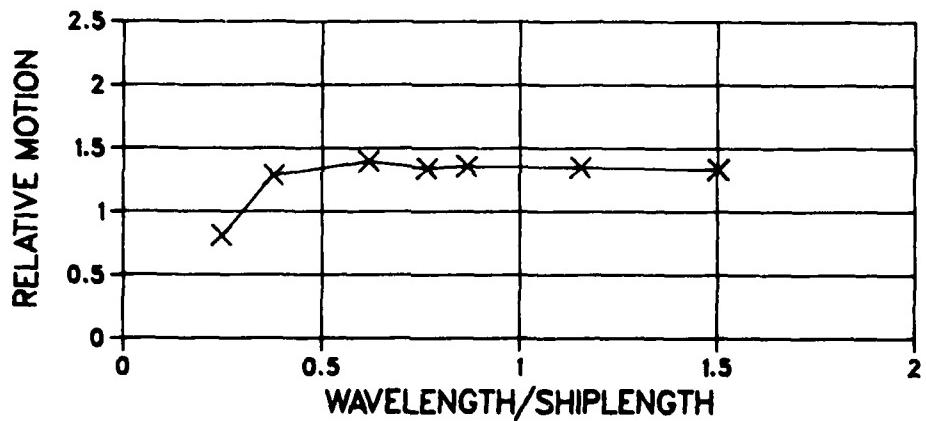


FORCED HEAVE. FN=0.2

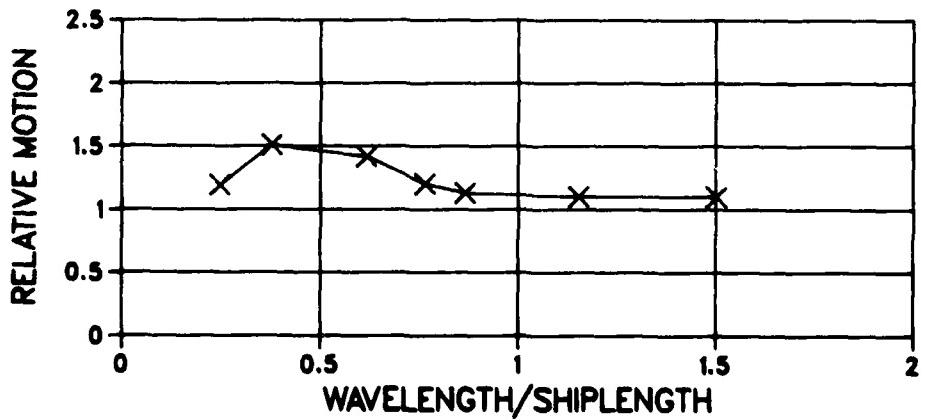
STA.18.5



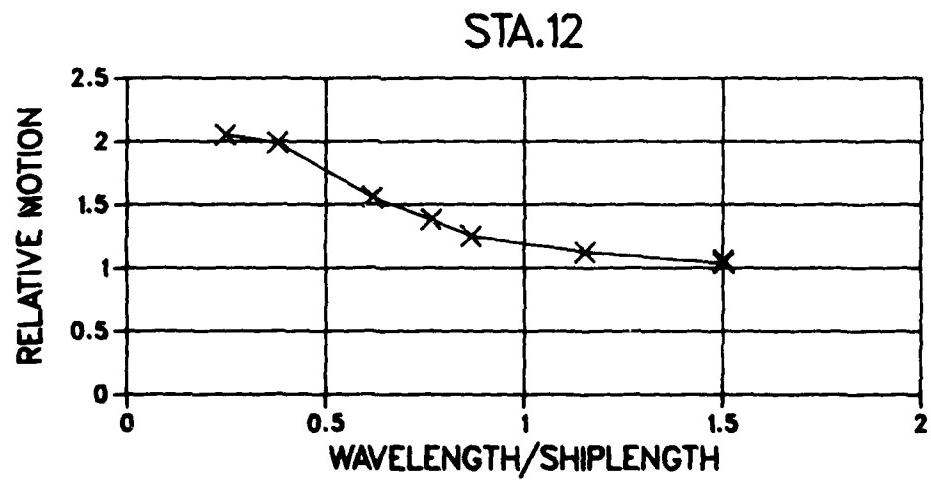
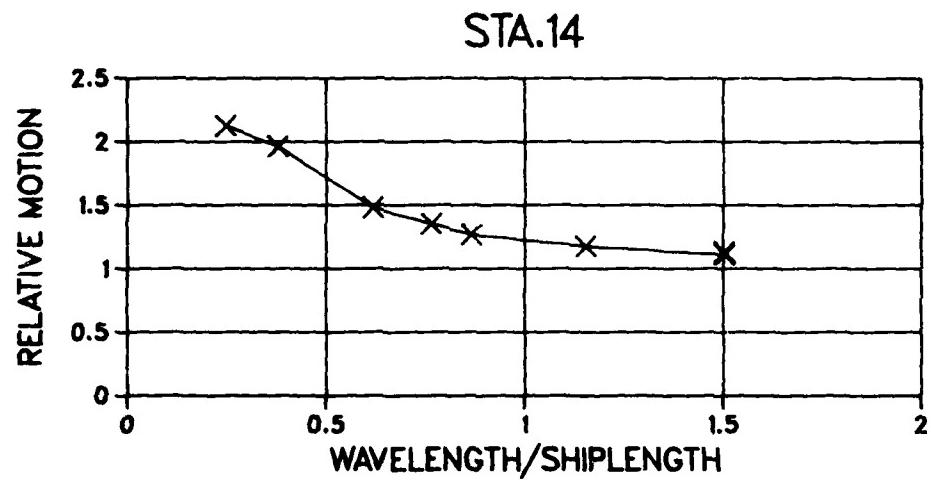
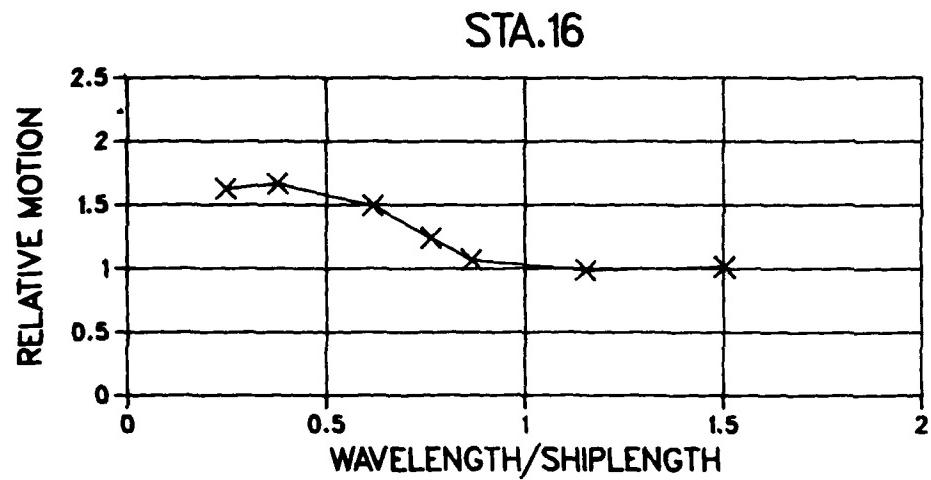
STA.18



STA.17

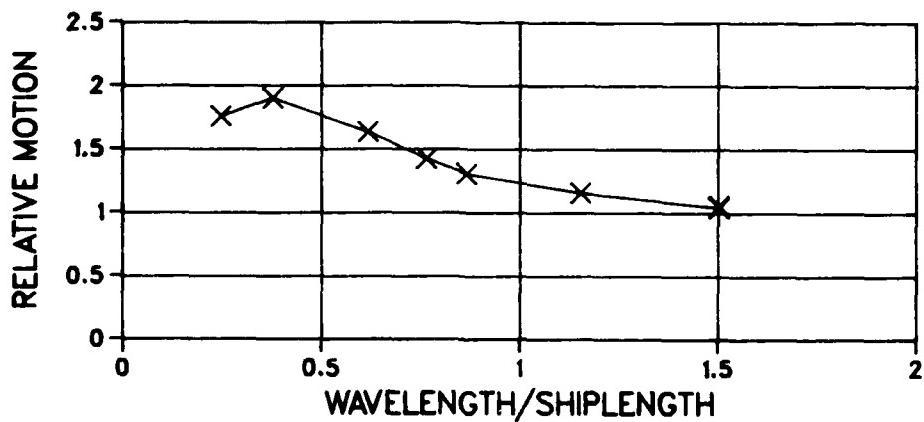


FORCED HEAVE. FN=0.2

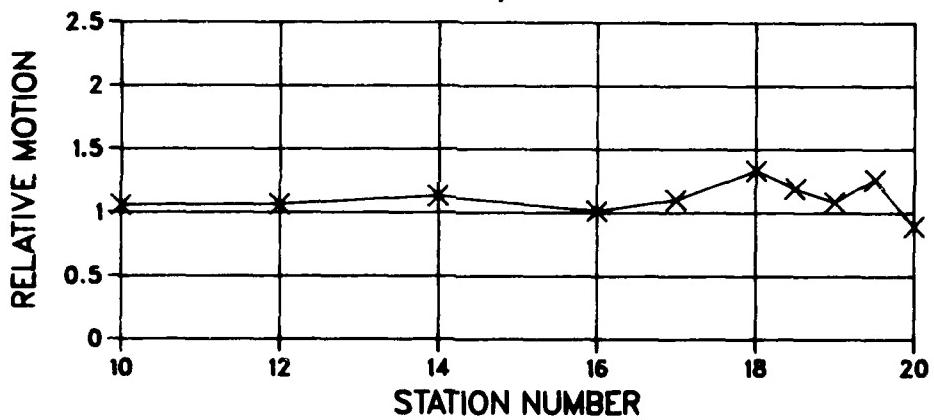


FORCED HEAVE. FN=0.2

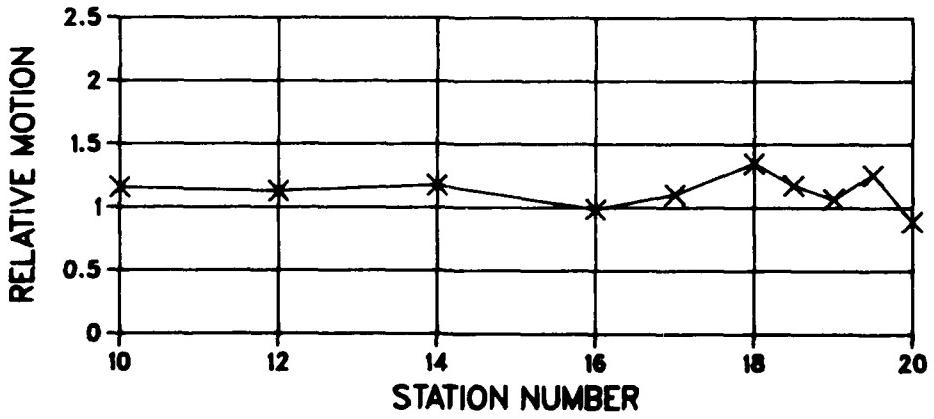
STA.10



LAMBDA/L = 1.503

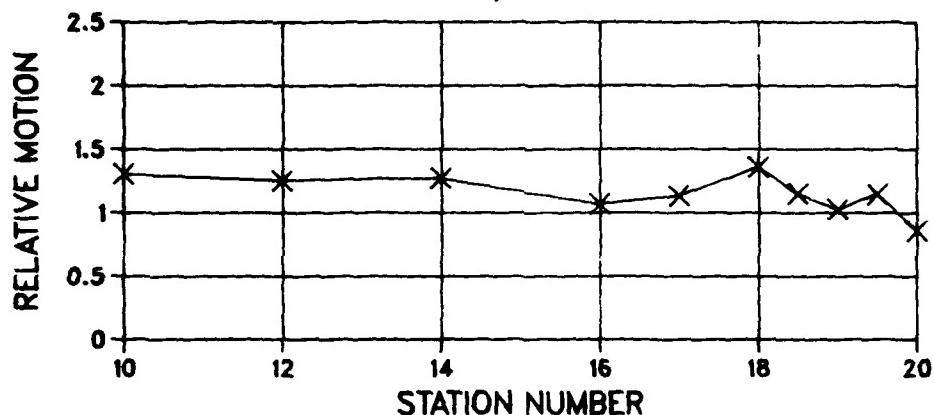


LAMBDA/L = 1.154

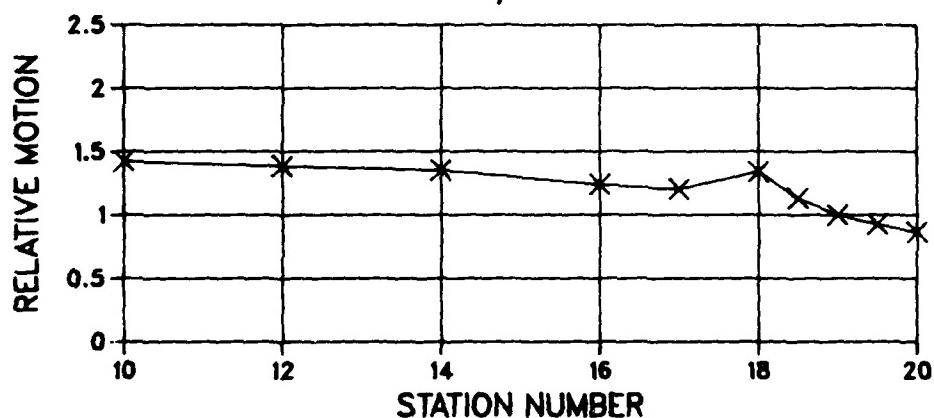


FORCED HEAVE. FN=0.2

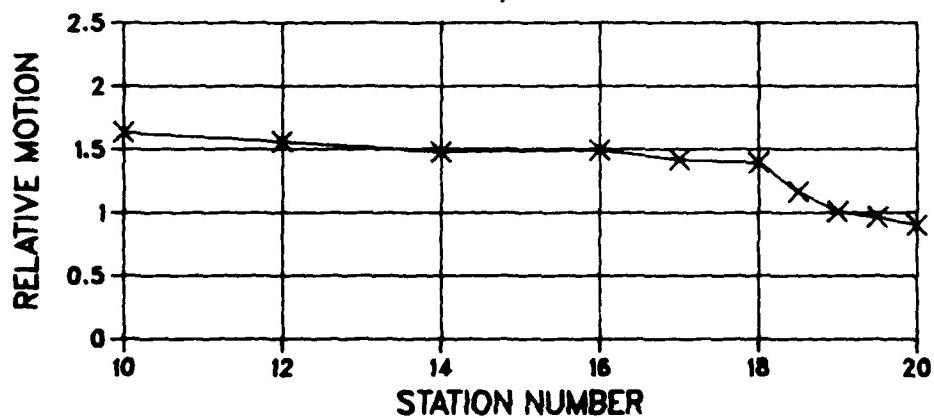
LAMBDA/L = 0.867



LAMBDA/L = 0.766

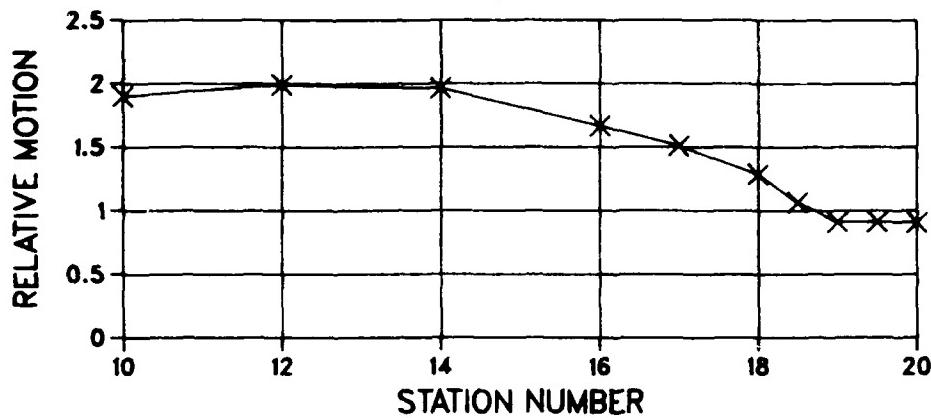


LAMBDA/L = 0.618

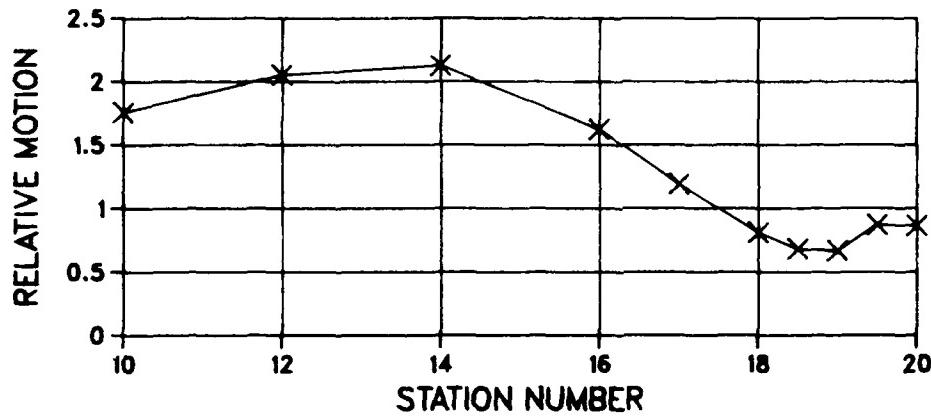


FORCED HEAVE. FN=0.2

LAMBDA/L = 0.379

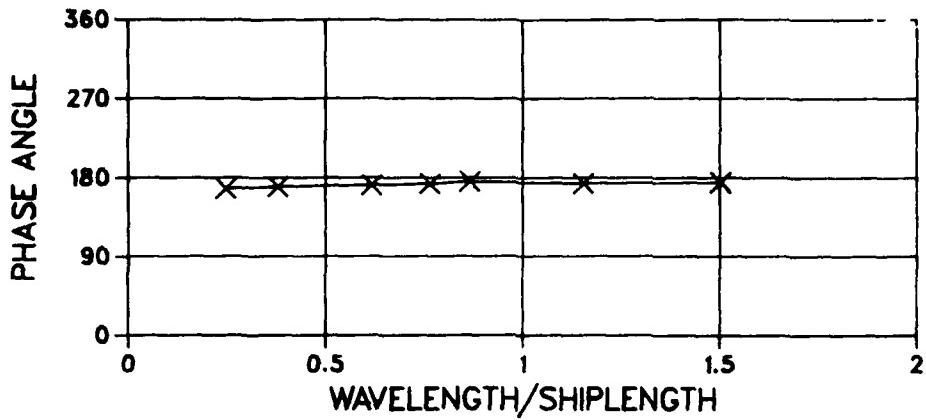


LAMBDA/L = 0.248

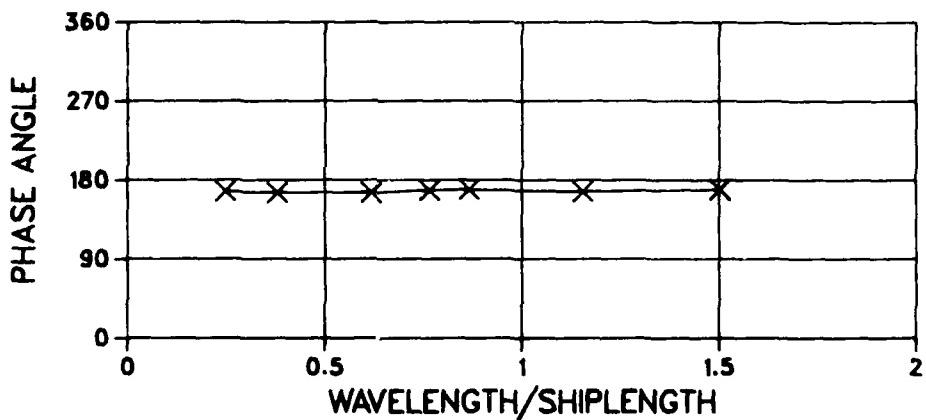


FORCED HEAVE. FN=0.2

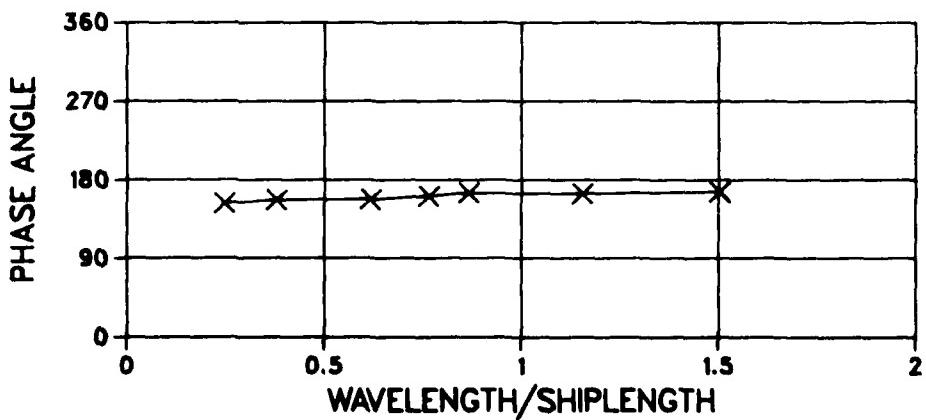
STA.20



STA.19.5

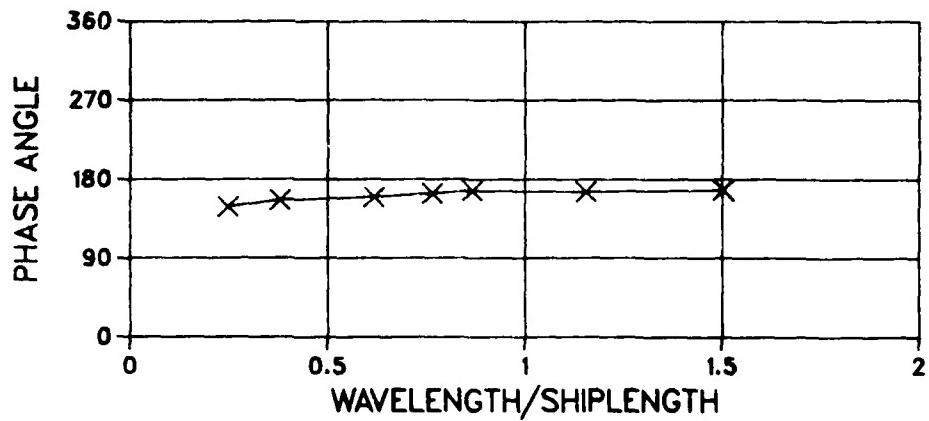


STA.19

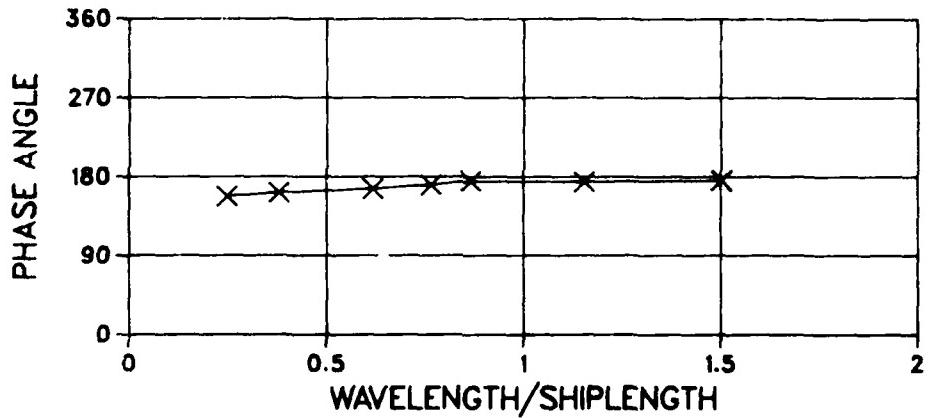


FORCED HEAVE. FN=0.2

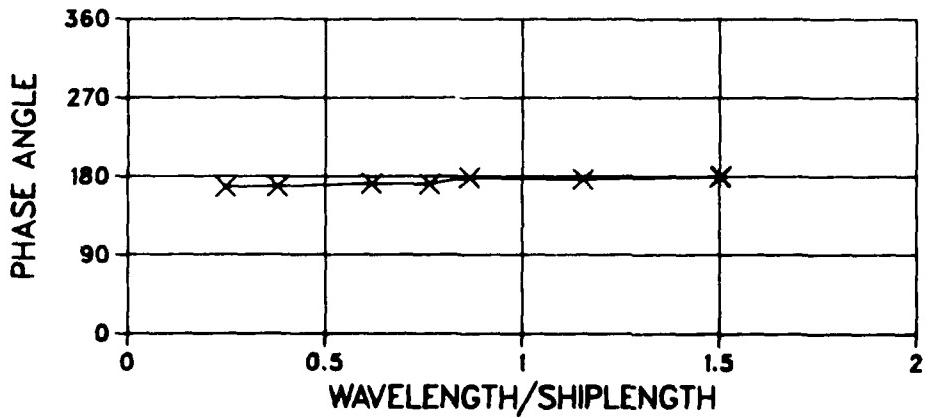
STA.18.5



STA.18

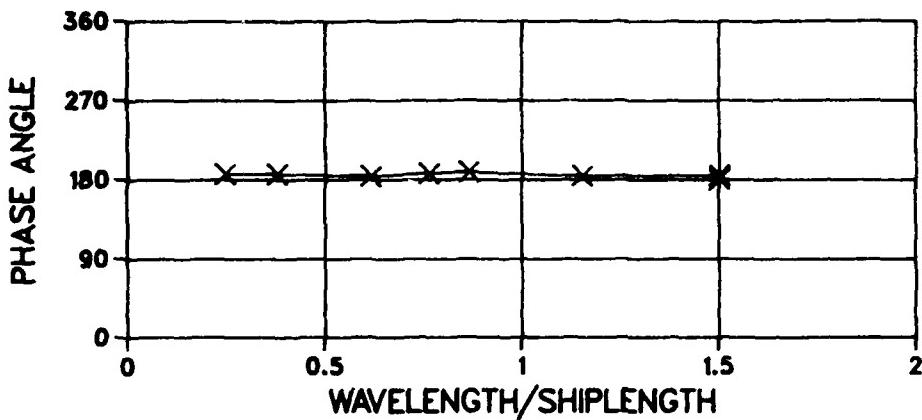


STA.17

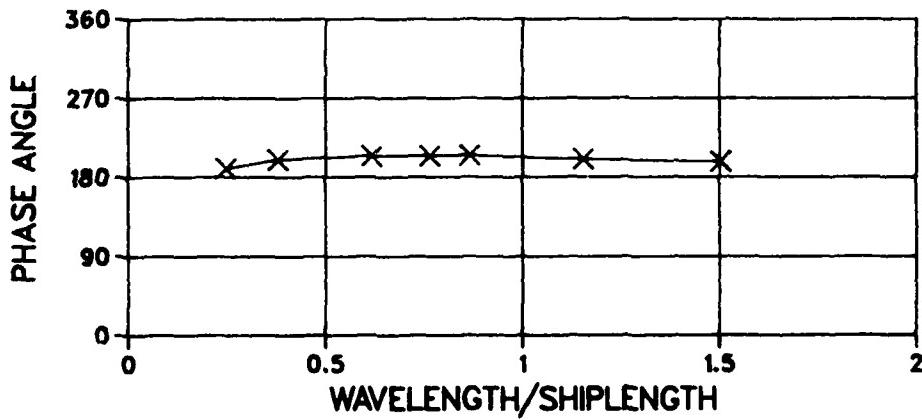


FORCED HEAVE. FN=0.2

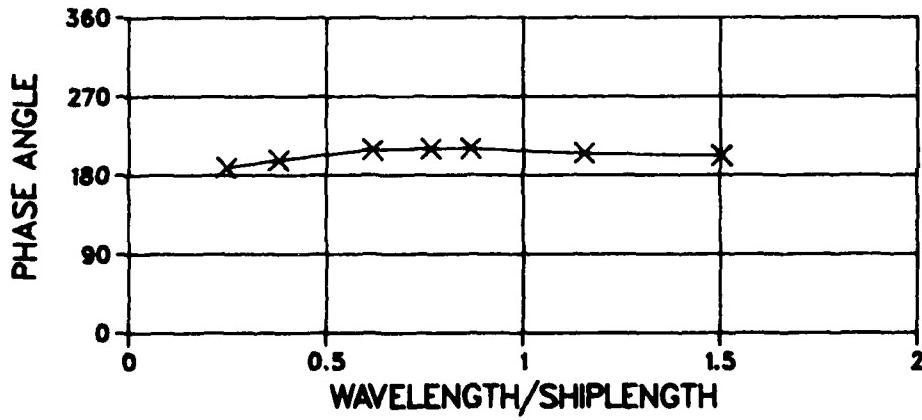
STA.16



STA.14

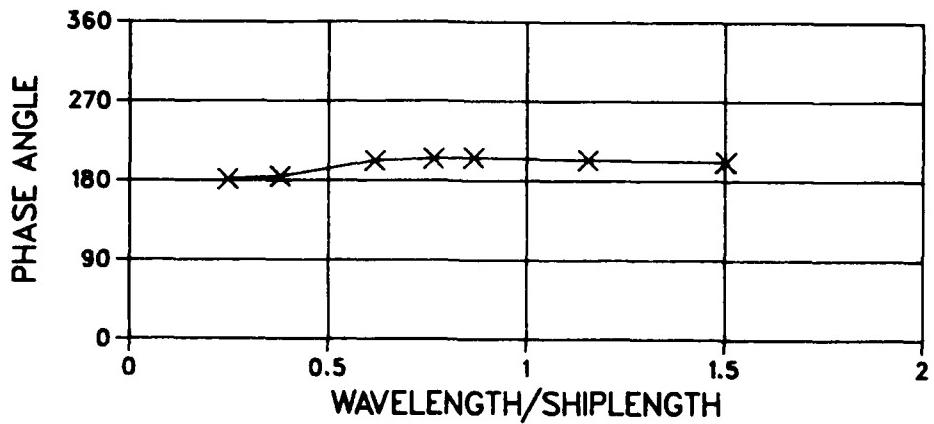


STA.12

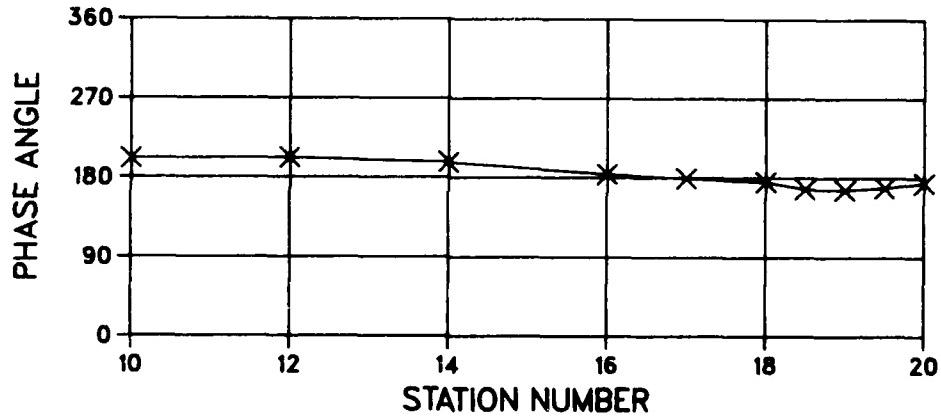


FORCED HEAVE. FN=0.2

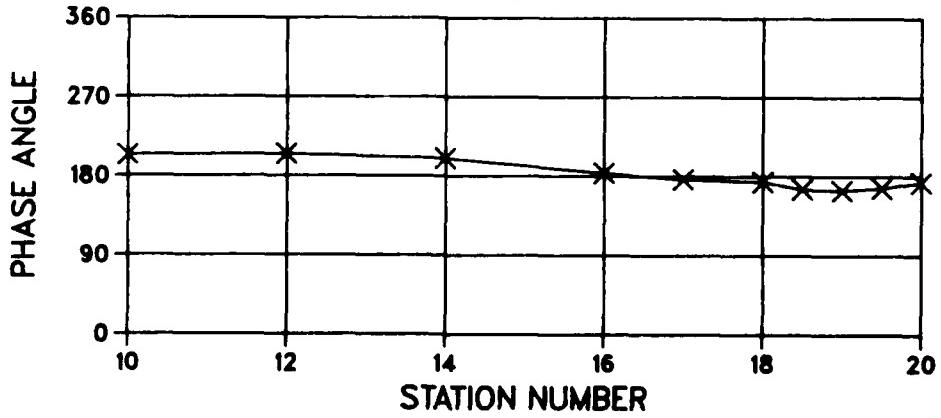
STA.10



LAMBDA/L = 1.503

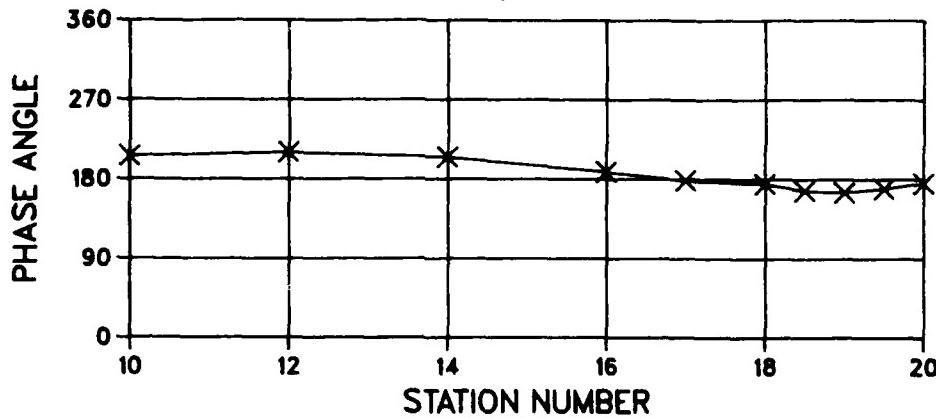


LAMBDA/L = 1.154

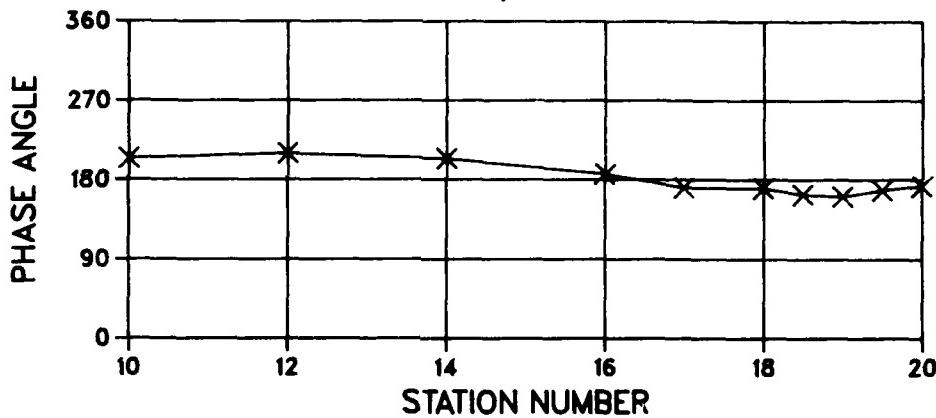


FORCED HEAVE. FN=0.2

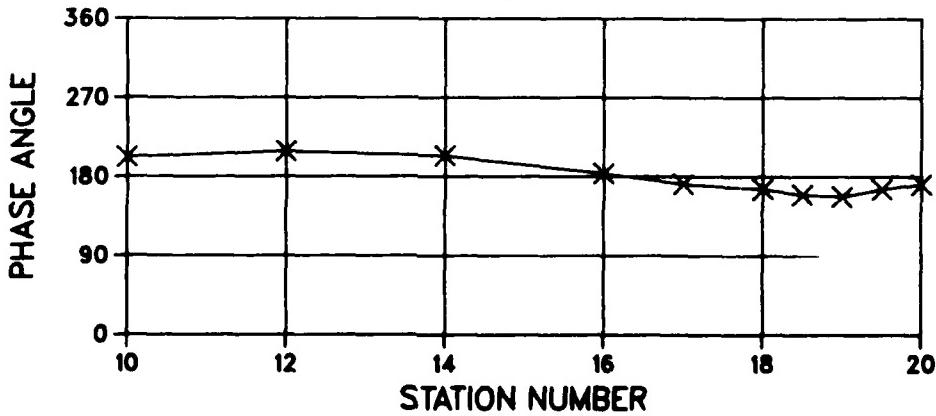
LAMBDA/L =0.867



LAMBDA/L =0.766

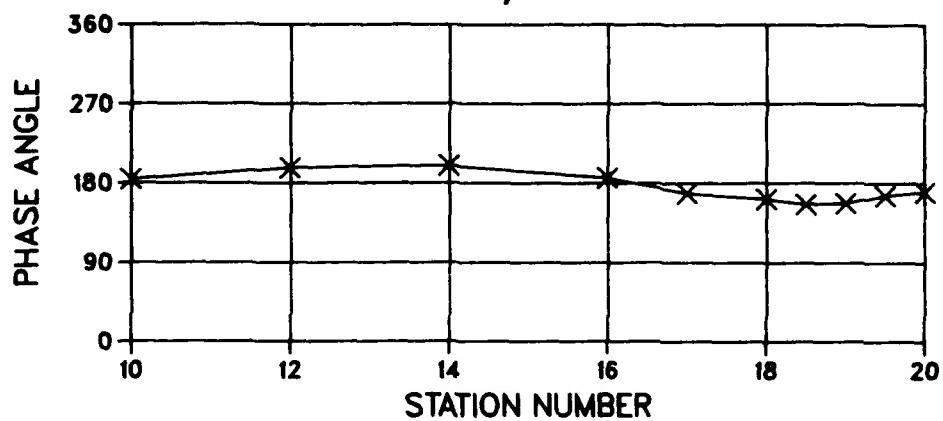


LAMBDA/L =0.618

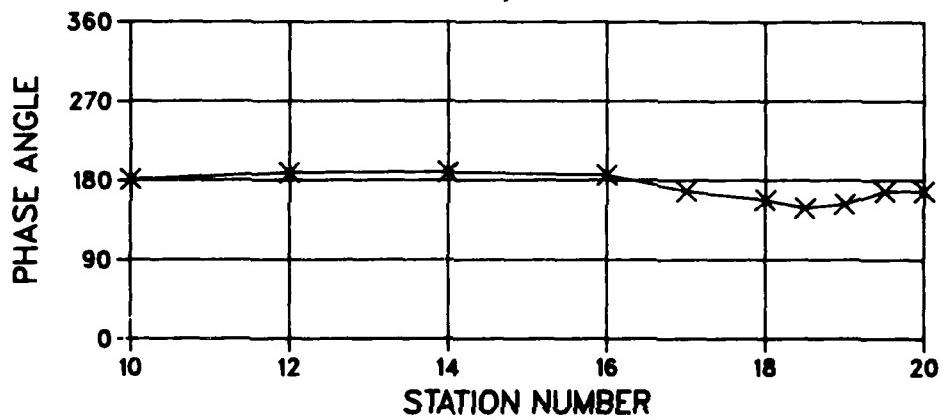


FORCED HEAVE. FN=0.2

LAMBDA/L =0.379

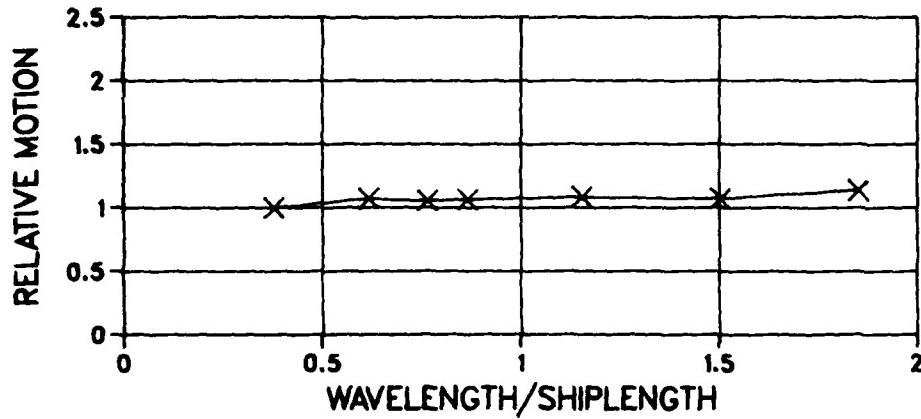


LAMBDA/L =0.248

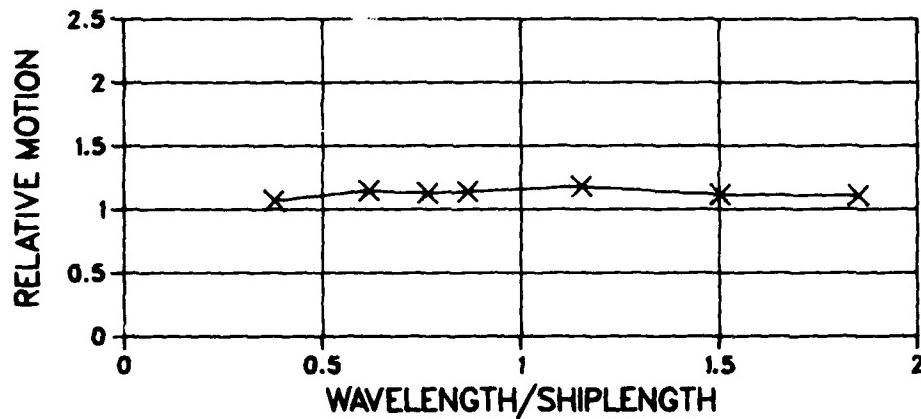


FORCED PITCH. FN=0.2

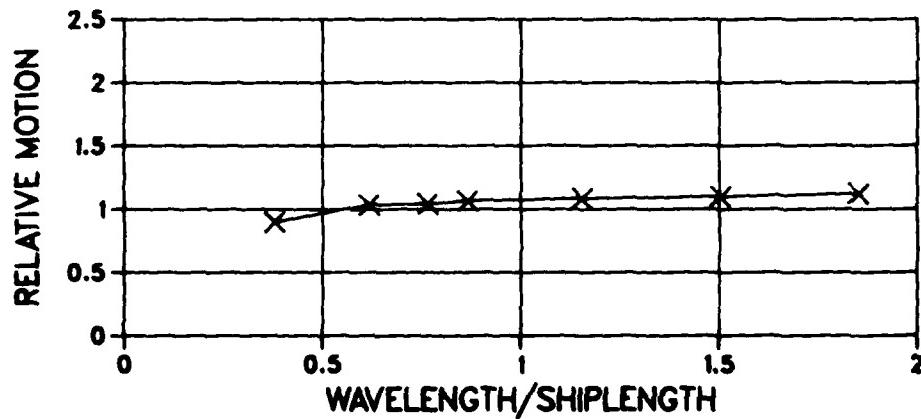
STA.20



STA.19.5

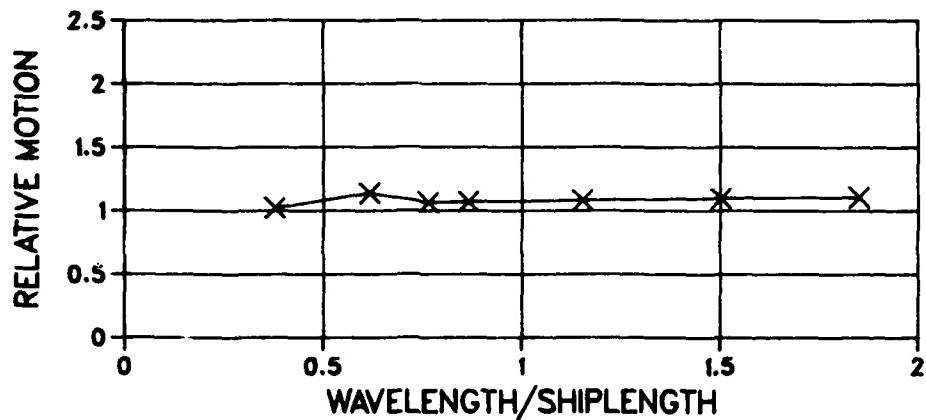


STA.19

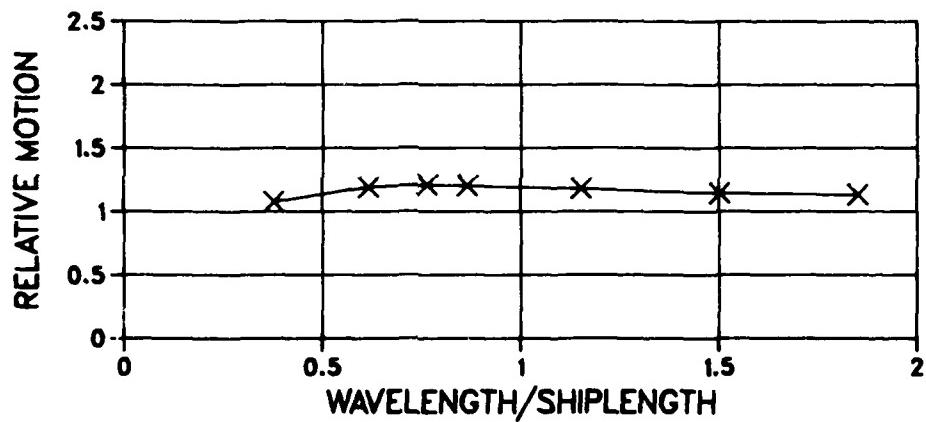


FORCED PITCH. FN=0.2

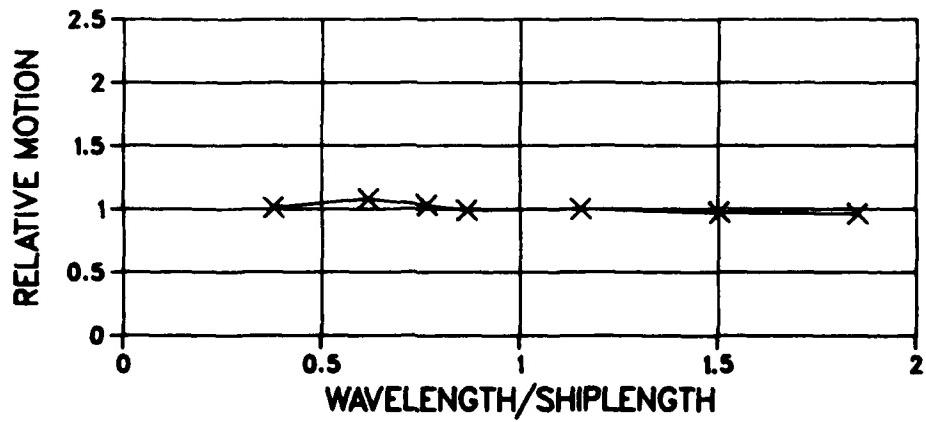
STA.18.5



STA.18

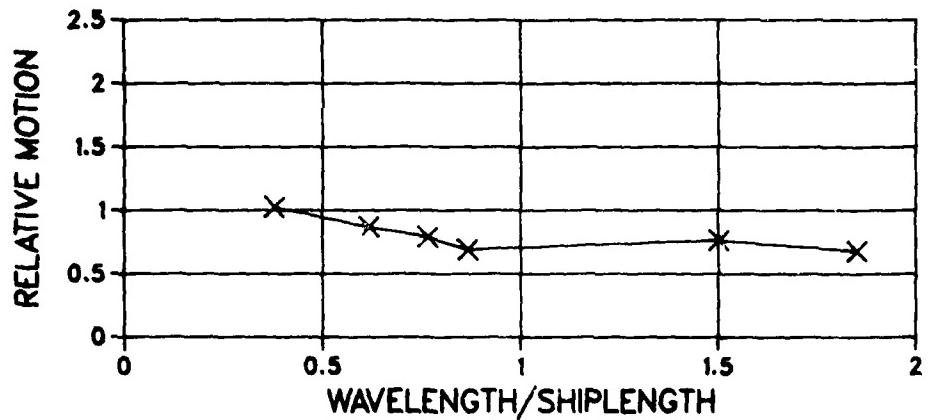


STA.17

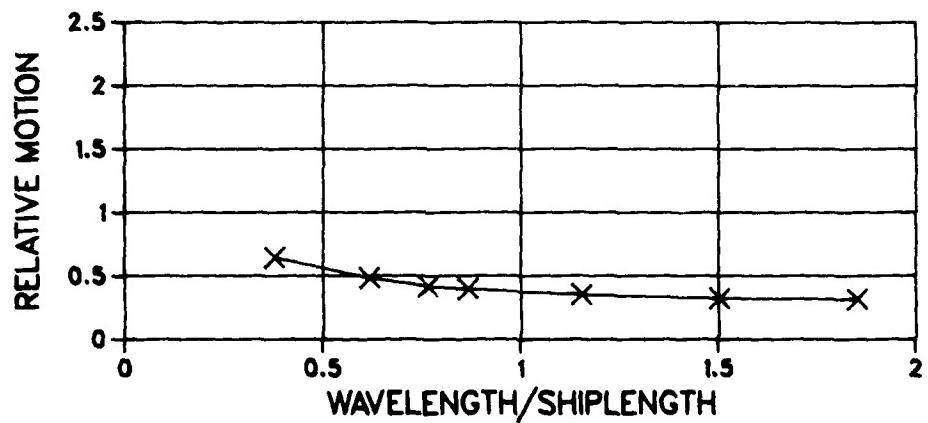


FORCED PITCH. FN=0.2

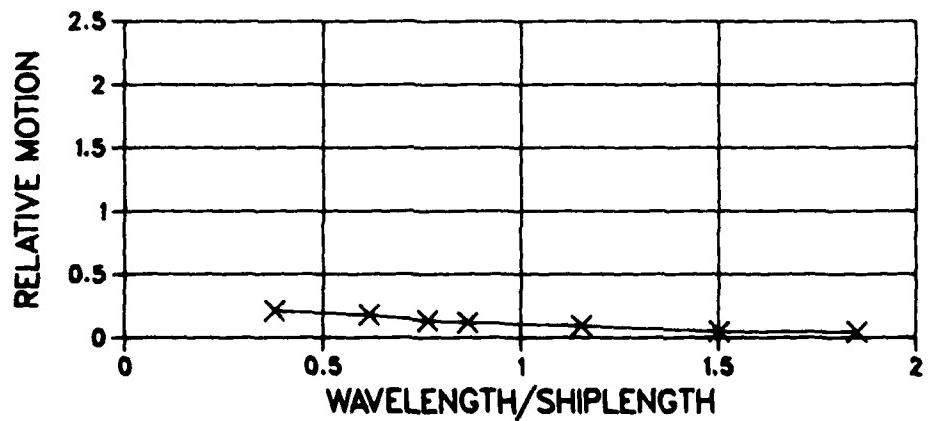
STA.16



STA.14

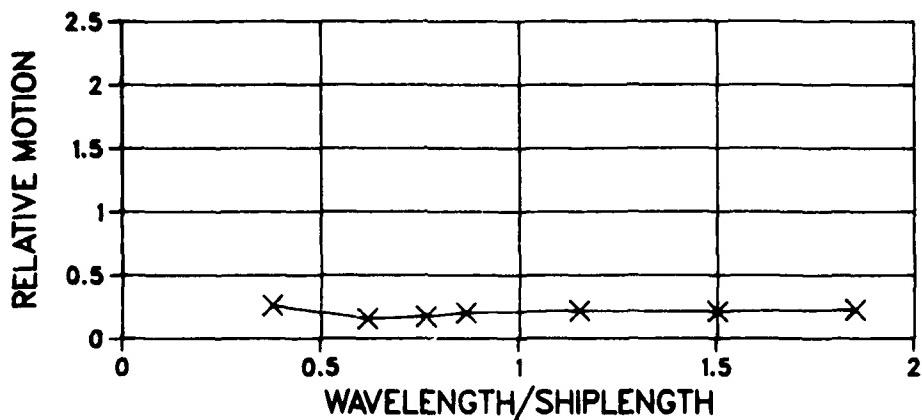


STA.12

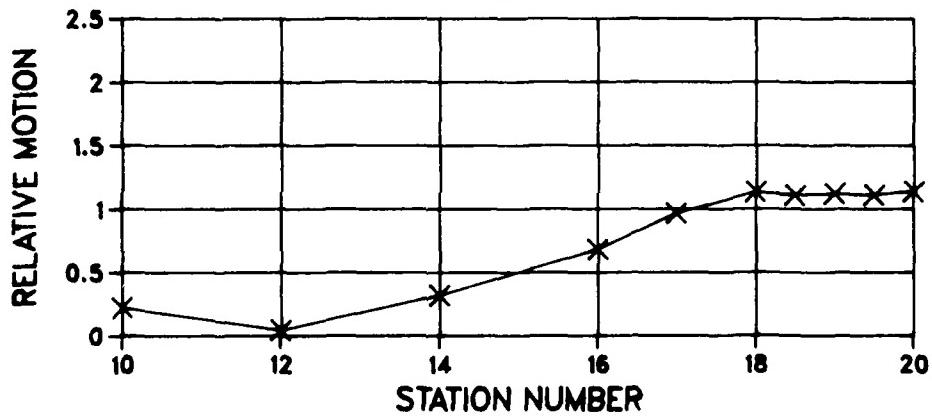


FORCED PITCH. FN=0.2

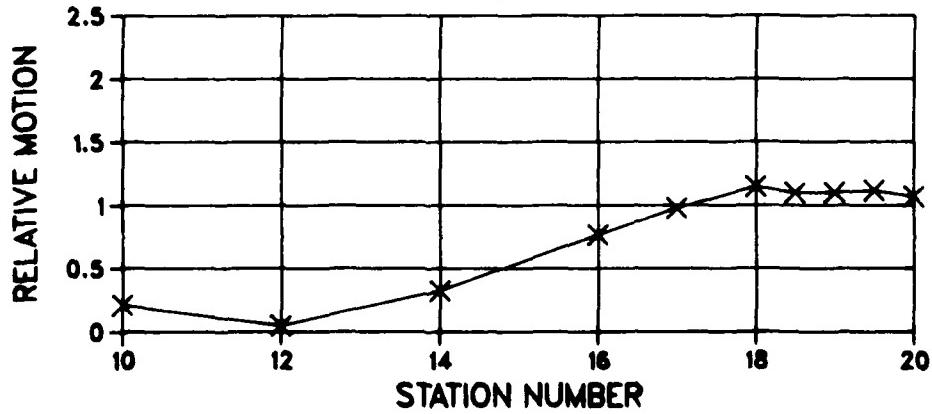
STA.10



LAMBDA/L = 1.852

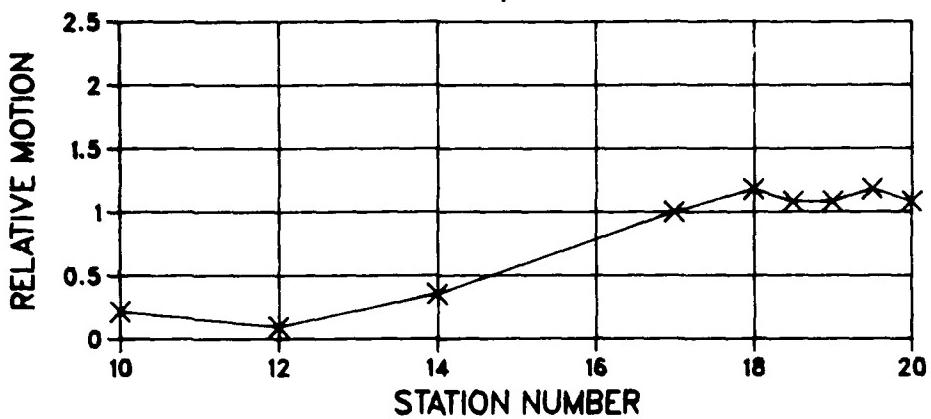


LAMBDA/L = 1.503

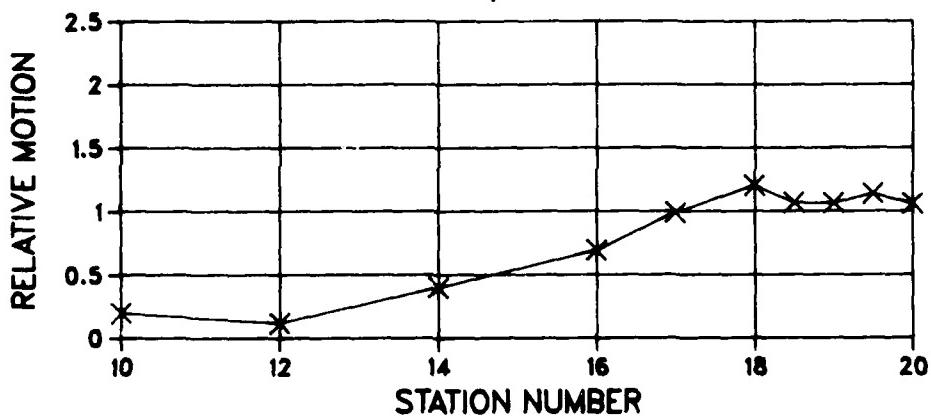


FORCED PITCH. FN=0.2

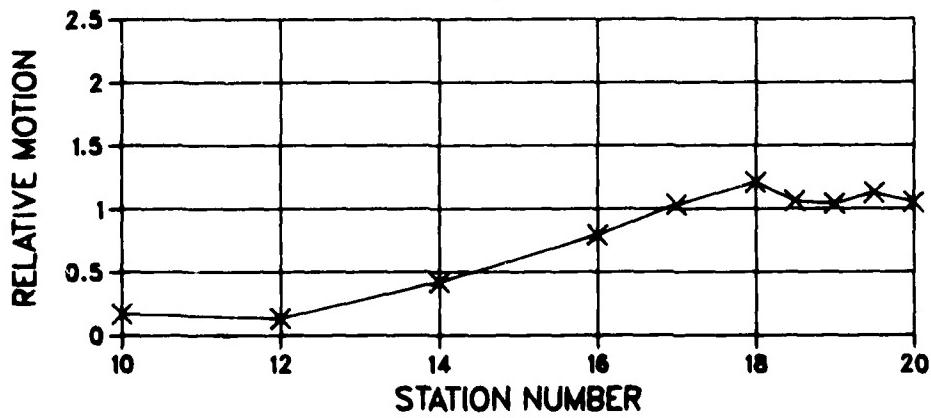
LAMBDA/L = 1.154



LAMBDA/L = 0.867

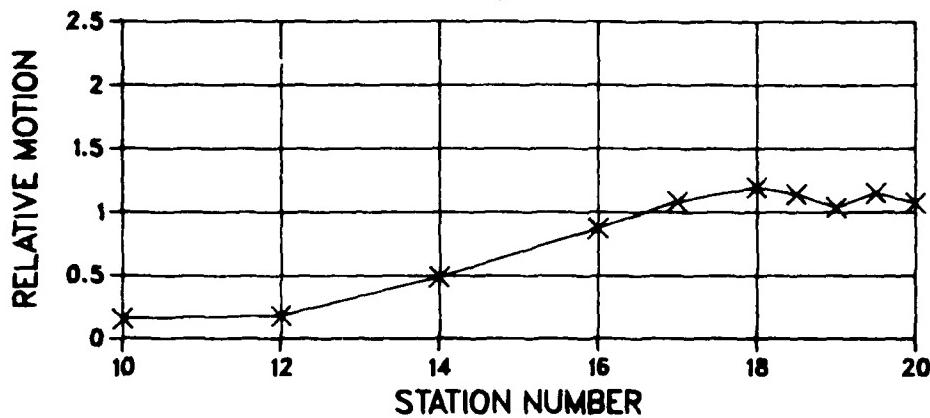


LAMBDA/L = 0.766

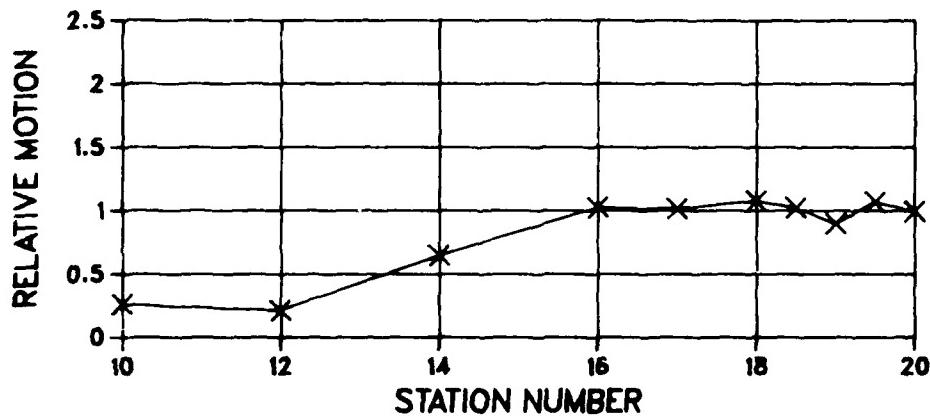


FORCED PITCH. FN=0.2

LAMBDA/L = 0.618

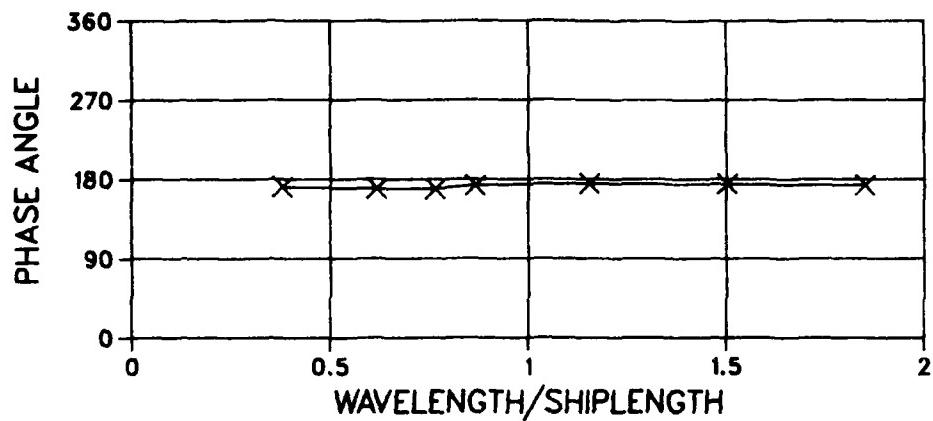


LAMBDA/L = 0.379

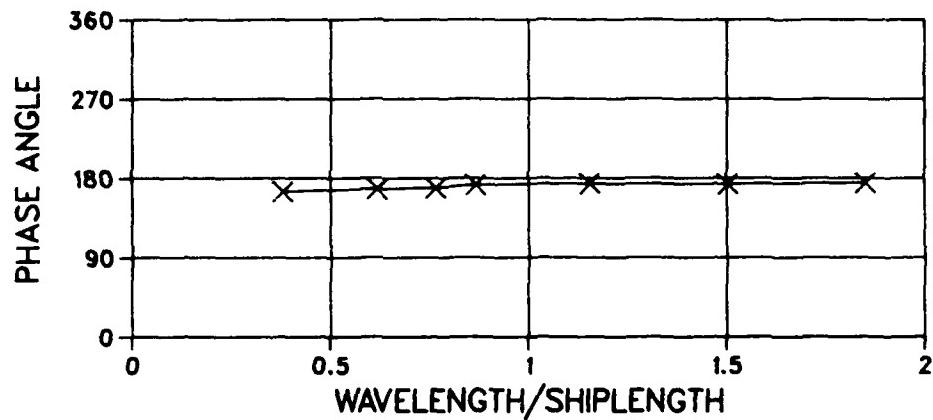


FORCED PITCH. FN=0.2

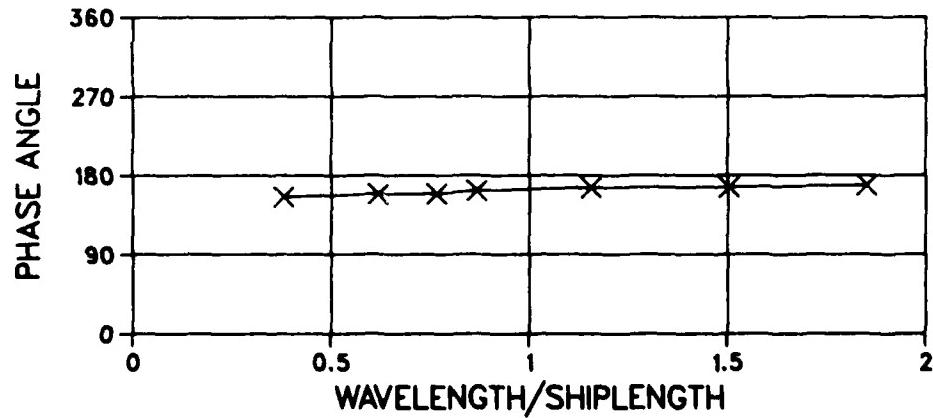
STA.20



STA.19.5

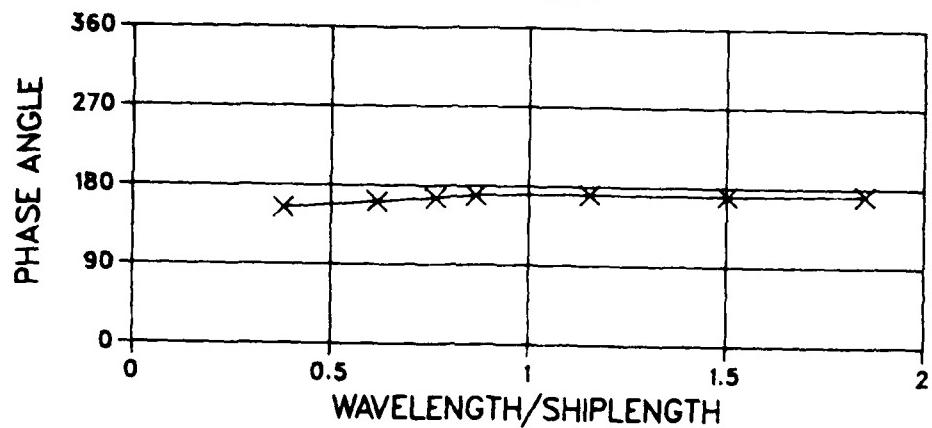


STA.19

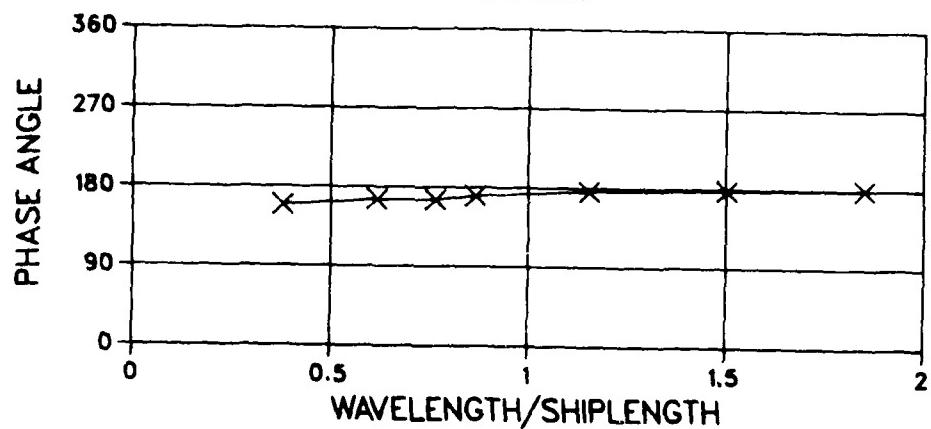


FORCED PITCH. FN=0.2

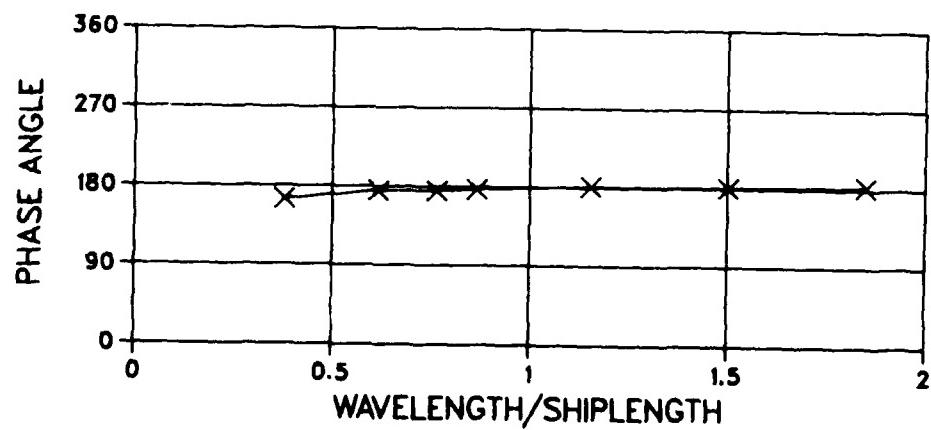
STA.18.5



STA.18

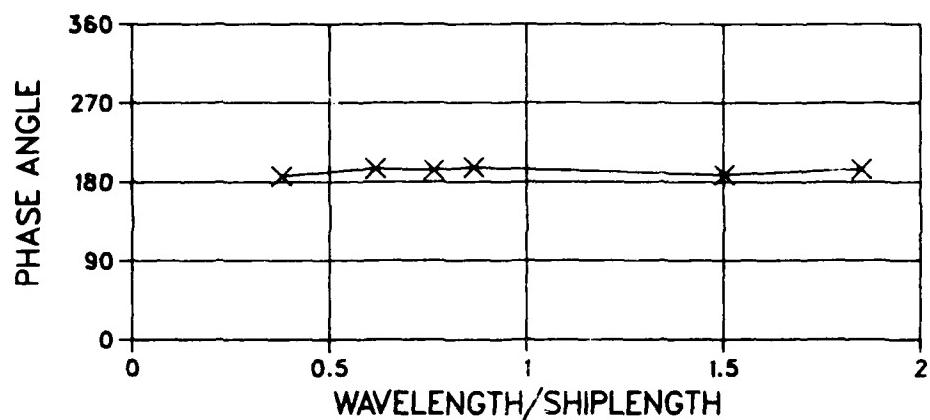


STA.17

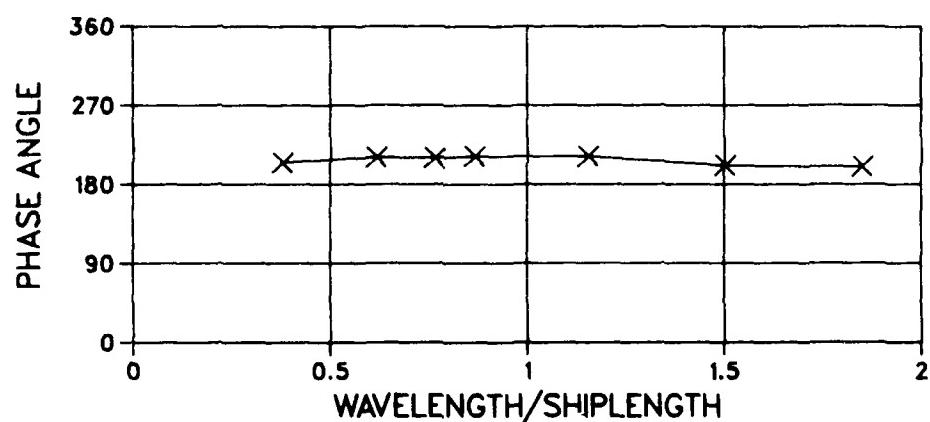


FORCED PITCH. FN=0.2

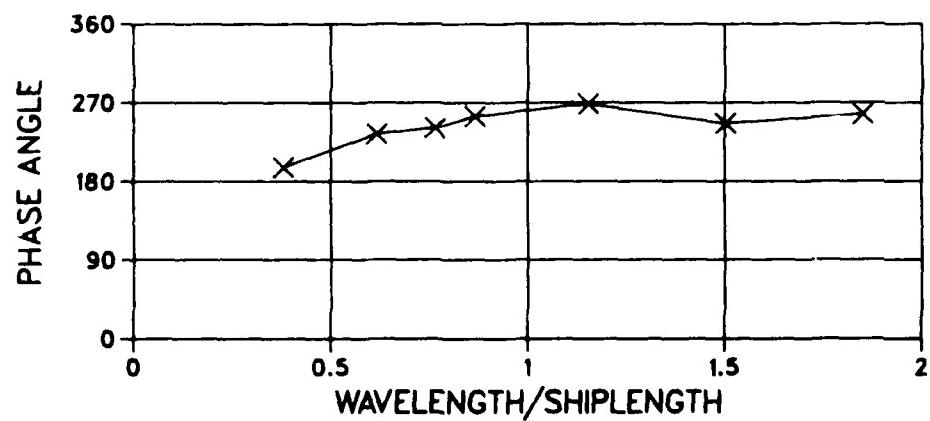
STA.16



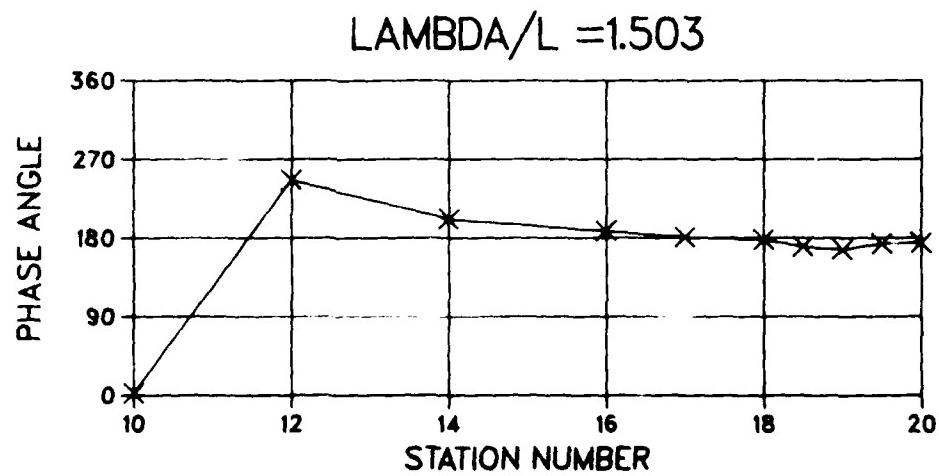
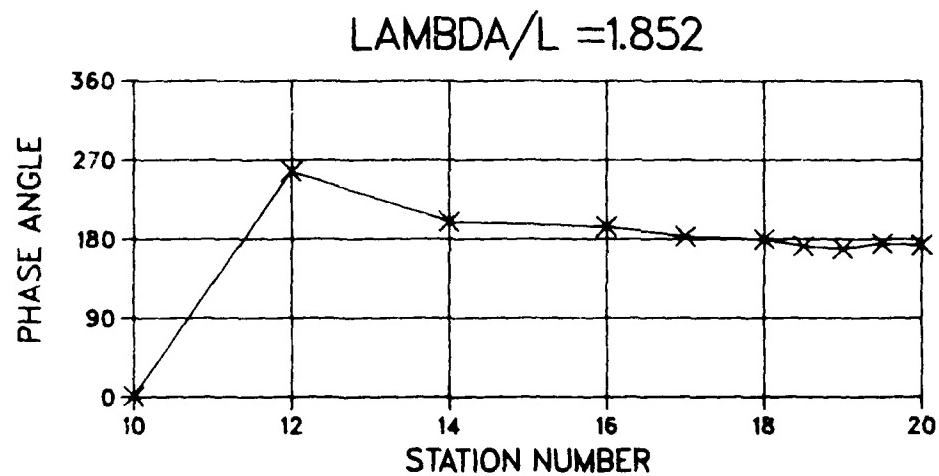
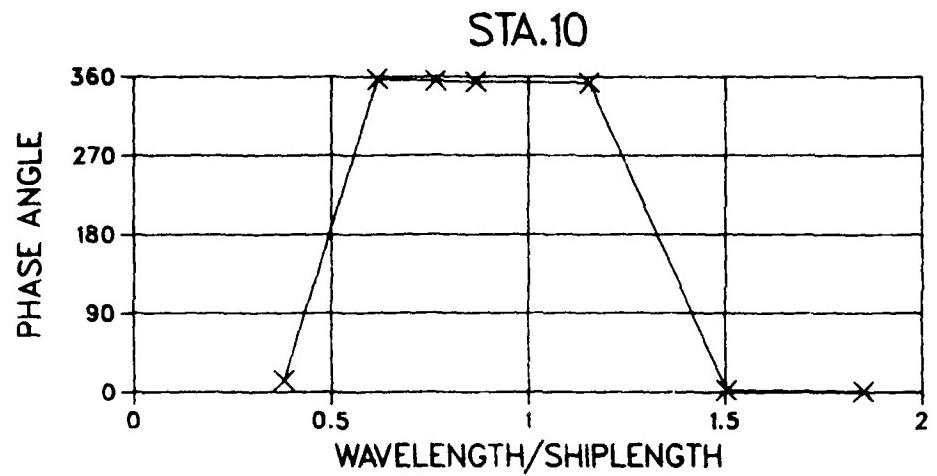
STA.14



STA.12

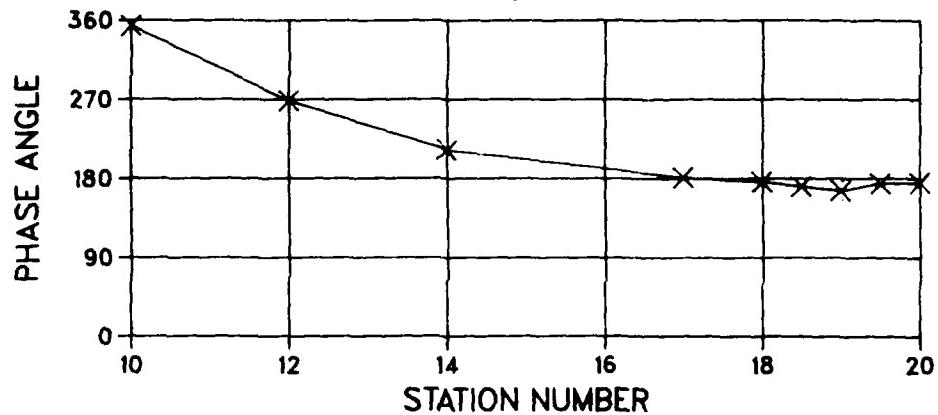


FORCED PITCH. FN=0.2

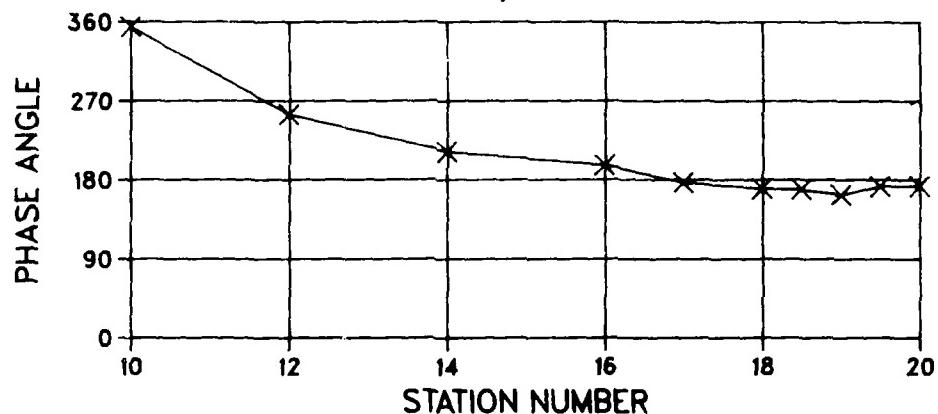


FORCED PITCH. FN=0.2

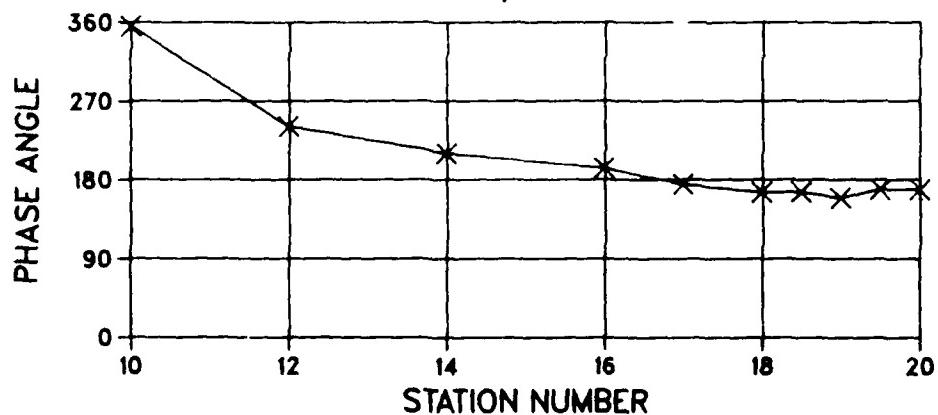
LAMBDA/L = 1.154



LAMBDA/L = 0.867

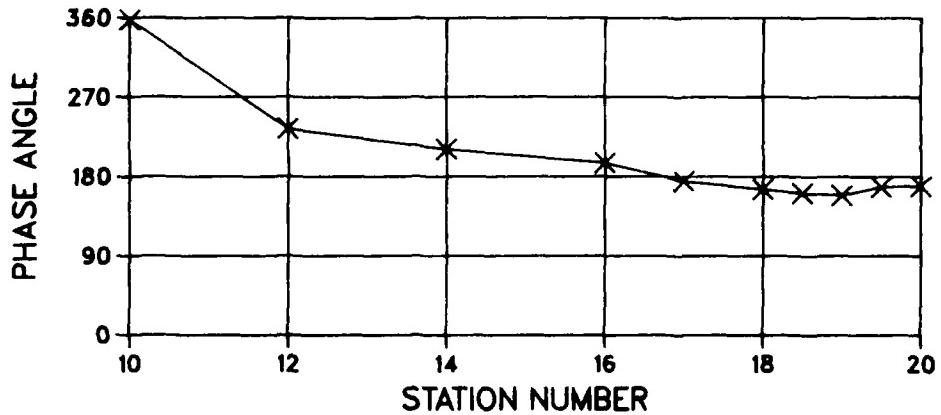


LAMBDA/L = 0.766

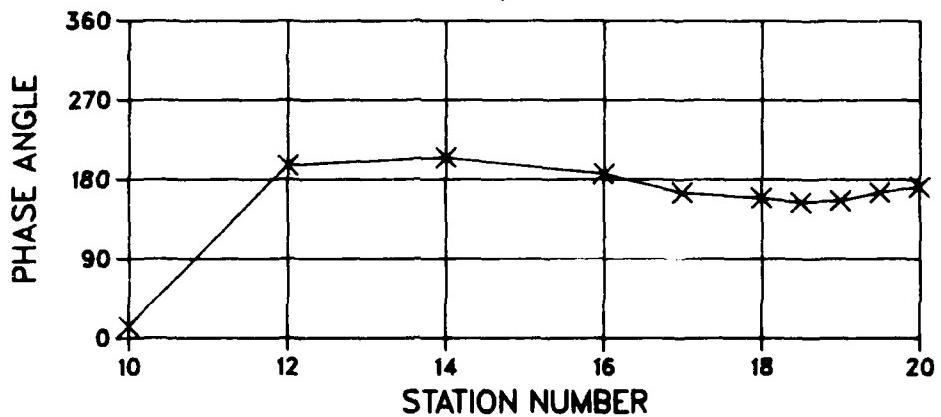


FORCED PITCH. FN=0.2

LAMBDA/L = 0.618

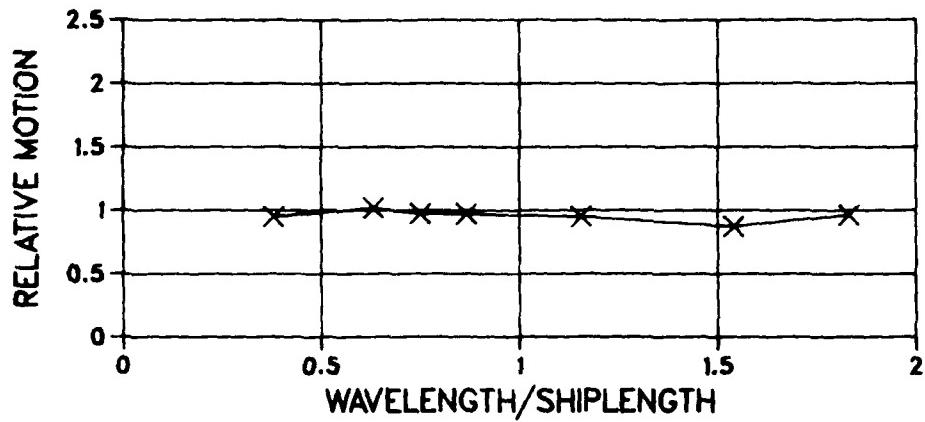


LAMBDA/L = 0.379

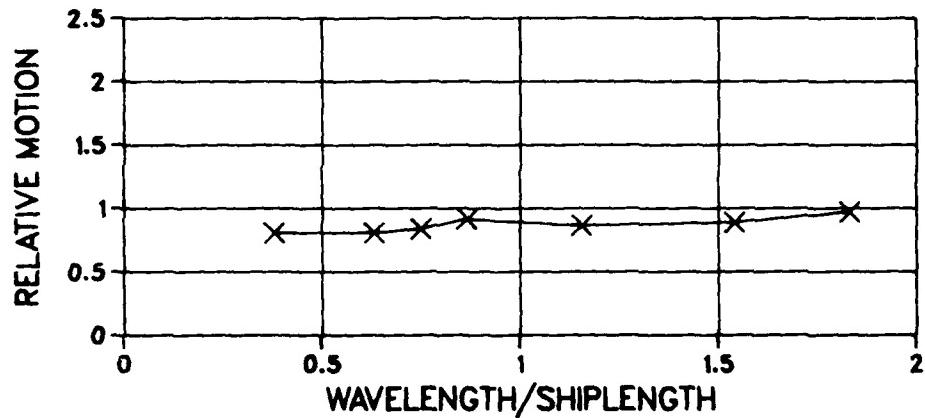


FORCED HEAVE. FN=0.3

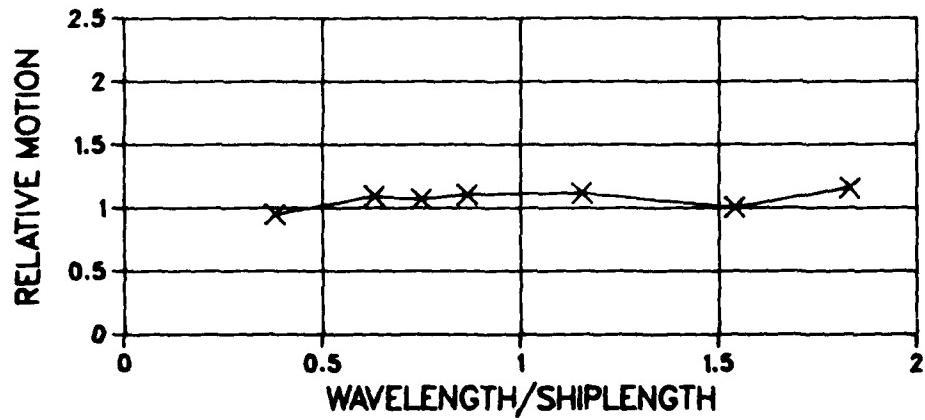
STA.20



STA.19.5

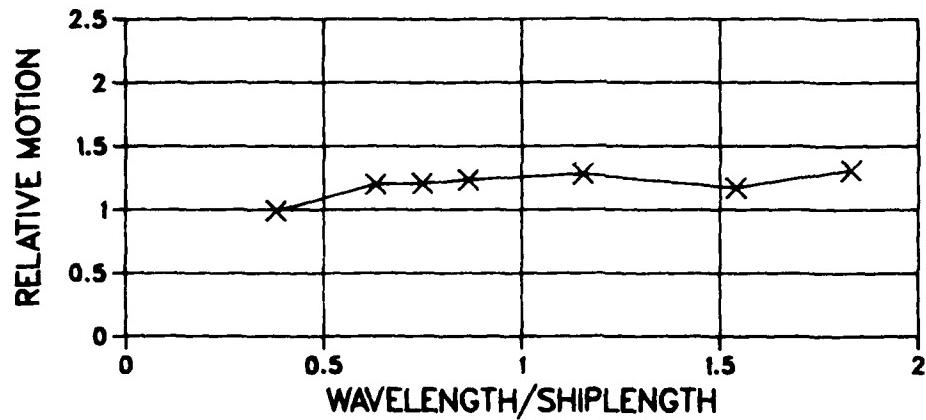


STA.19

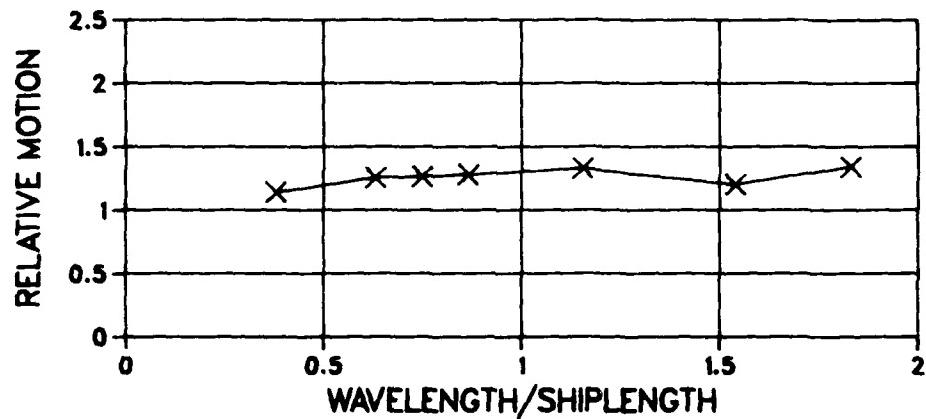


FORCED HEAVE. FN=0.3

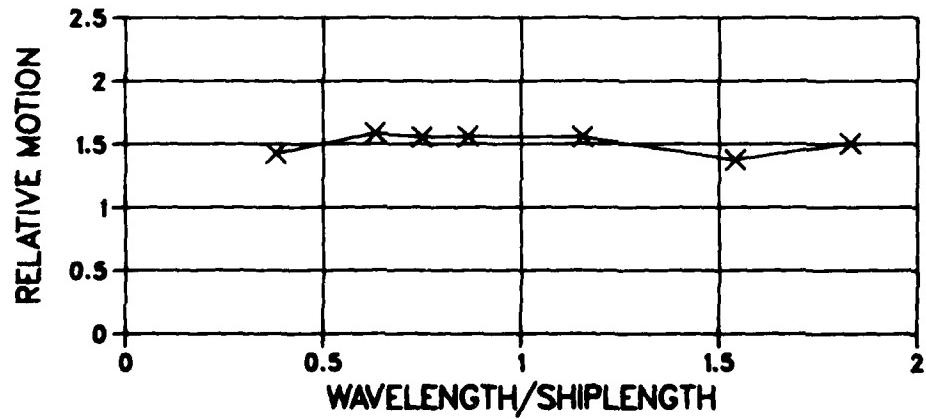
STA.18.5



STA.18

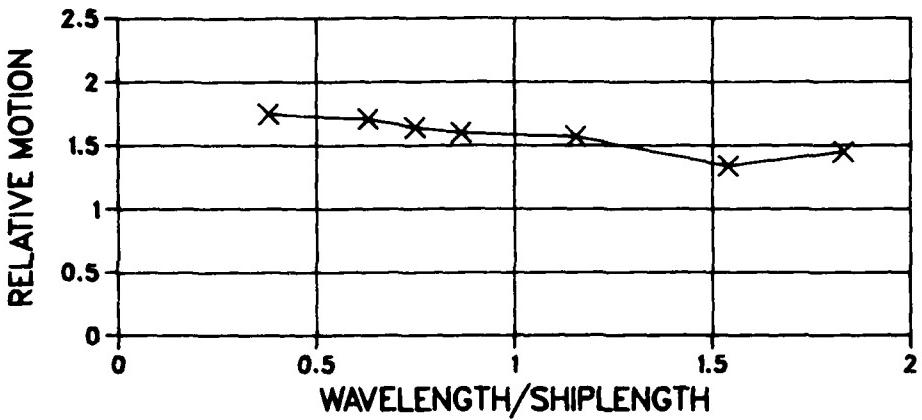


STA.17

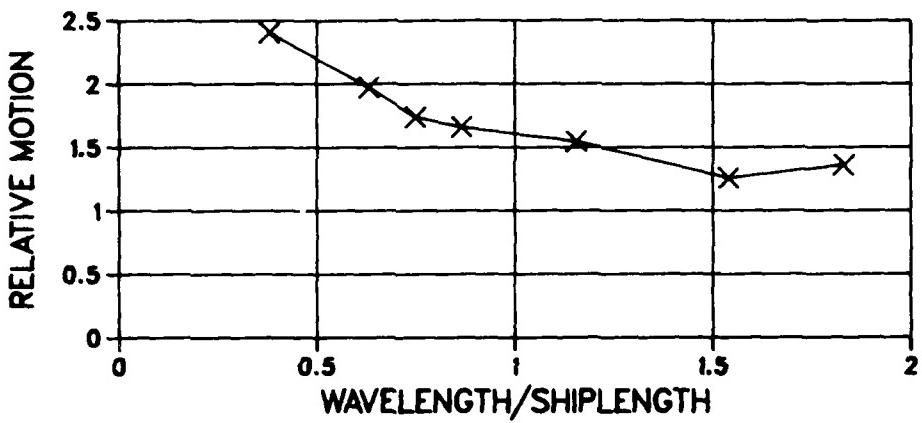


FORCED HEAVE. FN=0.3

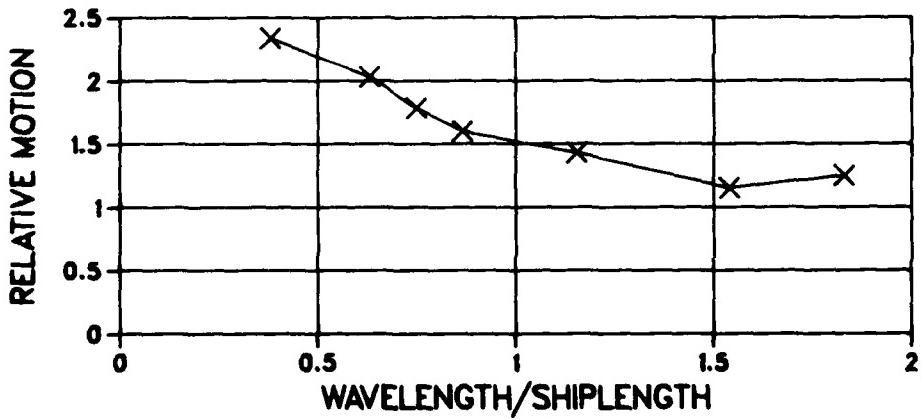
STA.16



STA.14

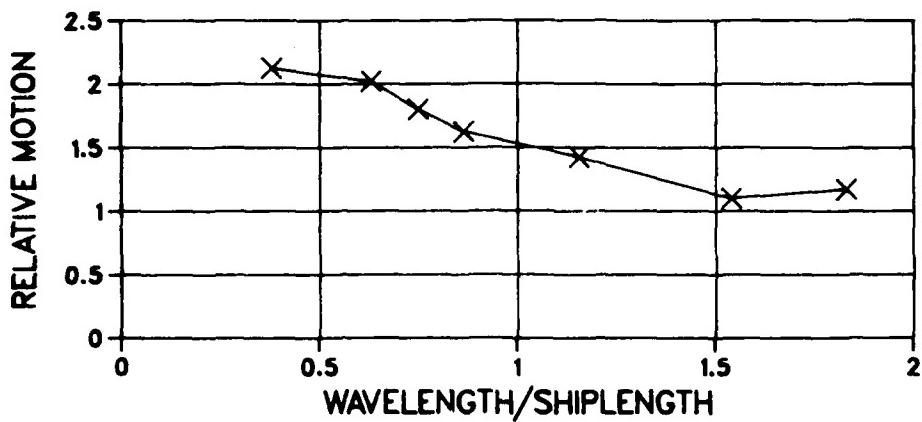


STA.12

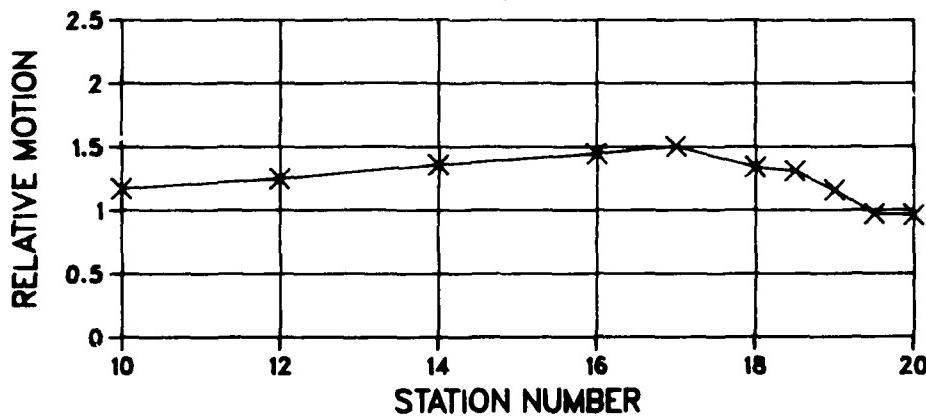


FORCED HEAVE. FN=0.3

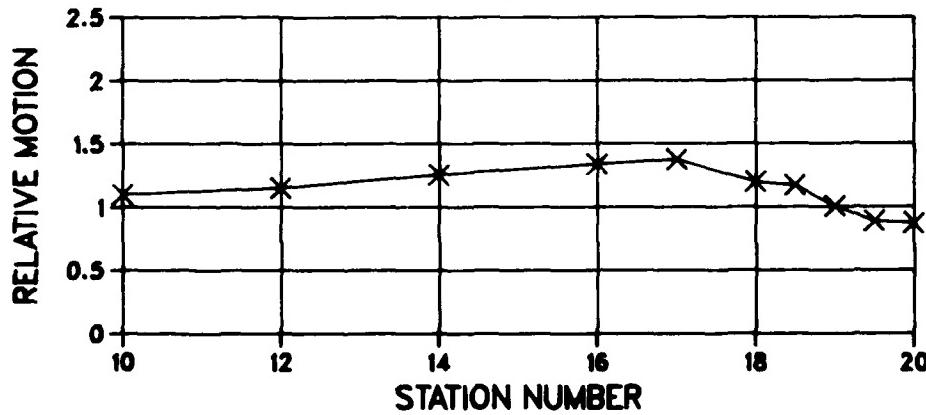
STA.10



$\Lambda/L = 1.831$

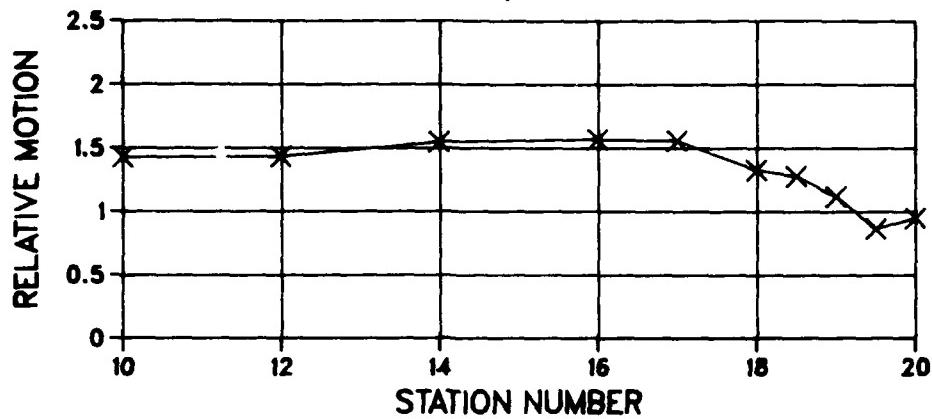


$\Lambda/L = 1.541$

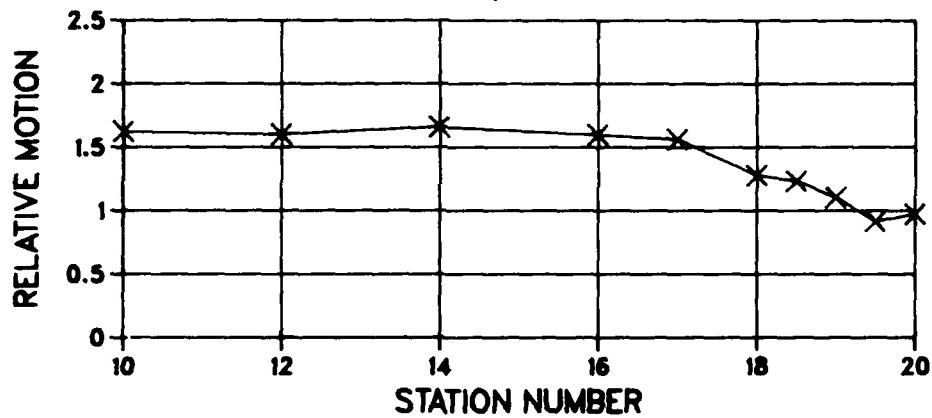


FORCED HEAVE. FN=0.3

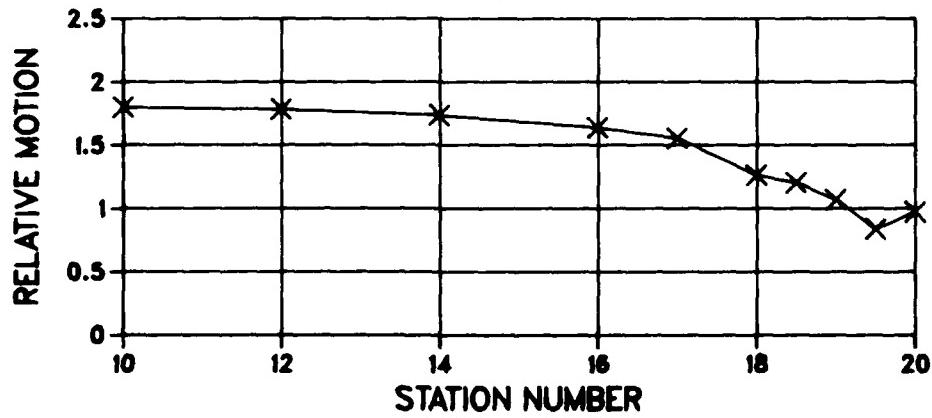
LAMBDA/L = 1.156



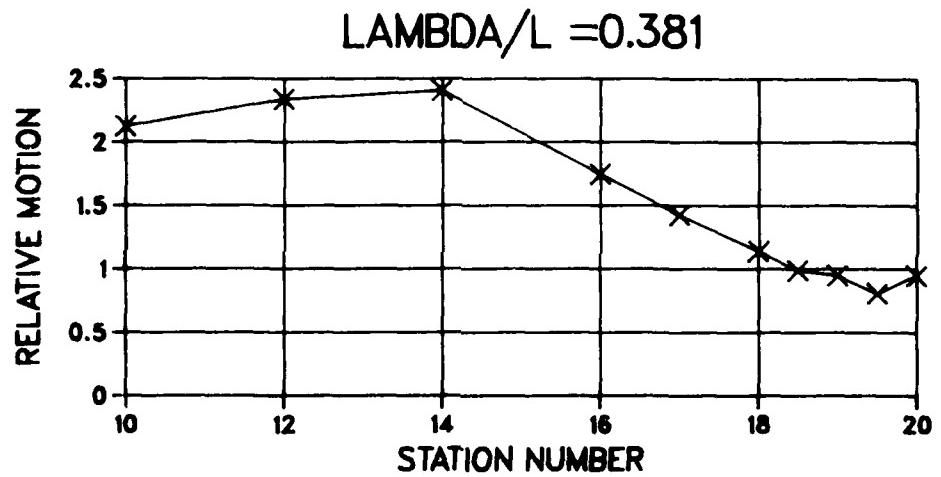
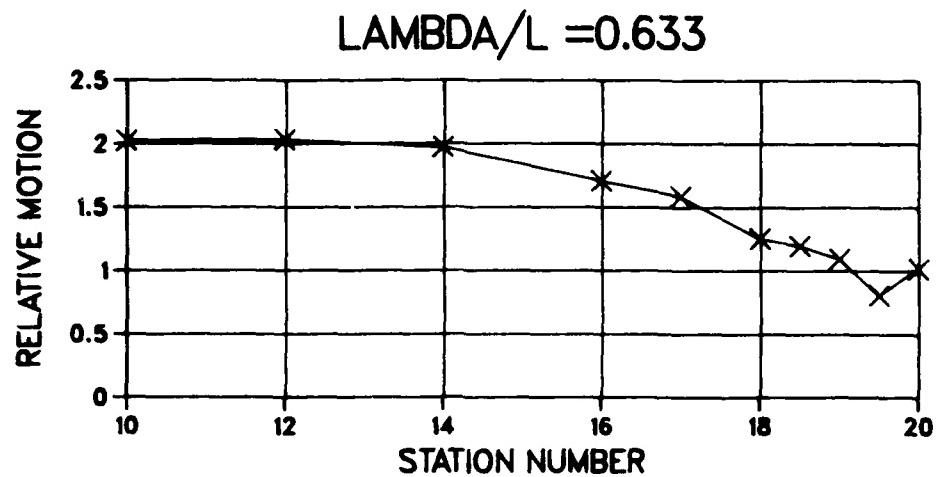
LAMBDA/L = 0.867



LAMBDA/L = 0.751

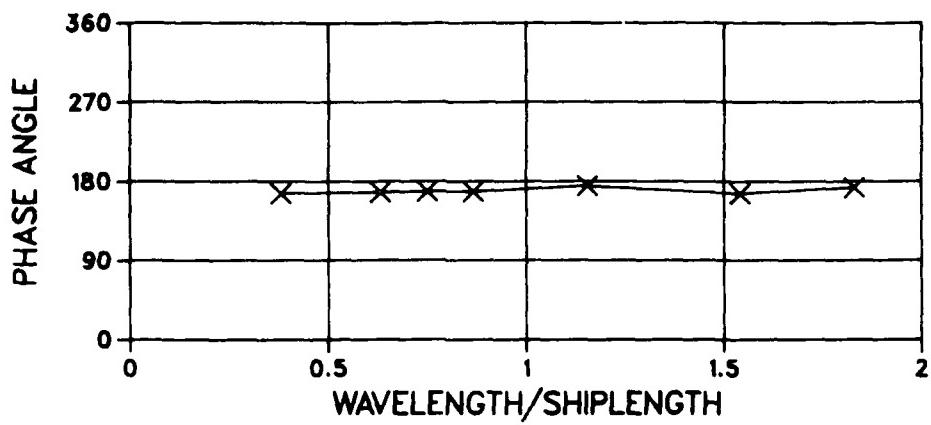


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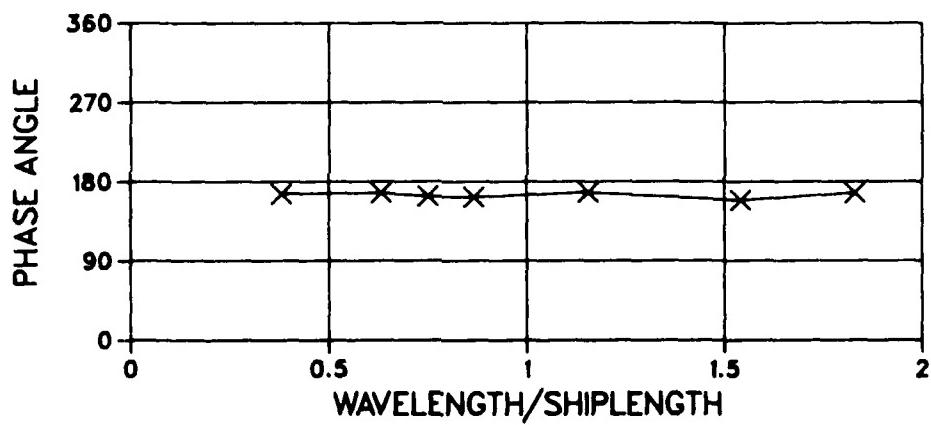


FORCED HEAVE. FN=0.3

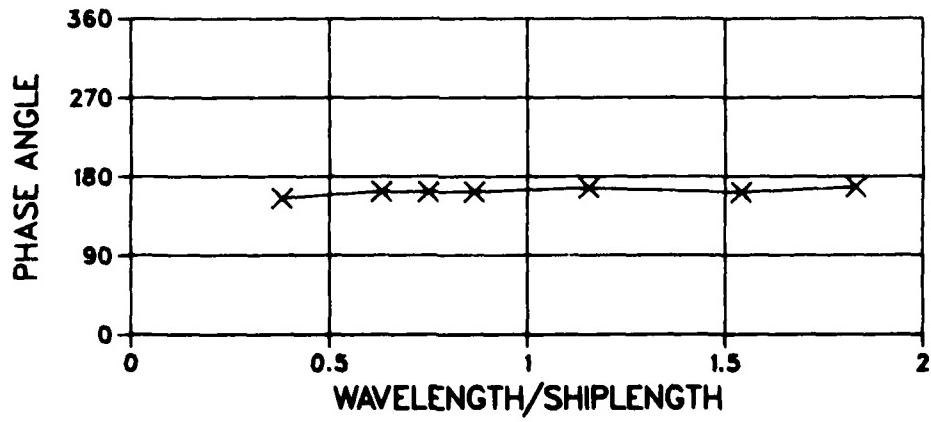
STA.20



STA.19.5

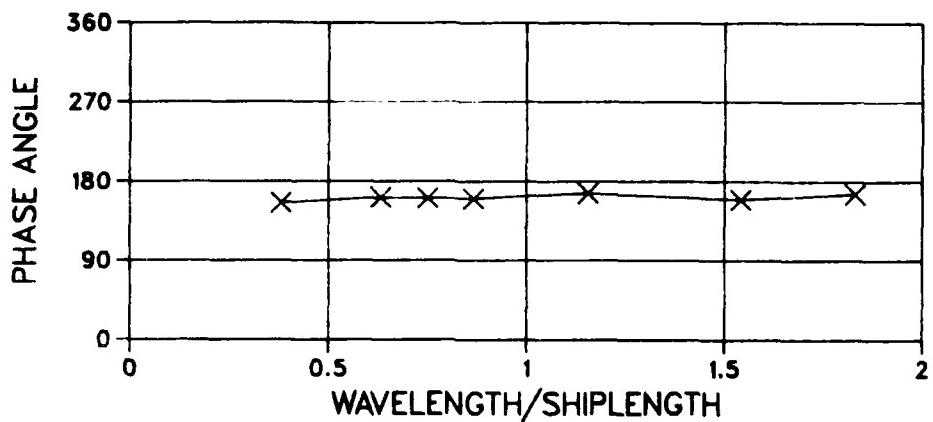


STA.19

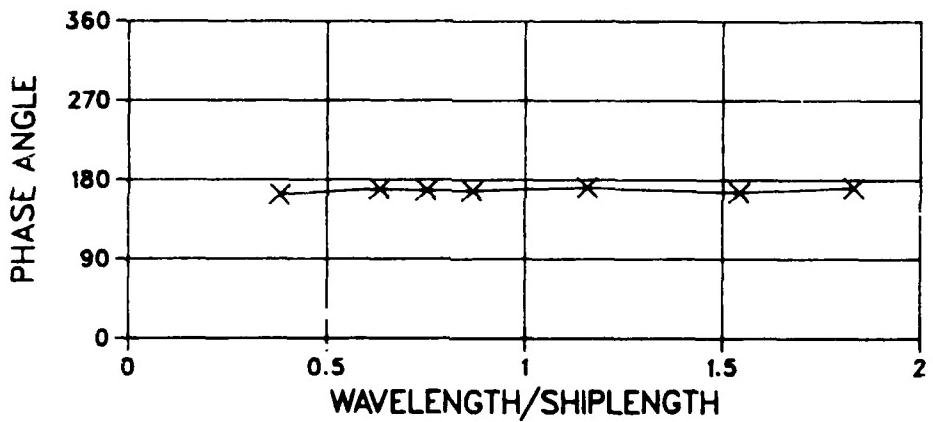


FORCED HEAVE. FN=0.3

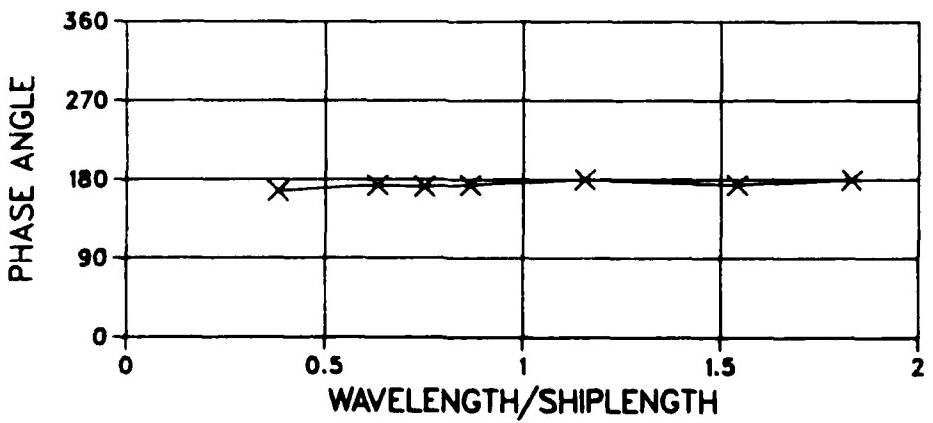
STA.18.5



STA.18

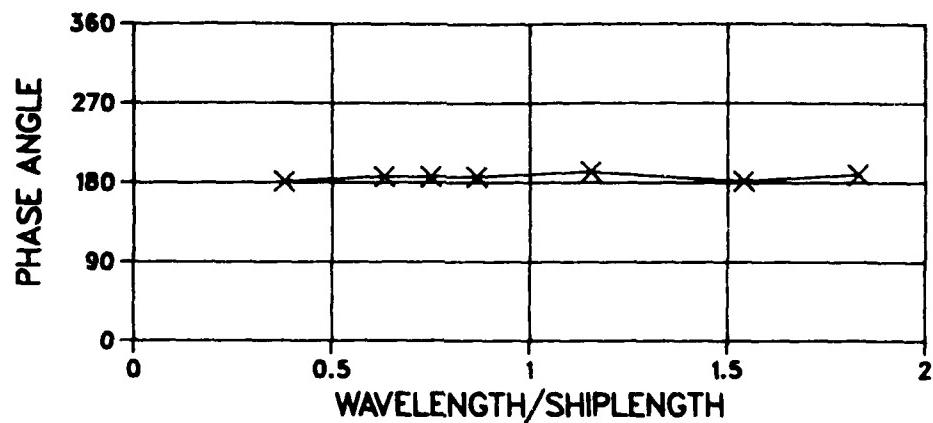


STA.17

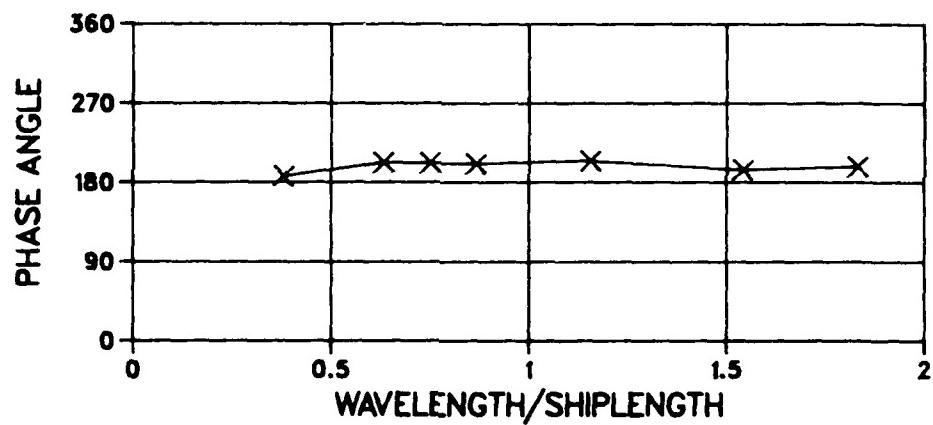


FORCED HEAVE. FN=0.3

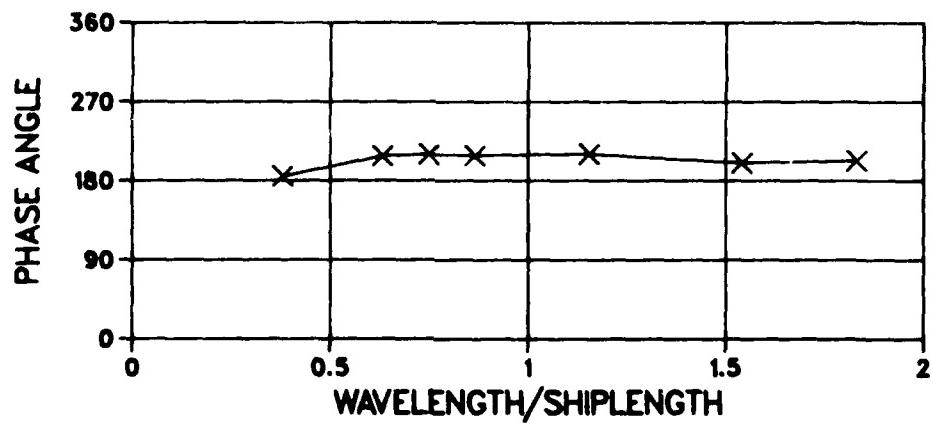
STA.16



STA.14

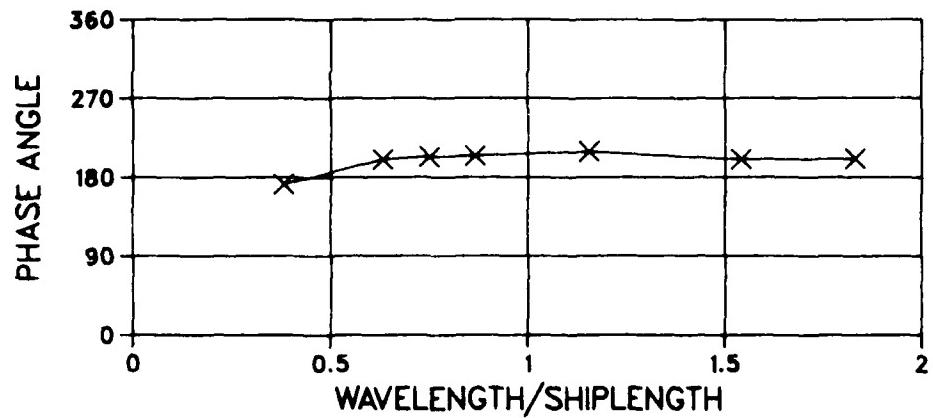


STA.12

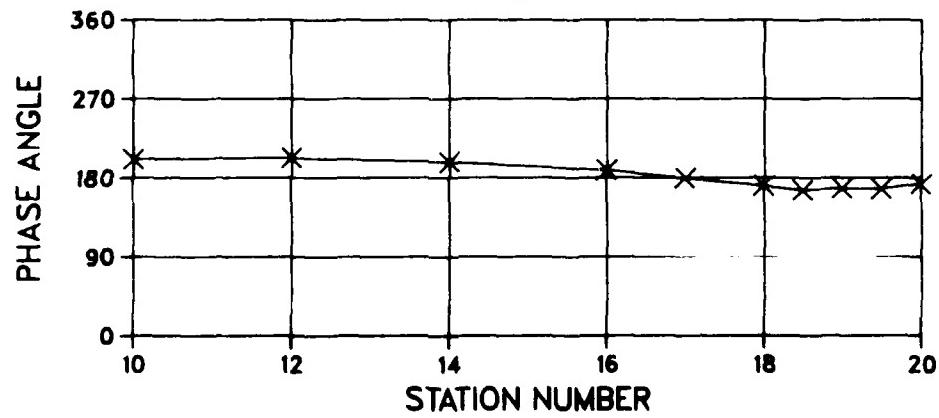


FORCED HEAVE. FN=0.3

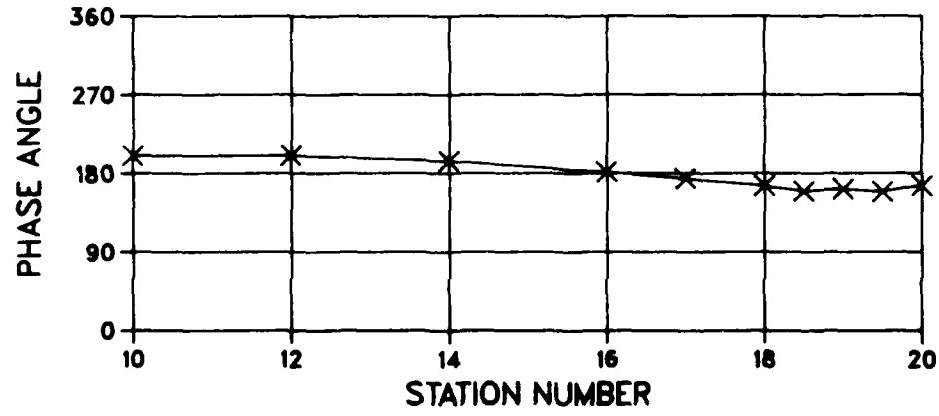
STA.10



LAMBDA/L = 1.831

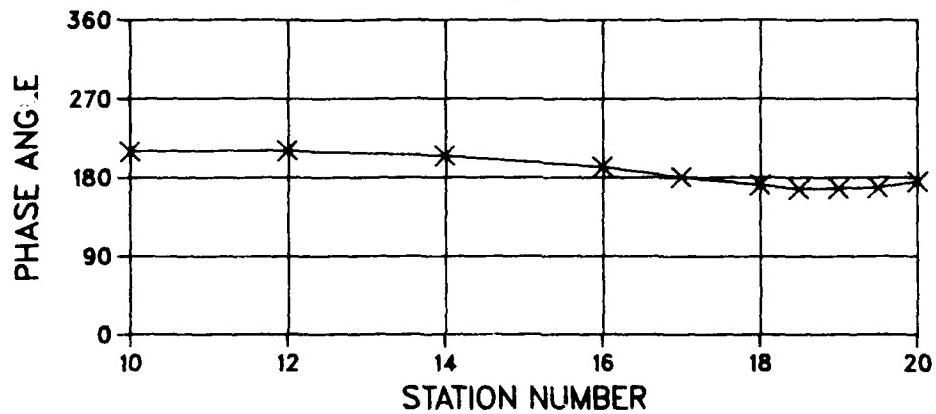


LAMBDA/L = 1.541

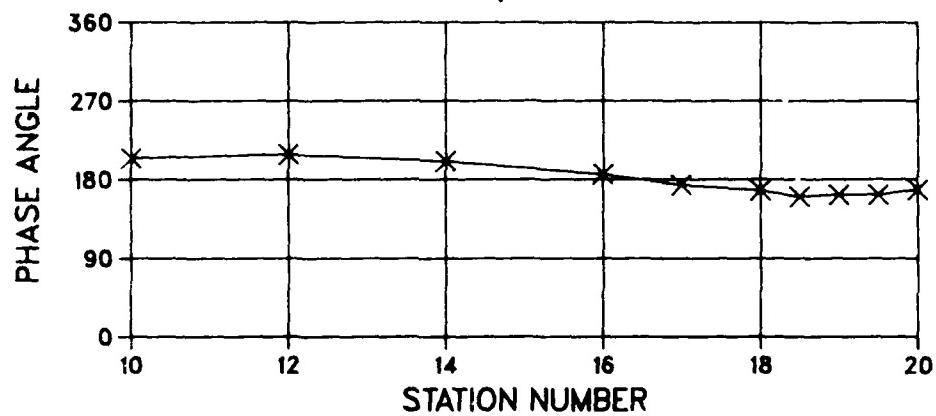


FORCED HEAVE. FN=0.3

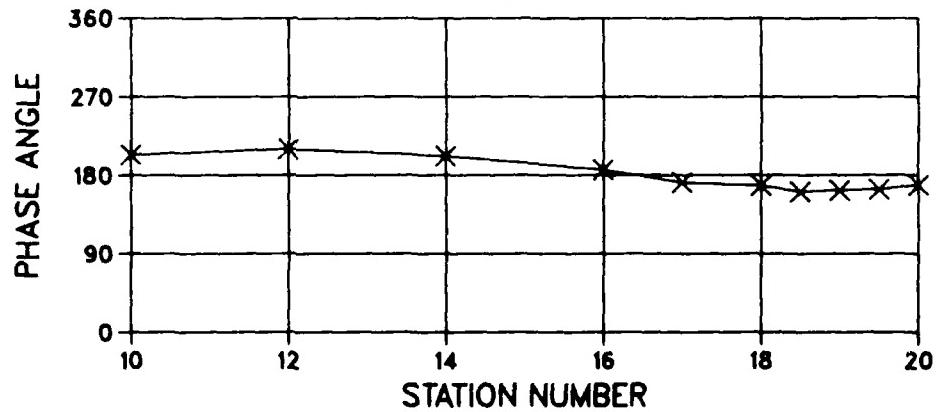
LAMBDA/L = 1.156



LAMBDA/L = 0.867

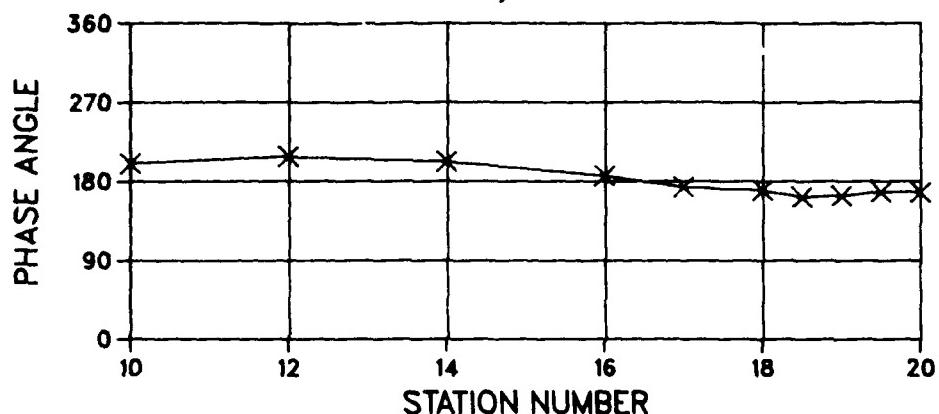


LAMBDA/L = 0.751

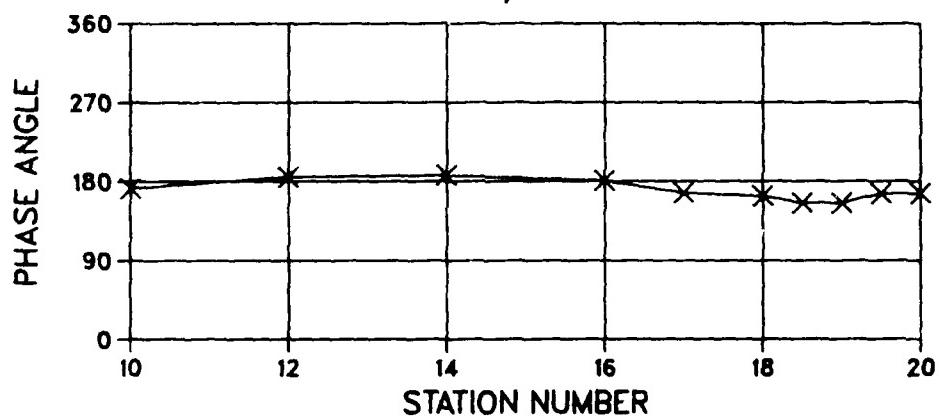


FORCED HEAVE. FN=0.3

LAMBDA/L = 0.633

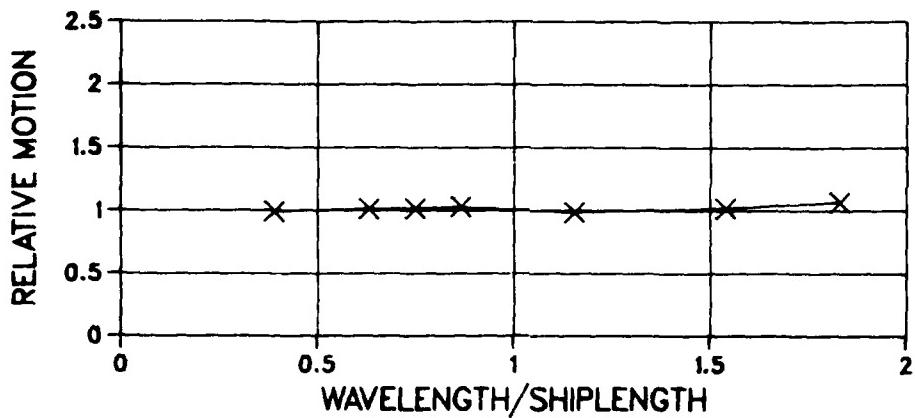


LAMBDA/L = 0.381

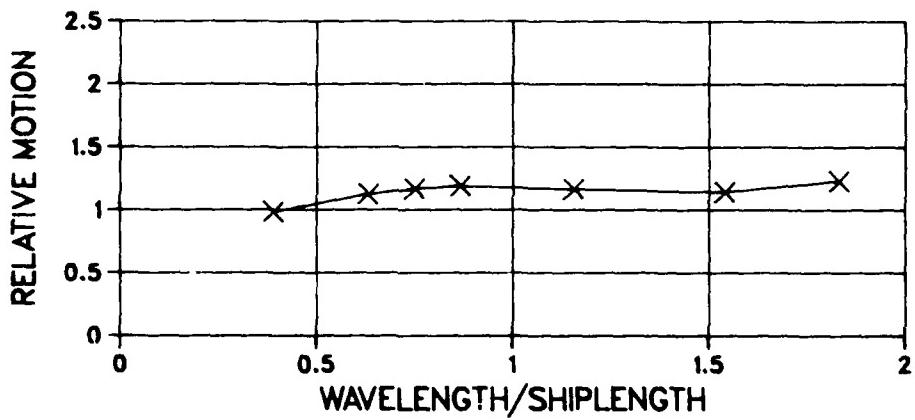


FORCED PITCH. FN=0.3

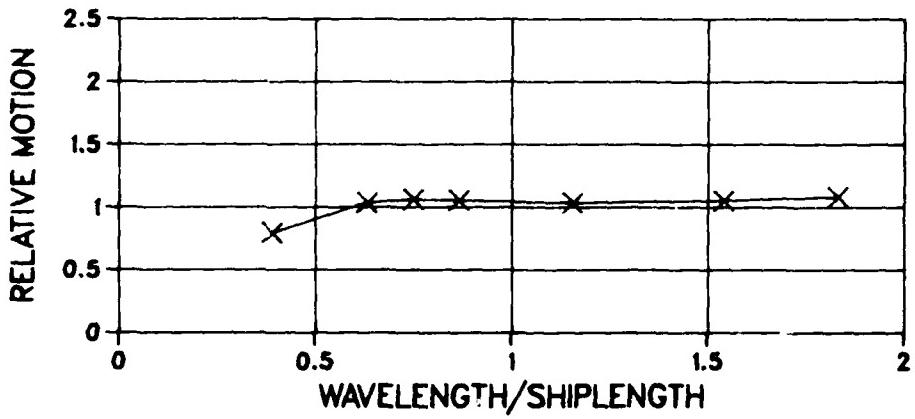
STA.20



STA.19.5

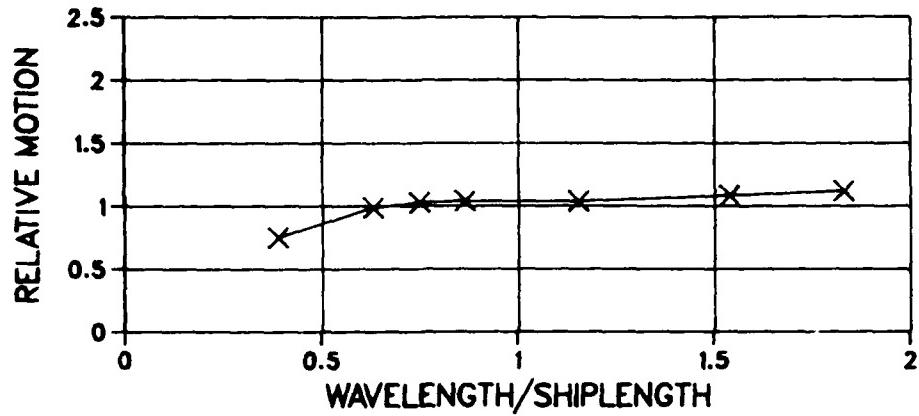


STA.19

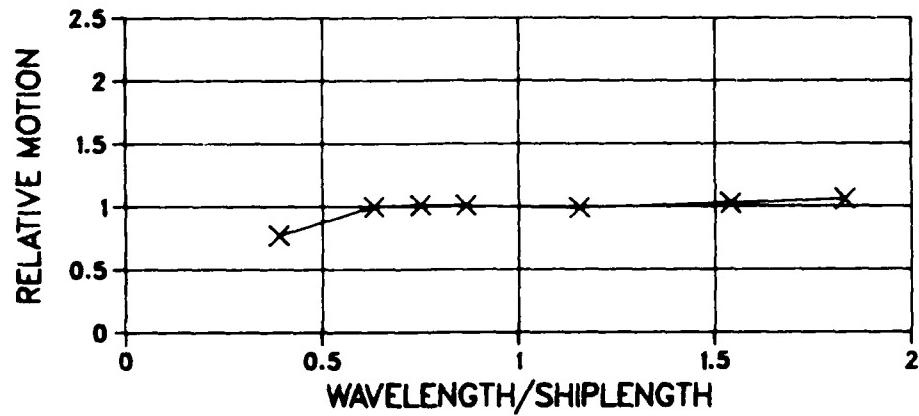


FORCED PITCH. FN=0.3

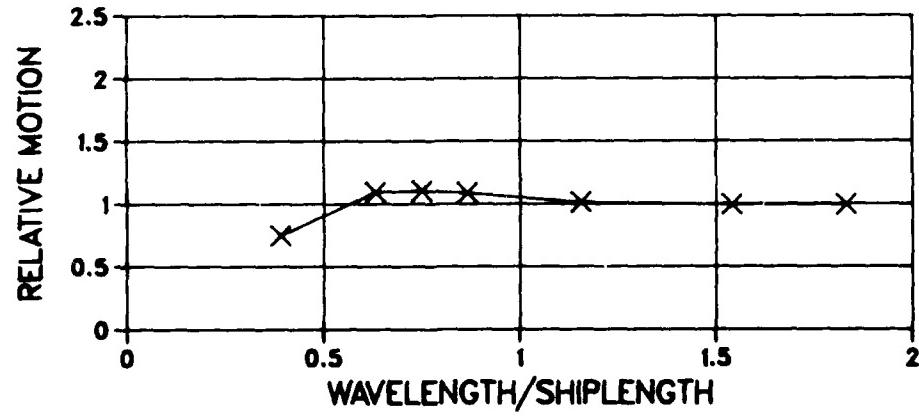
STA.18.5



STA.18

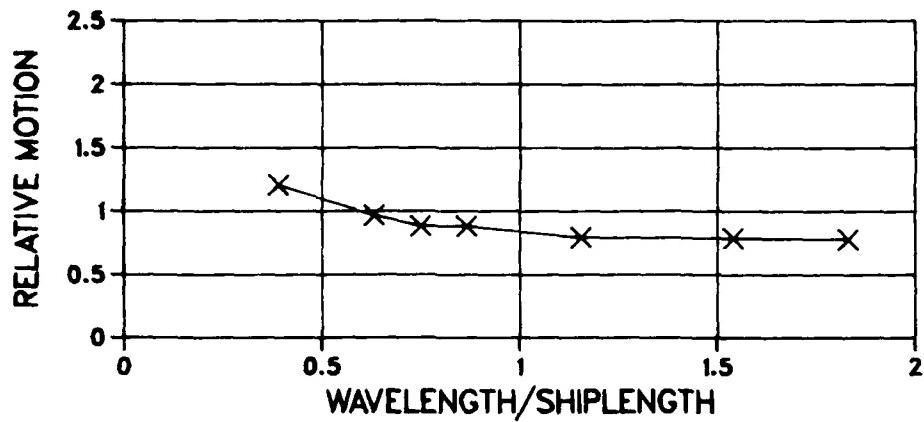


STA.17

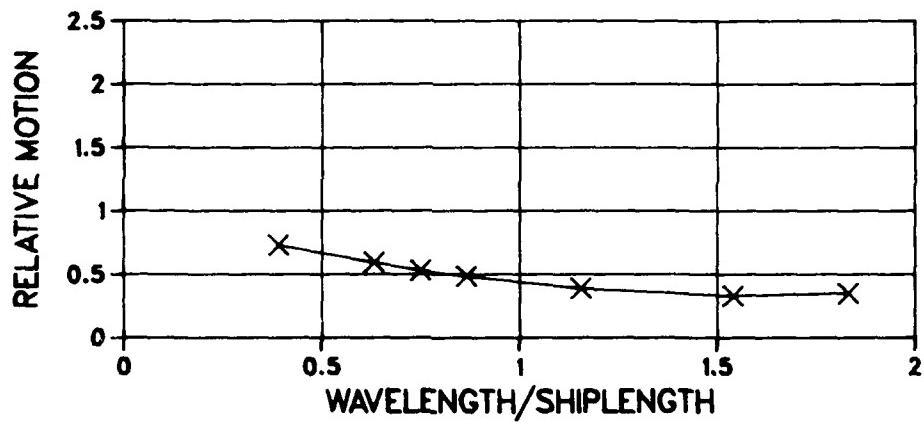


FORCED PITCH. FN=0.3

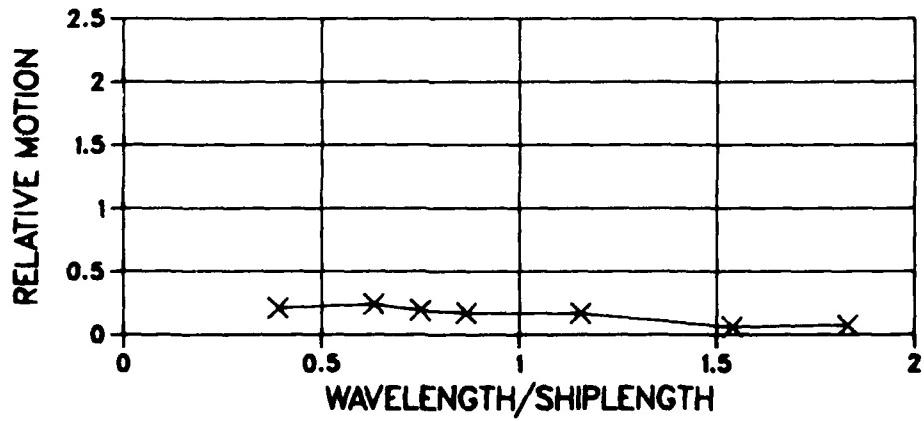
STA.16



STA.14

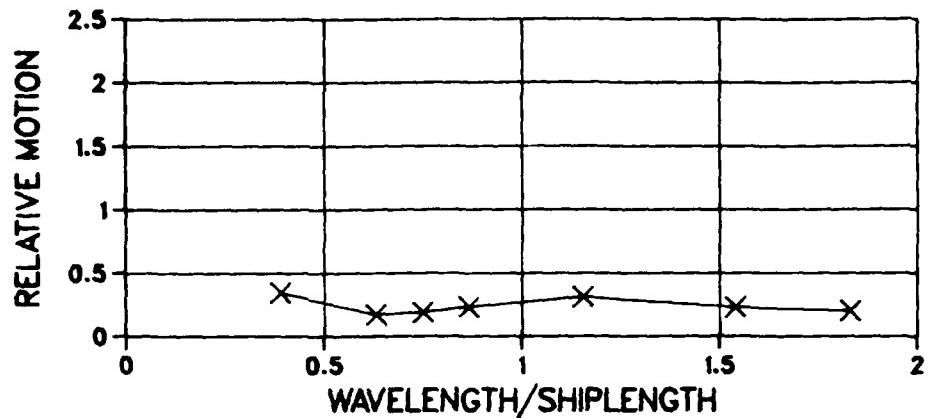


STA.12

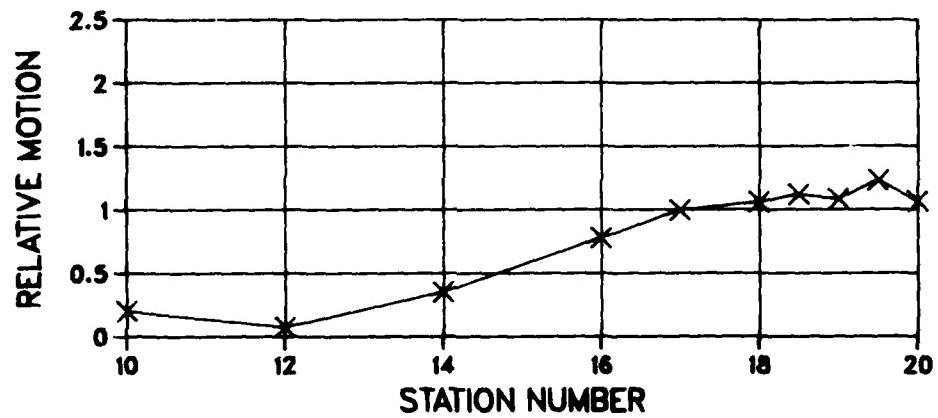


FORCED PITCH. FN=0.3

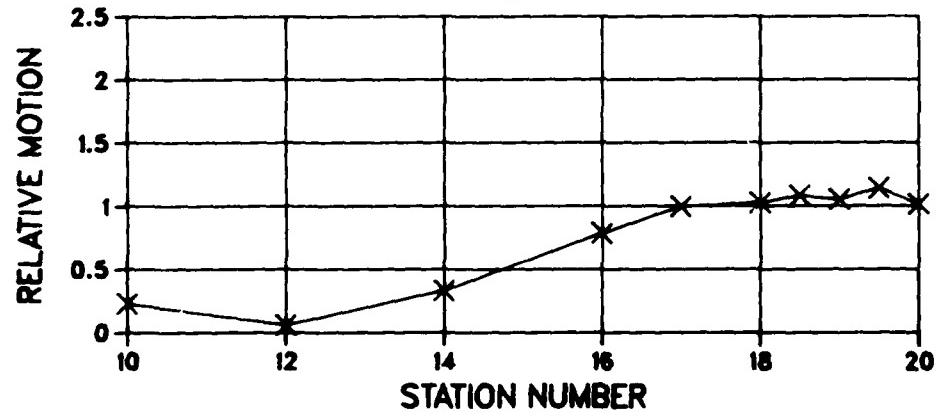
STA.10



$\Lambda/L = 1.831$

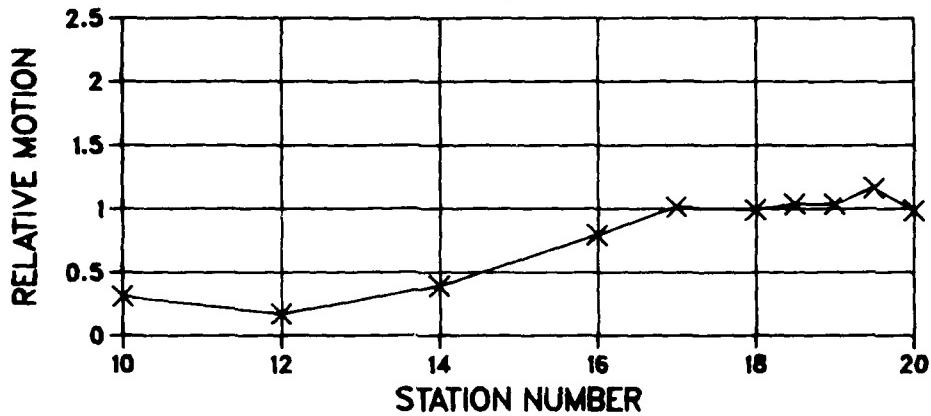


$\Lambda/L = 1.541$

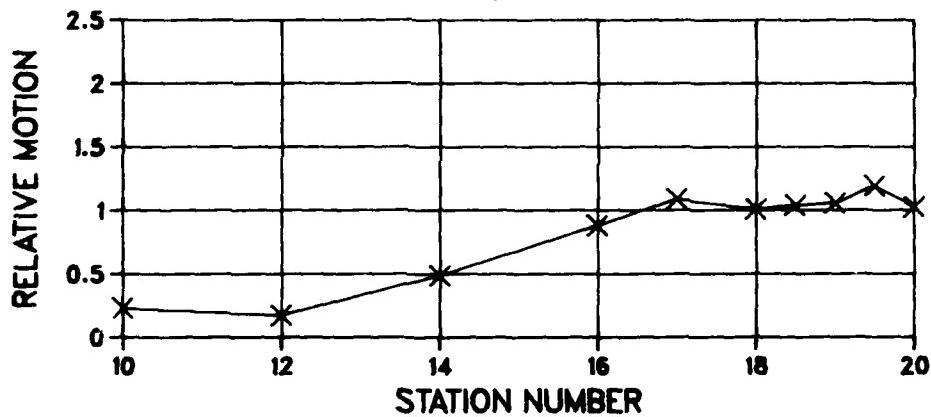


FORCED PITCH. FN=0.3

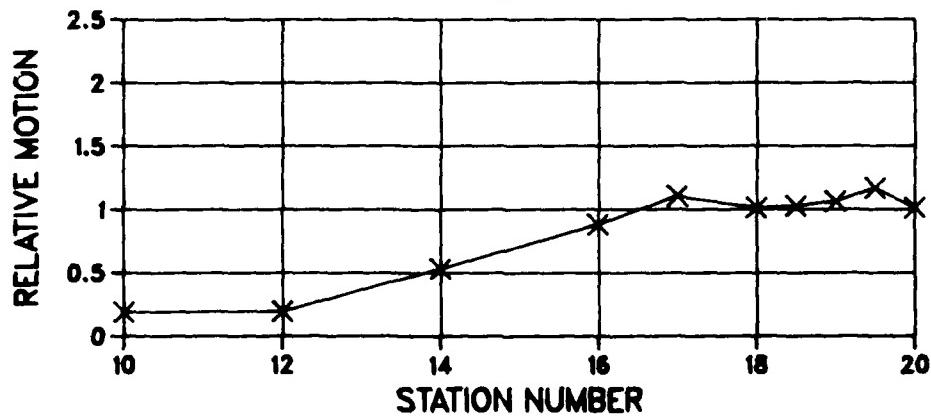
LAMBDA/L = 1.156



LAMBDA/L = 0.867

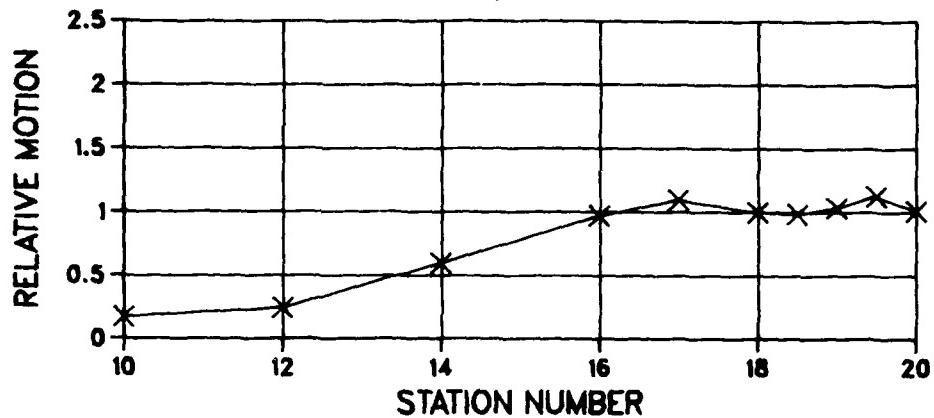


LAMBDA/L = 0.751

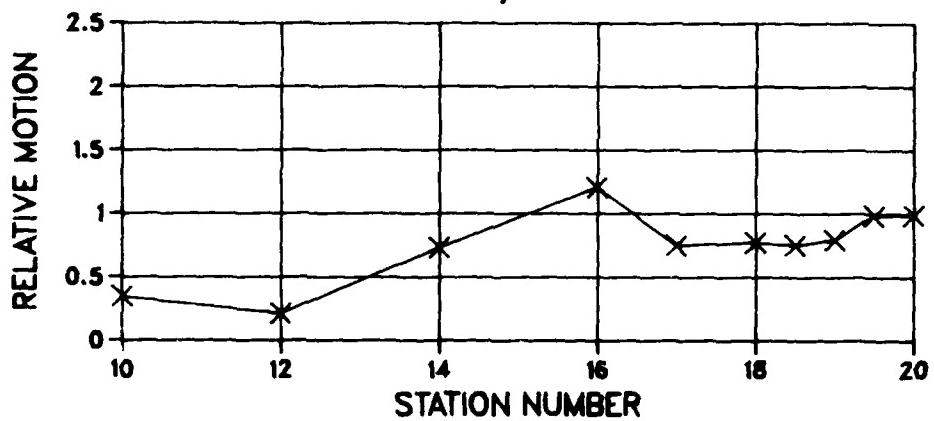


FORCED PITCH. FN=0.3

LAMBDA/L = 0.633

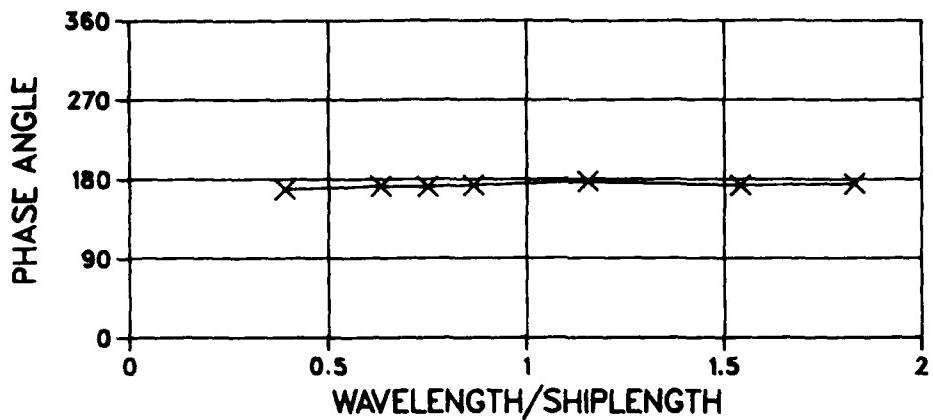


LAMBDA/L = 0.391

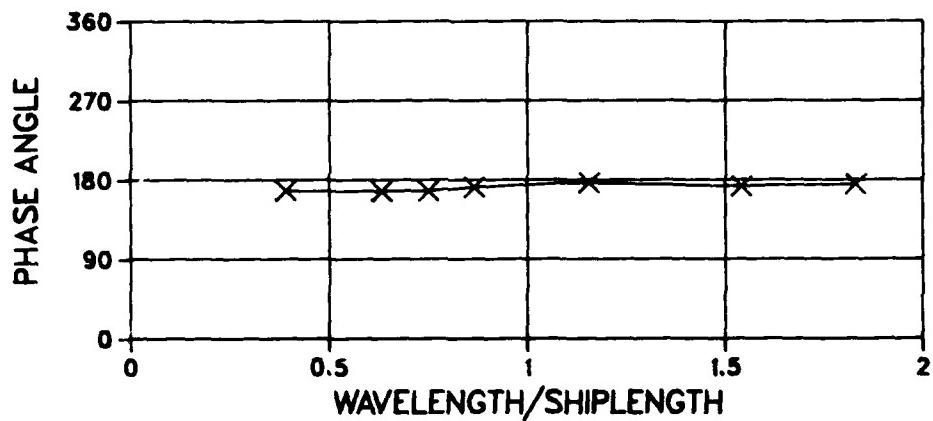


FORCED PITCH. FN=0.3

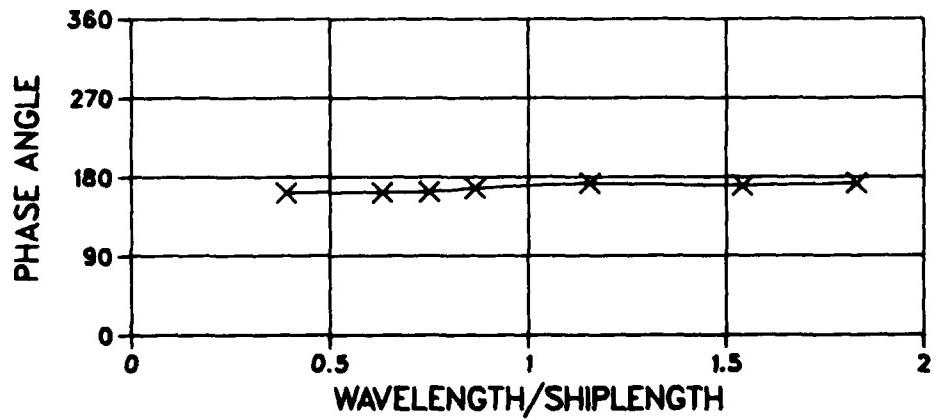
STA.20



STA.19.5

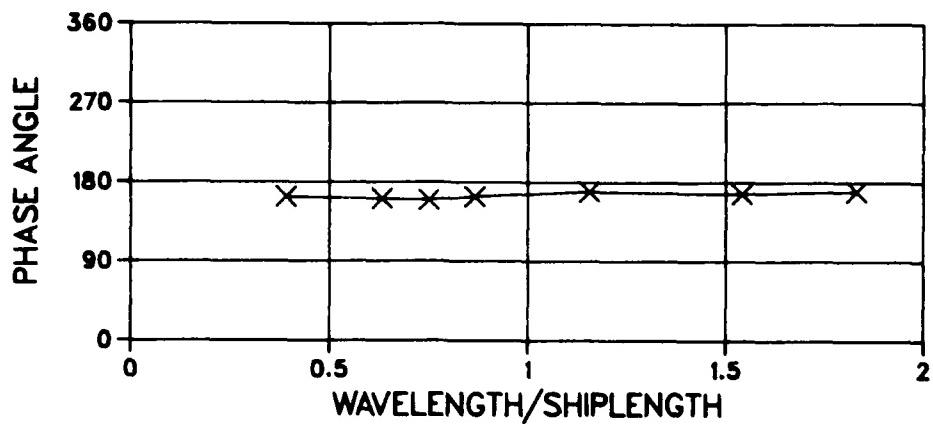


STA.19

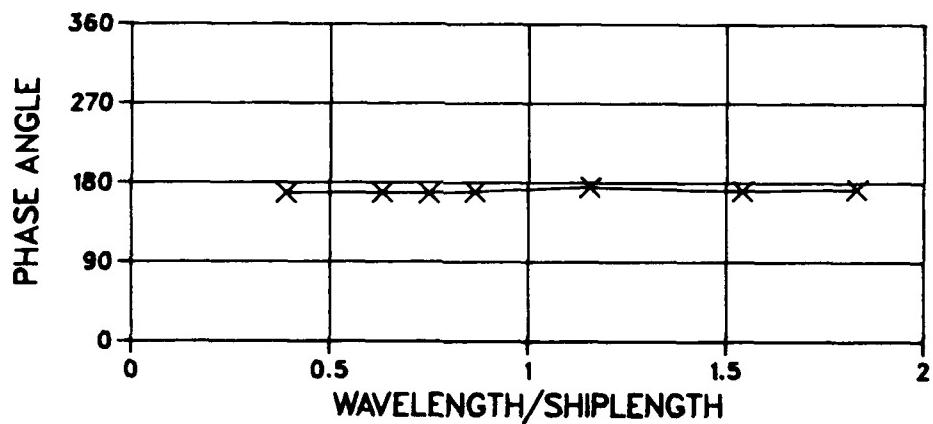


FORCED PITCH. FN=0.3

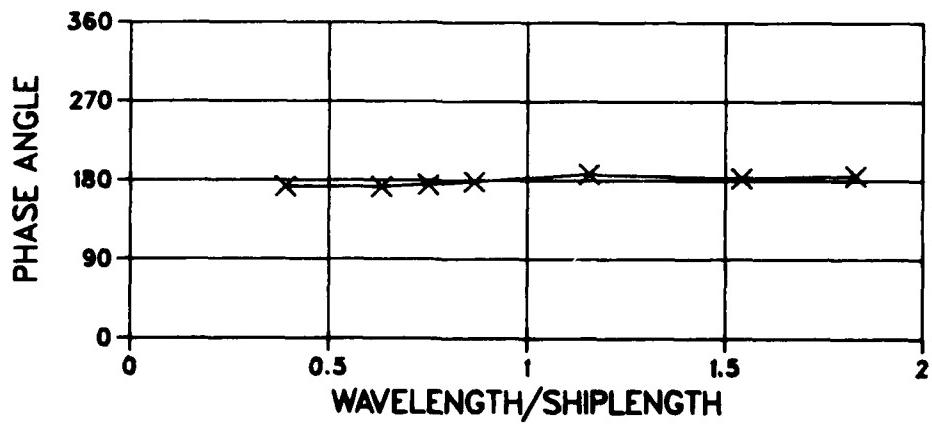
STA.18.5



STA.18

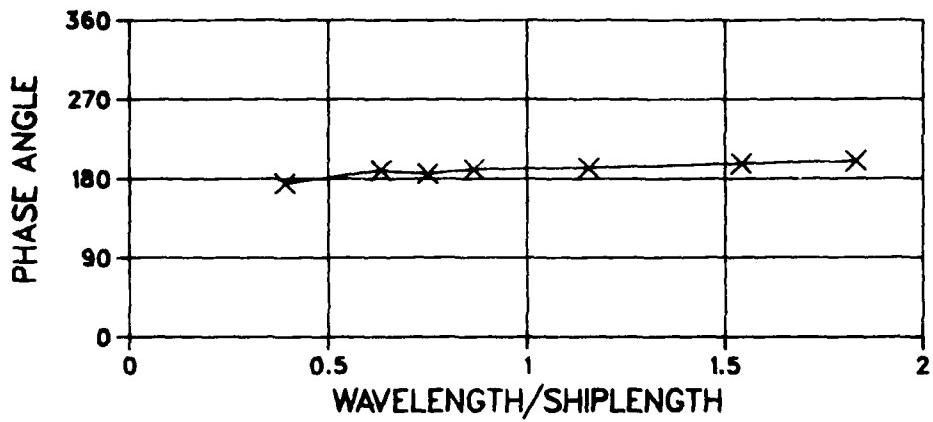


STA.17

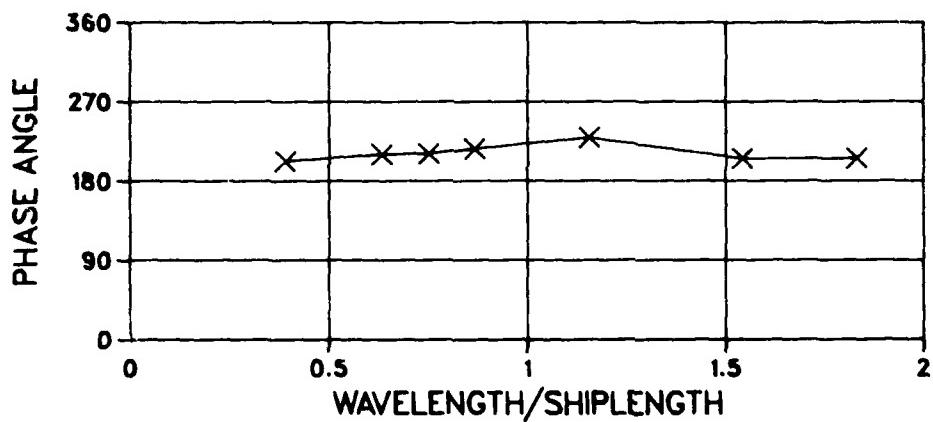


FORCED PITCH. FN=0.3

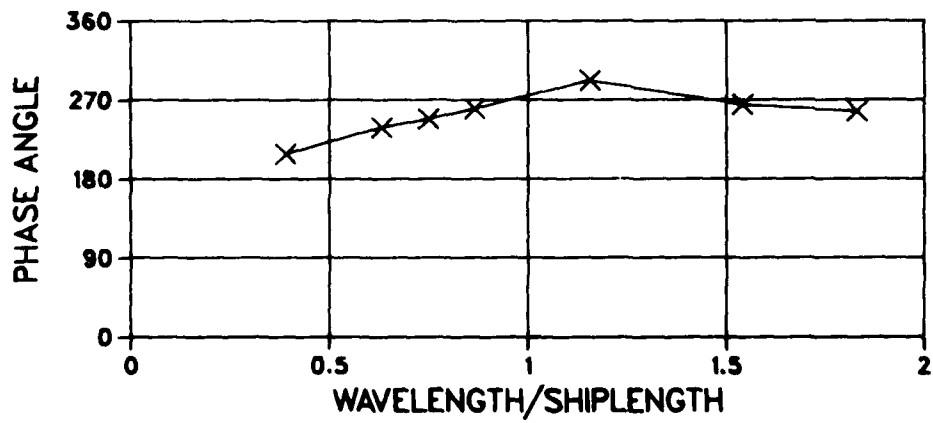
STA.16



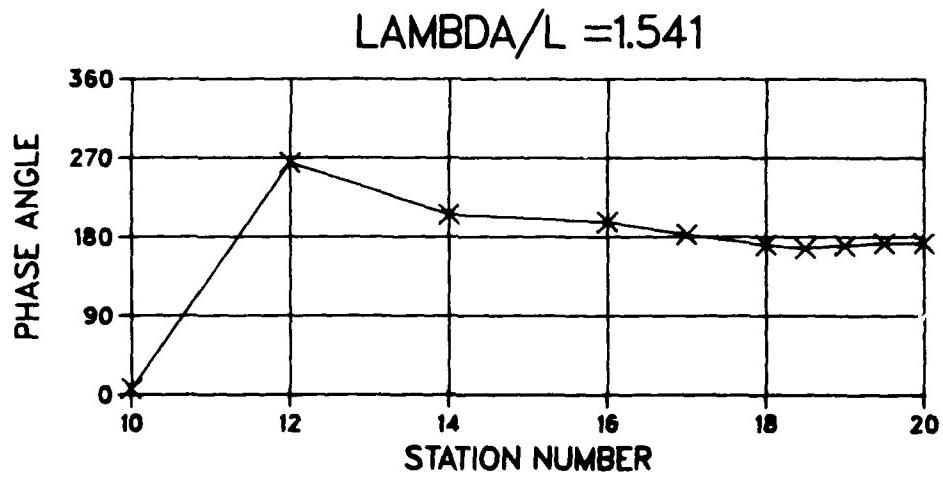
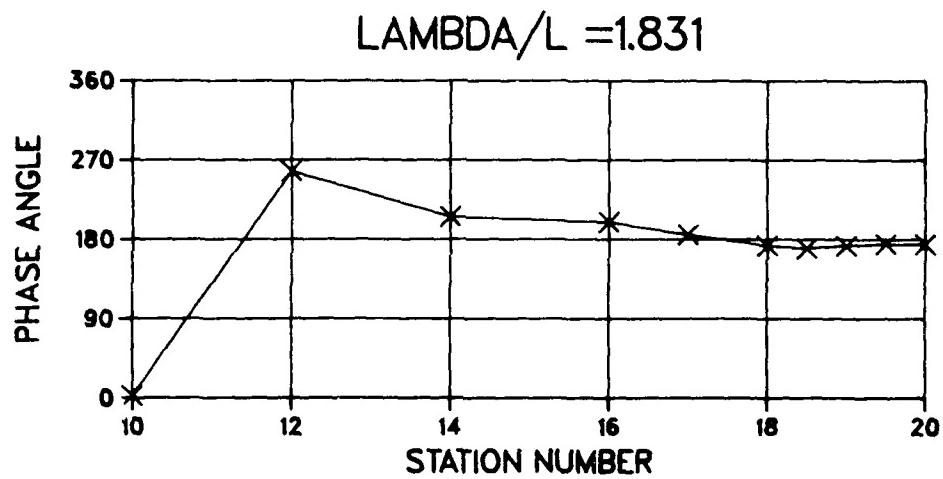
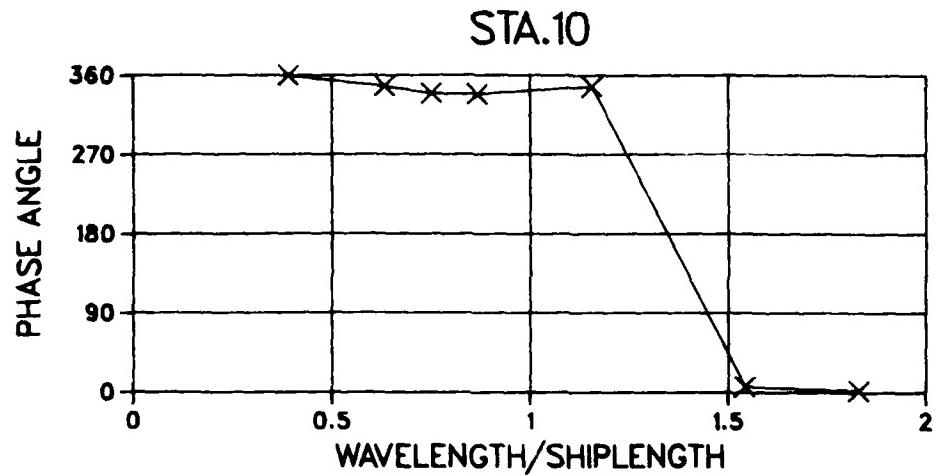
STA.14



STA.12

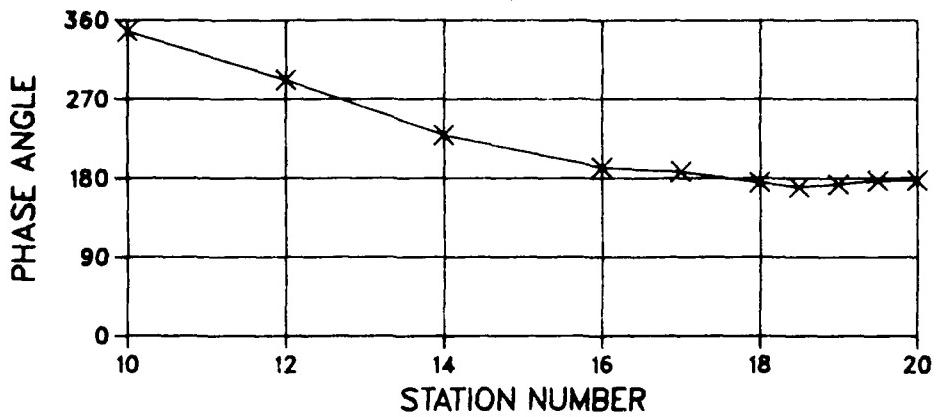


FORCED PITCH. FN=0.3

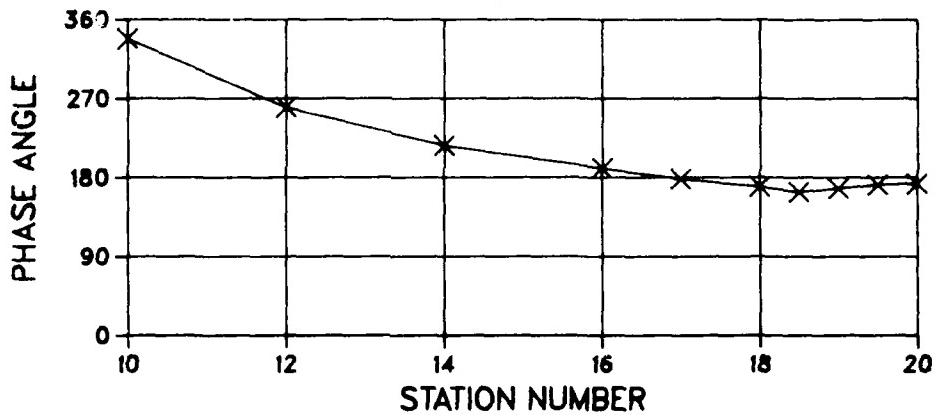


FORCED PITCH. FN=0.3

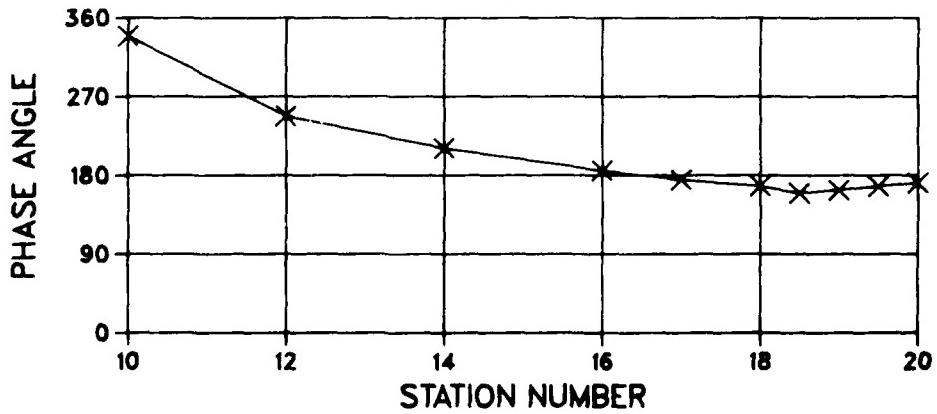
LAMBDA/L = 1.156



LAMBDA/L = 0.867

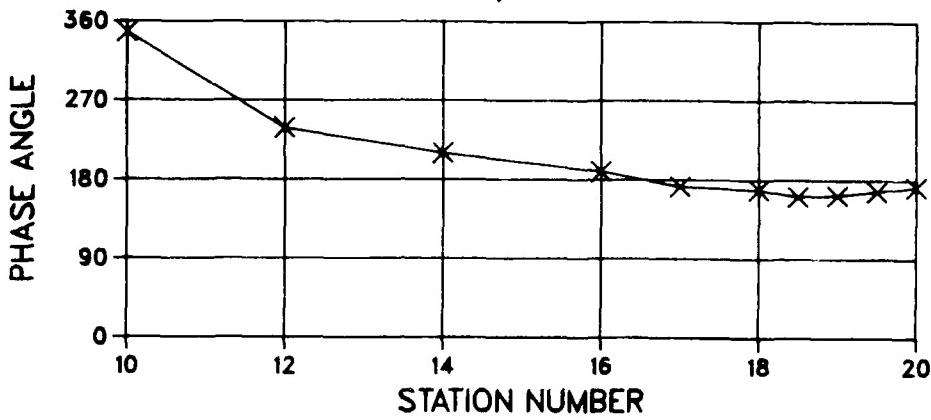


LAMBDA/L = 0.751

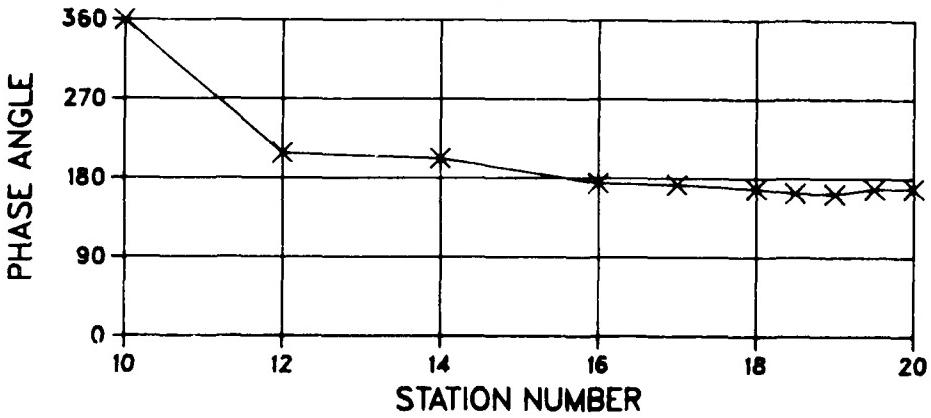


FORCED PITCH. FN=0.3

LAMBDA/L = 0.633

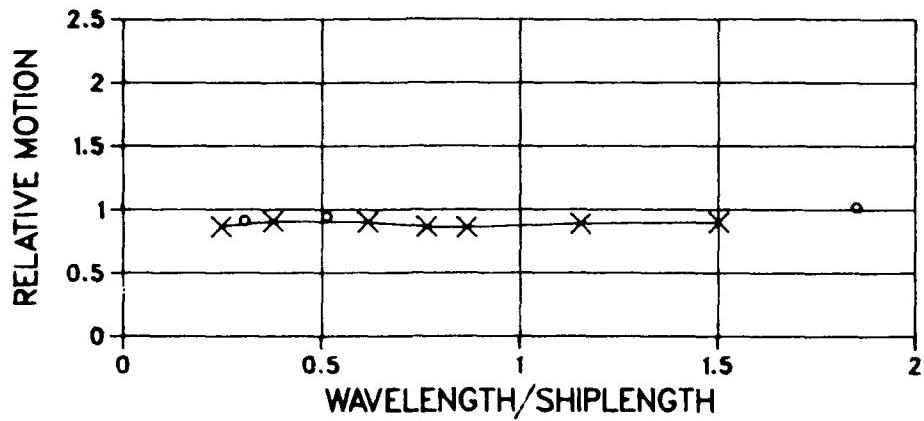


LAMBDA/L = 0.391

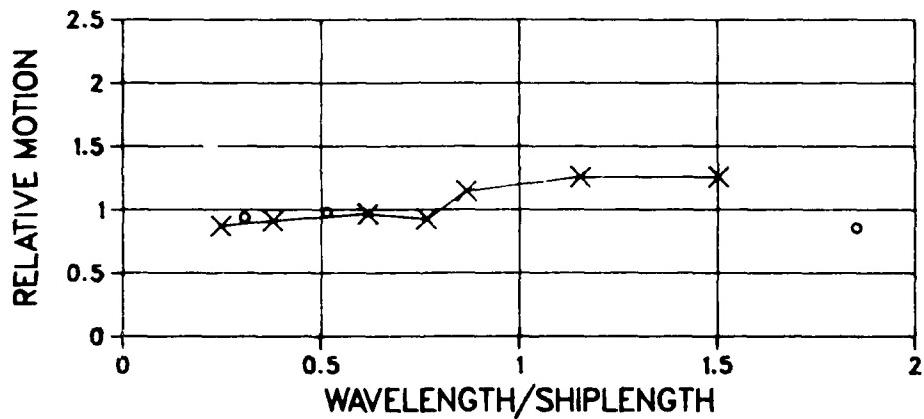


FORCED HEAVE. FN=0.2

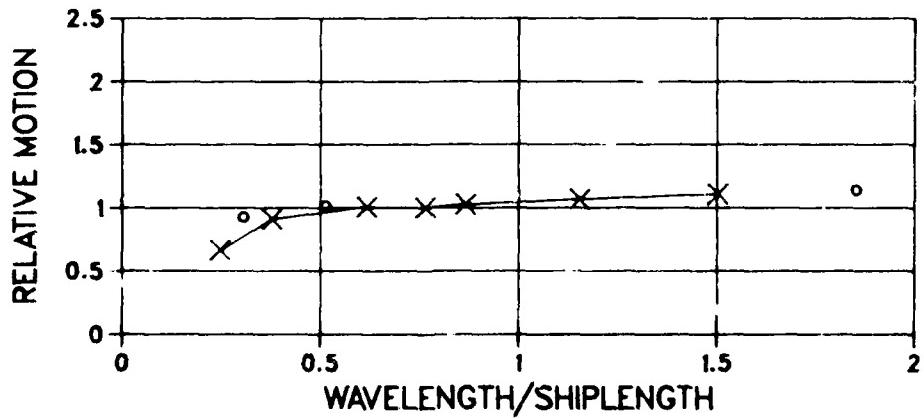
STA.20



STA.19.5



STA.19



X REGULAR
 O LINEARITY TEST

AD-A142 320

EXPERIMENTAL DETERMINATION OF THE WAVE ELEVATION NEXT
TO A MODEL OF THE S..(U) MICHIGAN UNIV ANN ARBOR DEPT
OF NAVAL ARCHITECTURE AND MARINE.. A W TROESCH ET AL.

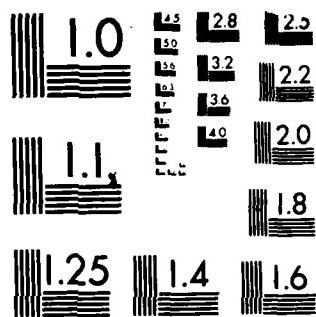
UNCLASSIFIED

25 MAR 84 N00014-83-K-0305

2/2
F/G 12/1

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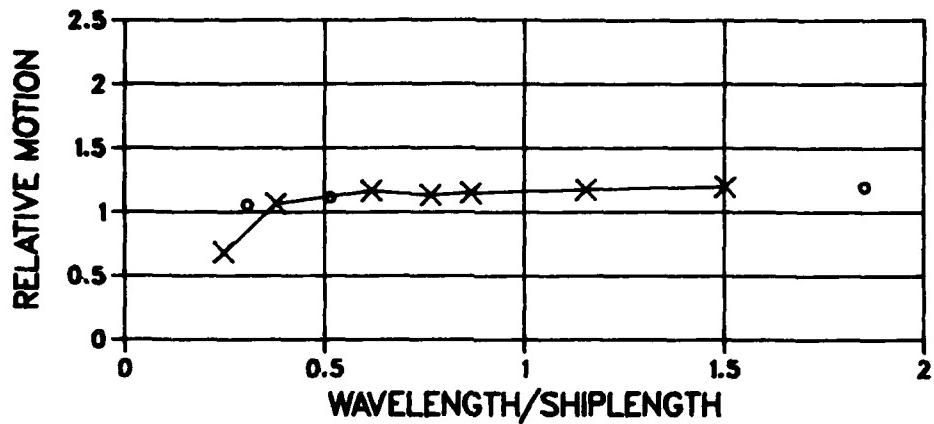
END
DATE
FILED
7-84
DTIC



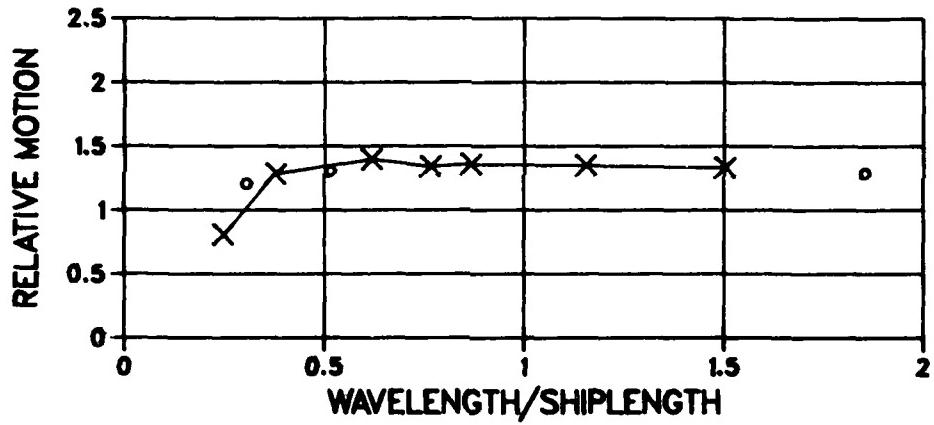
MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

FORCED HEAVE. FN=0.2

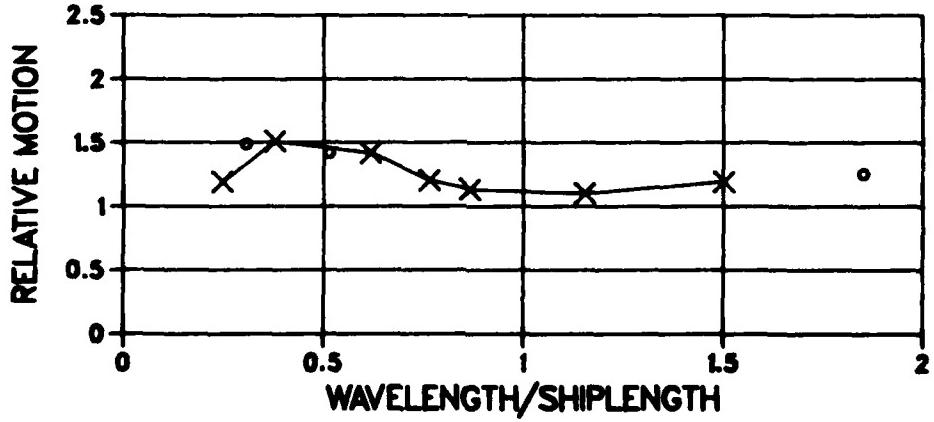
STA.18.5



STA.18



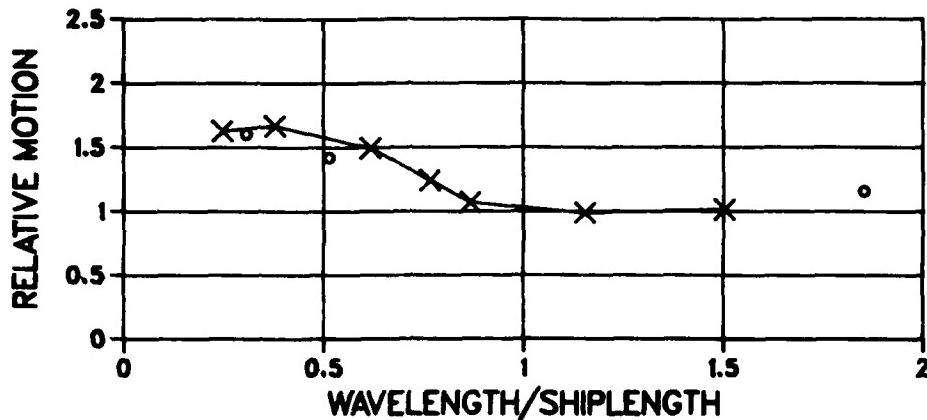
STA.17



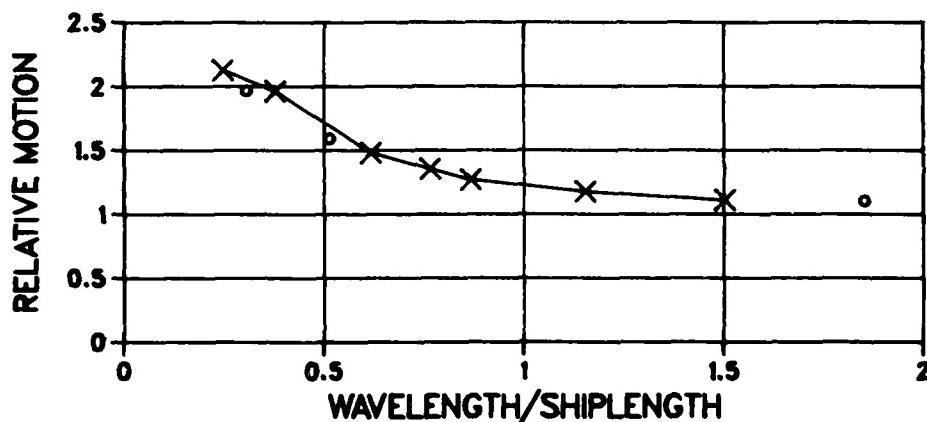
X REGULAR
O LINEARITY TEST

FORCED HEAVE. FN=0.2

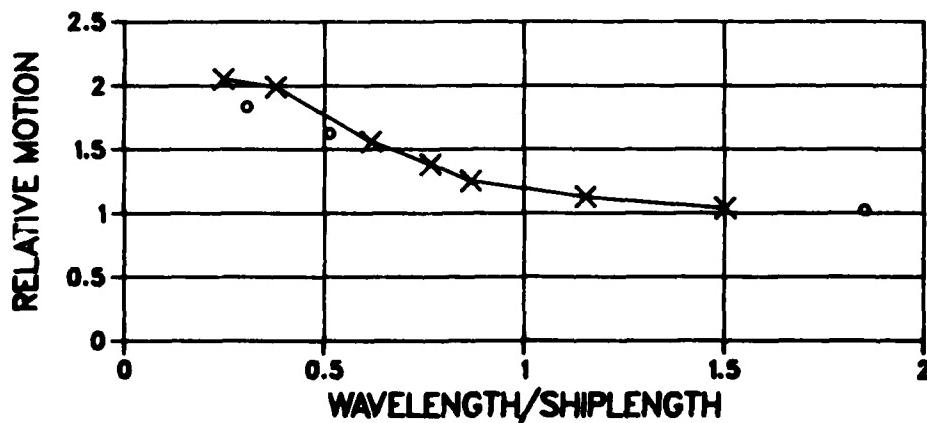
STA.16



STA.14



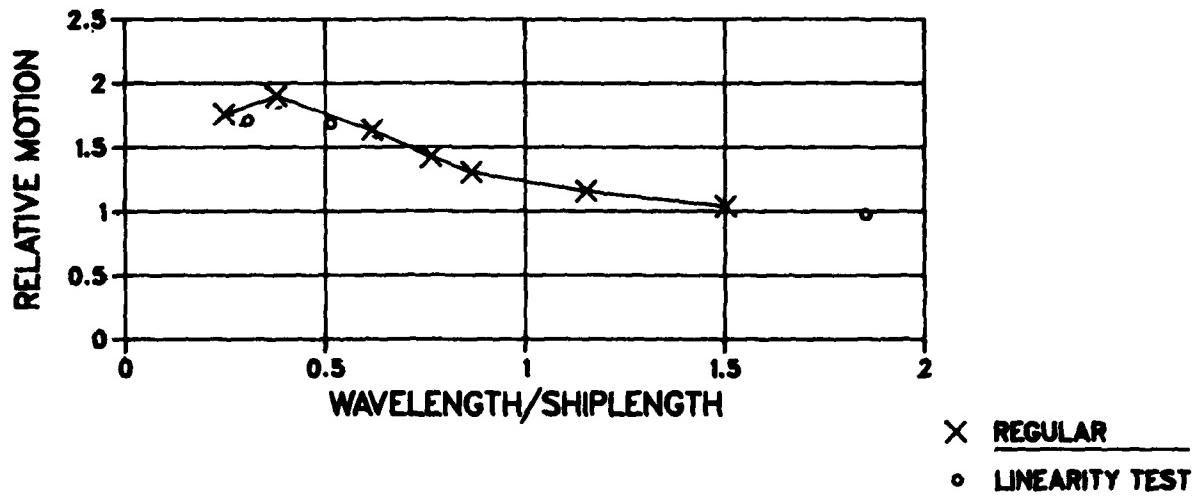
STA.12



X REGULAR
 o LINEARITY TEST

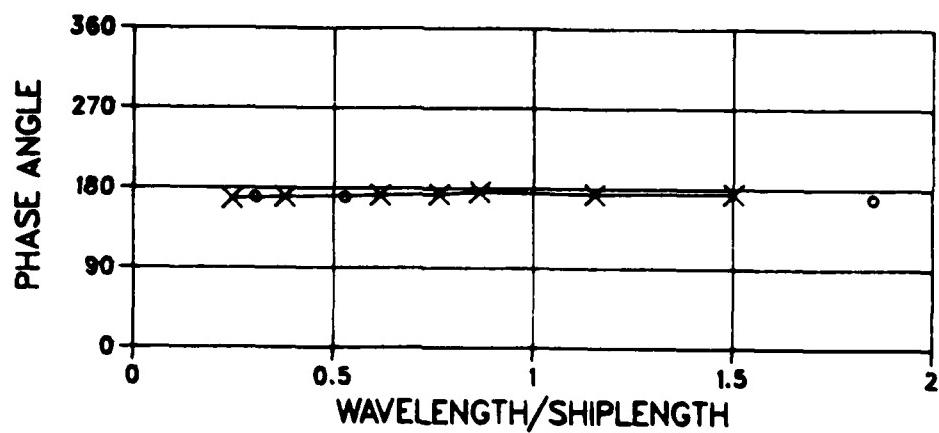
FORCED HEAVE. FN=0.2

STA.10

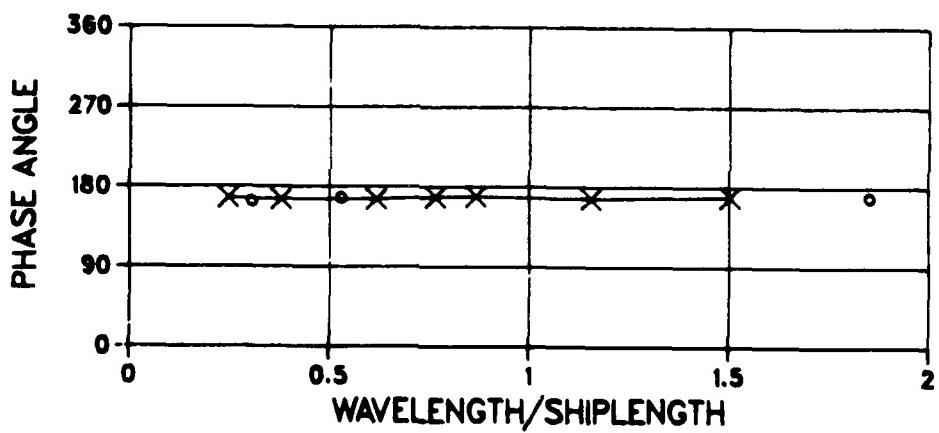


FORCED HEAVE. FN=0.2

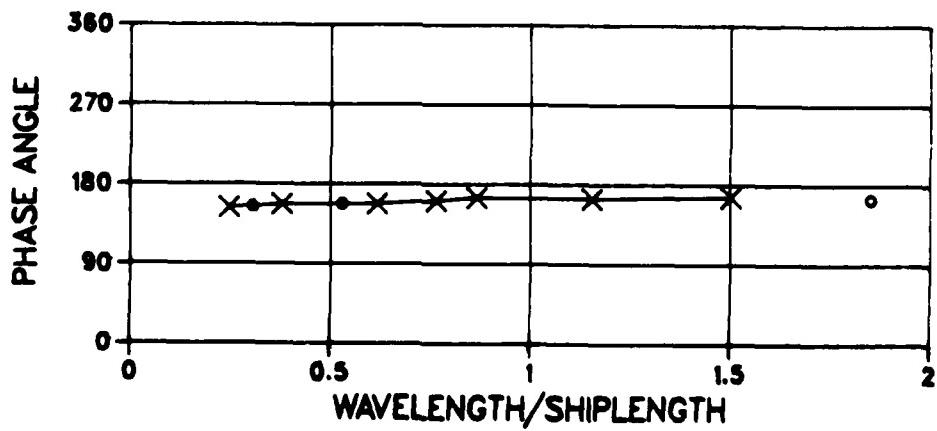
STA.20



STA.19.5



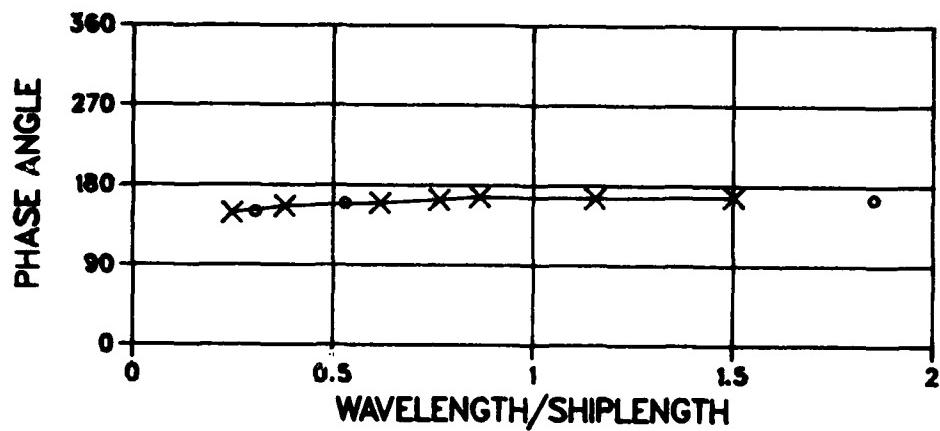
STA.19



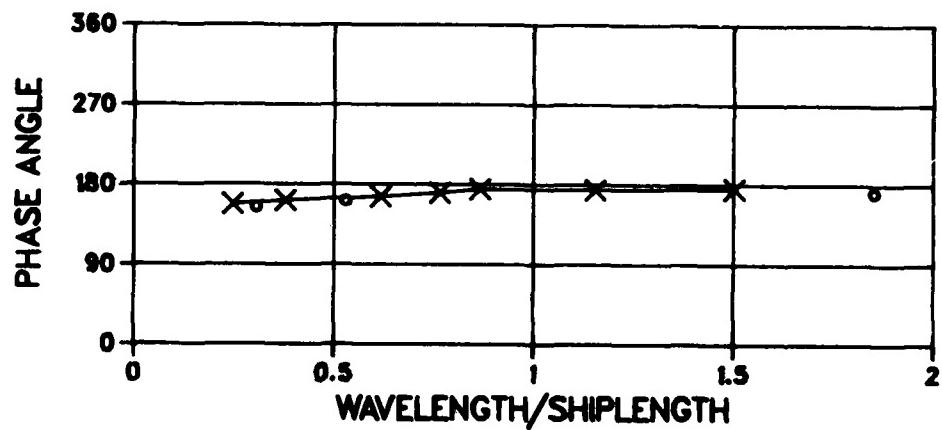
X REGULAR
O LINEARITY TEST

FORCED HEAVE. FN=0.2

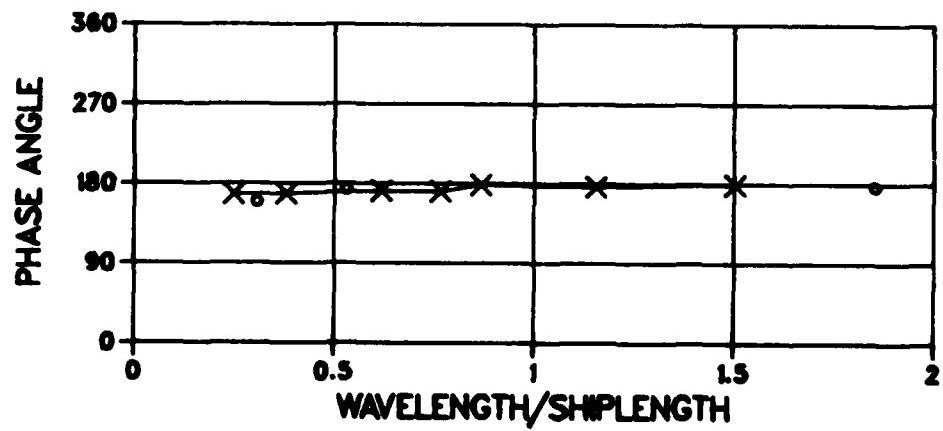
STA.18.5



STA.18



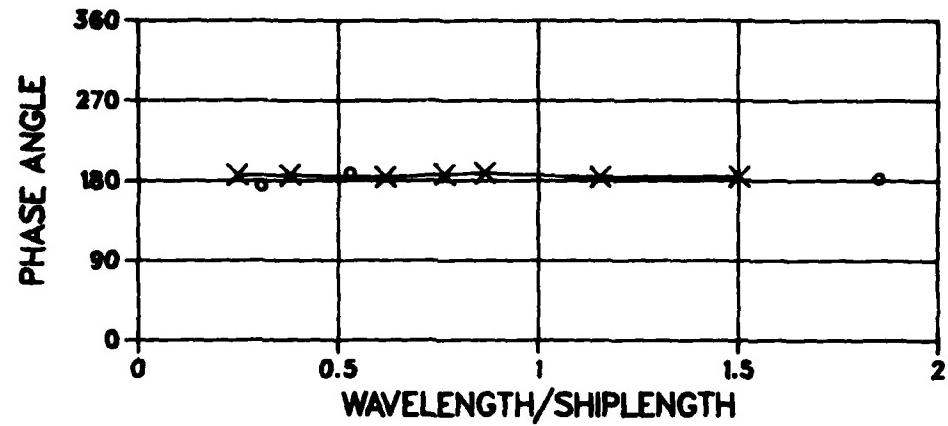
STA.17



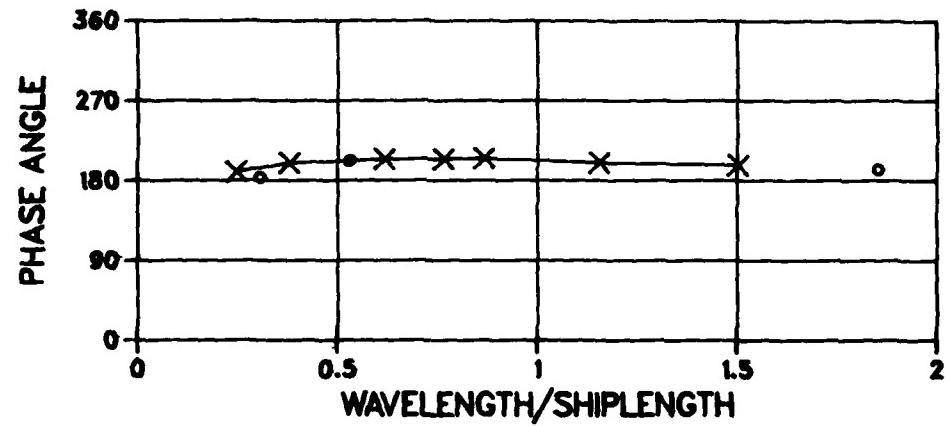
X REGULAR
○ LINEARITY TEST

FORCED HEAVE. FN=0.2

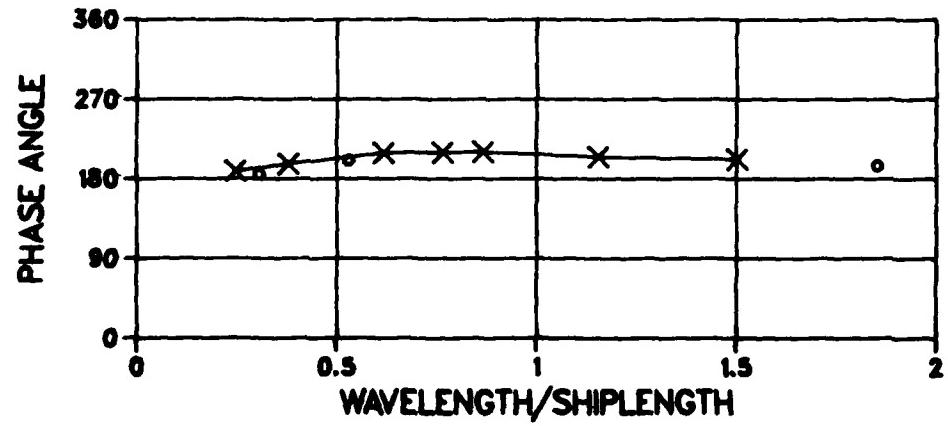
STA.16



STA.14



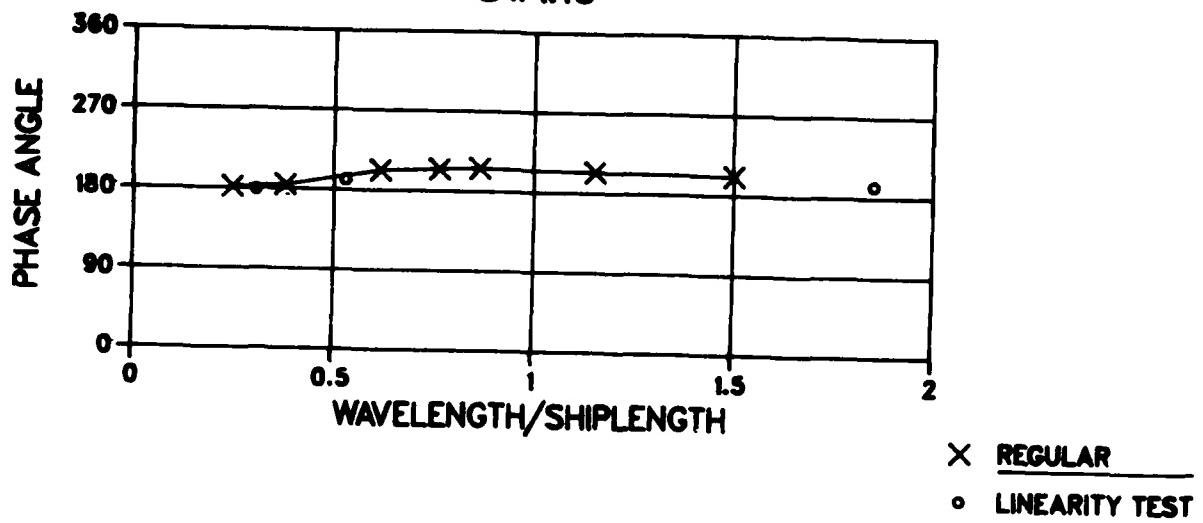
STA.12



X REGULAR
○ LINEARITY TEST

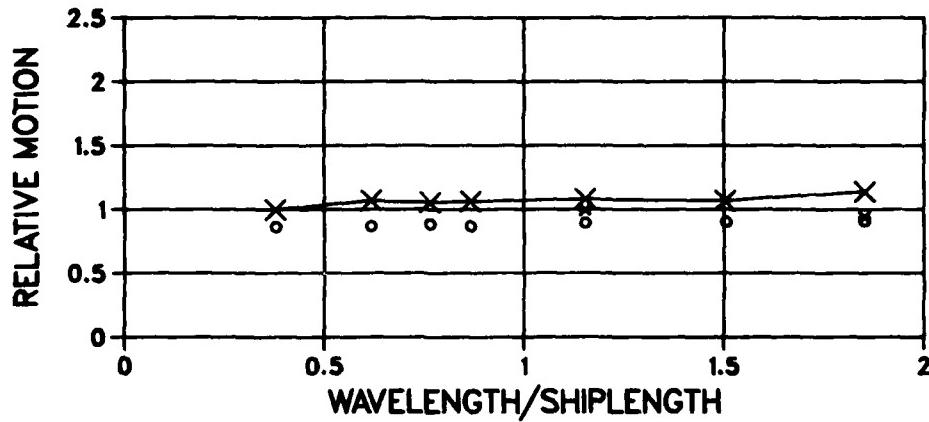
FORCED HEAVE. FN=0.2

STA.10

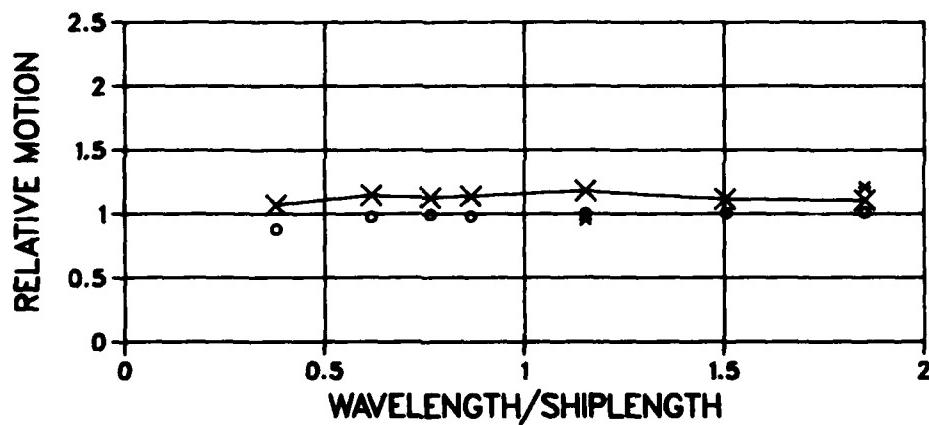


FORCED PITCH. FN=0.2

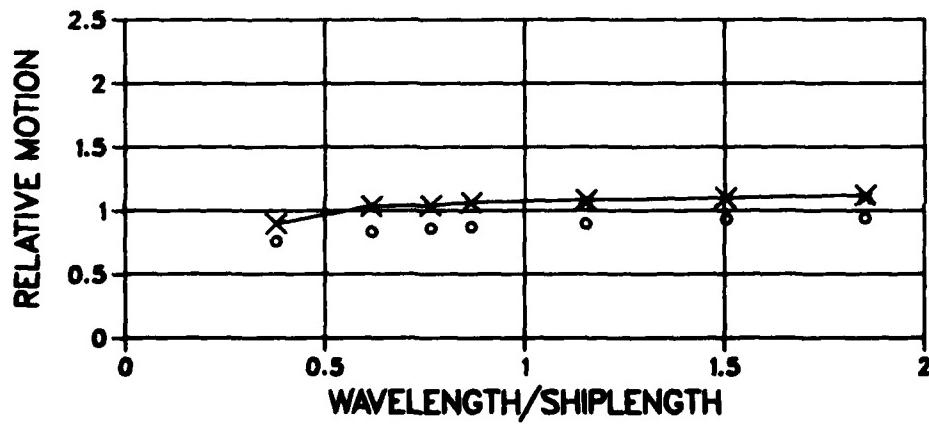
STA.20



STA.19.5



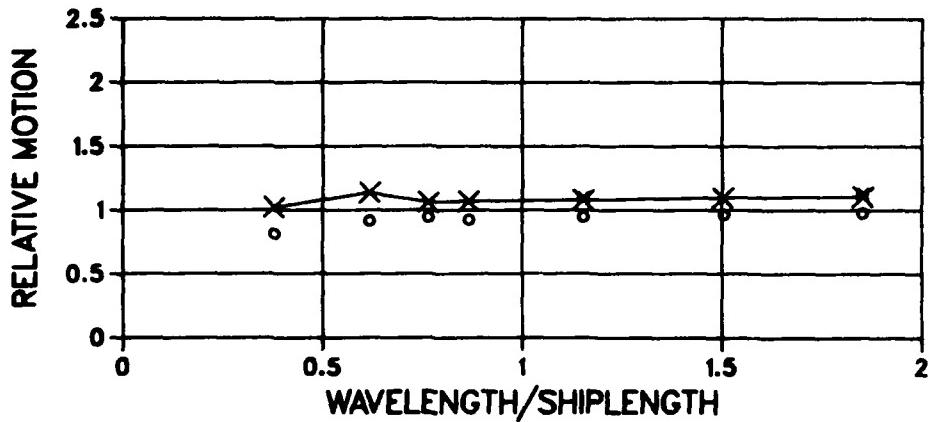
STA.19



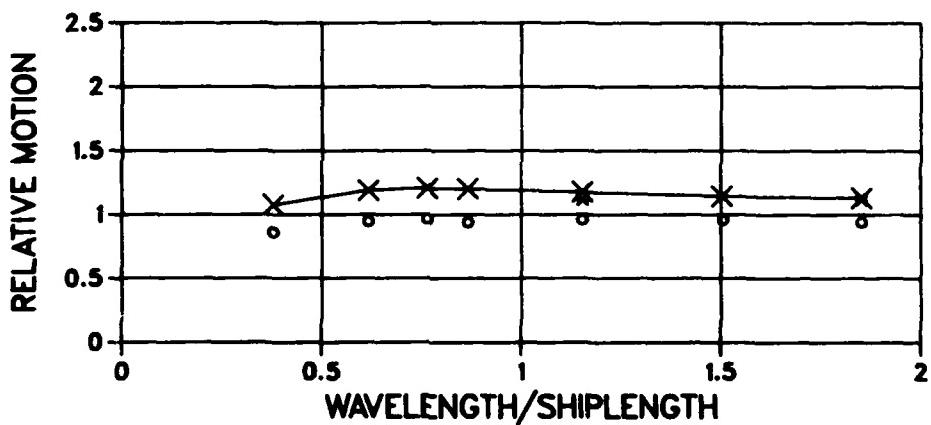
- × REGULAR
- LINEARITY TEST
- ✗ REPEATABILITY

FORCED PITCH. FN=0.2

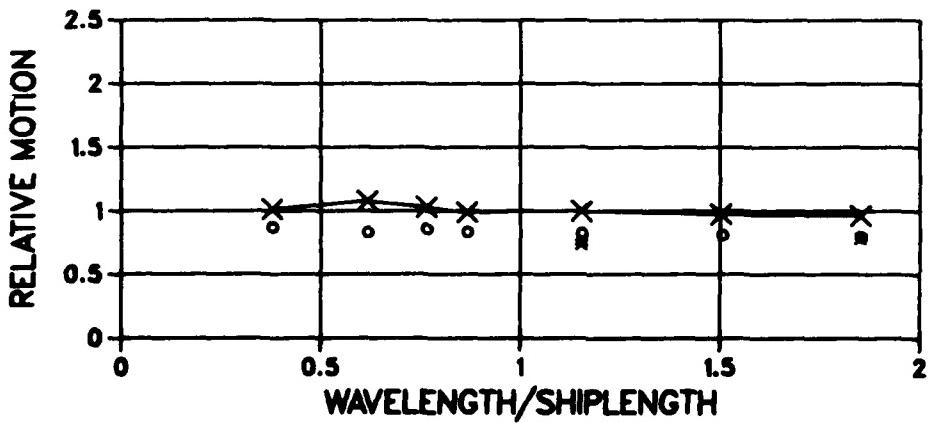
STA.18.5



STA.18



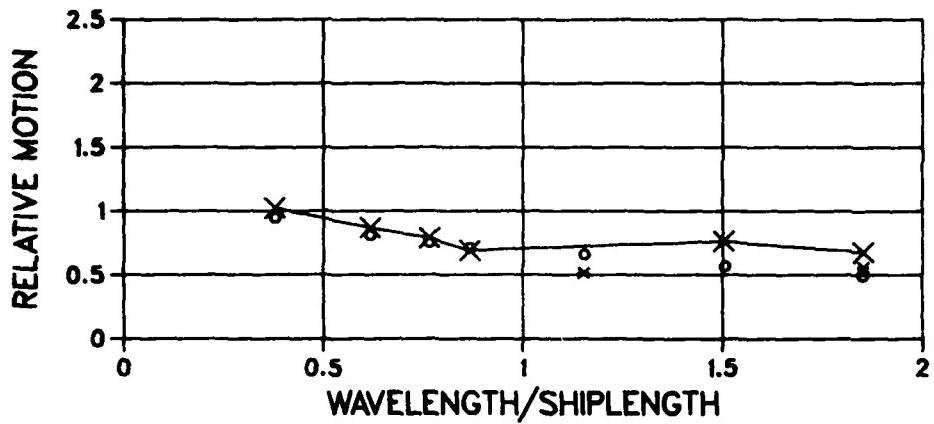
STA.17



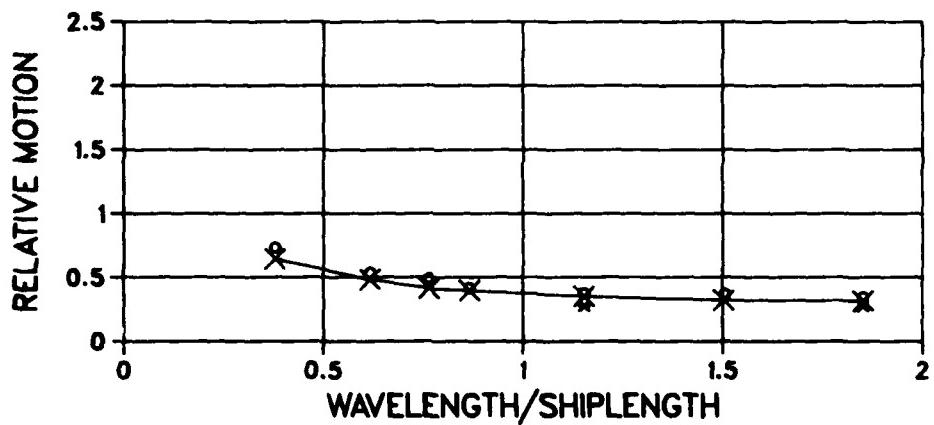
- × REGULAR
- LINEARITY TEST
- REPEATABILITY

FORCED PITCH. FN=0.2

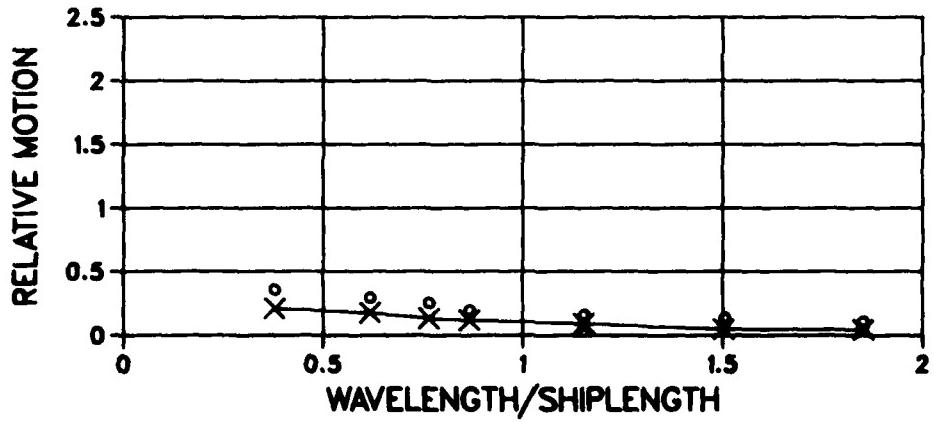
STA.16



STA.14



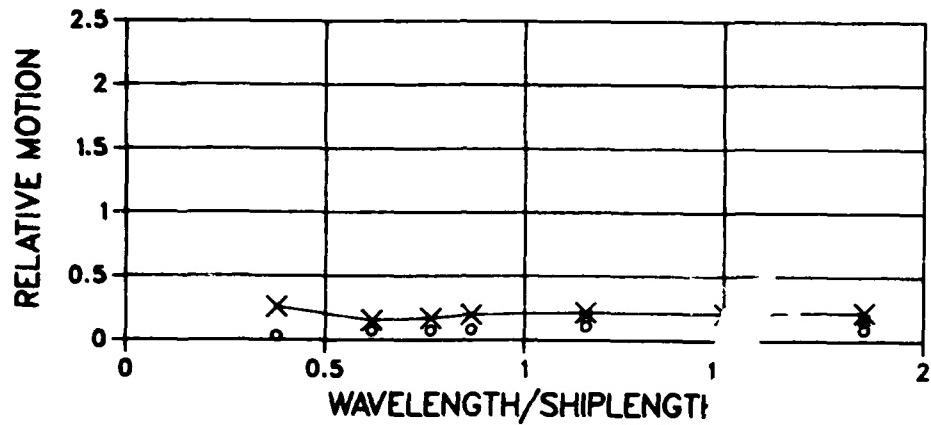
STA.12



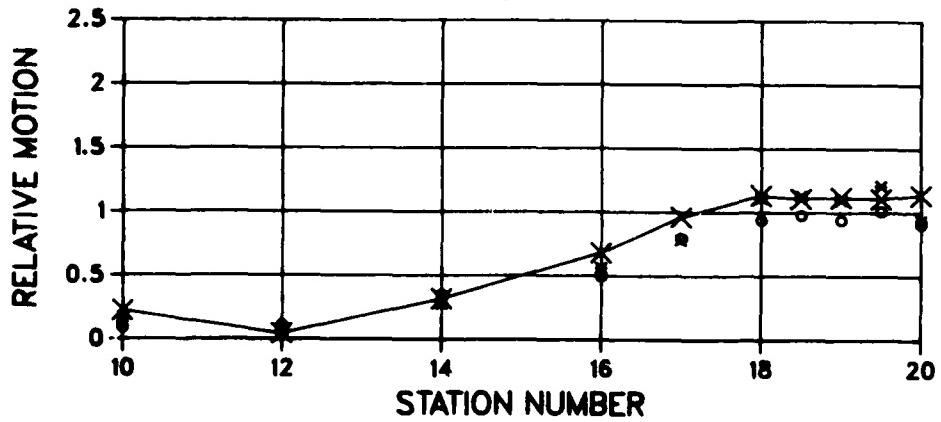
X REGULAR
 ○ LINEARITY TEST
 ▨ REPEATABILITY

FORCED PITCH. FN=0.2

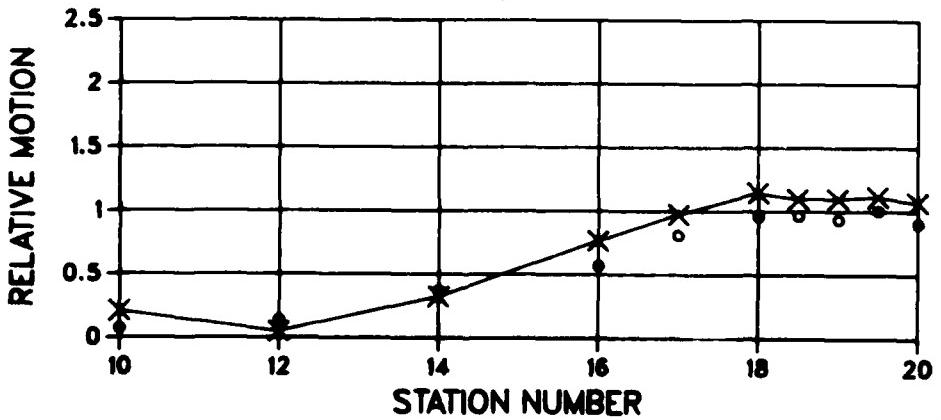
STA.10



$\lambda/L = 1.852$



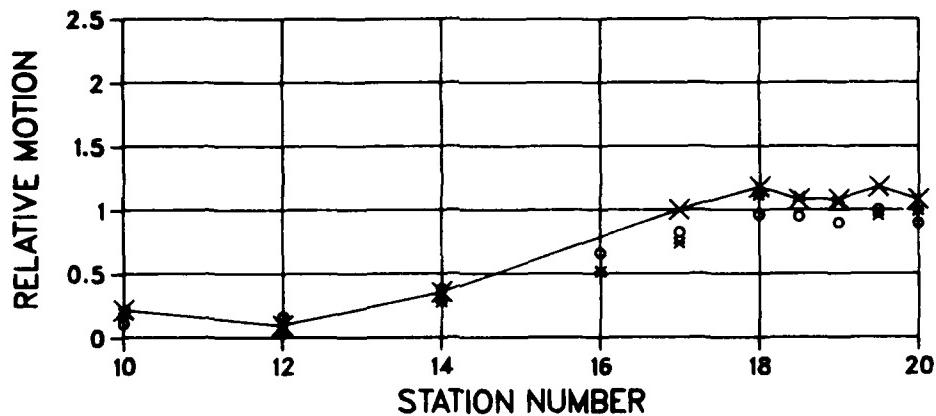
$\lambda/L = 1.503$



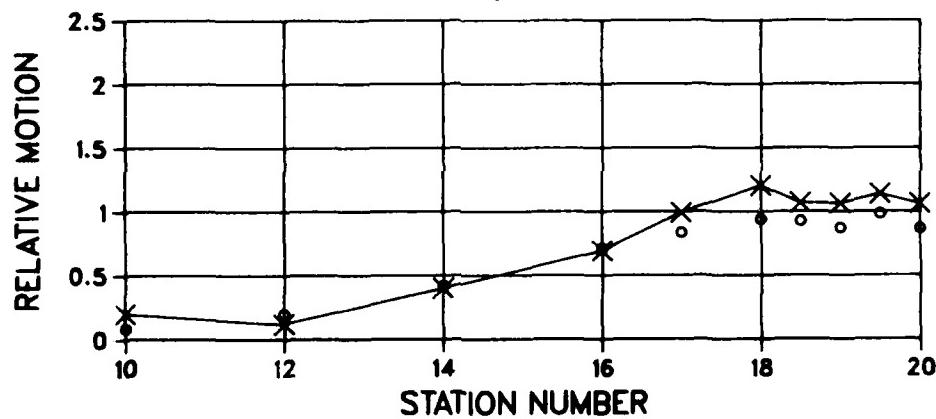
- × REGULAR
- LINEARITY TEST
- REPEATABILITY

FORCED PITCH. FN=0.2

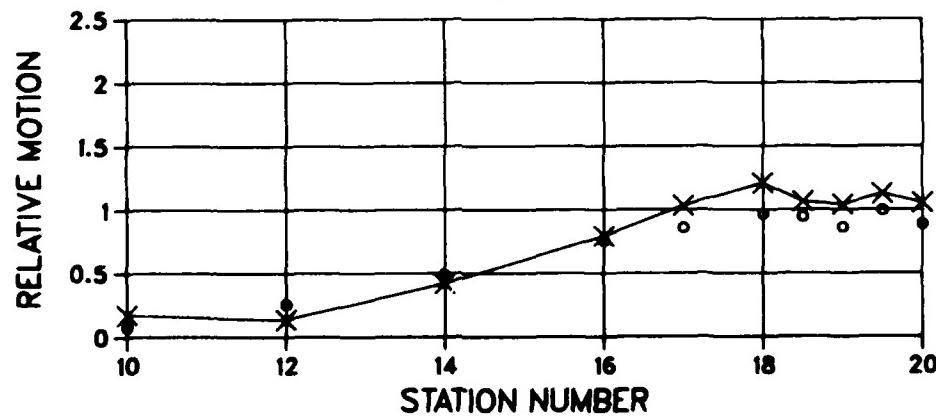
LAMBDA/L = 1.154



LAMBDA/L = 0.867



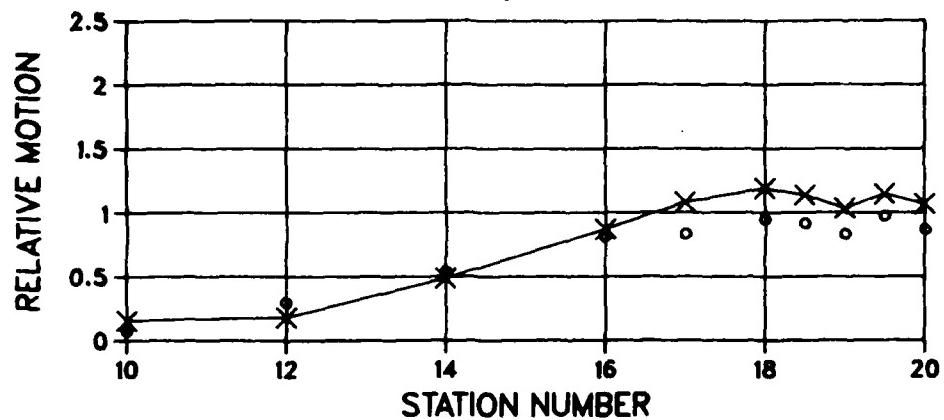
LAMBDA/L = 0.766



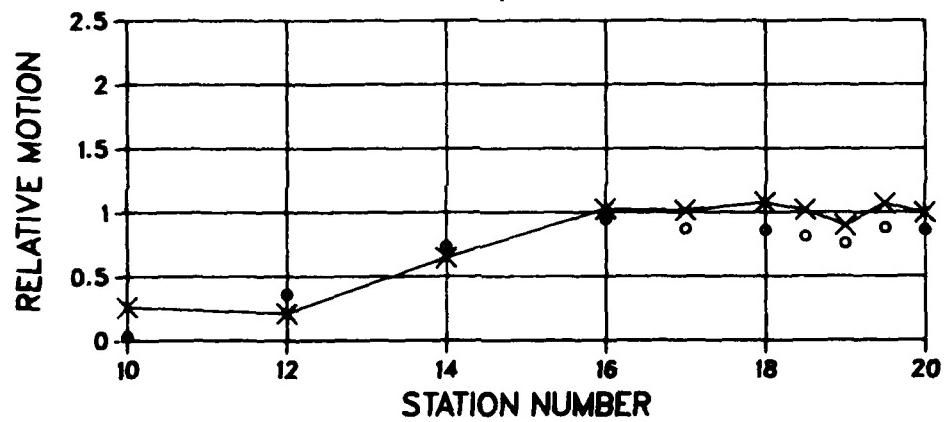
- × REGULAR
- LINEARITY TEST
- REPEATABILITY

FORCED PITCH. FN=0.2

LAMBDA/L = 0.618



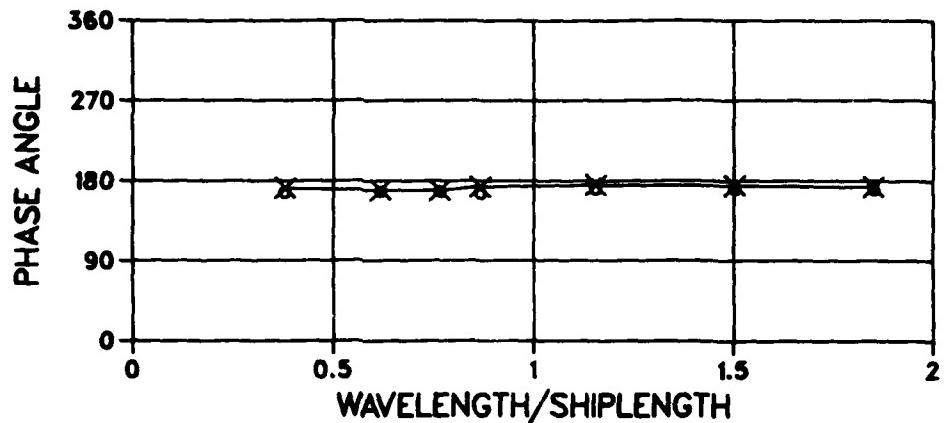
LAMBDA/L = 0.379



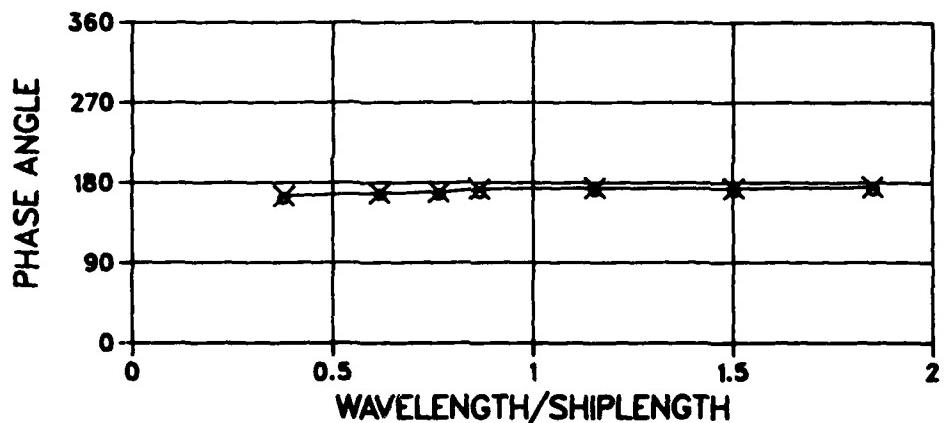
X REGULAR
○ LINEARITY TEST

FORCED PITCH. FN=0.2

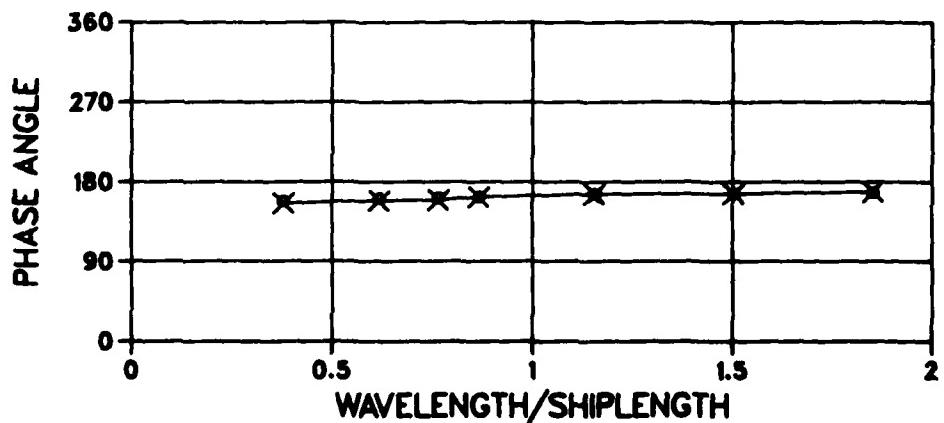
STA.20



STA.19.5



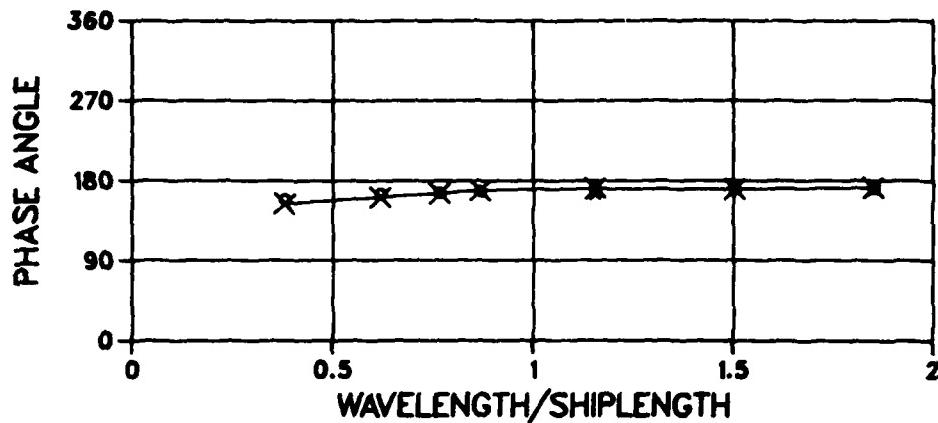
STA.19



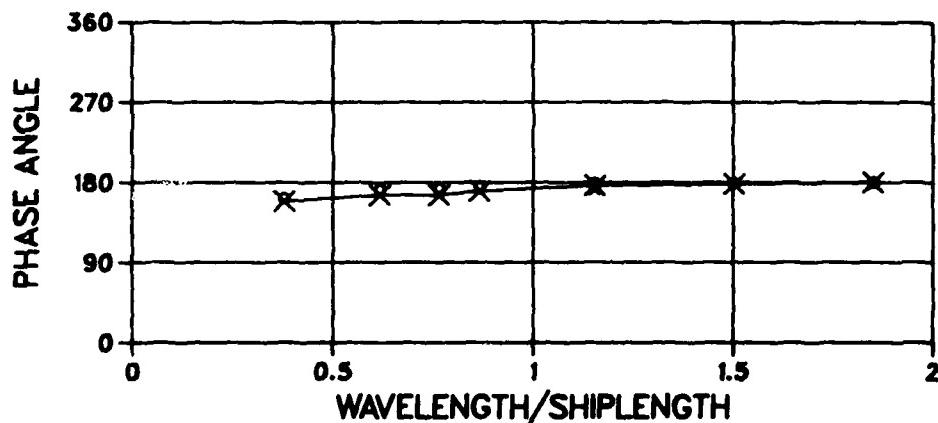
- × REGULAR
- LINEARITY TEST
- REPEATABILITY

FORCED PITCH. FN=0.2

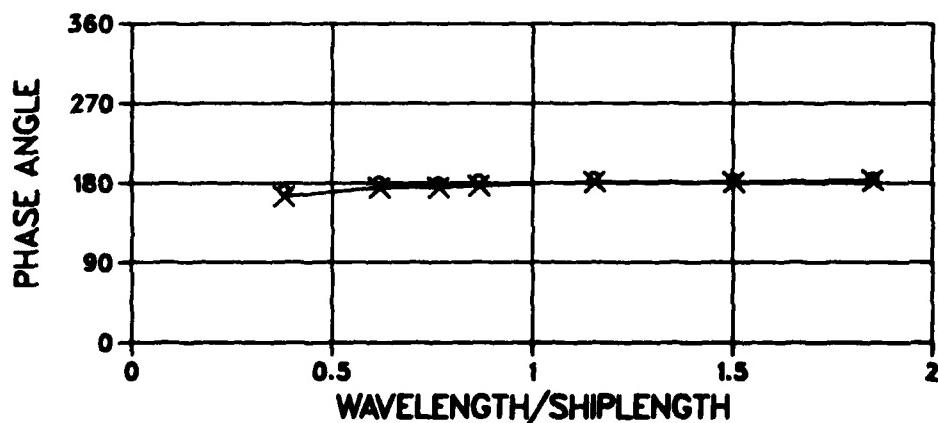
STA.18.5



STA.18



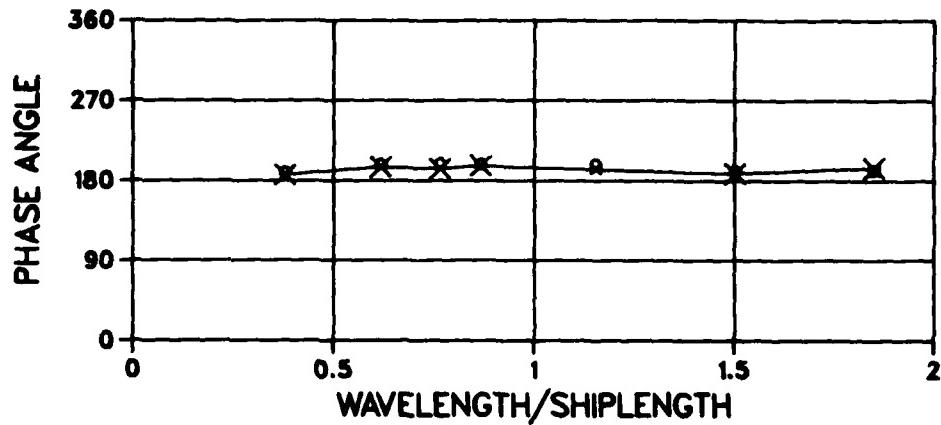
STA.17



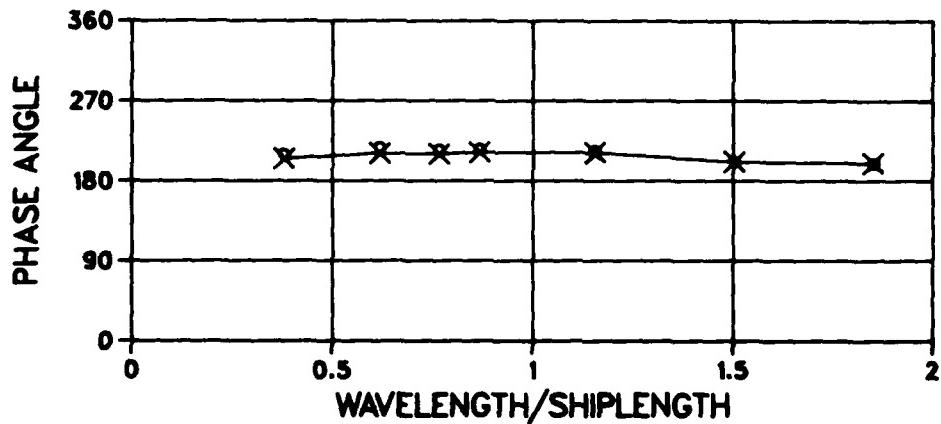
- × REGULAR
- LINEARITY TEST
- REPEATABILITY

FORCED PITCH. FN=0.2

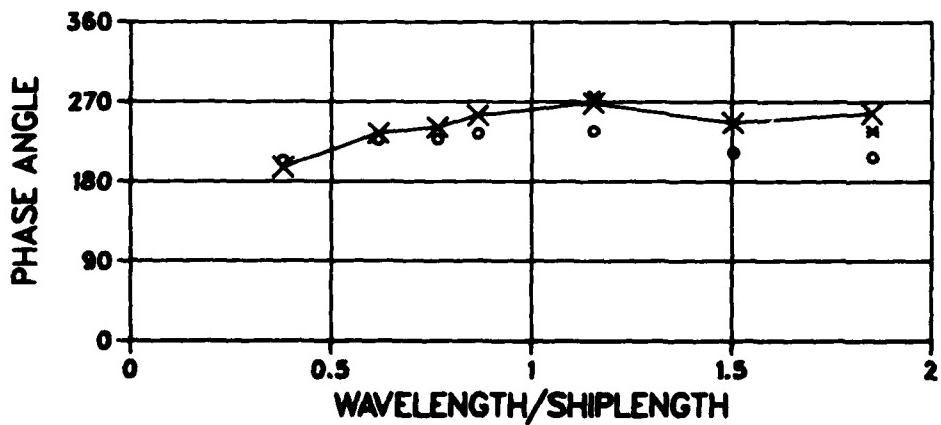
STA.16



STA.14



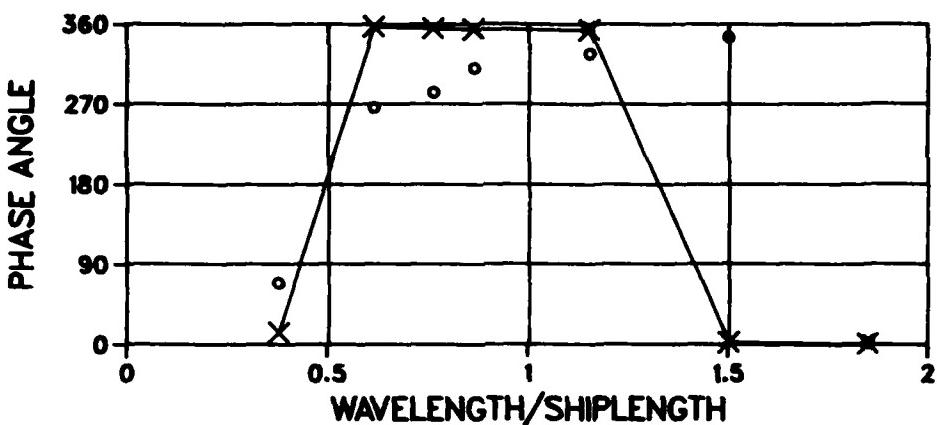
STA.12



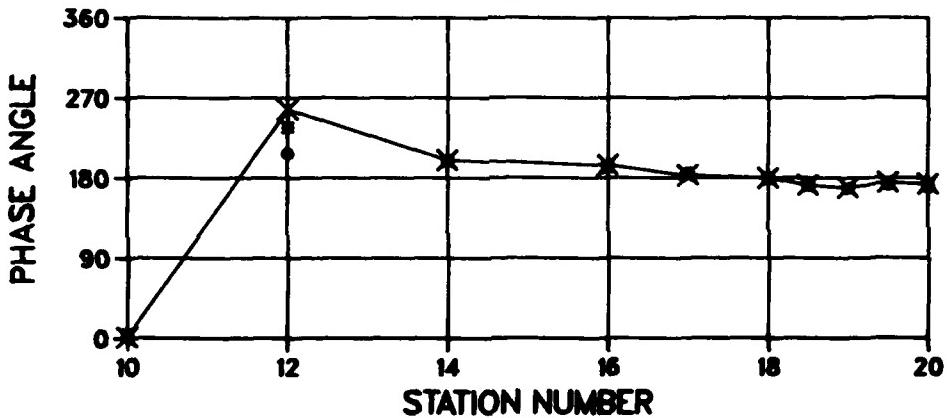
- × REGULAR
- LINEARITY TEST
- REPEATABILITY

FORCED PITCH. FN=0.2

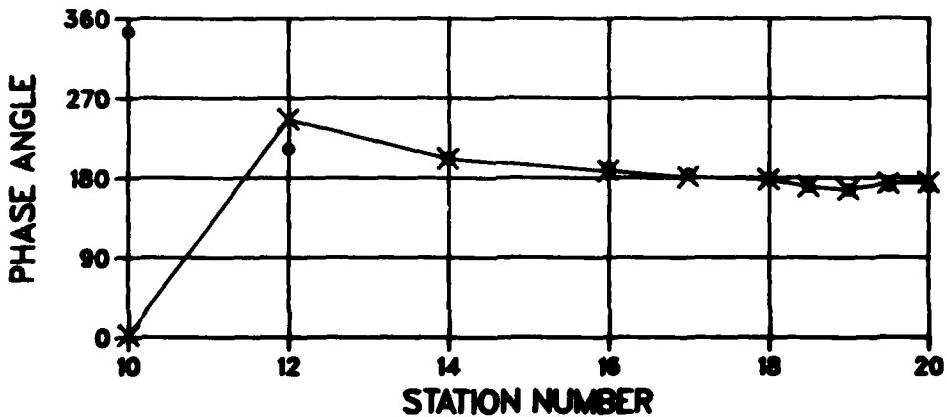
STA.10



LAMBDA/L = 1.852

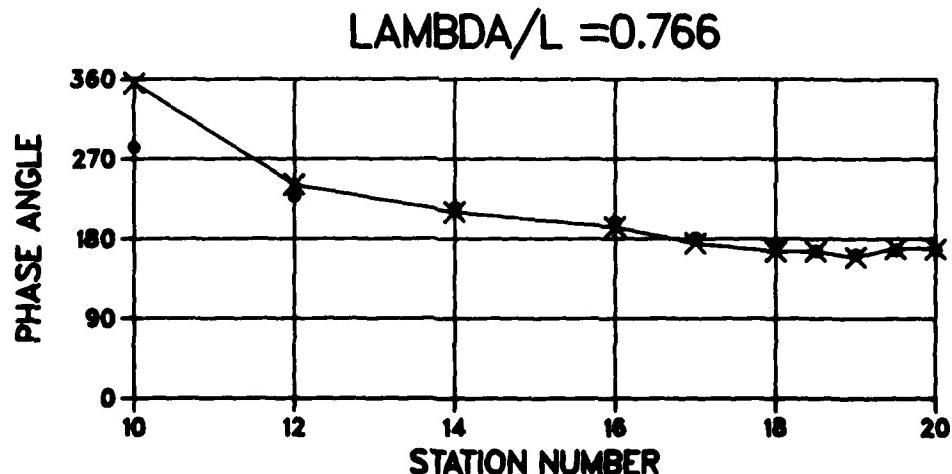
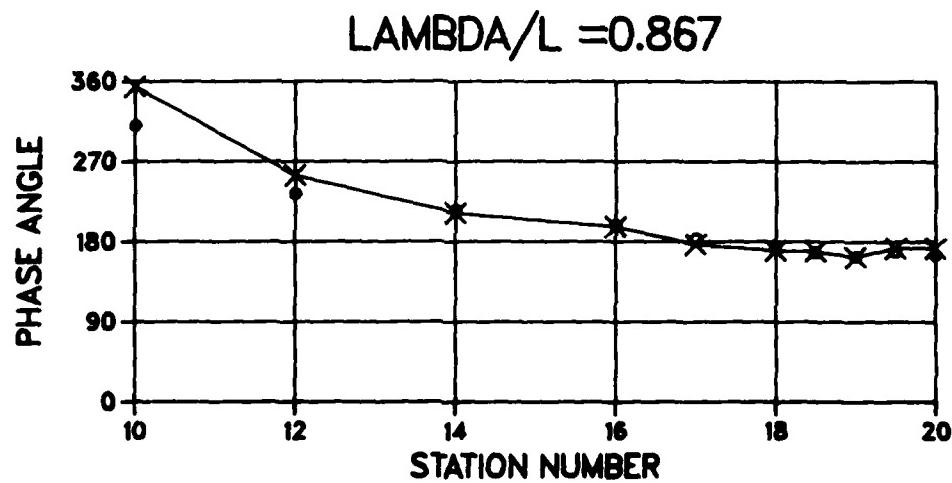
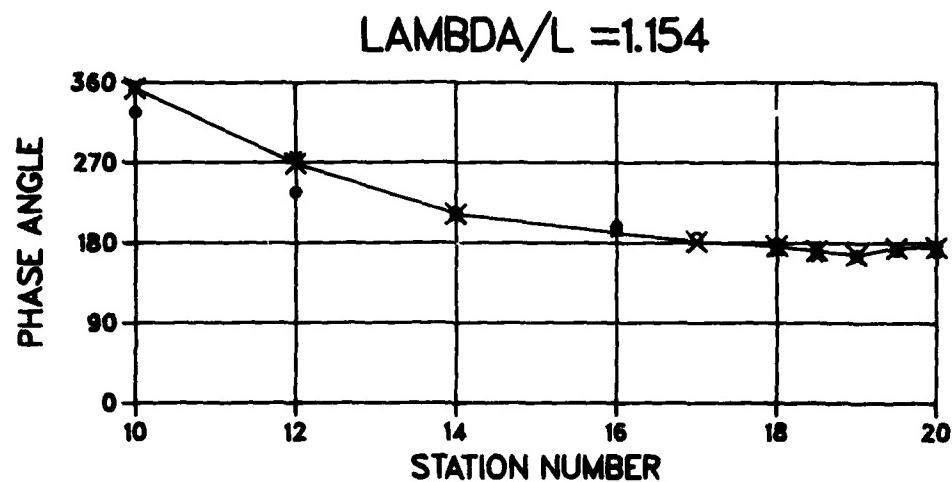


LAMBDA/L = 1.503



- \times REGULAR
- \circ LINEARITY TEST
- \bowtie REPEATABILITY

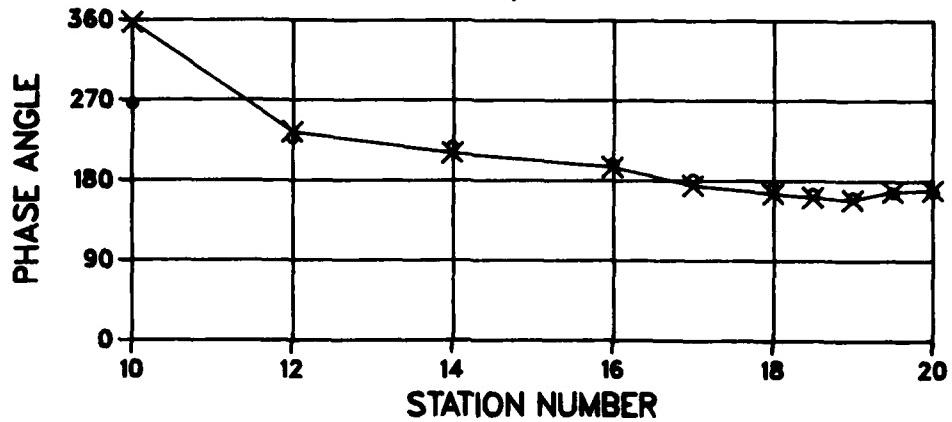
FORCED PITCH. FN=0.2



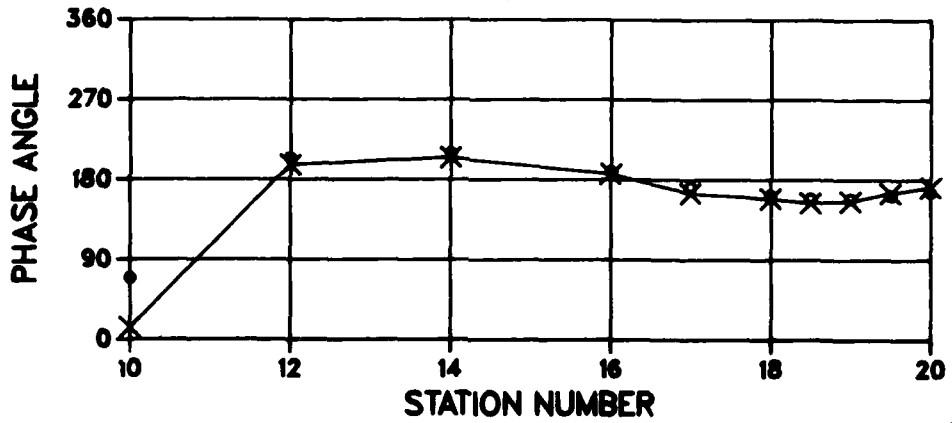
- × REGULAR
- LINEARITY TEST
- REPEATABILITY

FORCED PITCH. FN=0.2

LAMBDA/L = 0.618



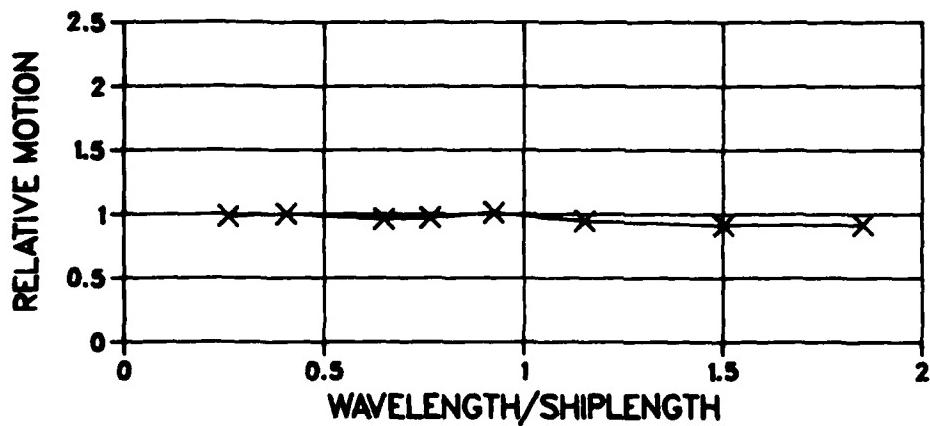
LAMBDA/L = 0.379



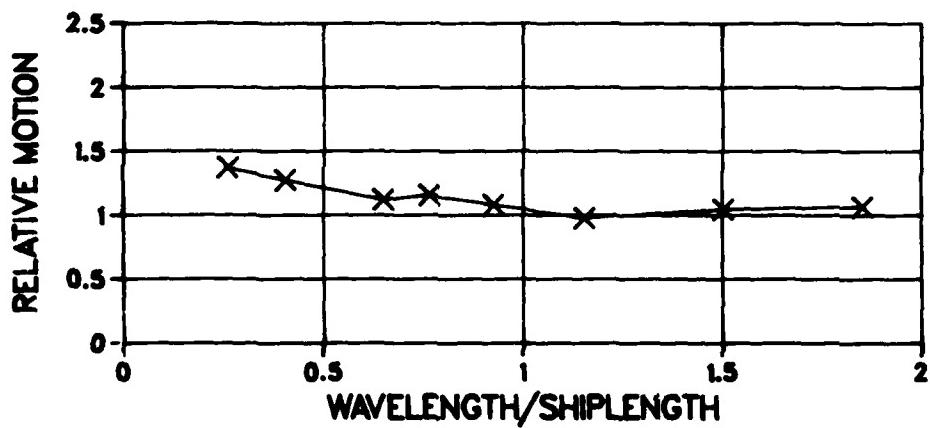
X REGULAR
o LINEARITY TEST

DIFFRACTION. FN=0.2

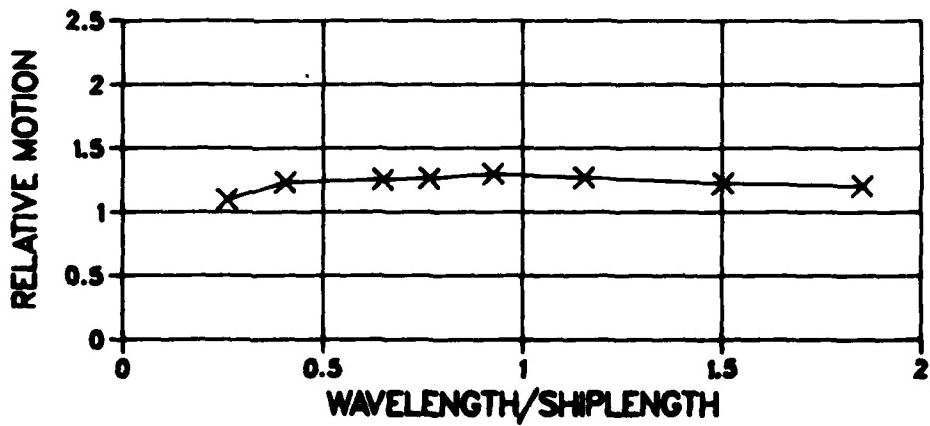
STA.20



STA.19.5

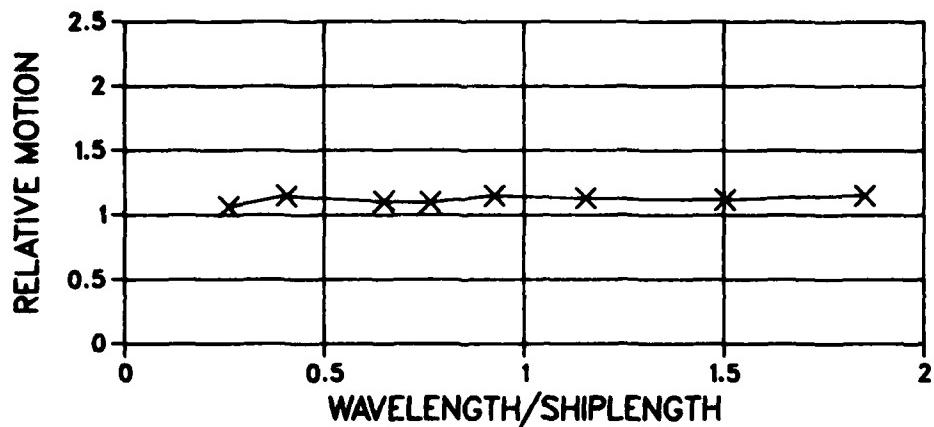


STA.19

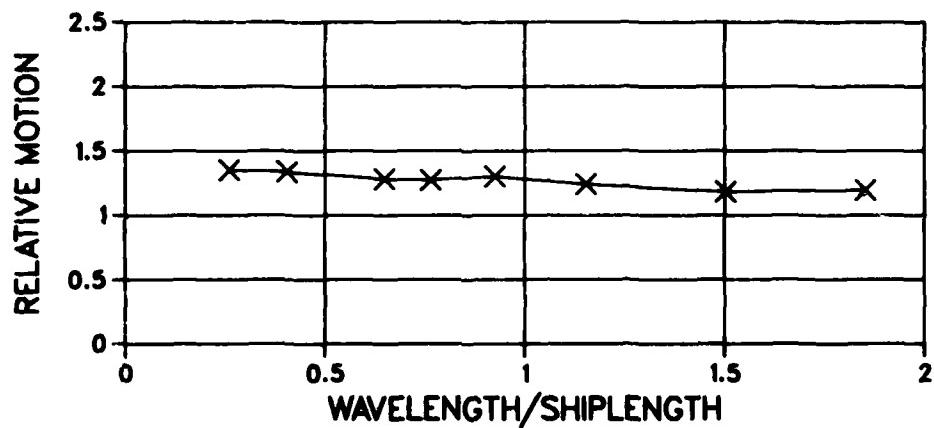


DIFFRACTION. FN=0.2

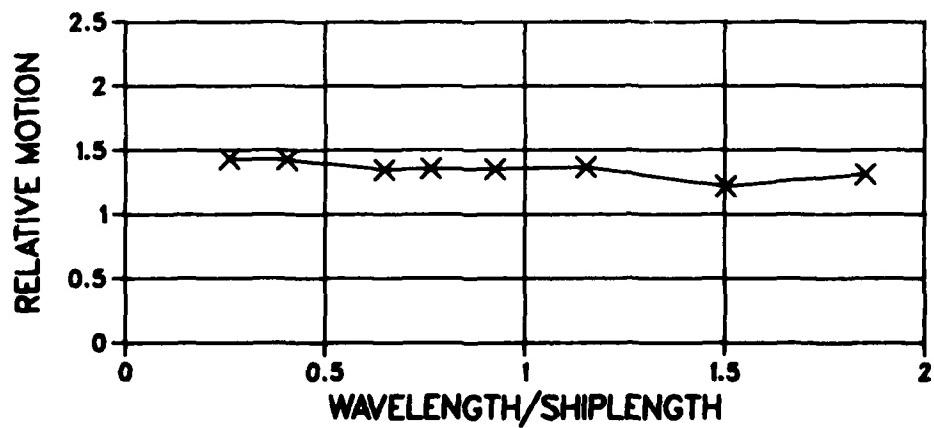
STA.18.5



STA.18

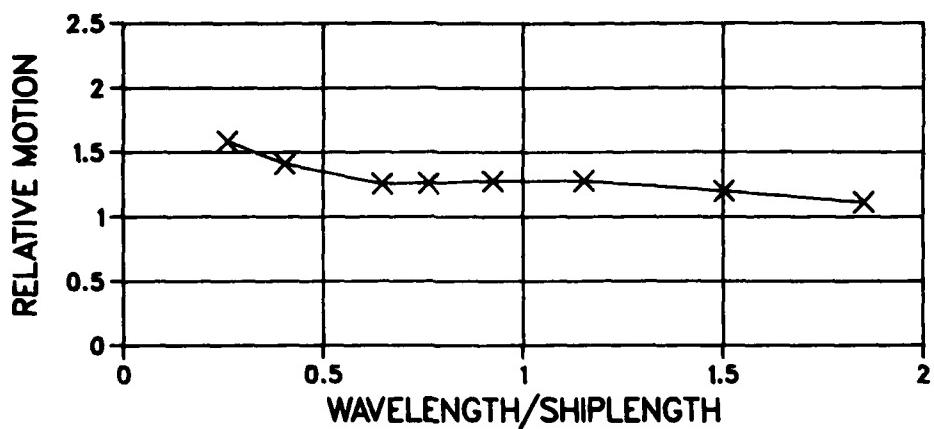


STA.17

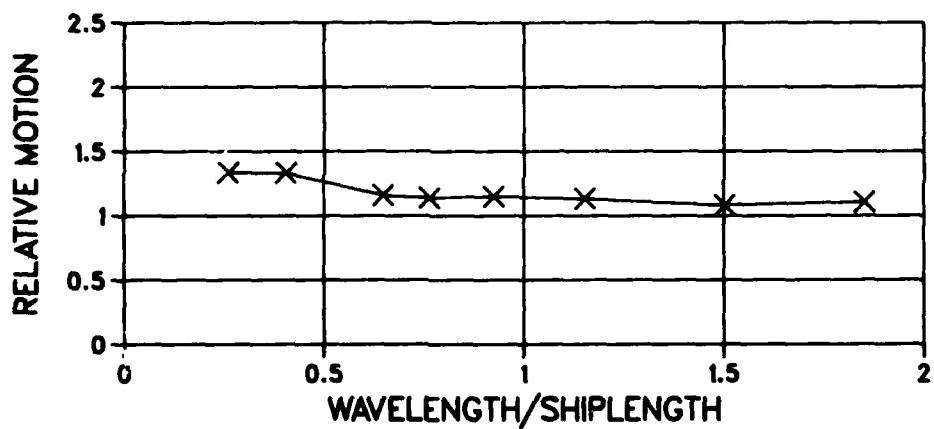


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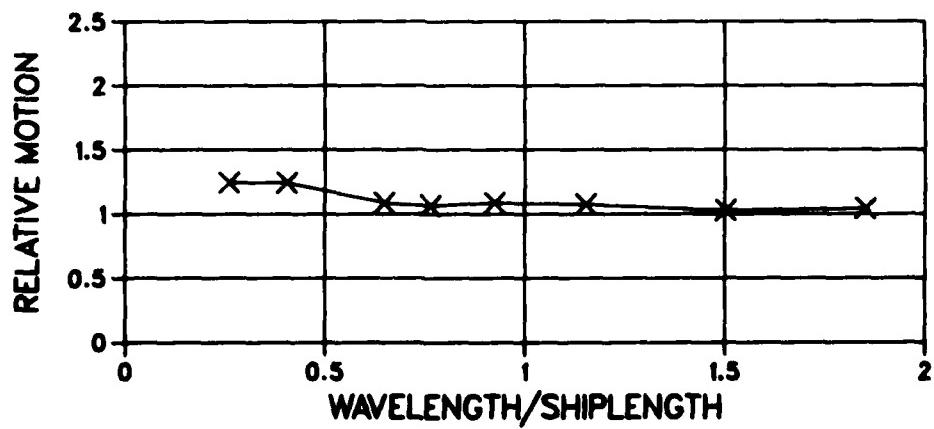
STA.16



STA.14

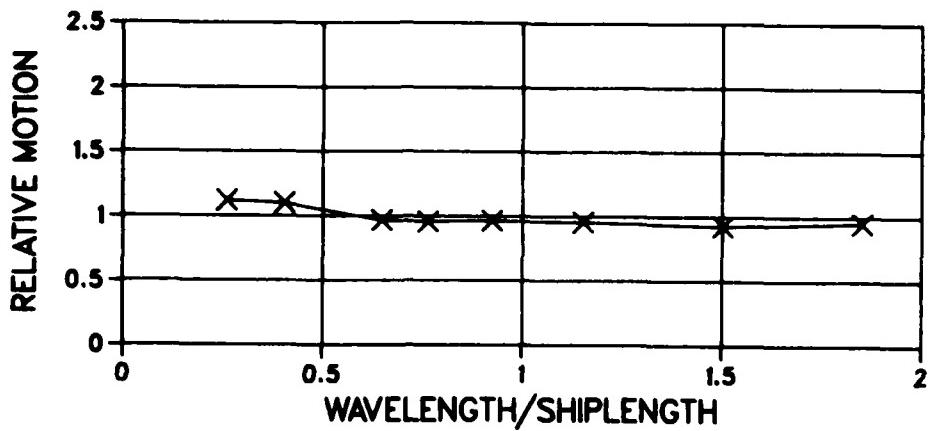


STA.12

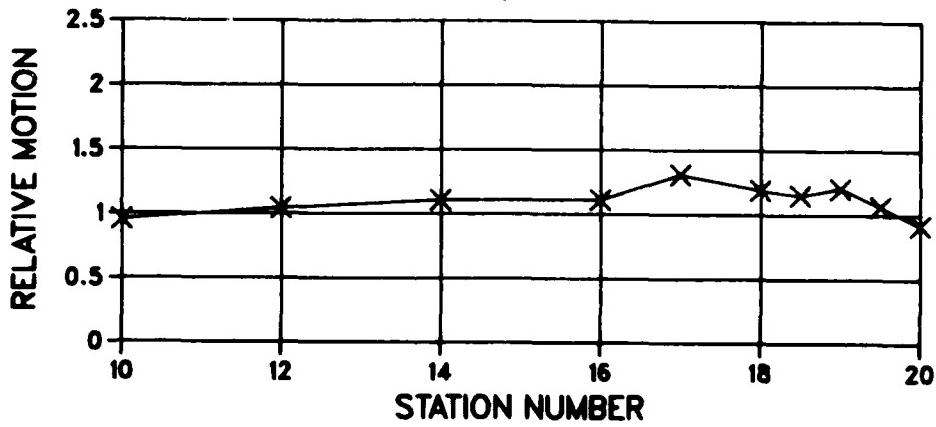


DIFFRACTION. FN=0.2

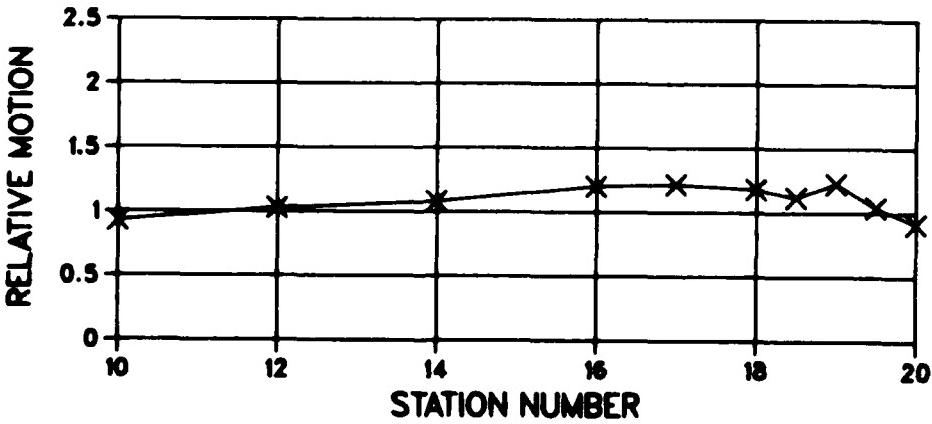
STA.10



LAMBDA/L = 1.852

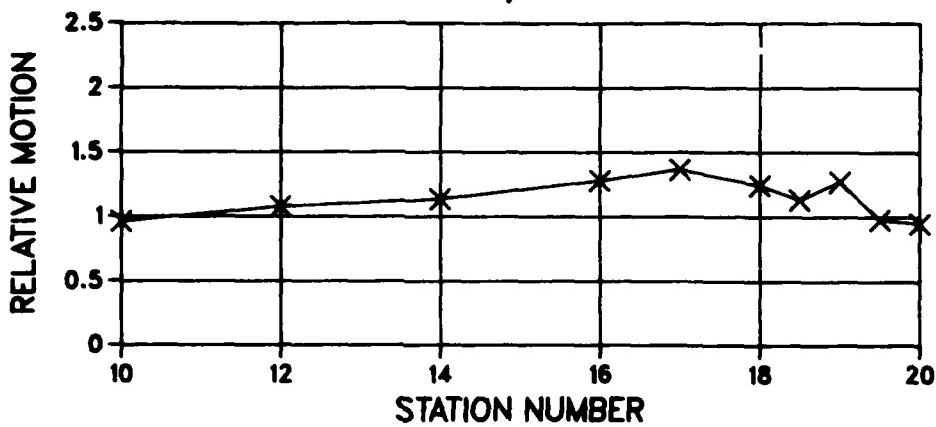


LAMBDA/L = 1.503

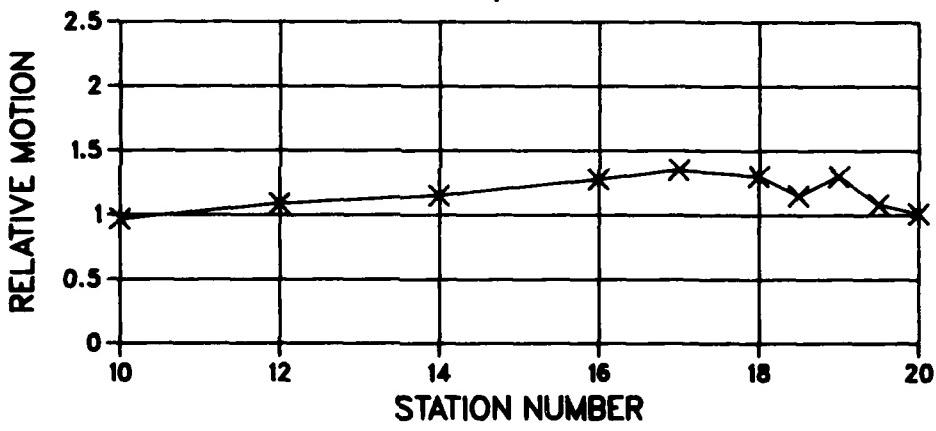


DIFFRACTION. FN=0.2

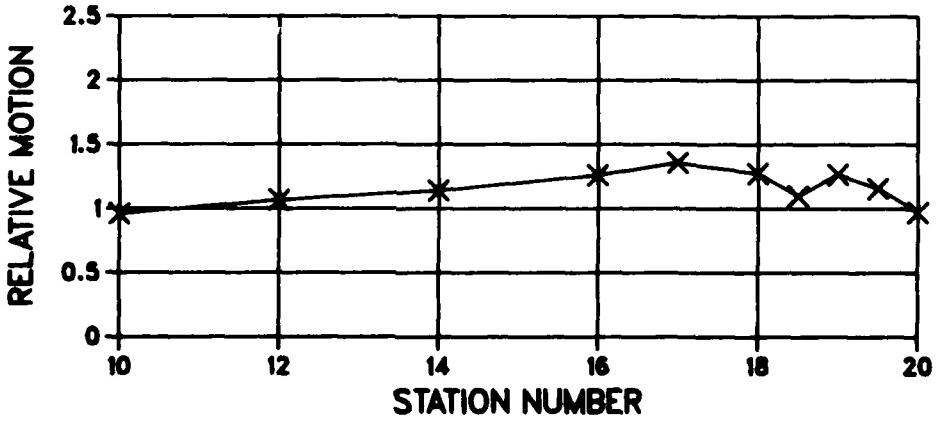
LAMBDA/L =1.154



LAMBDA/L =0.926

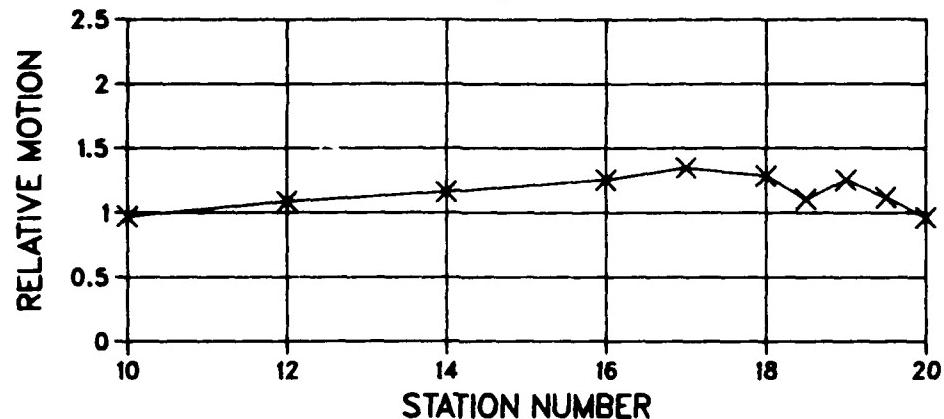


LAMBDA/L =0.766

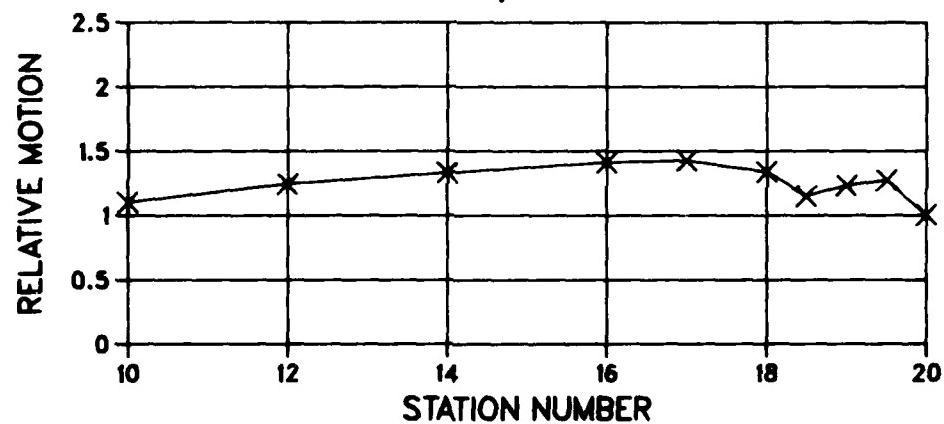


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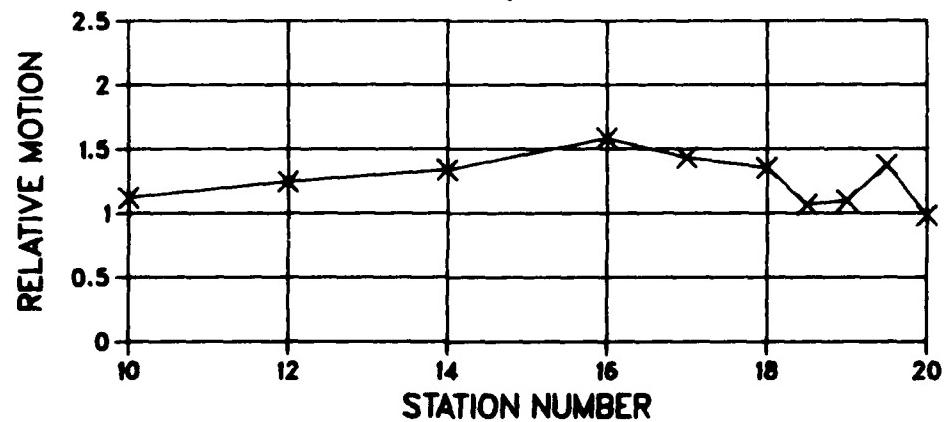
LAMBDA/L = 0.650



LAMBDA/L = 0.406

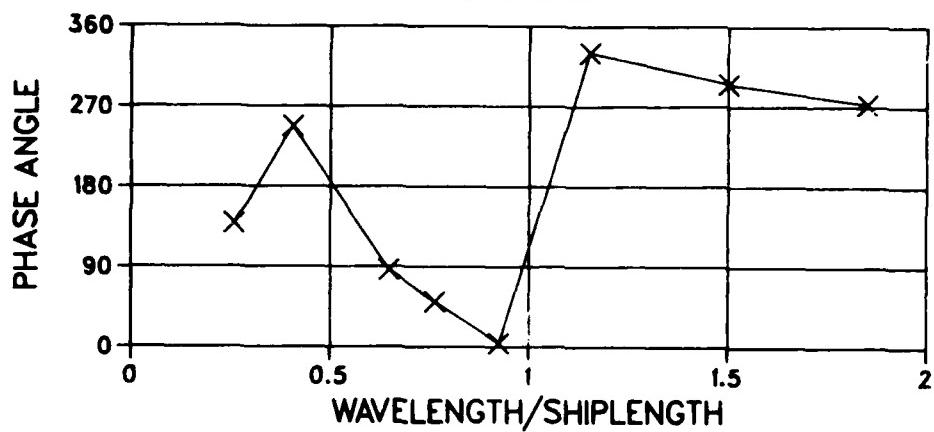


LAMBDA/L = 0.261

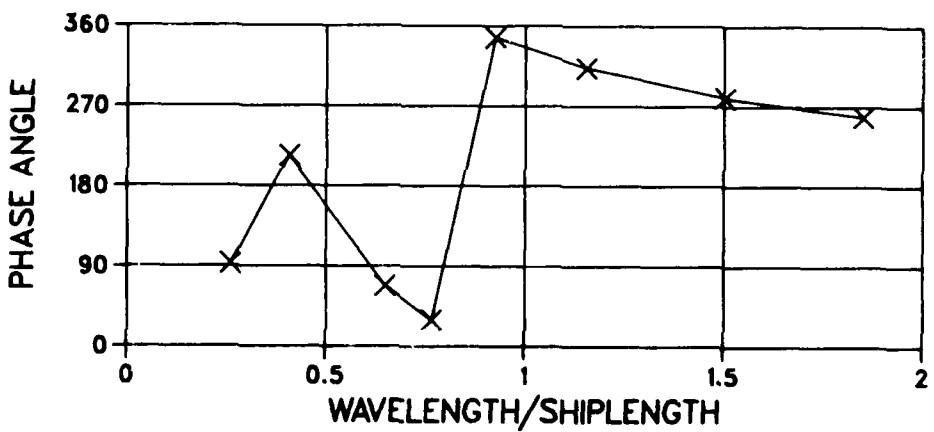


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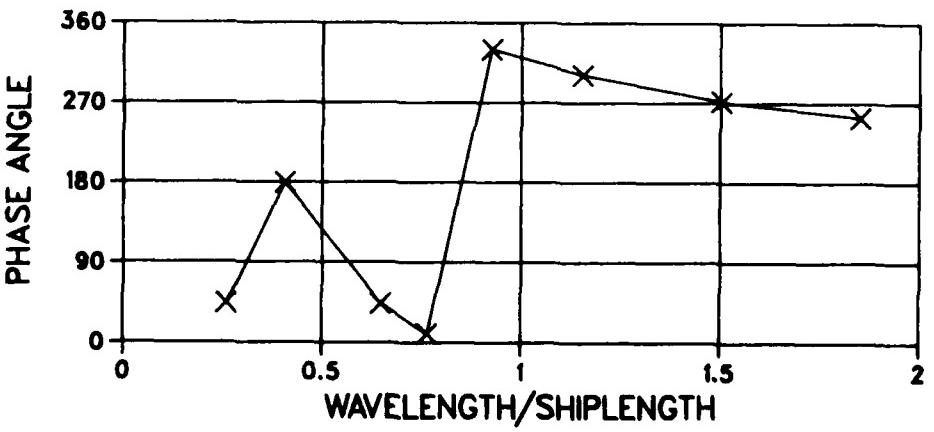
STA.20



STA.19.5

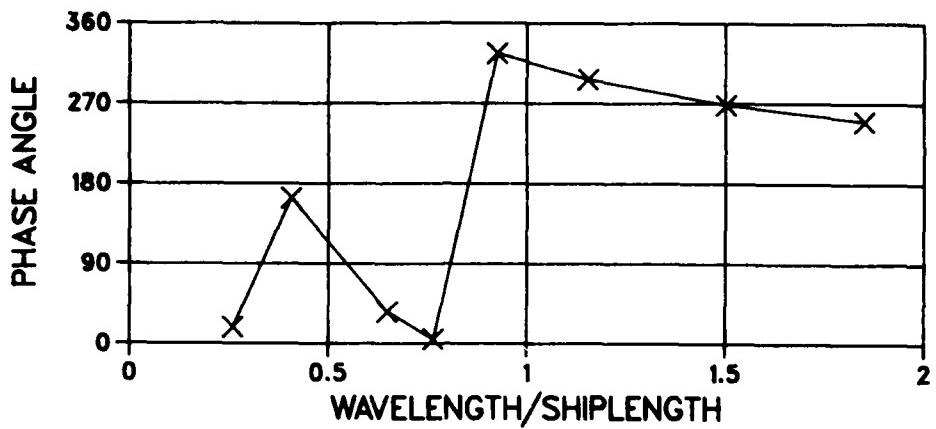


STA.19

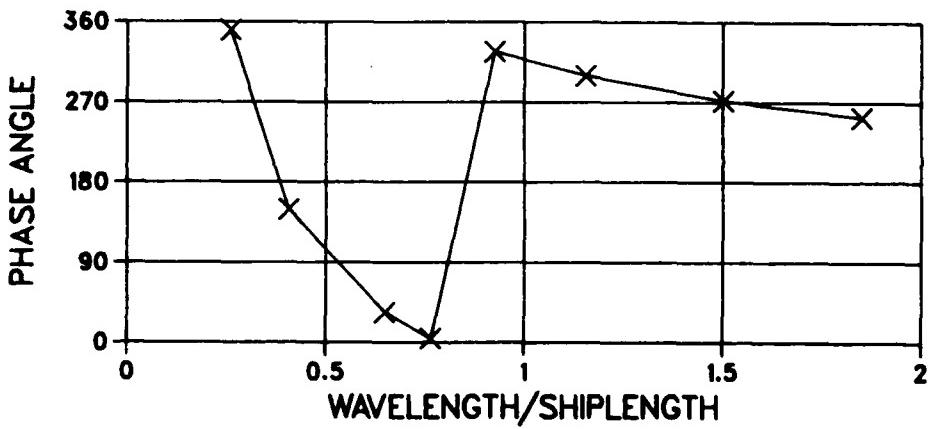


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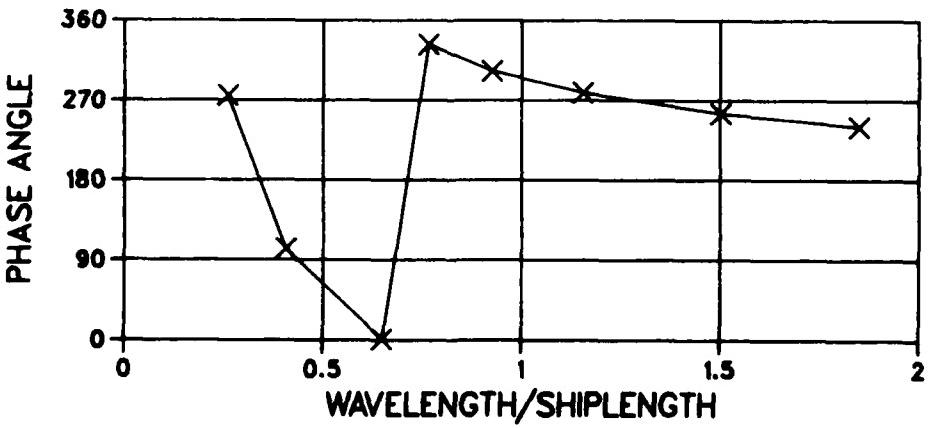
STA.18.5



STA.18

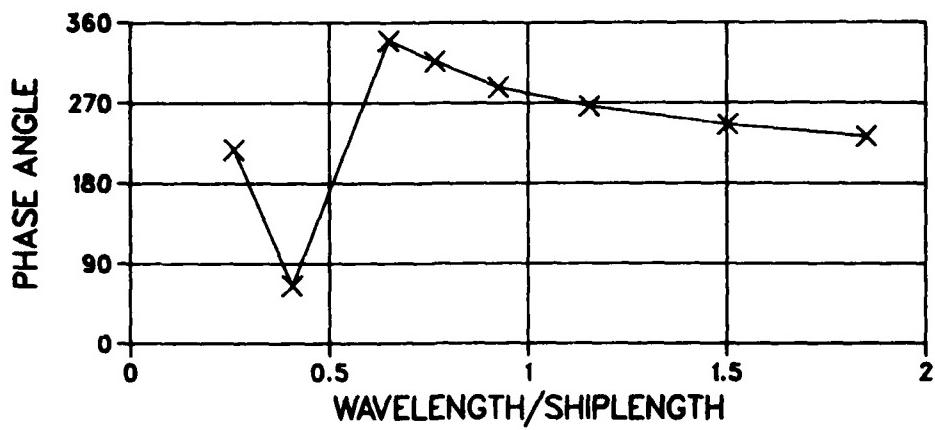


STA.17

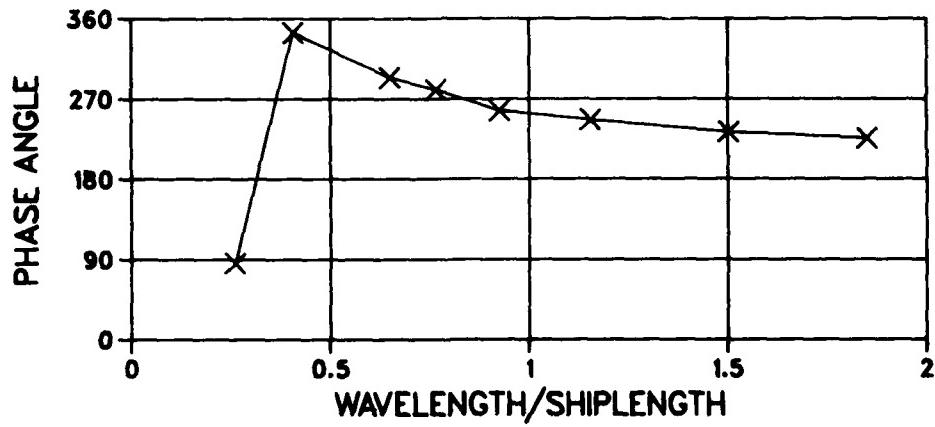


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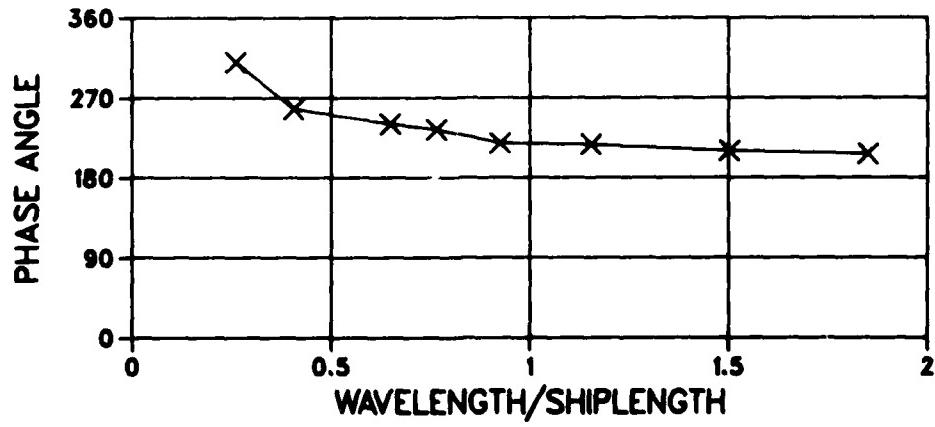
STA.16



STA.14

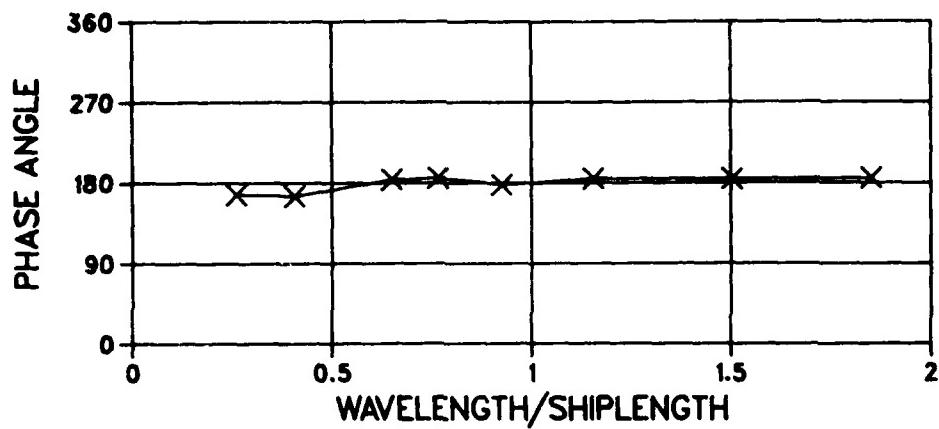


STA.12

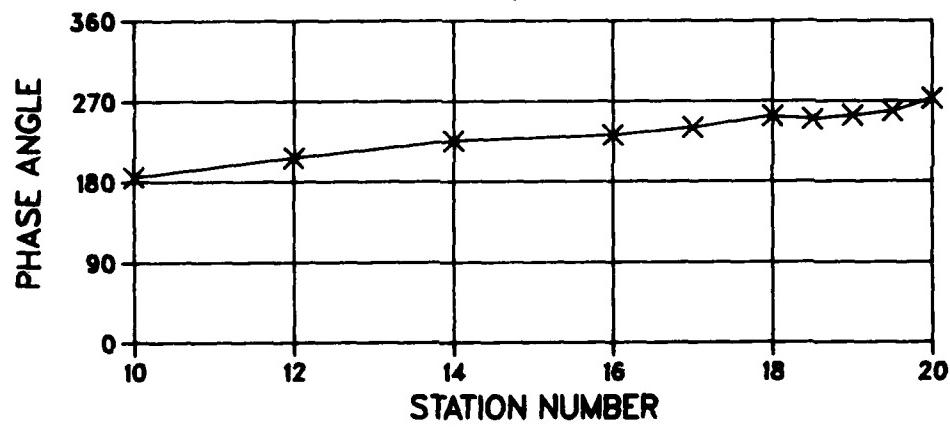


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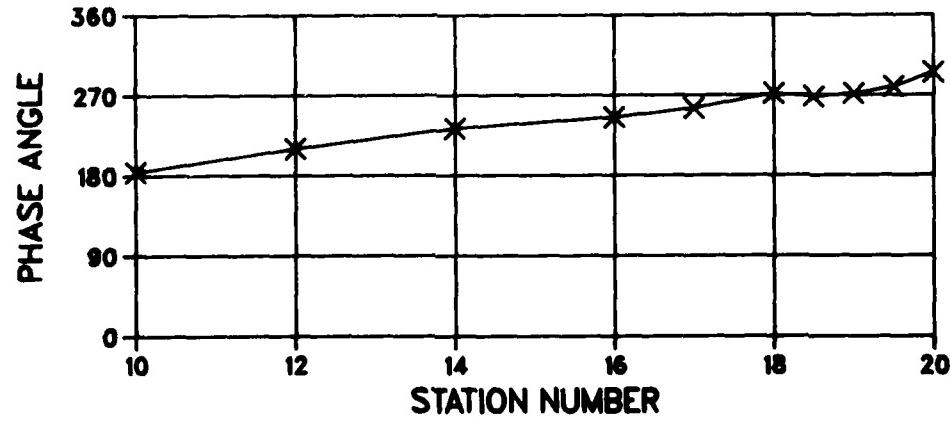
STA.10



LAMBDA/L = 1.852

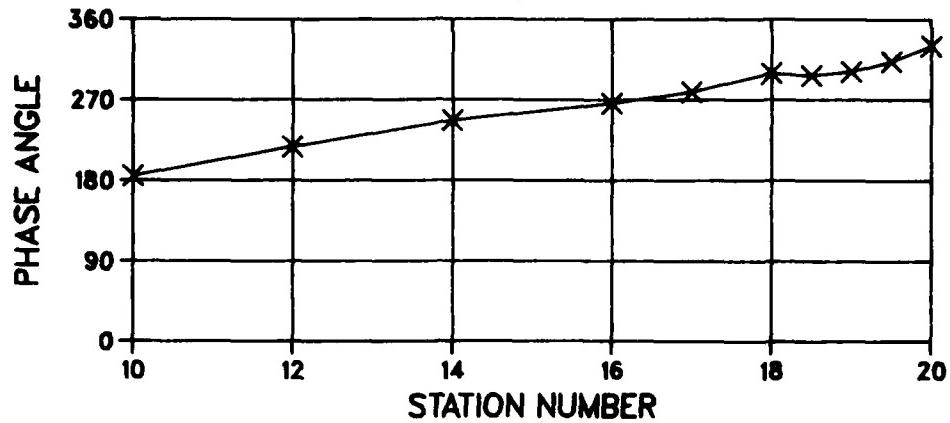


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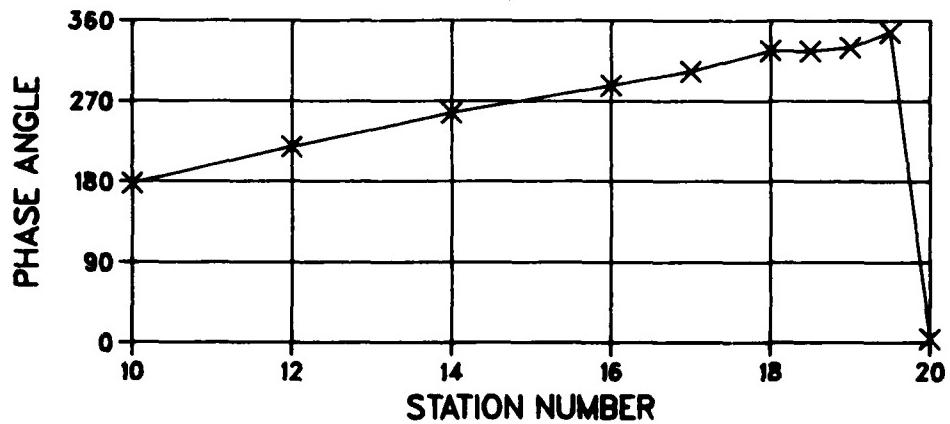


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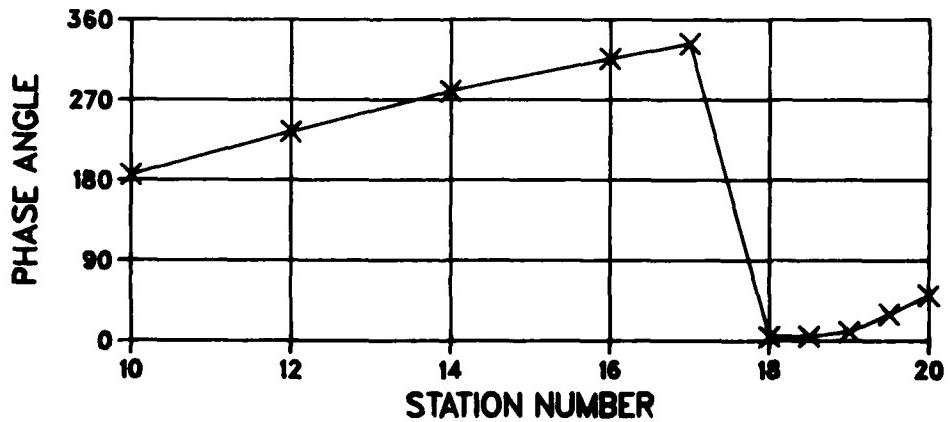
LAMBDA/L = 1.154



LAMBDA/L = 0.926

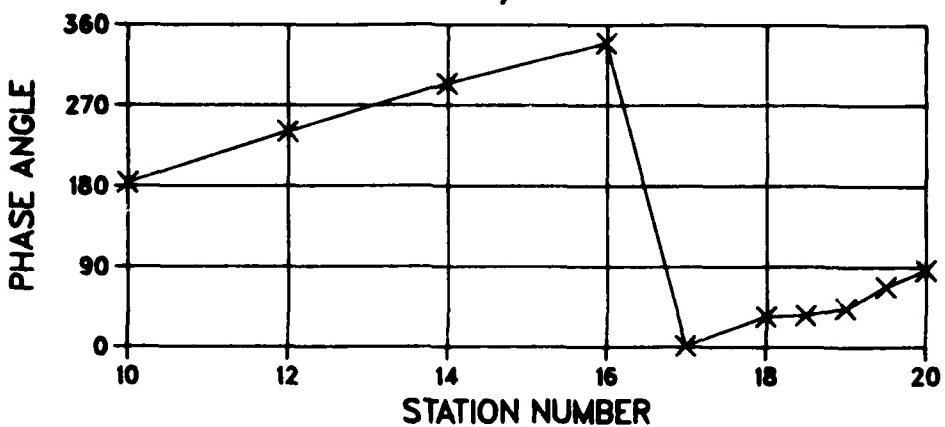


LAMBDA/L = 0.766

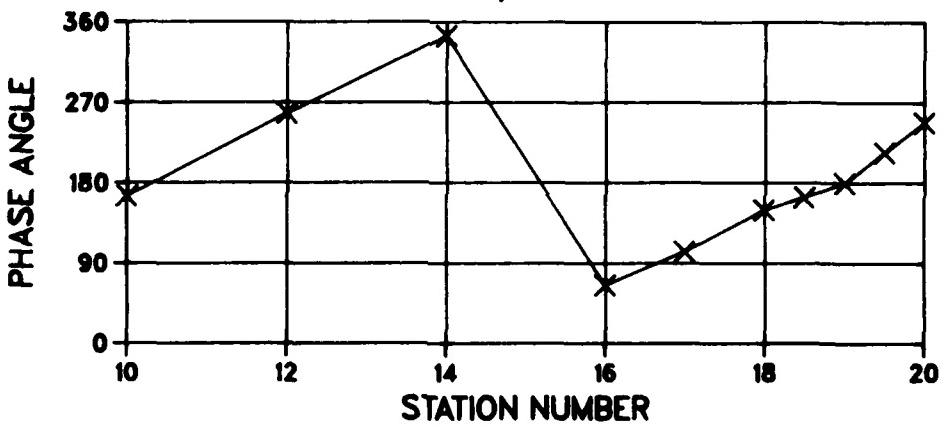


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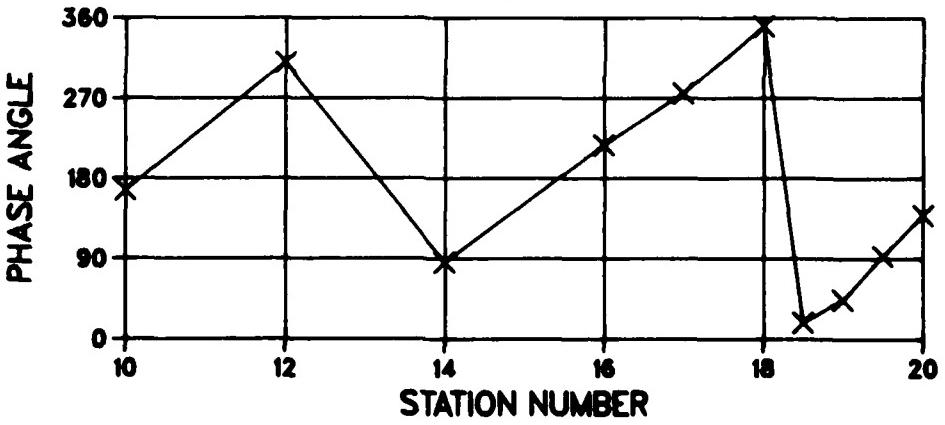
LAMBDA/L = 0.650



LAMBDA/L = 0.406

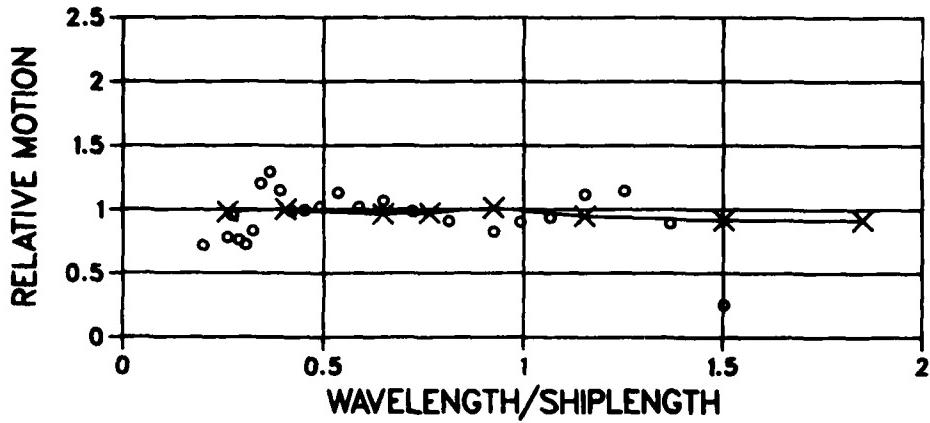


LAMBDA/L = 0.261

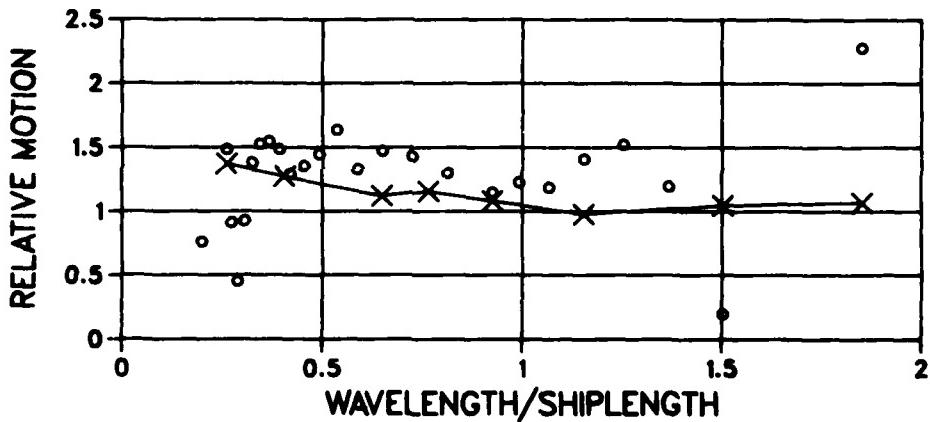


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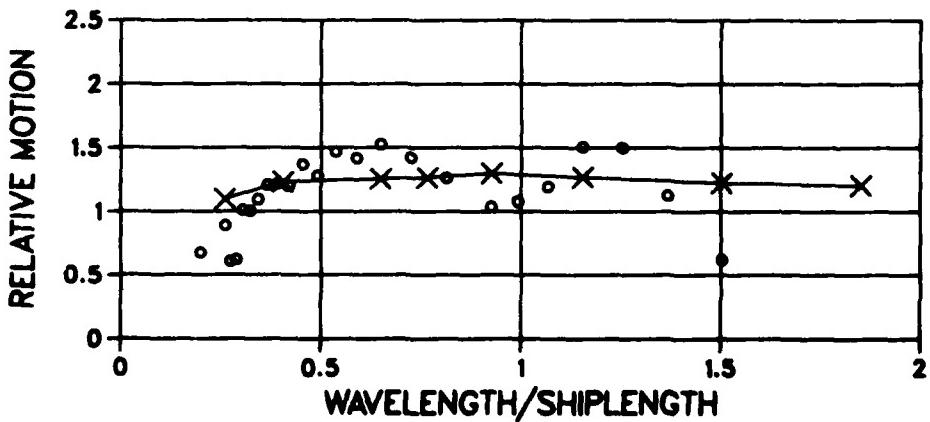
STA.20



STA.19.5



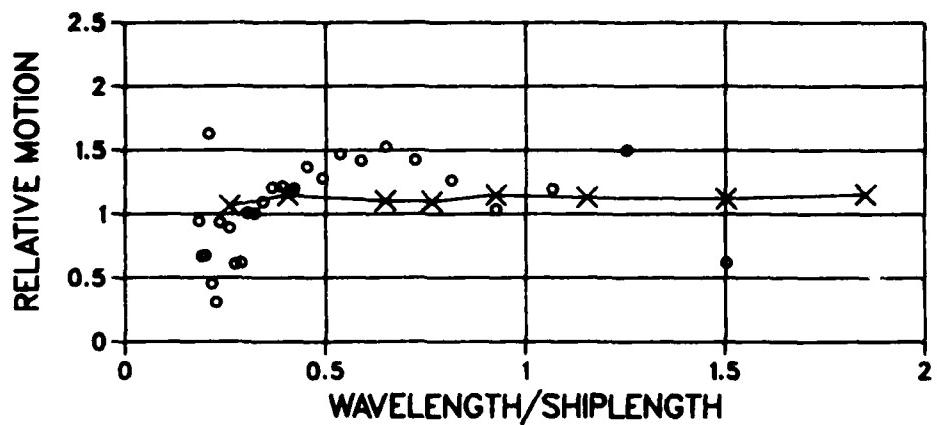
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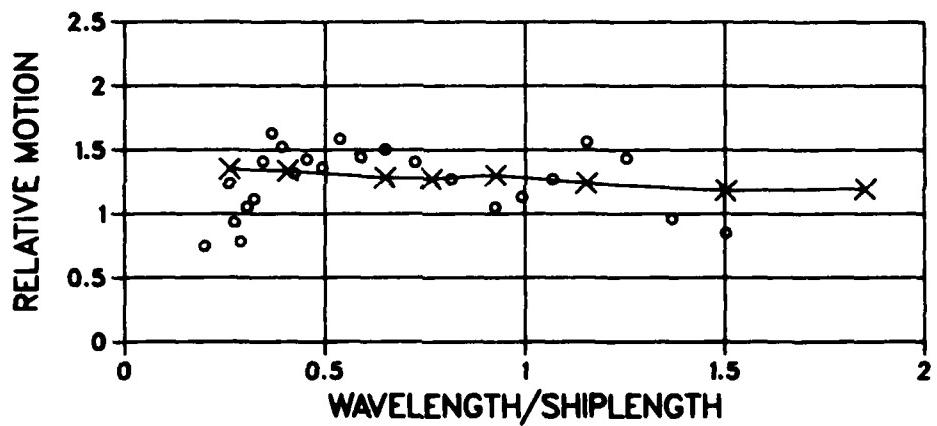
X REGULAR WAVES
 o TRANSIENT TEST

TRANSIENT TEST 1.

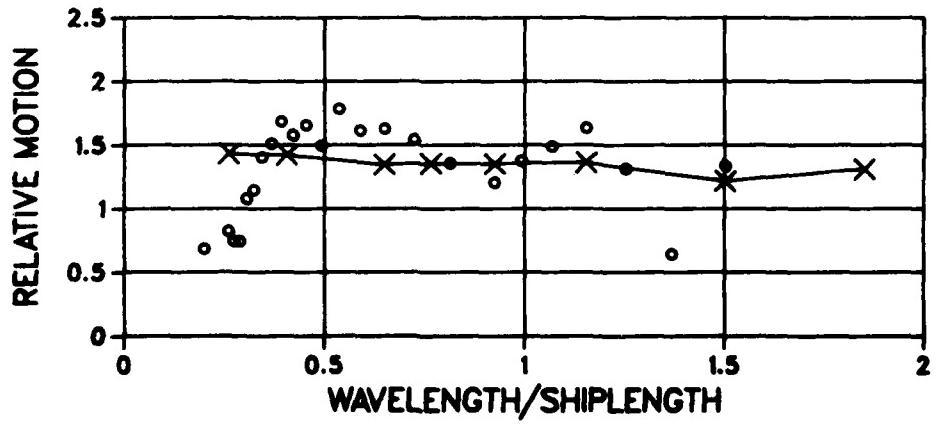
STA.18.5



STA.18



STA.17

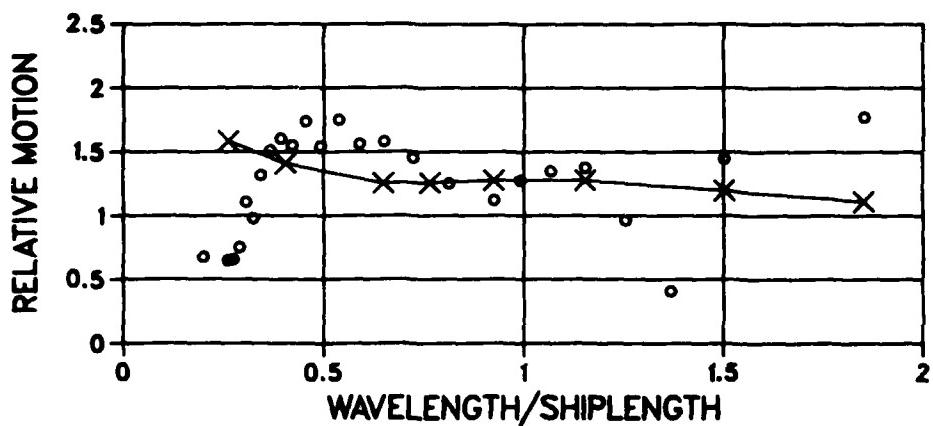


X REGULAR WAVES

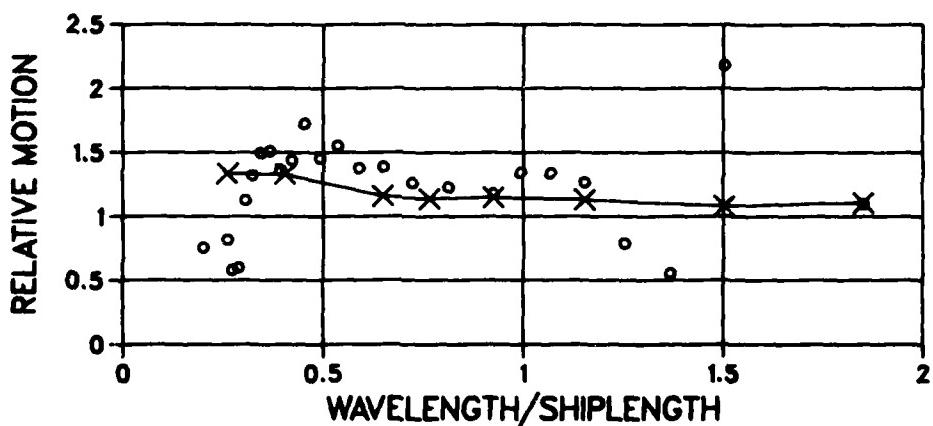
○ TRANSIENT TEST

TRANSIENT TEST 1.

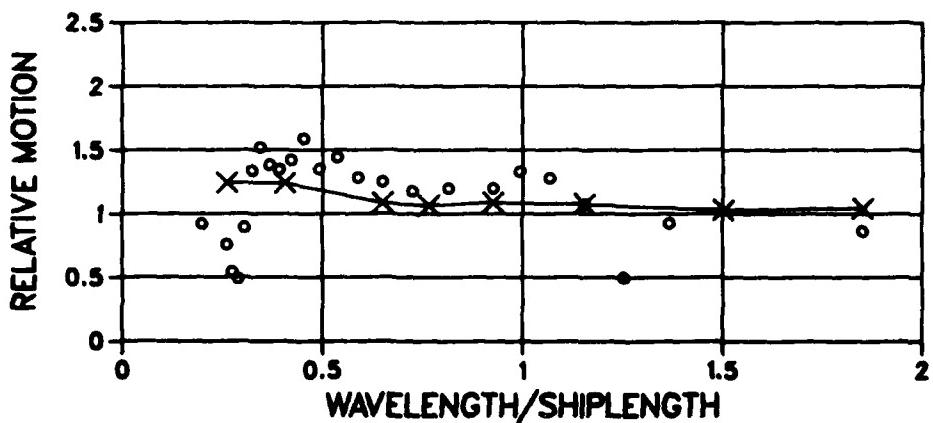
STA.16



STA.14



STA.12

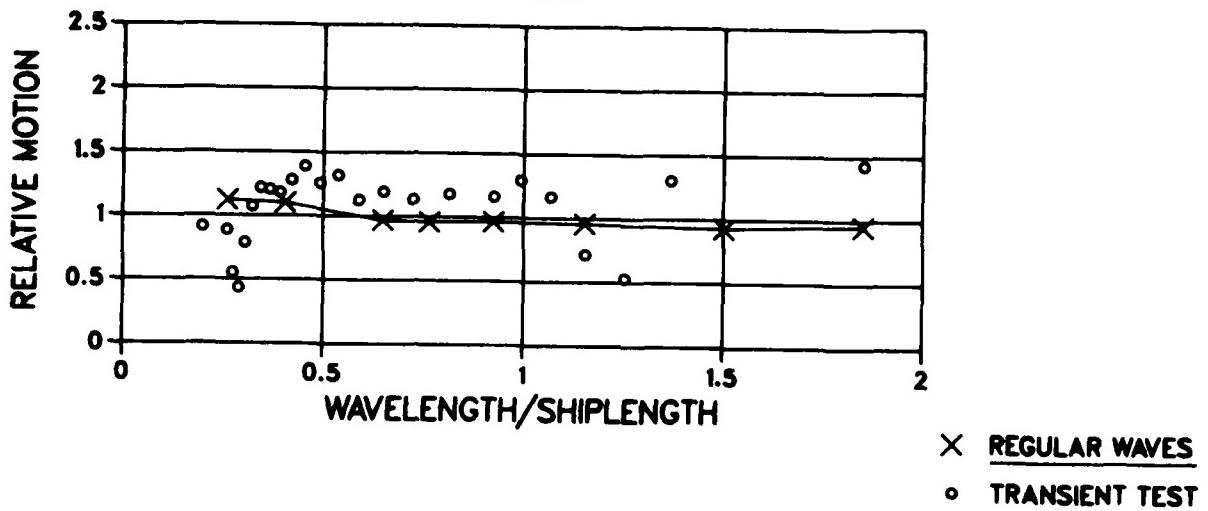


x REGULAR WAVES

o TRANSIENT TEST

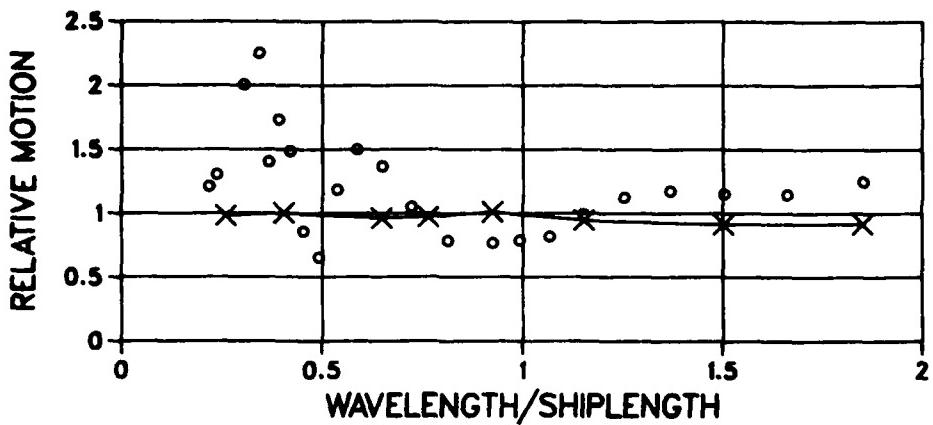
TRANSIENT TEST 1.

STA.10

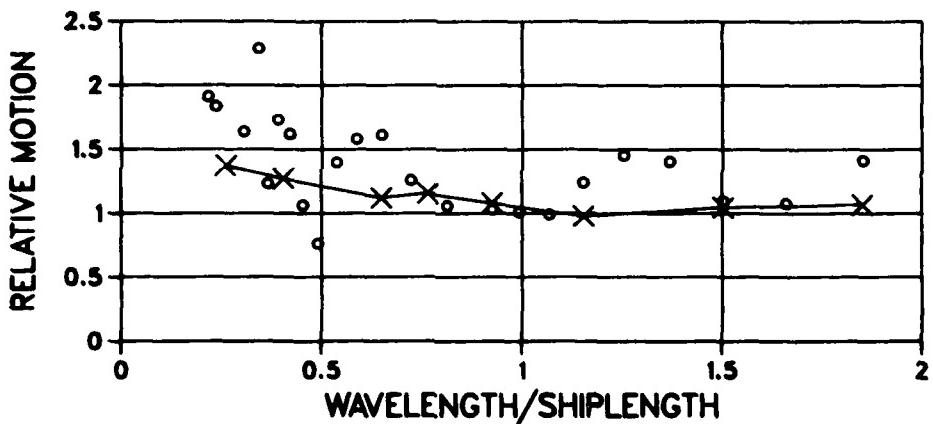


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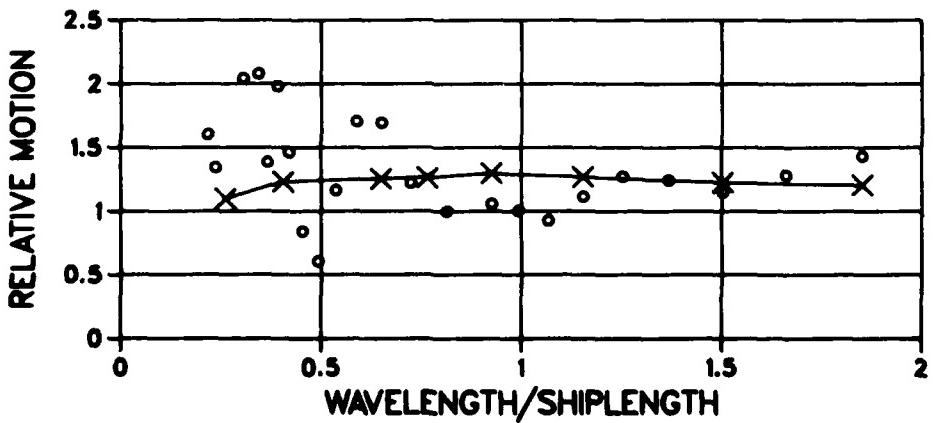
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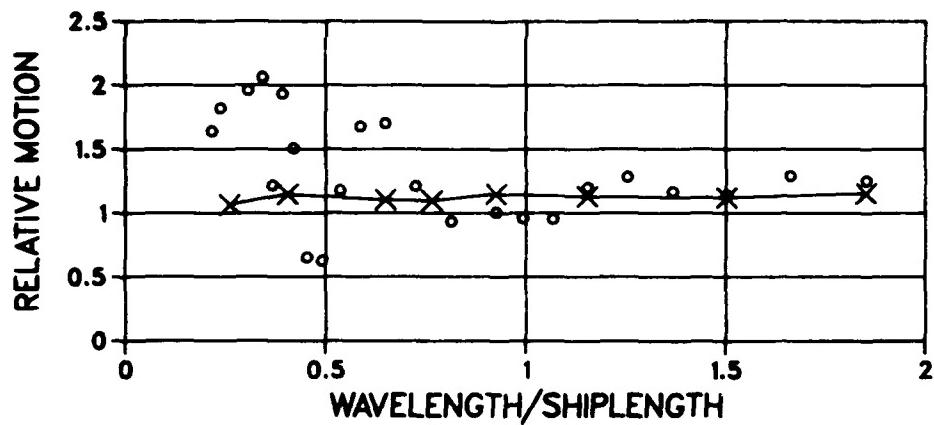
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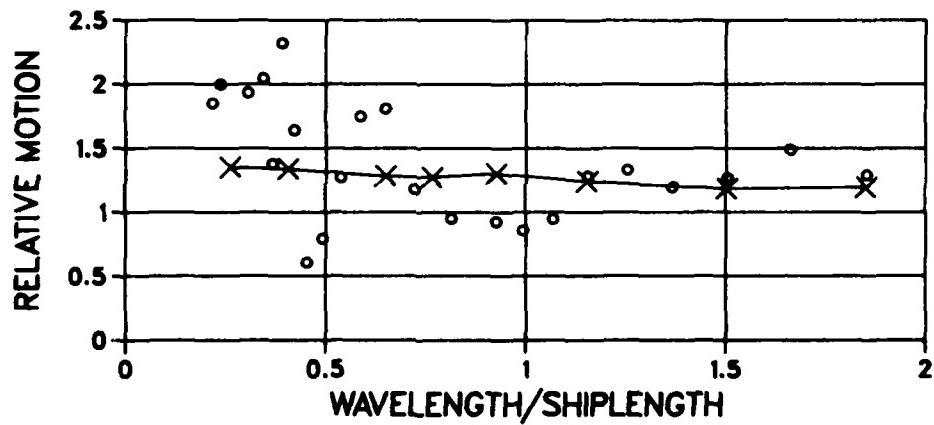
X REGULAR WAVES
 o TRANSIENT TEST

TRANSIENT TEST 2.

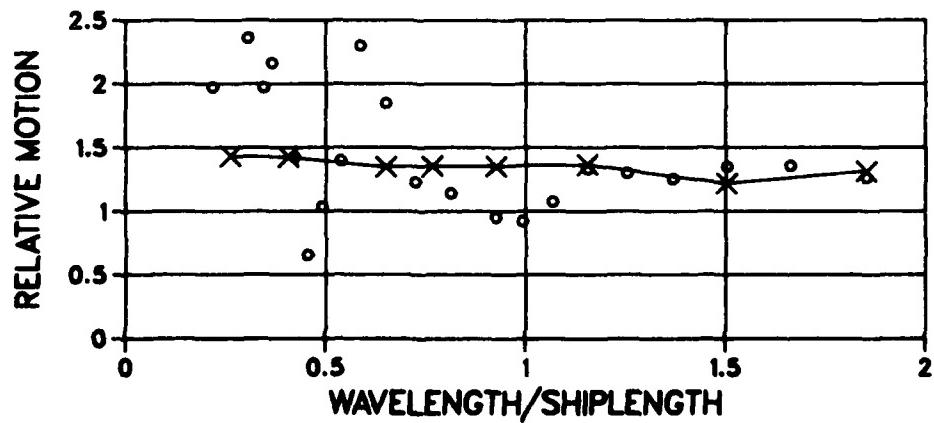
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STA.18



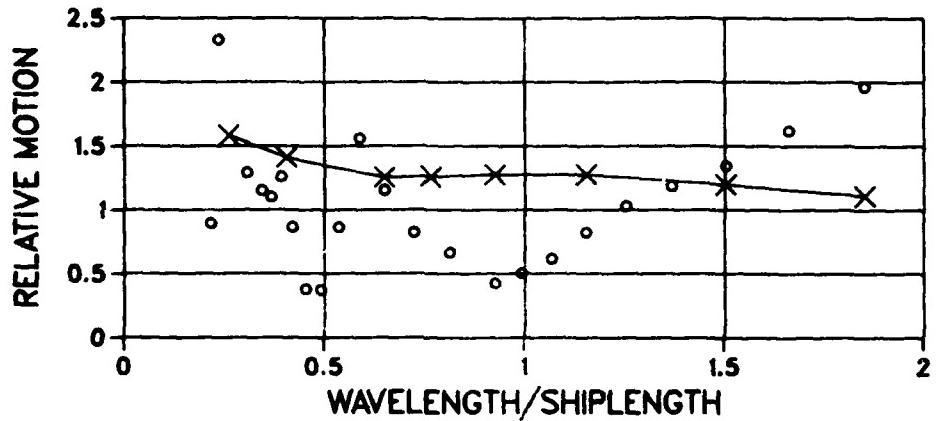
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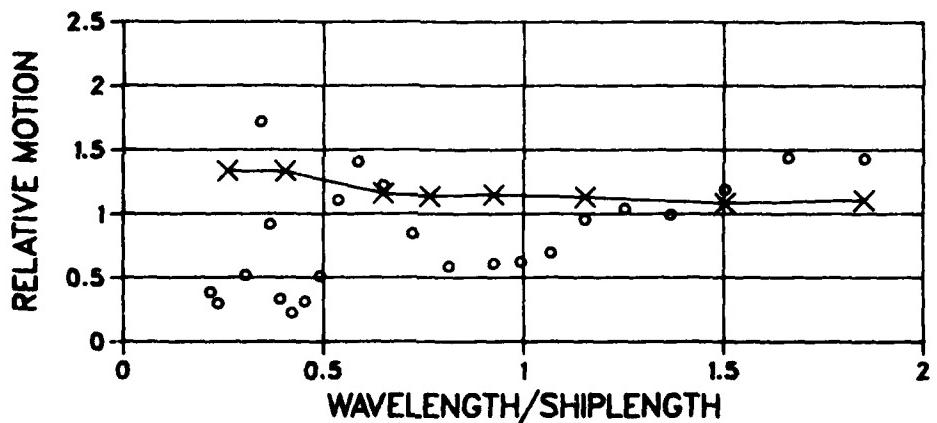
X REGULAR WAVES
 ○ TRANSIENT TEST

TRANSIENT TEST 2.

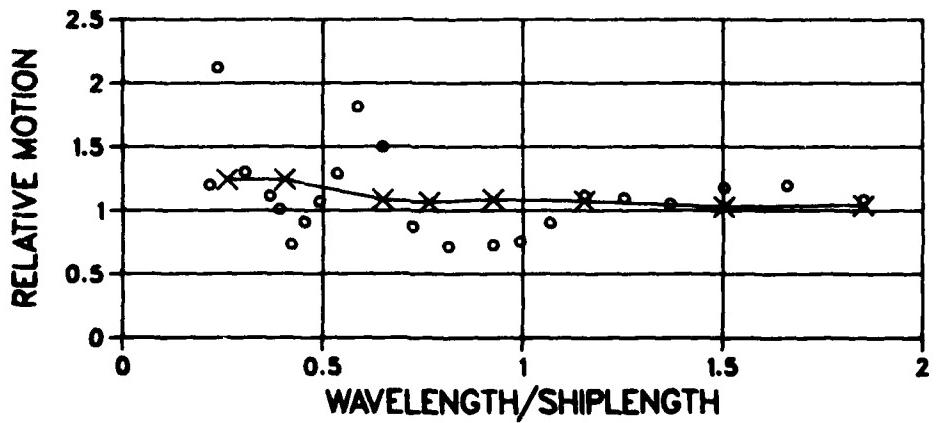
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STA.14



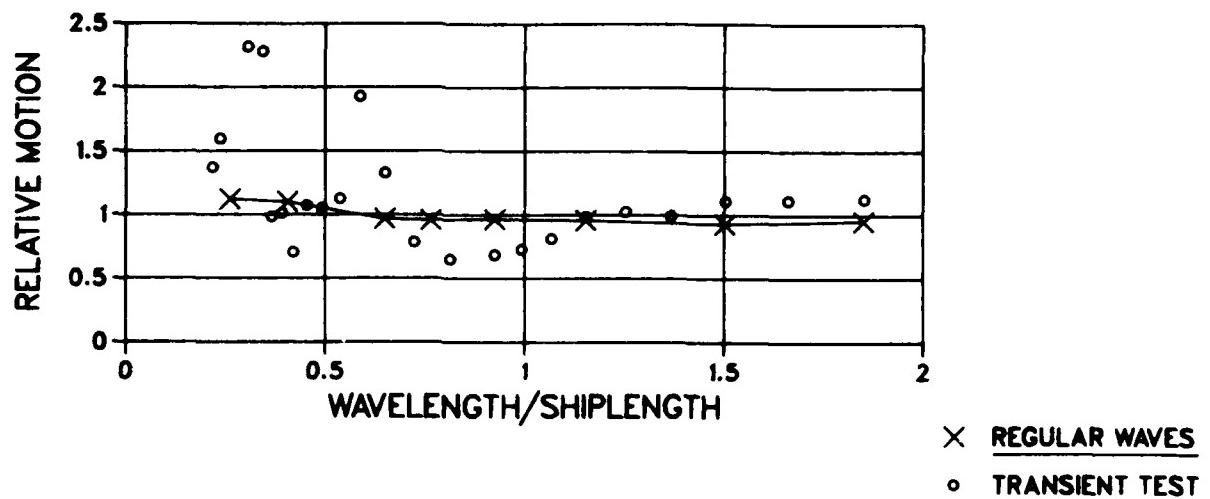
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X REGULAR WAVES
o TRANSIENT TEST

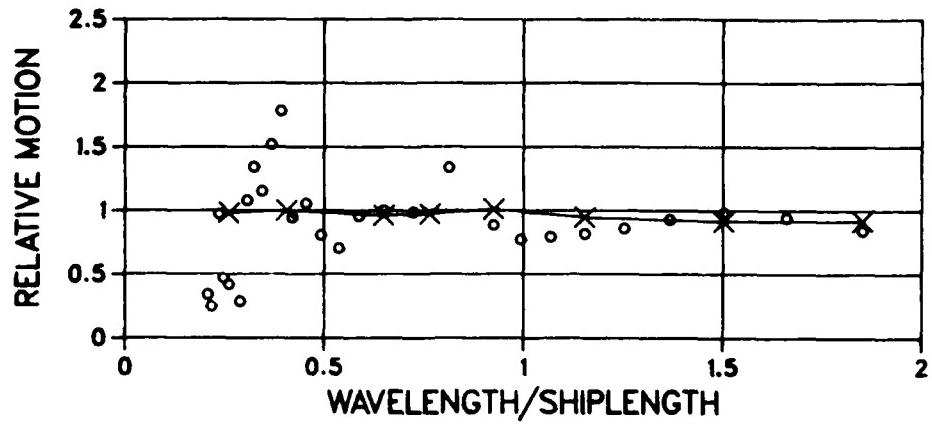
TRANSIENT TEST 2.

STA.10

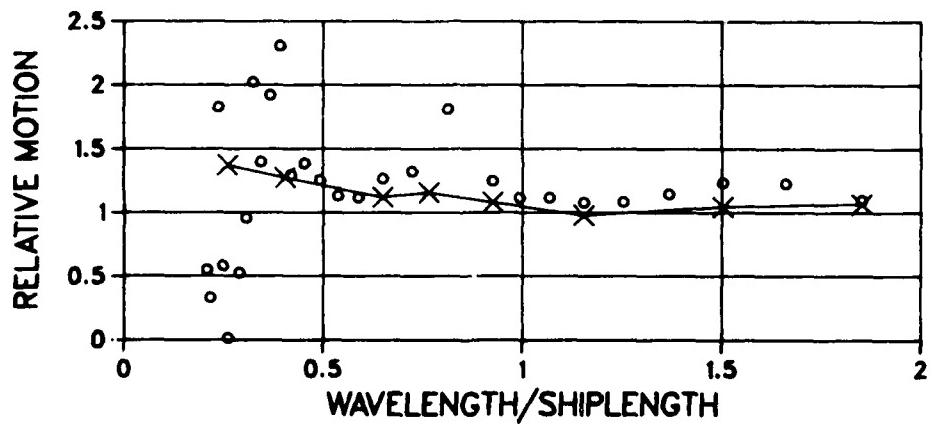


TRANSIENT TEST 3.

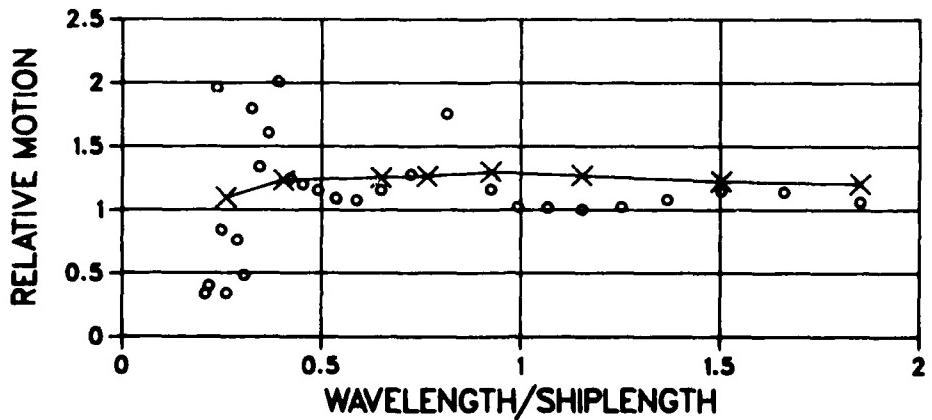
STA.20



STA.19.5



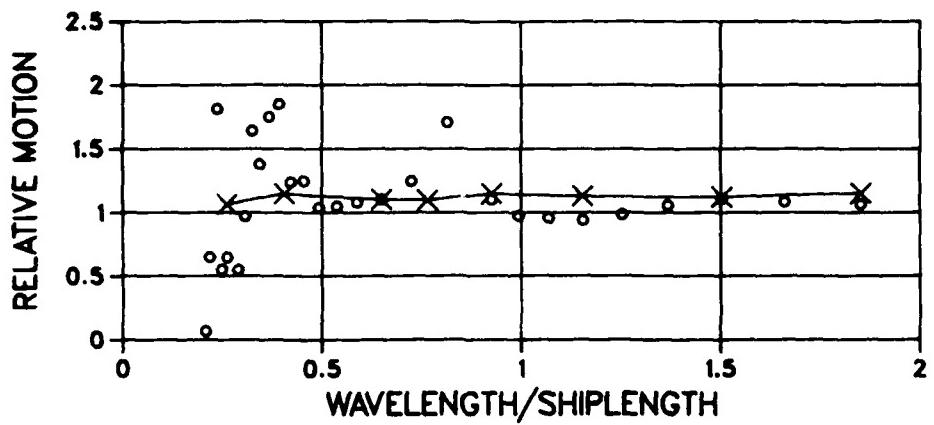
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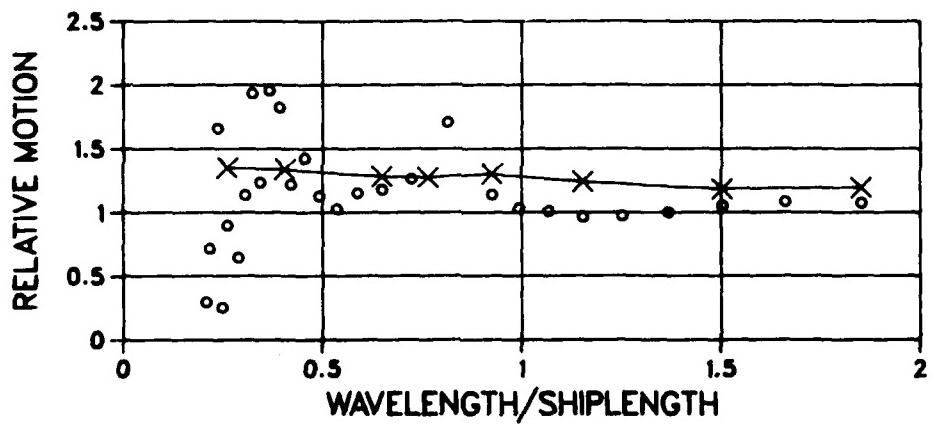
X REGULAR WAVES
 o TRANSIENT TEST

TRANSIENT TEST 3.

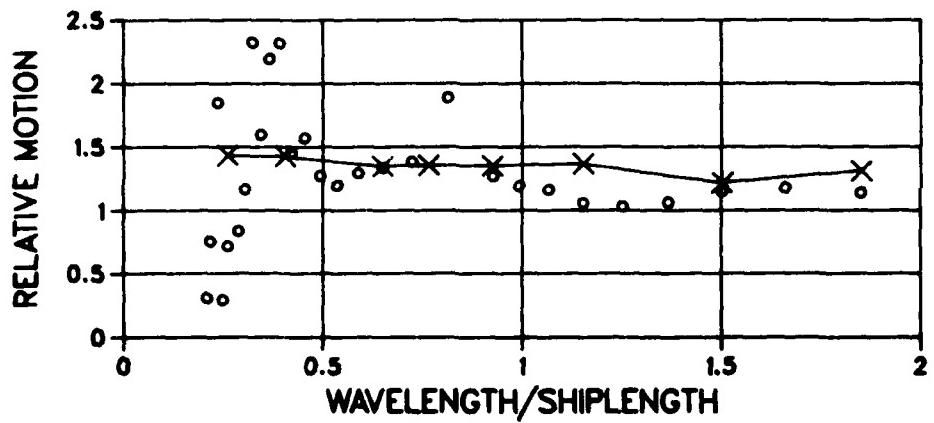
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STA.18



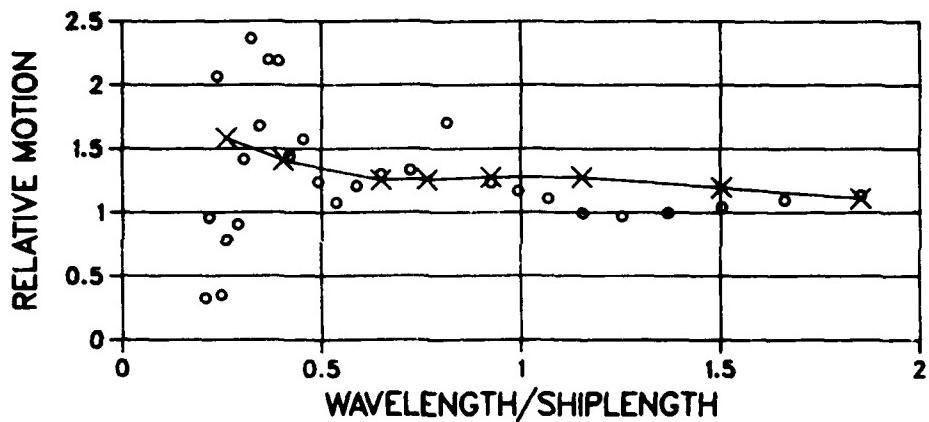
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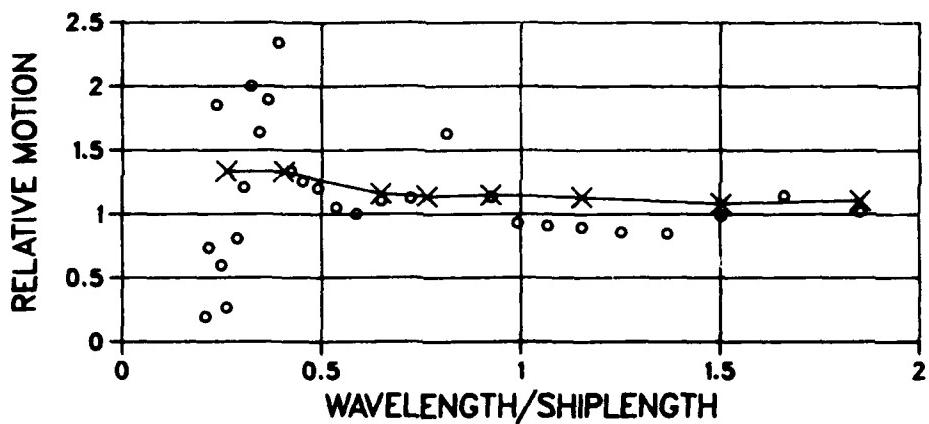
X REGULAR WAVES
o TRANSIENT TEST

TRANSIENT TEST 3.

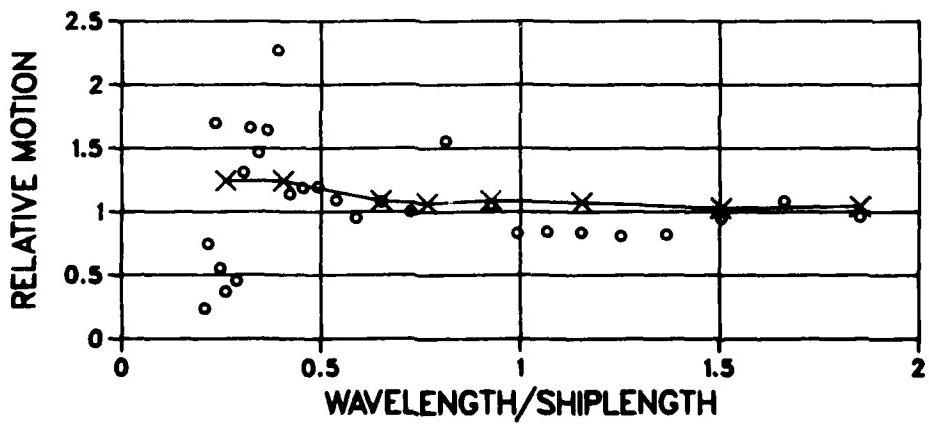
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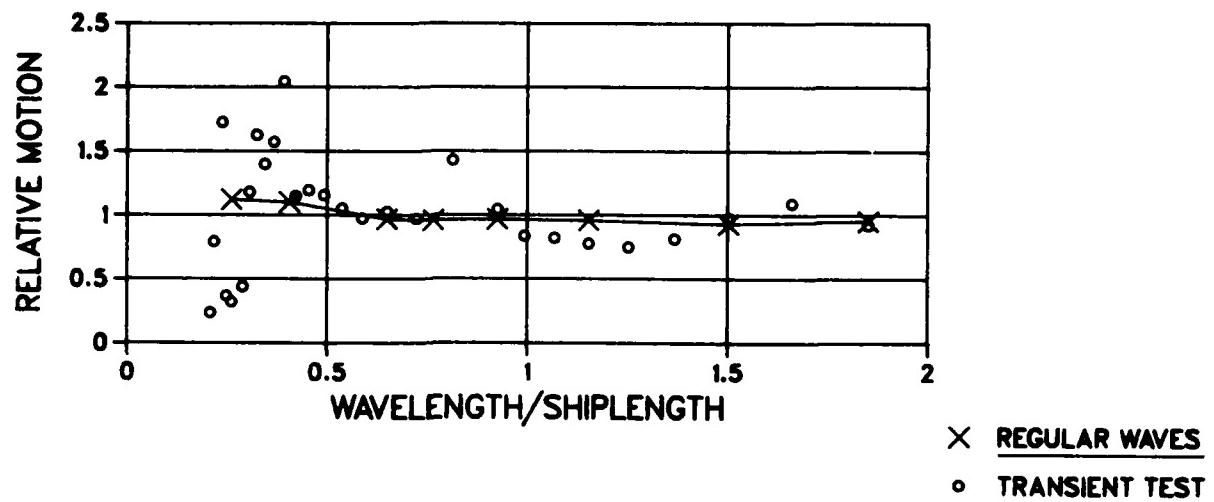
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X REGULAR WAVES
○ TRANSIENT TEST

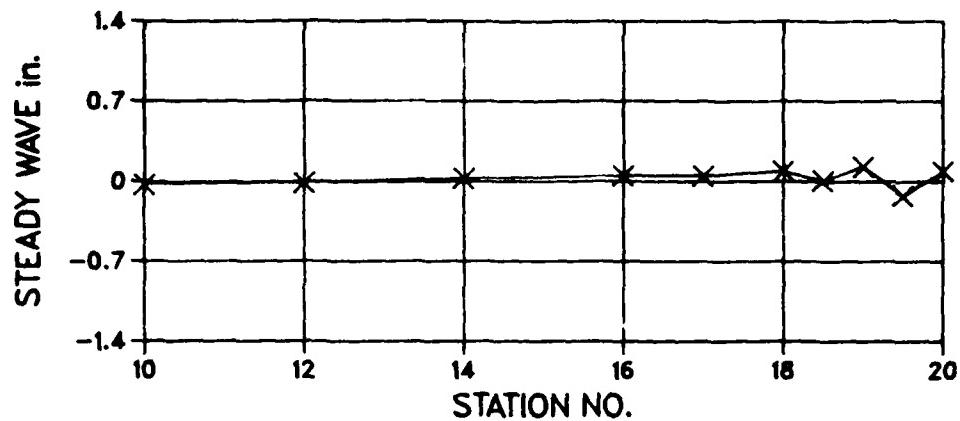
TRANSIENT TEST 3.

STA.10

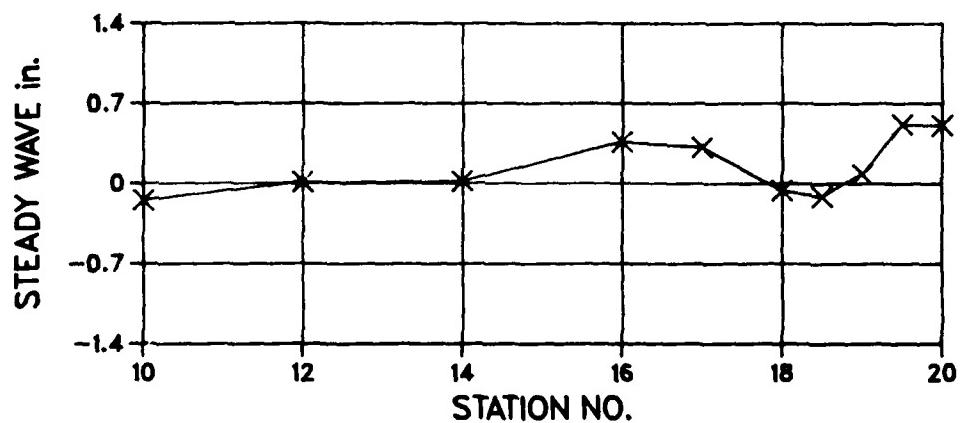


STEADY WAVE PROFILES

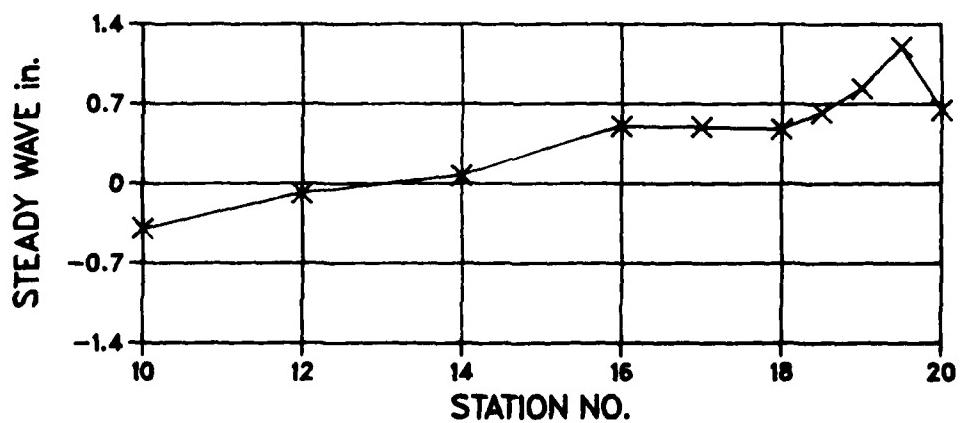
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$FN=0.2$



$FN=0.3$



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